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systems Division**

LRRR 300 Corner Array
Dynamic Analysis

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ATM-936	
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DATE 11 January 1971	

This ATM presents the results of the structural dynamics analysis performed on the Lunar Ranging Retro Reflector 300 Corner Array.

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1.0	Summary
2.0	Description of Dynamic Models
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4.0	Dynamic Loads

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- (2) "Vibration Tests of Single Retro-Reflector Assemblies", Arthur D. Little, Inc., Cambridge, Massachusetts, December 31, 1970
- (3) Leonard Meirovitch, Analytical Methods in Vibrations, The MacMillan Company, New York, 1967
- (4) AL 410900, "Exhibit B, Design and Performance Specification for the Retro-Reflector Array Panels," Bendix Aerospace Systems Division, Ann Arbor, Michigan, November 1, 1970



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1.0 Summary

1.1 Motivation of Analysis

The vibrations Analysis of the Lunar Ranging Retro Reflector 300 Corner Array (LRRR 300) was performed to answer four basic questions.

These questions are:

1. What loads are imposed on the LRRR 300 structure during qualification and acceptance testing, during launch and boost and during lunar landing? Are the loads assumed by the stress group adequate to ensure structural reliability?
2. What is the worst vibration environment seen by any single corner reflector mount? Does this vibration environment impose loads on any corner reflector mount that will cause either the reflector or the mount to fail?
3. What vibration test spectrum should be used at the small array attachment fittings when the small array is separately tested?
4. If the experiment is flown with the upper array removed (the off loaded condition) will the structure and the corner reflector mounts survive?

To answer the four above questions, it was necessary to analyze three different mathematical models of the LRRR 300 structure. One math model predicted vibration response perpendicular to the plane of the two arrays. The second model predicted vibration response in the plane of the arrays. The final model was constructed to find vibration response of the experiment in the off loaded condition.

1.2 Summary of Conclusions Drawn from Analysis

The analysis that was performed to answer the questions of paragraph 1.1 and the reasoning behind the answers make up the body of this report. The four paragraphs that follow give very brief summaries of the answers arrived at.



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1. With the two arrays connected, vibration responses were such that the dynamic loads and deflections seen in the array structure were small (see Table 4.3.1). The dynamic environment therefore produced low stress levels in the array structure. The stress levels seen in the array structure are discussed in the "Structural Analysis Report" (1). In addition, analysis showed that the reaction loads carried through the LM interface fittings were well under the loads that the fittings were designed to carry (see Table 4.3.2 and (1)).
2. The analysis also showed that, when the experiment is subjected to its worst vibration environment (launch and boost qualification test), the worst 3 sigma response at a corner reflector mount in the array plane was 40 g. The worst 3 sigma response at a corner reflector mount perpendicular to the array plane was 48 g. Single corner reflector mounts have been subjected to sinusoidal vibration levels in excess of 50 g peak for considerable periods of time (2).
3. Vibration levels on the lower array at the attachments of the upper array were calculated both in the array plane and perpendicular to it. Using these calculated vibration levels, test levels for the upper array are proposed which are envelopes of the calculated responses. These envelope test levels are shown in Figures 3.1.7 and 3.2.7 of this report.
4. Analysis of the lower array in the "off loaded condition" showed that acceleration responses on the lower array are reduced by removing the upper array. Thus removing the upper array reduces loads seen in the lower array structure and reduces the acceleration levels seen by the corner reflector mounts.



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2.0 Description of Dynamic Models

2.1 Dynamic Model Perpendicular to the Array Plane

The dynamic model of the structure perpendicular to the array was a lumped mass type model. The large array was modeled with 23 lumped masses and the small array was modeled with 8 lumped masses. Each lumped mass point was given one degree of freedom, motion perpendicular to the array plane. With 1 constraint, this gave a total of 30 degrees of freedom for the model. The locations of the lumped mass points on the lower array are shown in Figure 2.1.1 and on the upper array in Figure 2.1.2.

The structure consists of the corner reflector support cavity structure with a network of cross beams below. The upper array is about half as large as the lower array and is attached over one side of the lower array at four points as shown in figure 2.1.3.

The stiffness of the arrays between lumped mass points was evaluated by developing an equivalent beam stiffness for the cavity support structure and the lower ribs. The structure of each array was then modeled as a grid of beams, and a stiffness matrix for each separate array was found using Bendix Program No. 0313, Rigid Grid Analysis by Direct Element Method. The interface fittings, the hinges, and the clevis fittings were assumed to be very stiff compared to the arrays in this direction. The fittings were modeled as rigid links. The stiffness matrices found by this method are listed in Figure 2.1.4.

