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Fabrication of Construction Materials from Lunar and Martian Regolith Using Thermite Reactions with Magnesium

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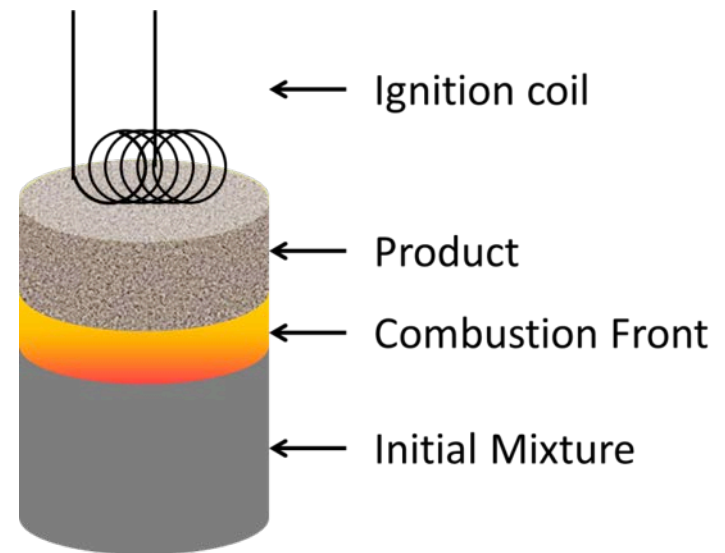
In-Situ Production of Construction Materials from Lunar and Martian Regolith

- In future lunar and Mars missions, construction materials will be needed for landing/launching pads, radiation shielding, and other structures.
- Fabrication methods:
 - Lunar concrete
 - Water or sulfur recovered from regolith
 - Thermoplastic brought from Earth
 - Microwave heating of regolith
 - Needs lots of energy
 - Energetic additives enabling a self-sustained combustion
 - Low energy needed



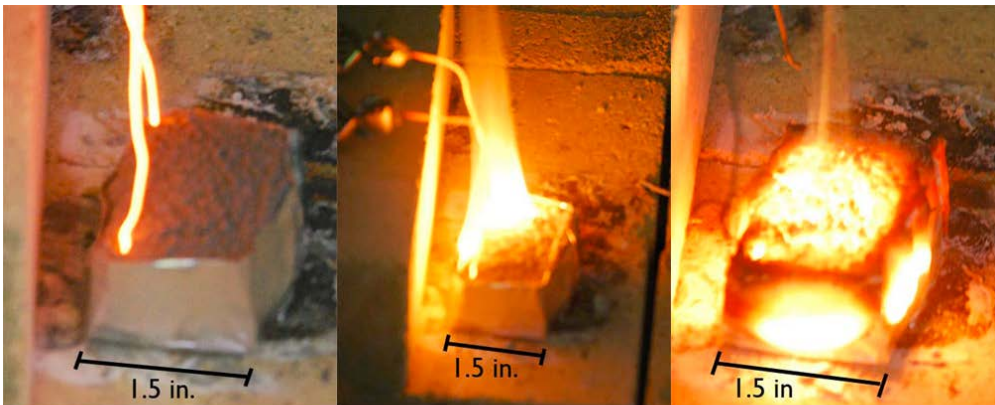
Self-Propagating High-Temperature Synthesis (SHS)

- Upon ignition of a mixture, exothermic reactions cause a self-sustained propagation of the combustion wave.
- Advantages:
 - Low energy for ignition
 - High temperatures generated by the reaction heat release
- Used for synthesis of many ceramic materials.



Combustion in Regolith-based Mixtures

Research Team	Energetic Additive	Additive Content (wt%)
Martirosyan and Luss (2006)	Ti + B	>40
Corrias et al. (2012)	FeTiO_3 + Al	>70
Faierson et al. (2010)	Al	>37



a

b

c

- JSC-1A/Al mixture
- Large external energy is supplied by a long heating wire embedded in the mixture
- No self-sustained combustion

Prior Research of Our Team

- Combustion of mixtures of JSC-1A lunar regolith simulant with **magnesium**
- Magnesium is easier to ignite than aluminum
- Thermodynamic calculations of the adiabatic flame temperatures and combustion products.
 - For Mg, the temperatures are higher than for Al.
 - Maximum adiabatic temperature: 1417 °C at 26 wt% Mg (equal to the melting point of Si)

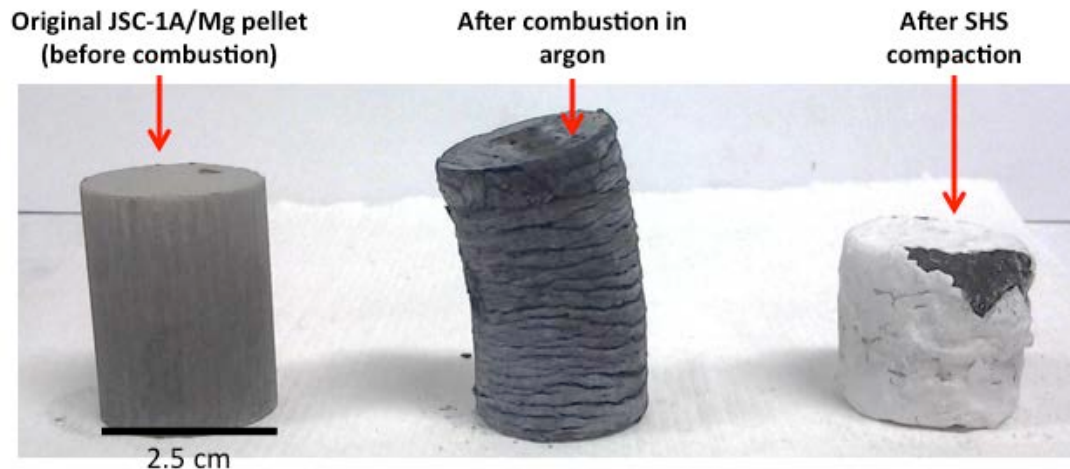


Combustion at 26 wt% Mg



Prior Research of Our Team (contd.)

