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This ATM is being issued to summarize the results of the anomaly study instigated as a result of the Apollo 14 problem with Dome Removal Tool insertion on the Fuel Cask Body Release Assembly.

Approved by:

L. R. Lewis, Manager

ALSEP Engineering



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I. STUDY PLAN

- A. Check the design tolerances of the flight and trainer hardware.
- B. Review the training and flight hardware configuration.
- C. Survey the CF² procedures.
- D. Review training sequence versus design intent.
- E. Submit recommendations.

II. FINDINGS

- A. Flight and Trainer Design Comparison The trainer and flight hardware designs were compared and it was found that all dimensional requirements were the same for both models.
- B. Trainer and Flight Hardware Configuration

l. KSC Trainer

- a. The dimensional checks that were made are shown in attachment I with the out-of-tolerance item circled.
- b. The spline lock is a metal pin rather than the ceramic bead string.
- c. The GLFC dome holes for alignment of the Body Release Assembly are extremely oversize, elongated and broken out.
- d. The Body Release Assembly had several discrepancies above and beyond those noted on the dimensional check:
 - (1) The assembly is an aluminum construction.
 - (2) The release levers are badly worn.
 - (3) The angle brackets are full slotted per design but are badly worn.
 - (4) The positive stop appears to have rewelded.
 - (5) The angle brackets have been broken off and reassembled to the bands.
 - (6) The plunger is badly worn and deformed (both internally and externally).
 - (7) As reflected in the dimensional check the spring force is extremely low.

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2. MSC Trainer

- a. The dimensional checks that were made are shown in Attachment II, with the out-of-tolerance items circled.
- b. The Body Release Assembly had several discrepancies above and beyond those noted on the dimensional check:
 - (1) The locking assembly is free to rotate without depressing the plunger.
 - (2) The locking plate is badly worn.
 - (3) The plunger is badly worn and distorted (both internally and externally).
 - (4) As reflected in the dimensional check the spring force is very low.

3. Flight Hardware

a. All dimensions taken are within print requirements.

C. CF² Procedure Review

- 1. It was noted during CF² that the Dome Removal Tool appeared to be difficult to install.
- 2. The tool was returned to Ann Arbor and reworked.
- 3. During the delta CF² both astronauts performed fit checks using the tool on both the flight and the flight spare cask and found the fit acceptable.

D. Training Exercise Review

1. A Trainer deployment on 22 March 1971 at KSC was witnessed and it was evident that the installation of the DRT was a poor simulation of the flight hardware requirements.

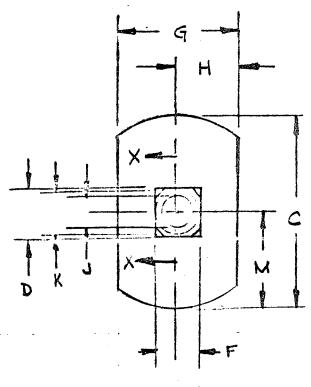


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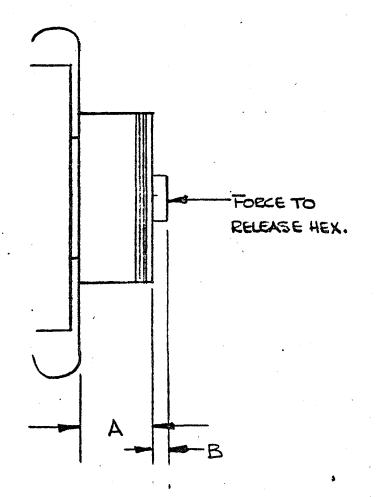
III. RECOMMENDATIONS

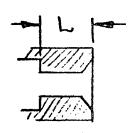
- A. It was agreed at a meeting with J. Briley, NASA/MSC that the KSC Trainer should be considered the prime training hardware consisting of the best possible components.
- B. The aluminum body release assembly on the KSC Trainer should be replaced with a new titanium assembly.
- C. The titanium body release assembly on the MSC Trainer should have piece part replacement in the area of the DRT interface.
- D. The non-conforming wire spline lock on the KSC Trainer should be replaced with proper ceramic bead construction.
- E. The Norco lock assembly of the Trainer DRT should be replaced.
- F. An astronaut decal should be added to the DRT to reflect the offset requirement for engagement.

MODEL NO 5/NG FLIGHT



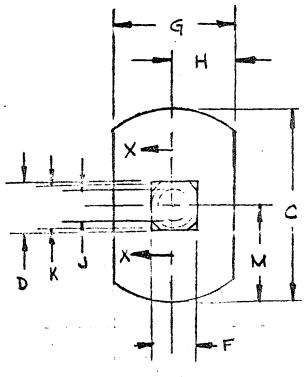
A .570 B .120 1.526 D .401 E .400 1.135 H .554 X .374 L .306 M .743 FORCE 7.5-80



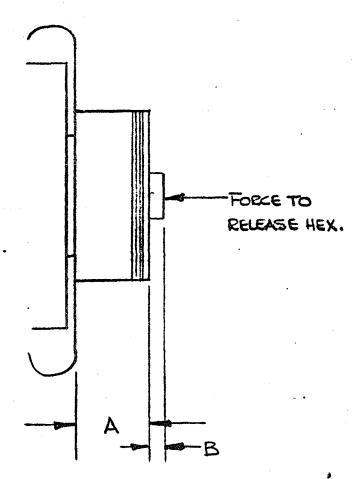


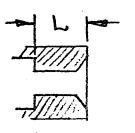
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MODEL NO S/N8 FLIGHT SPARE



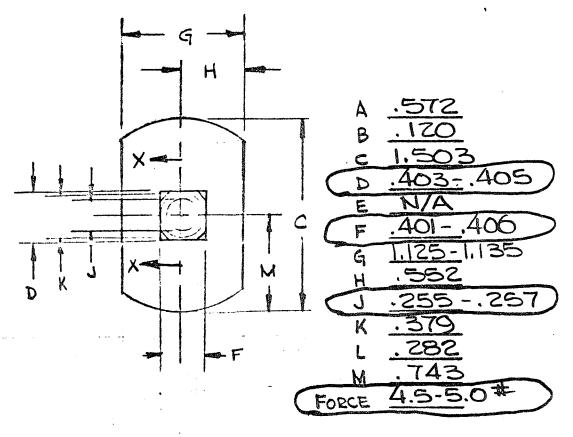
A .563
B .128
C 1.509
D .400
E N/A
F .400
G 1.137
H .557
J .253
K .250
L .282
M .743
FORCE T.O-7.2*

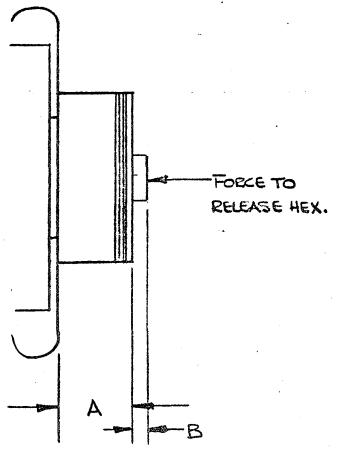


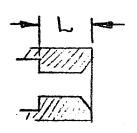


SECTION X-X

MODEL NO KSC TRAINER







SECTION X-X

MODEL NO MSC TRAINER