The mass matrix for the whole system was then evaluated. Thirty one lumped mass points were chosen as shown on Figure 2.1.1 and 2.1.2. The mass lumped at each location is shown in Table 2.1.1.

Again referring to Figures 2.1.1 and 2.1.2, it is seen that mass locations 3 and 16 are repeated on both arrays. This because the two arrays are connected by hinges at these points and the upper and lower arrays have the same motion. The clevis locations on both arrays are also connected, but no mass is lumped at the clevis locations on the upper array.

The stiffness matrix for the upper array was calculated with the four attachments between the arrays fixed in a plane. Since a plane is determined by 3 points, the displacement of any one of the 4 attachment fittings is determined by the other 3. Therefore 30 independent coordinates describe the system. The dependent coordinate chosen here is the displacement of the mass lumped at the rear clevis fitting (shown dotted). The mass matrix for the independent system was calculated

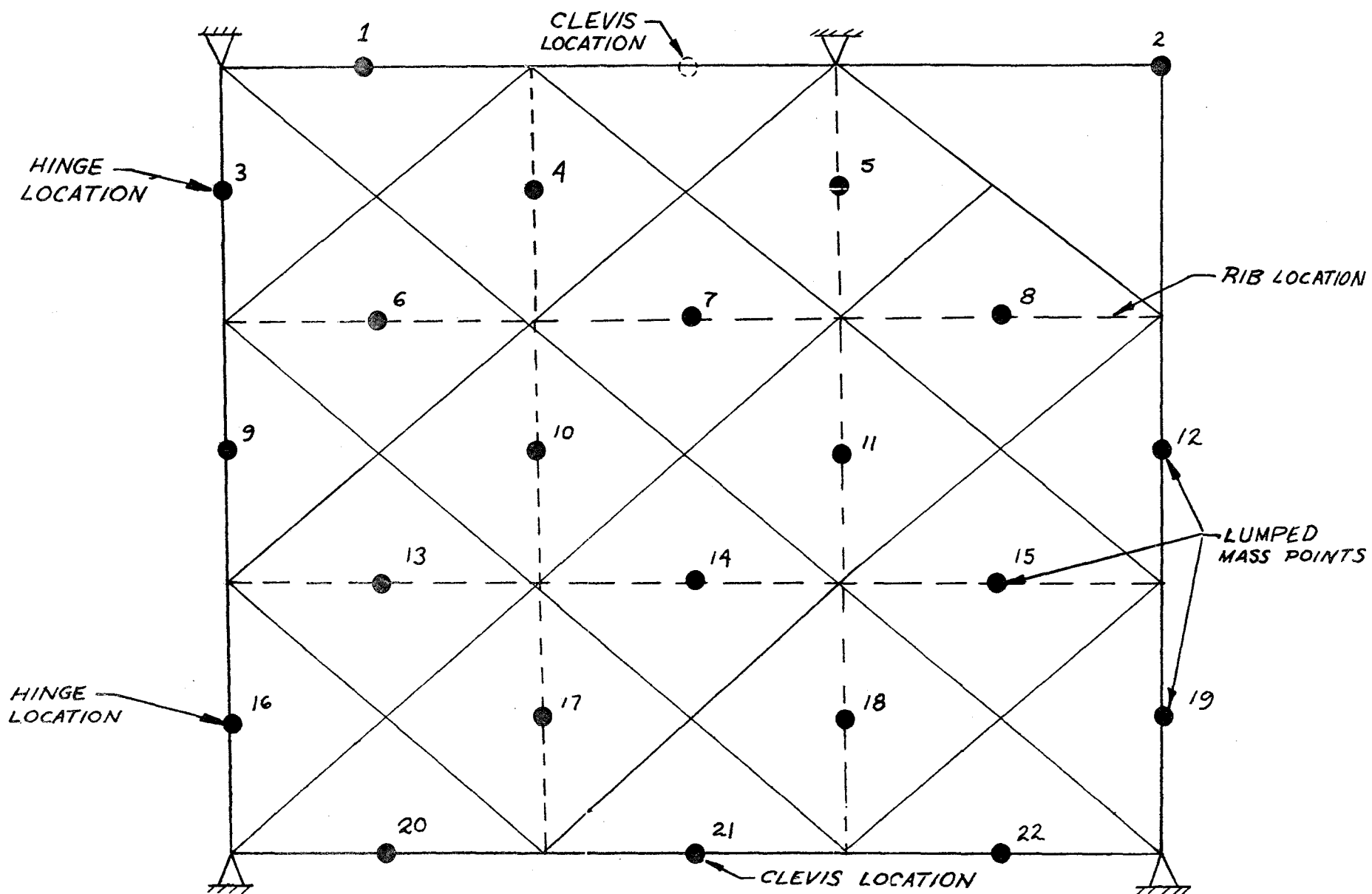


Figure 2.1.1