- Preheating (100°C) decreased content to **8 wt% Mg**.
- SHS compaction increased the density of the products by **66%** as compared to conventional SHS in argon.
- The compressive strength for SHS compaction products is about **10 MPa** – twice stronger than common bricks (5 MPa).



Present Research: Objectives

- Study combustion of **Martian** regolith simulants with Mg and compare it with combustion of JSC-1A **lunar** regolith simulant with Mg.
 - Martian regolith simulants: **JSC-Mars-1A** and **Mojave Mars**
- Clarify the **mechanisms of reactions** that occur during combustion of the lunar and Martian regolith simulants with Mg.



Regolith Simulant Compositions

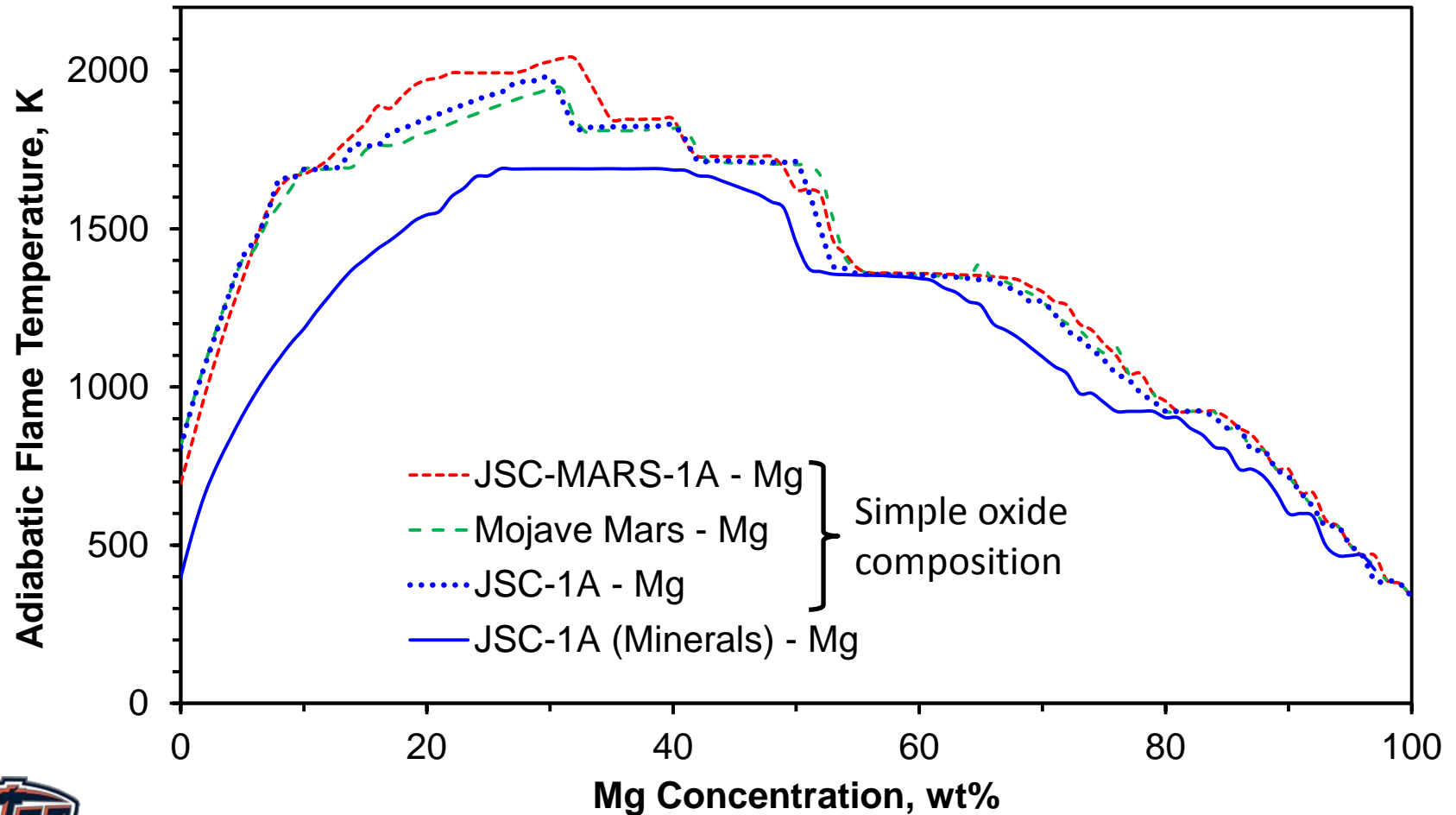
Mineral composition is known only for JSC-1A.
Simple oxide compositions are shown here:

Compound	Concentration, wt%		
	JSC-1A	JSC-Mars-1A	Mojave Mars
SiO₂	45.7	43.48	49.4
Al ₂ O ₃	16.2	22.09	17.1
Fe₂O₃	12.4	16.08	10.87
CaO	10	6.05	10.45
MgO	8.7	4.22	6.08
Na ₂ O	3.2	2.34	3.28
TiO ₂	1.9	3.62	1.09

Main potential reactions:



Thermodynamic Calculations for Combustion of Regolith Simulants with Mg



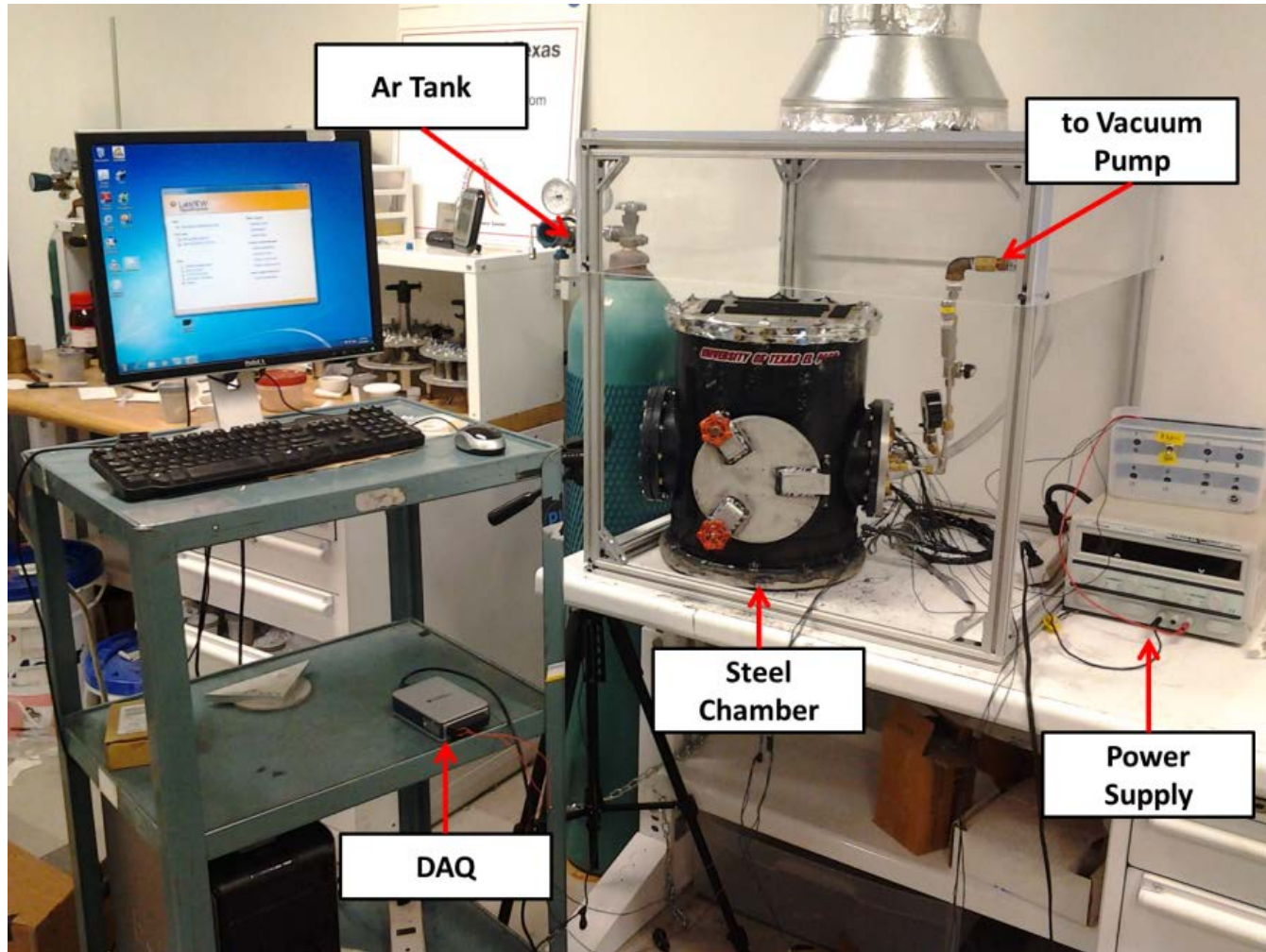
Sample Preparation

- Regolith is milled in a planetary ball mill (1100 rpm, 40 min).
- Regolith is mixed with Mg (10, 20, 30.., wt%).
- The mixtures are compacted into pellets.
 - Mass: 5 g
 - Diameter: 1.3 cm
 - Force: 2 metric tons
- Channel drilled for thermocouple.



Compacted
Powder

Experimental Setup



Mars Regolith/Mg Combustion



JSC-Mars-1A/Mg pellet (30 wt% Mg)



Mojave Mars/Mg pellet (30 wt% Mg)

Conclusions from Combustion Experiments

- JSC-Mars-1A combustion was much more vigorous than for Mojave Mars.
 - Relatively fast, steady propagation of combustion and a uniform structure of the product
- Different combustion behaviors may be related to different **$\text{SiO}_2/\text{Fe}_2\text{O}_3$ ratios**.
- To clarify reaction mechanisms in regolith/Mg mixtures, thermoanalytical experiments should be conducted.



Thermoanalytical Experiments

- To investigate reaction mechanisms of regolith/Mg mixtures.
 - Differential scanning calorimeter (Netzsch DSC 404 F1 Pegasus)
- Examined mixtures:

Regolith Simulant Mixtures

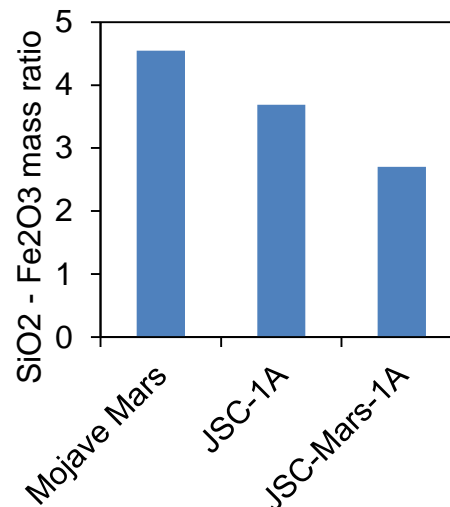
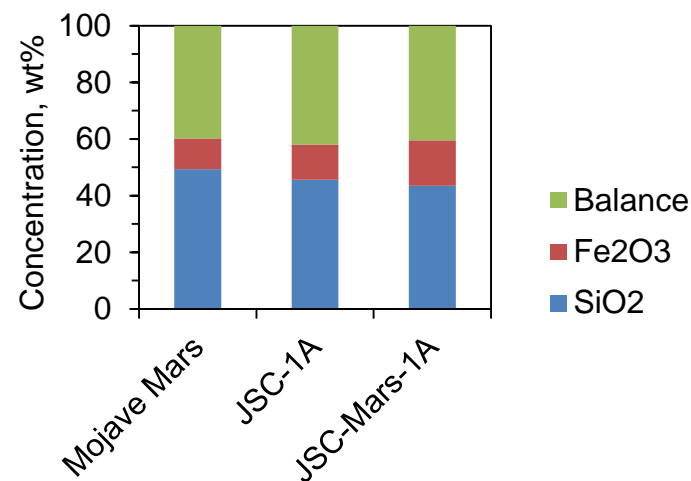
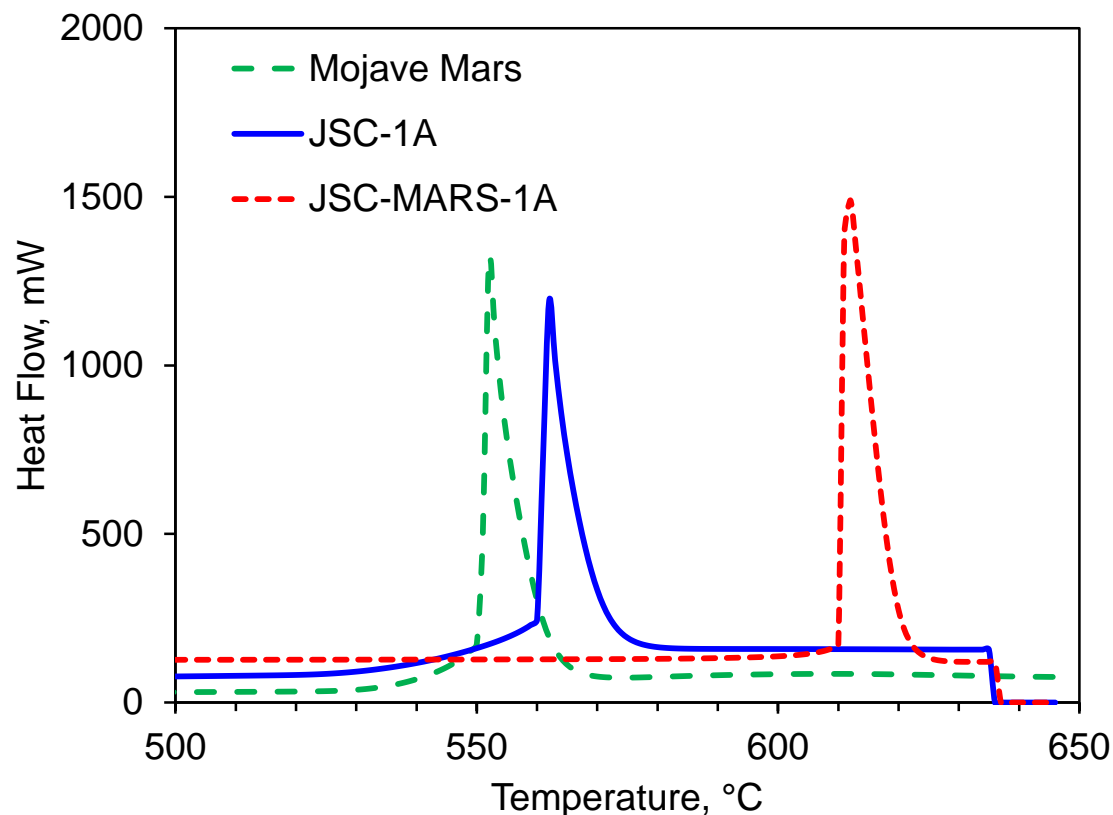
- 26 wt% Mg / 74 wt% JSC-Mars-1A
- 26 wt% Mg / 74 wt% JSC-1A
- 26 wt% Mg / 74 wt% Mojave Mars

Simple Oxide Mixtures

- Mg / $\text{SiO}_2\text{-Fe}_2\text{O}_3$ ($\text{SiO}_2\text{-Fe}_2\text{O}_3$ ratio: 0.5)
- Mg / $\text{SiO}_2\text{-Fe}_2\text{O}_3$ ($\text{SiO}_2\text{-Fe}_2\text{O}_3$ ratio: 1)
- Mg / $\text{SiO}_2\text{-Fe}_2\text{O}_3$ ($\text{SiO}_2\text{-Fe}_2\text{O}_3$ ratio: 2)