OUT OF PLANE DYNAMIC MODEL
LOWER ARRAY

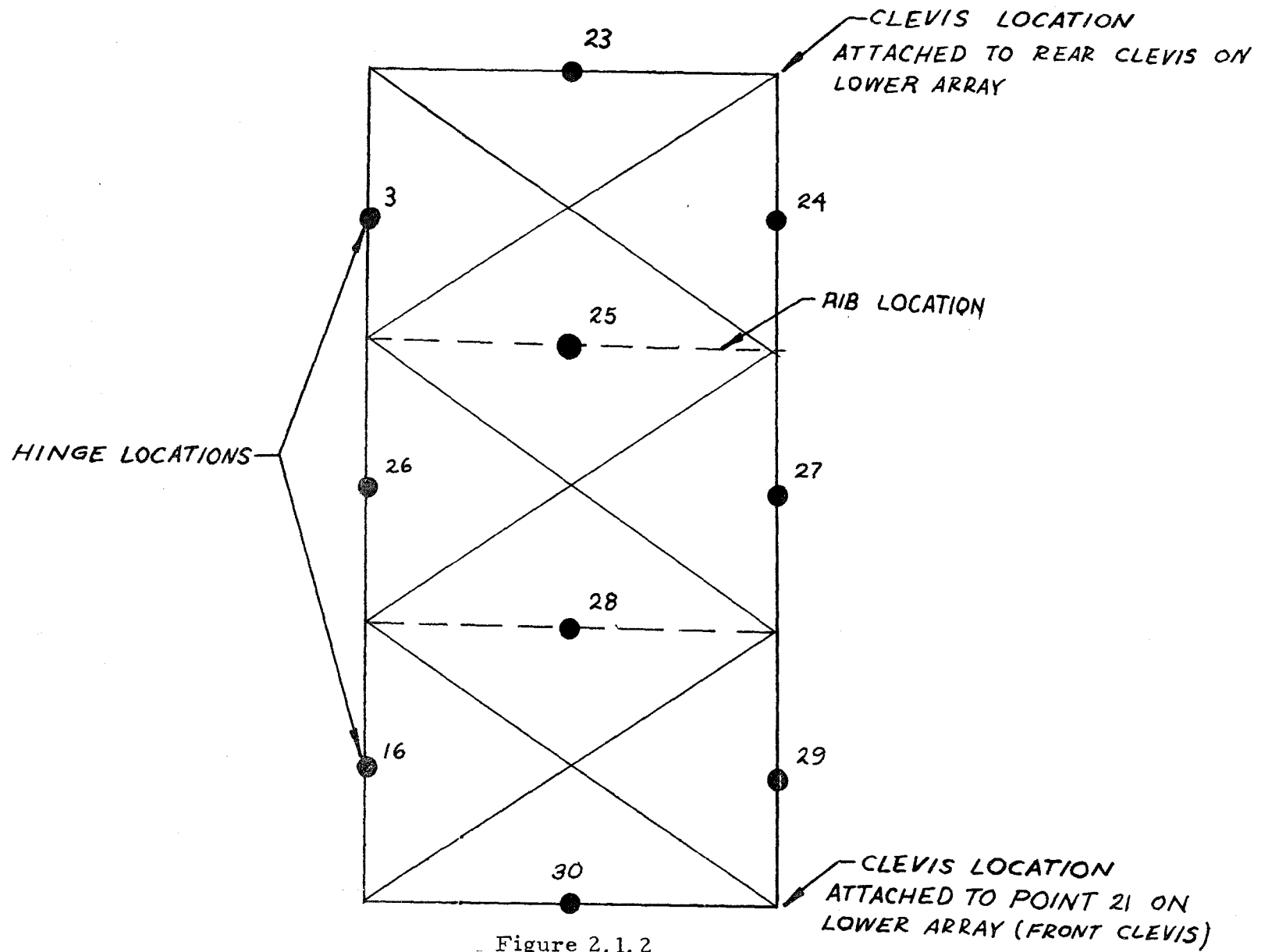


Figure 2.1.2

OUT OF PLANE DYNAMIC MODEL
UPPER ARRAY

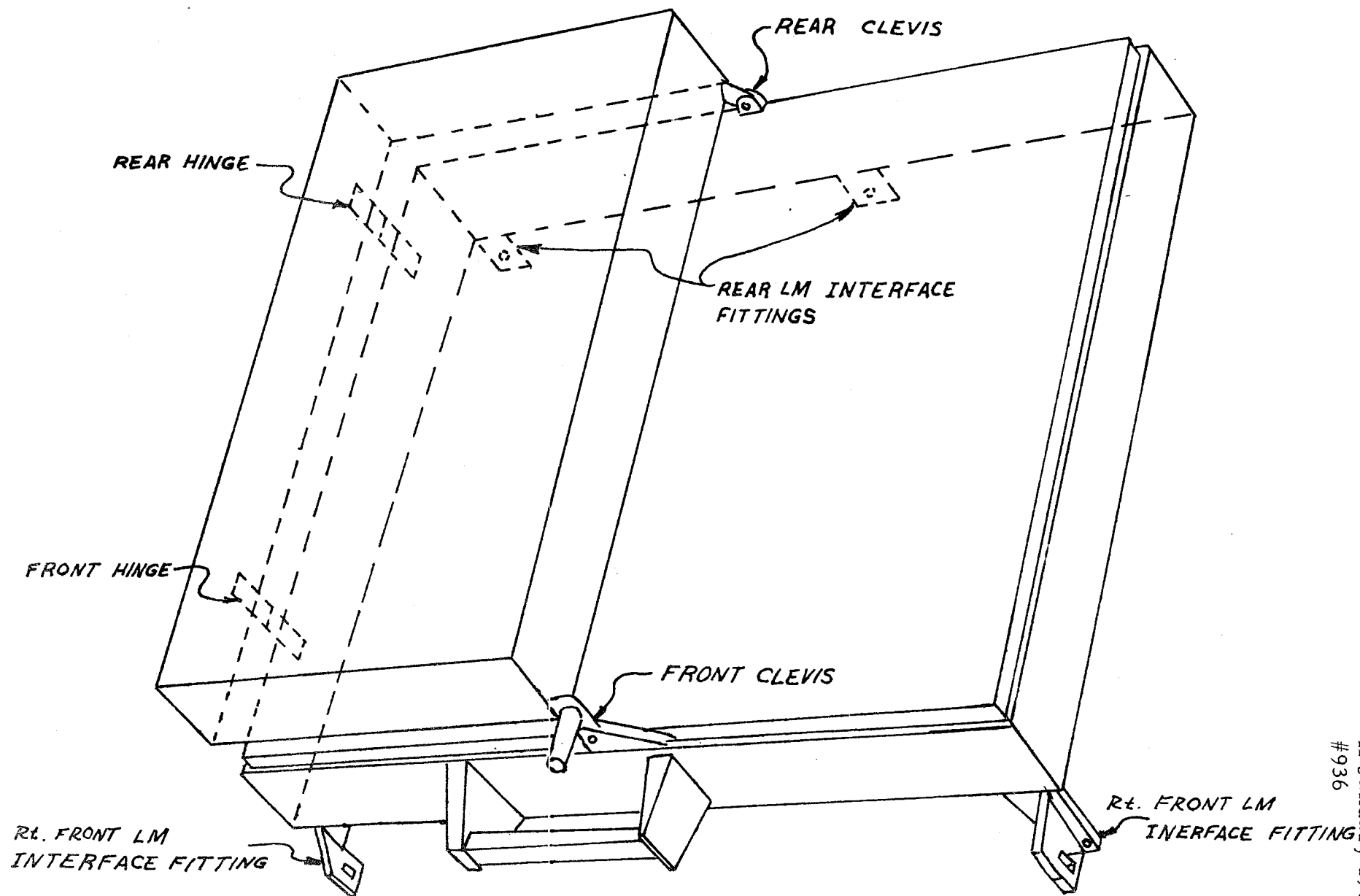


Figure 2.1.3
Fittings Between Array

SMALL K MATRIX

Figure 2.1.4

ROW 1

0.526007E 06	0.619771E 04	-0.258751E 05	-0.586831E 06	-0.332139E 05	0.137783E 06	0.151968E 06	0.792879E 03
-0.179200E 05	0.108560E 06	-0.247187E 05	0.832466E 03	-0.182693E 05	-0.183464E 05	0.368200E 04	0.839743E 04
-0.158291E 05	0.586827E 04	-0.137728E 04	0.420651E 04	0.241131E 04	-0.186492E 04		

ROW 2

0.619755E 04	0.451707E 05	0.207798E 04	0.416698E 05	0.422768E 05	-0.104759E 05	-0.110820E 05	-0.997391E 05
0.135560E 04	-0.854196E 04	0.325356E 05	0.178036E 04	0.166135E 04	-0.311063E 04	0.794736E 04	-0.755985E 03
0.205895E 04	-0.768521E 04	0.236573E 05	-0.715156E 03	0.966373E 03	0.163914E 04		

ROW 3

-0.258758E 05	0.207816E 04	0.796714E 06	0.124173E 06	-0.110109E 05	-0.402136E 06	0.114931E 05	-0.178587E 03
-0.308350E 06	0.126711E 06	-0.336583E 04	0.183522E 03	0.530359E 05	-0.198692E 05	0.156840E 04	0.135222E 06
-0.379419E 05	0.415821E 04	-0.722393E 03	0.760988E 04	0.586764E 04	-0.199680E 04		

ROW 4

-0.586831E 06	0.416697E 05	0.124171E 06	0.137515E 07	0.157231E 06	-0.656782E 06	-0.719730E 06	-0.419751E 04
0.926028E 05	0.623579E 05	0.127005E 06	-0.423432E 04	0.455632E 05	0.430710E 05	-0.168227E 05	-0.324326E 05
0.465149E 05	-0.213435E 05	0.600157E 04	-0.122832E 05	-0.591309E 04	0.655343E 04		

ROW 5

-0.332139E 05	0.422763E 05	-0.110095E 05	0.157230E 06	0.222509E 07	0.107649E 05	-0.935605E 06	-0.781997E 06
-0.360921E 04	0.172812E 06	-0.128168E 06	0.183392E 06	-0.247955E 05	0.732523E 05	0.622218E 05	0.742454E 04
-0.319240E 05	0.784836E 05	-0.729841E 05	0.100412E 05	-0.105942E 05	-0.202815E 05		

ROW 6

0.137783E 06	-0.104762E 05	-0.402131E 06	-0.656782E 06	0.107701E 05	0.142312E 07	0.283088E 06	0.181667E 04
-0.349771E 06	-0.833102E 06	-0.148442E 05	0.738673E 02	0.139974E 06	0.108985E 06	-0.209923E 04	0.523022E 05
0.583956E 05	-0.166617E 05	0.161691E 04	-0.170422E 05	-0.118346E 05	0.551046E 04		

ROW 7

0.151971E 06	-0.110823E 05	0.114928E 05	-0.719737E 06	-0.935610E 06	0.283094E 06	0.196371E 07	0.337456E 06
-0.952442E 04	-0.919661E 06	-0.867934E 06	-0.154802E 05	0.109928E 06	0.229610E 06	0.101326E 06	-0.211802E 05
0.623256E 05	0.539382E 05	-0.175639E 05	-0.152709E 05	-0.290374E 05	-0.129226E 05		

ROW 8

0.794101E 03	-0.997381E 05	-0.177781E 03	-0.419823E 04	-0.781992E 06	0.181503E 04	0.337455E 06	0.126060E 07
0.552431E 03	-0.249023E 05	-0.732704E 06	-0.514559E 06	-0.755012E 02	0.928844E 05	0.161395E 06	0.745074E 03
-0.125785E 05	0.337266E 05	0.124712E 06	0.386537E 04	-0.804031E 04	-0.117650E 05		

ROW 9

-0.179195E 05	0.135555E 04	-0.308351E 06	0.926002E 05	-0.361156E 04	-0.349767E 06	-0.952150E 04	0.551710E 03
0.624256E 06	0.248675E 06	-0.755613E 04	0.239801E 03	-0.352765E 06	-0.140040E 05	-0.536774E 03	-0.307700E 06
0.103841E 06	0.129098E 04	0.212649E 03	-0.157028E 05	-0.141233E 05	0.238393E 04		