Comparable composition with simple oxides

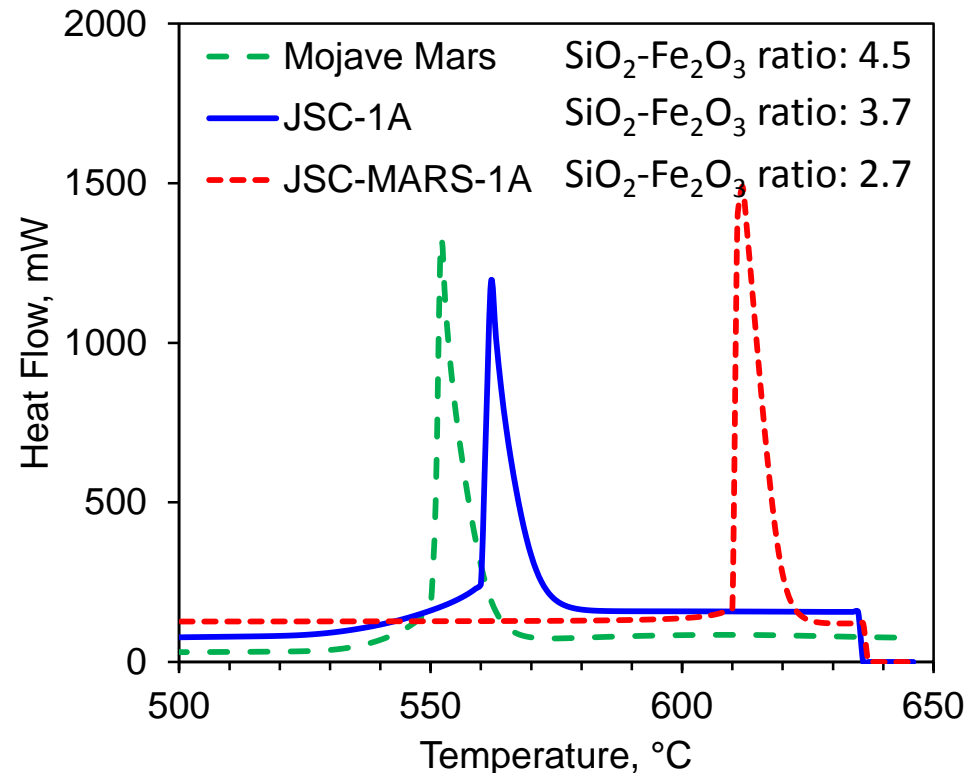
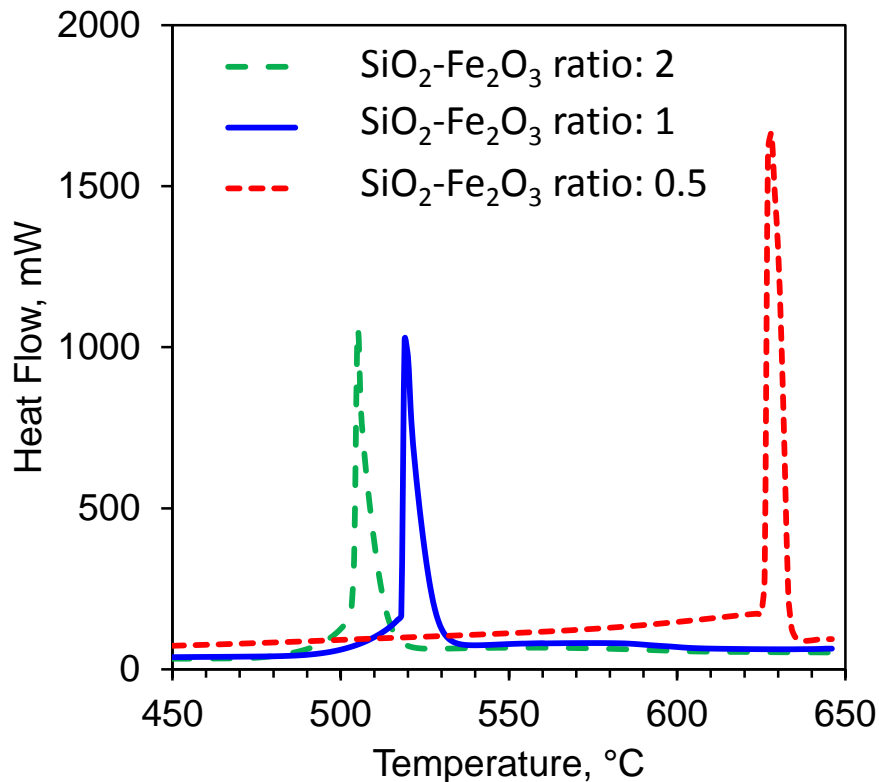
DSC of Regolith Simulants Mixed with Mg



**DSC peaks correlate with SiO₂-Fe₂O₃ ratio:
Increase in this ratio decreases the
temperature of the peak.**



DSC of Mg-Fe₂O₃-SiO₂ Thermites



**Temperature order of the peaks correlates with SiO₂-Fe₂O₃ ratio.
This explains the different peak temperatures of the three simulants.**



Conclusions from Thermoanalytical Experiments

- Iron oxide plays a primary role in combustion of **iron-rich** JSC-Mars-1A simulant with Mg.
 - The iron-rich regolith exhibits higher temperatures and more vigorous combustion owing to the higher exothermicity of $\text{Mg-Fe}_2\text{O}_3$ reaction.
- The effect of silica is significant in combustion of **iron-lean** JSC-1A and Mojave Mars simulants
 - It is easier to ignite the iron-lean regolith simulants because Mg-SiO_2 reaction occurs at a lower temperature.



Summary

- Combustion-based methods for the fabrication of construction materials from lunar and Martian regolith have an advantage of low energy consumption.
- Mixtures of lunar and Martian regolith simulants with Mg exhibit a self-sustained combustion, leading to formation of ceramic materials.
- The reaction mechanisms in these mixtures involve thermite reactions of Mg with silica and iron oxide.
 - Iron oxide ensures intensive combustion.
 - Silica facilitates the ignition.



Acknowledgements

- The NASA Office of Education for support of this research
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Thank you!