ROW 10

0.108556E 06	-0.854108E 04	0.126708E 06	0.623666E 05	0.172810E 06	-0.833105E 06	-0.919660E 06	-0.249055E 05
0.248678E 06	0.219461E 07	0.351175E 06	-0.721879E 04	-0.816865E 06	-0.888690E 06	-0.118297E 05	0.123250E 06
-0.630851E 03	0.123725E 06	-0.411665E 04	0.981496E 05	0.655984E 05	-0.344919E 05		

ROW 11

-0.249232E 05	0.325351E 05	-0.336795E 04	0.127015E 06	-0.128169E 06	-0.148442E 05	-0.867942E 06	-0.732695E 06
-0.755619E 04	0.351173E 06	0.220552E 07	0.283476E 06	-0.122681E 05	-0.892186E 06	-0.829805E 06	-0.414727E 04
0.123871E 06	0.800029E 03	0.105419E 06	-0.344836E 05	0.655688E 05	0.981232E 05		

ROW 12

0.831960E 03	0.178041E 04	0.183608E 03	-0.423385E 04	0.183386E 06	0.721615E 02	-0.164810E 05	-0.514555E 06
0.238248E 03	-0.722023E 04	0.289482E 06	0.501962E 06	-0.158248E 03	-0.202744E 05	-0.331019E 06	0.400066E 02
0.248836E 04	0.903959E 05	-0.256256E 06	0.174709E 04	-0.120499E 05	-0.130786E 05		

Fig 2.1.4 (c)

[illegible]

DW									14
-0.183419E 05	-0.311653E C4	-0.198653E 05	0.430621E 05	0.732547E 05	0.108985E 06	0.229605E 06	0.928766E 05		
-0.140068E 05	-0.888678E 06	-0.892174E 05	-0.202712E 05	0.312462E 06	0.196914E 07	0.314761E 06	0.581071E 04		
-0.840621E 06	-0.840577E C6	0.849010E 04	0.116442E 06	0.238029E 06	0.116378E 06				

[illegible][illegible][illegible][illegible]

0M	19								
-0.137815E 04	0.236575E 05	-0.721833E 03	0.600419E 04	-0.729789E 05	0.161447E 04	-0.175655E 05	0.124711E 06		
-0.213244E 03	-0.411779E 04	0.105416E 06	-0.256299E 06	-0.232679E 04	0.848874E 04	-0.415332E 06	0.938044E 03		
-0.347049E 04	0.146537E 06	0.774816E 06	0.461449E 04	-0.205537E 05	-0.240676E 05				

[illegible][illegible][illegible]

K_V MATRIX

Figure 2.14 (cont.)