BACK-UP SLIDES



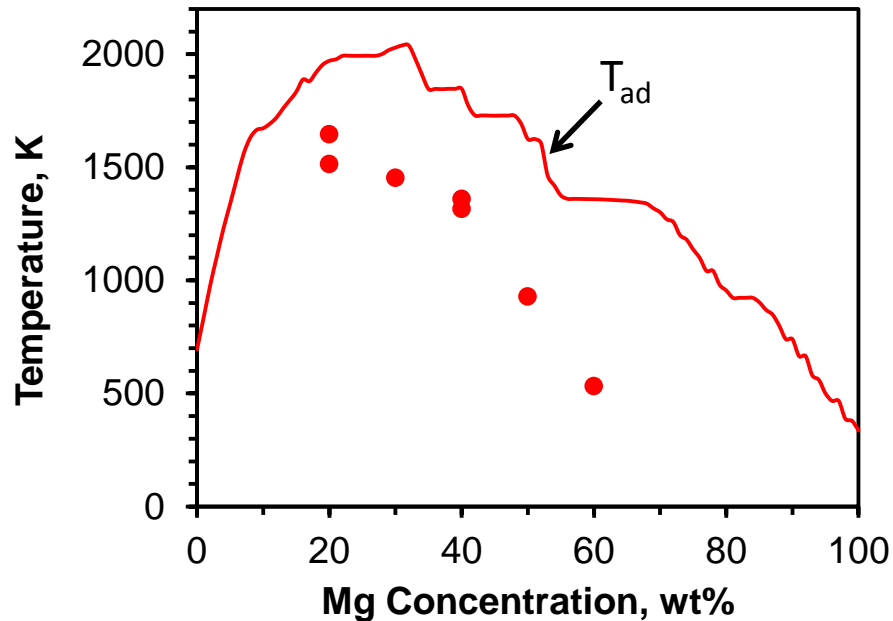
Spin Combustion



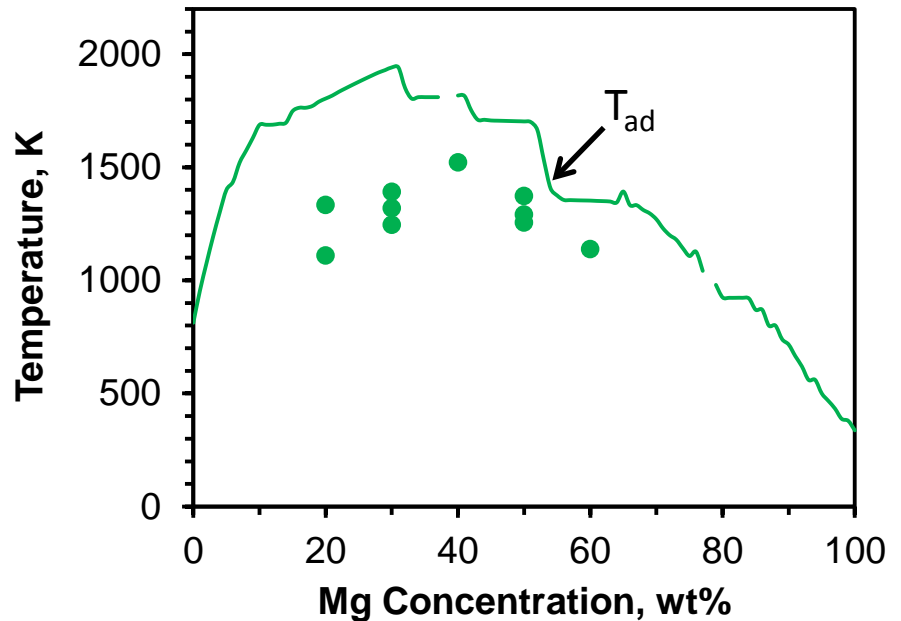
Two counter-propagating hot spots

Maximum Temperature in the Combustion Wave

JSC-Mars-1A



Mojave Mars

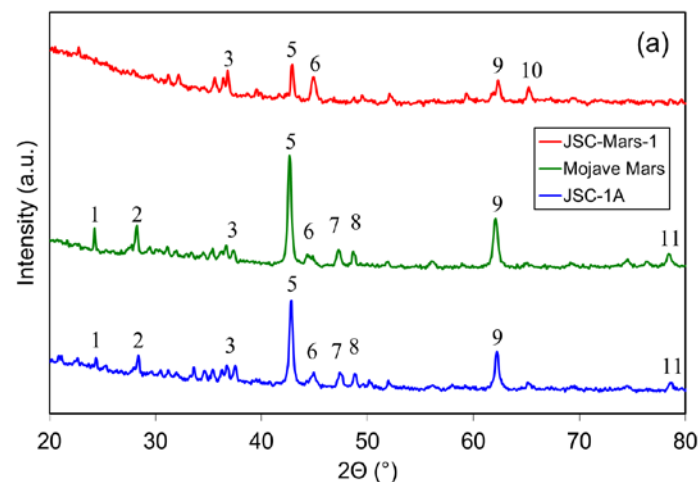


- Reasonable agreement between experimental values and calculated adiabatic flame temperatures

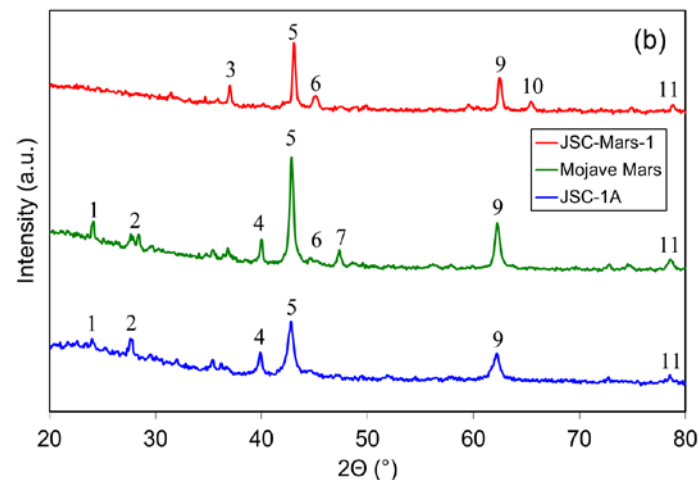
XRD of Combustion Products

Peak	Phases			
1	$\text{Ca}_2\text{MgSi}_2\text{O}_7$	CaMgSiO_4	Mg_2Si	
2	MgAl_2O_4	CaMgSiO_4	FeSi	Mg_2Si
3	MgAl_2O_4	MgO	MgO_2	
4	Al_2O_3	FeSi	Fe	Mg_2Si
5	MgO			
6	MgAl_2O_4	FeSi		
7	Al_2O_3	Mg_2Si		
8	Al_2O_3			
9	MgO	FeSi		
10	MgAl_2O_4	$\text{Ca}_2\text{MgSi}_2\text{O}_7$	Al_2O_3	Fe
11	MgO			

20 wt% Mg / 80 wt% Regolith



26 wt% Mg / 74 wt% Regolith

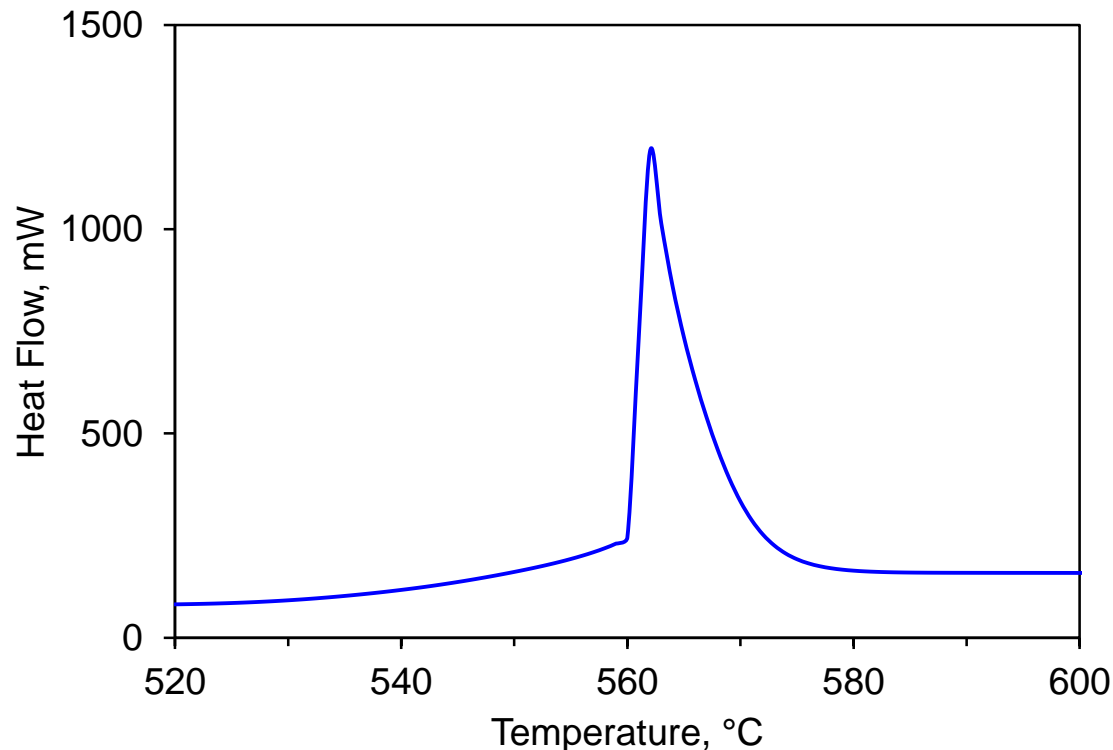


Combustion Products from THERMO

Compound		State	Concentration, wt%					
			20 wt% Mg			26 wt% Mg		
			80 wt% JSC-1A	80 wt% JSC-Mars-1A	80 wt% Mojave Mars	74 wt% JSC-1A	74 wt% JSC-Mars-1A	74 wt% Mojave Mars
MgO		S	29.4	2.0	7.9	32.8	30.4	31.3
Al ₂ Si ₂ O ₁₃	Al ₂ O ₃ ·(SiO ₅) ₂	S	31.0	-	-	25.7	-	-
Al ₂ MgO ₄	Al ₂ O ₃ ·MgO	S	-	25.2	19.3	-	23.3	17.9
Mg ₂ SiO ₄	(MgO) ₂ ·SiO ₂	S	-	45.8	42.5	-	9.9	18.3
Ca ₂ SiO ₄	(CaO) ₂ ·SiO ₂	S	-	7.6	13.0	-	2.5	12.0
CaMgSi ₂ O ₆	CaO·MgO·(SiO ₂) ₂	L	-	-	-	-	11.5	-
CaMgO ₂	CaO·MgO	S	10.9	-	-	9.2	-	-
FeSi		S	10.6	-	-	9.8	-	-
FeSi		L	-	13.8	9.3	-	12.8	8.6
Si		S	8.1	-	-	5.5	-	-
Si		L	-	0.9	4.8	-	5.0	8.6
Si ₃ Ti ₅		S	1.3	2.4	0.7	1.2	2.2	0.7
Mg ₂ Si		L	-	-	-	6.8	-	-
Al ₂ Ca		L	3.0	-	-	3.7	-	-
Mg		G	3.8	0.7	0.2	3.5	0.9	0.5
Na		G	1.9	1.4	2.3	1.8	1.3	2.1
SiO		G	-	0.2	-	-	0.2	0.1

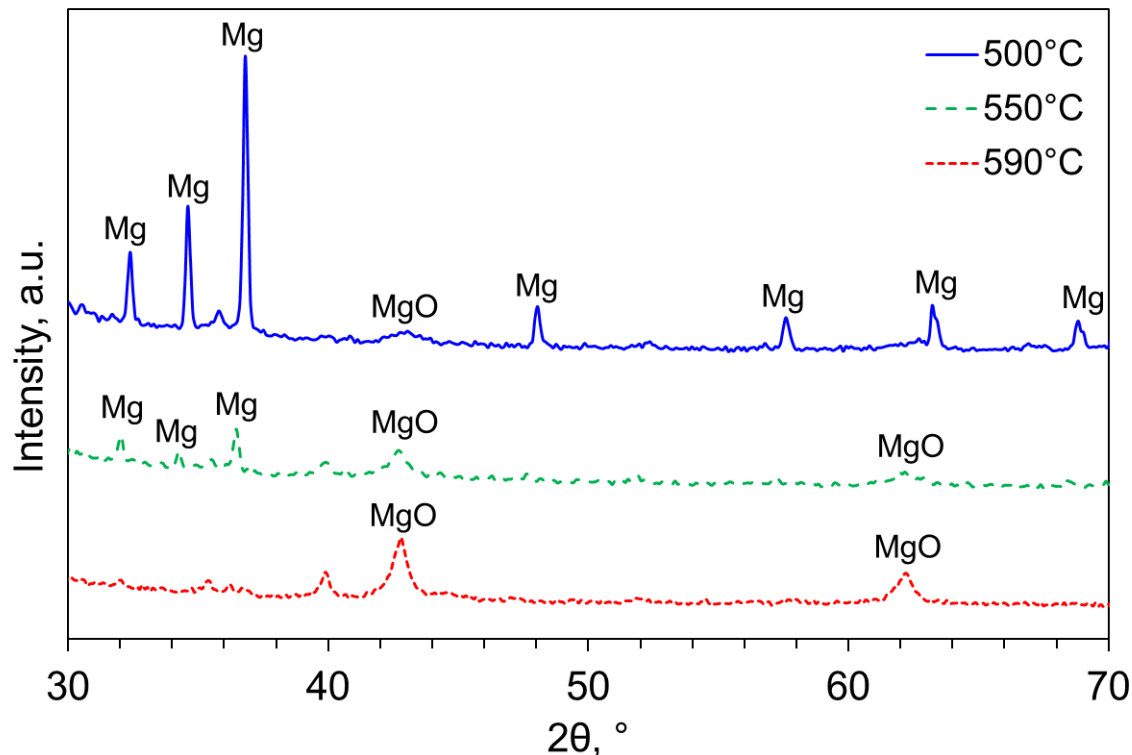
DSC of JSC-1A/Mg (26 wt% Mg)

- Differential scanning calorimetry (DSC) curve
- Heating rate: 10°C/min
- DSC curve shows exothermic peak at 560°C.

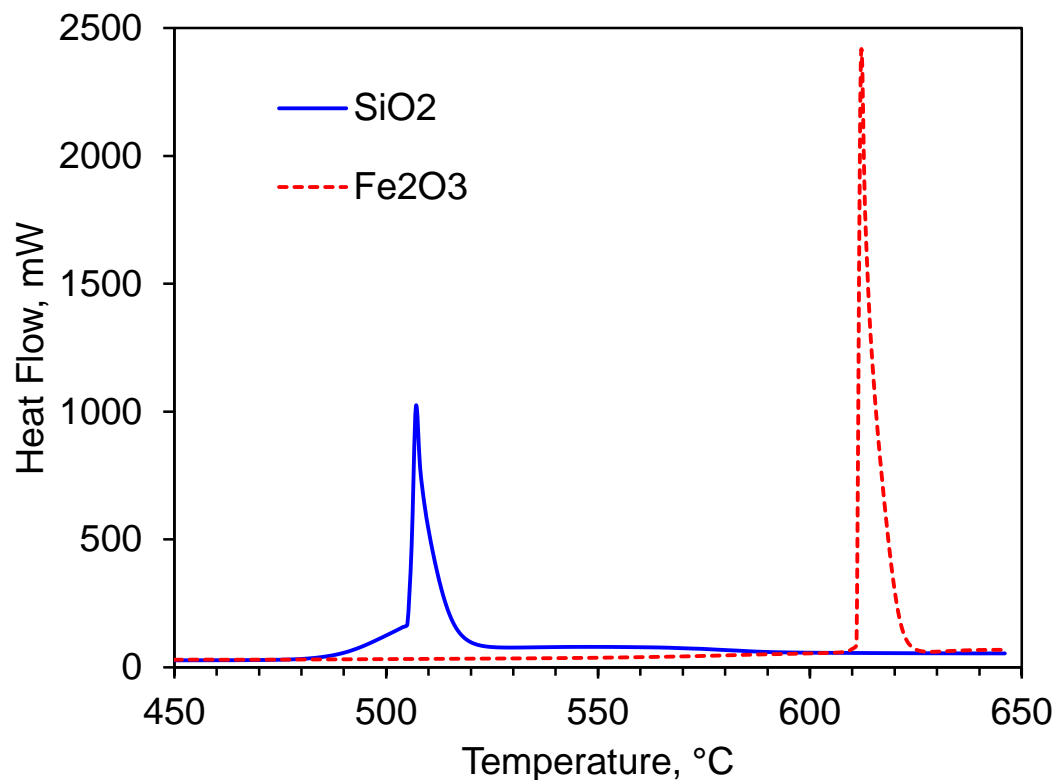


XRD of Reaction Products at Different Temperatures

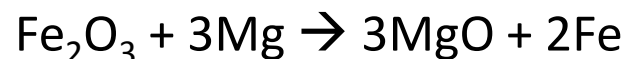
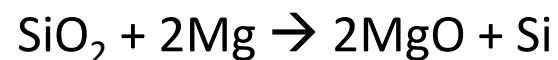
- To investigate reaction, analysis was stopped at 500°C, 550°C, and 590°C.
- Heating rate: 5°C/min
- The reaction is complete at a temperature between 550°C and 590°C.
- Magnesium is solid throughout reaction ($T_{\text{melting, Mg}} = 650^\circ\text{C}$).



DSC of Mg-Fe₂O₃ and Mg-SiO₂ Thermites



Stoichiometric mixtures of



The peak temperature for Mg-Fe₂O₃ is higher than for Mg-SiO₂.