ROW 1	0.219645E 06	-0.518918E C5	-0.143117E 05	0.366075E 05	0.214080E 04	0.296894E 05	0.710589E 03	-0.247245E 04
	-0.219645E 06	0.518918E 05	0.143117E 05	-0.366075E 05	-0.214080E 04	-0.296894E 05	-0.710589E 03	0.247245E 04
ROW 2	-0.518919E 05	0.798519E 06	-0.201561E 06	0.510754E 05	-0.280859E 06	0.333917E 05	0.133188E 06	0.215182E 04
	0.518919E 05	-0.798519E 06	0.201561E 06	-0.510754E 05	0.280859E 06	-0.333917E 05	-0.133188E 06	-0.215182E 04
ROW 3	-0.143121E C5	-0.201562E C6	0.701185E 06	-0.205677E 06	-0.182932E 06	-0.120458E 06	0.402414E 05	0.275100E 05
	0.143121E 05	0.201562E C6	-0.701185E 06	0.205677E 06	0.182932E 06	0.120458E 06	-0.402414E 05	-0.275100E 05
ROW 4	0.366075E C5	0.510757E C5	-0.205677E 06	0.271213E 06	0.129620E 06	-0.179171E 06	0.340586E 05	0.558781E 05
	-0.366075E 05	-0.510757E 05	0.205677E 06	-0.271213E 06	-0.129620E 06	0.179171E 06	-0.340586E 05	-0.558781E 05
ROW 5	0.214081E C4	-0.280859E C6	-0.182933E 06	0.129620E 06	0.411428E 06	-0.162511E 06	-0.269076E 06	0.292225E 04
	-0.214081E 04	0.280859E C6	0.182933E 06	-0.129620E 06	-0.411428E 06	0.162511E 06	0.269076E 06	-0.292225E 04
ROW 6	0.296896E C5	0.333914E C5	-0.120457E 06	-0.179172E 06	-0.162511E 06	0.711210E 06	-0.196401E 06	-0.134845E 05
	-0.296896E 05	-0.333914E C5	0.120457E 06	0.179172E 06	0.162511E 06	-0.711210E 06	0.196401E 06	0.134845E 05
ROW 7	0.710660E 03	0.133187E 06	0.402413E 05	0.340586E 05	-0.269076E 06	-0.196401E 06	0.617360E 06	-0.419941E 05
	-0.710660E 03	-0.133187E 06	-0.402413E 05	-0.340586E 05	0.269076E 06	0.196401E 06	-0.617360E 06	0.419941E 05
ROW 8	-0.247247E 04	0.215187E C4	0.275100E 05	0.558781E 05	0.292227E 04	-0.134843E 05	-0.419940E 05	0.218861E 06
	0.247247E C4	-0.215187E C4	-0.275100E 05	-0.558781E 05	-0.292227E 04	0.134843E 05	0.419940E 05	-0.218861E 06
ROW 9	-0.219645E 06	0.518918E C5	0.143117E 05	-0.366075E 05	-0.214080E 04	-0.296894E 05	-0.710589E 03	0.247245E 04
	0.219645E 06	-0.518918E 05	-0.143117E 05	0.366075E 05	0.214080E 04	0.296894E 05	0.710589E 03	-0.247245E 04
ROW 10	0.518919E 05	-0.798519E 06	0.201561E 06	-0.510754E 05	0.280859E 06	-0.333917E 05	-0.133188E 06	-0.215182E 04
	-0.518919E 05	0.798519E 06	-0.201561E 06	0.510754E 05	-0.280859E 06	0.333917E 05	0.133188E 06	0.215182E 04
ROW 11	0.143121E C5	0.201562E 06	-0.701185E 06	0.205677E 06	0.182932E 06	0.120458E 06	-0.402414E 05	-0.275100E 05
	-0.143121E 05	-0.201562E 06	0.701185E 06	-0.205677E 06	-0.182932E 06	-0.120458E 06	0.402414E 05	0.275100E 05
ROW 12	-0.366075E C5	-0.510757E C5	0.205677E 06	-0.271213E 06	-0.129620E 06	0.179171E 06	-0.340586E 05	-0.558781E 05
	0.366075E 05	0.510757E C5	-0.205677E 06	0.271213E 06	0.129620E 06	-0.179171E 06	0.340586E 05	0.558781E 05
ROW 13	-0.214081E C4	0.280859E C6	0.182933E 06	-0.129620E 06	-0.411428E 06	0.162511E 06	0.269076E 06	-0.292225E 04
	0.214081E 04	-0.280859E C6	-0.182933E 06	0.129620E 06	0.411428E 06	-0.162511E 06	-0.269076E 06	0.292225E 04
ROW 14	-0.296896E C5	-0.333914E C5	0.120457E 06	0.179172E 06	0.162511E 06	-0.711210E 06	0.196401E 06	0.134845E 05
	0.296896E 05	0.333914E 05	-0.120457E 06	-0.179172E 06	-0.162511E 06	0.711210E 06	-0.196401E 06	-0.134845E 05
ROW 15	-0.710660E 03	-0.133187E 06	-0.402413E 05	-0.340586E 05	0.269076E 06	0.196401E 06	-0.617360E 06	0.419941E 05
	0.710660E 03	0.133187E 06	0.402413E 05	0.340586E 05	-0.269076E 06	-0.196401E 06	0.617360E 06	-0.419941E 05

Ku Matrix con.
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Figure 2.1.1 (cont.)

0.247247E 04	-0.215187E 04	-0.275100E 05	-0.558781E 05	-0.292227E 04	0.134843E 05	0.419940E 05	-0.218861E 06
-0.247247E 04	0.215187E 04	0.275100E 05	0.558781E 05	0.292227E 04	-0.134843E 05	-0.419940E 05	0.218861E 06

Table 2.11
Lumped Masses

MASS LUMPED FOR LOWER ARRAY

MASS LUMPED FOR UPPER ARRAY

Station	Weight (Lbs)	Attached Weight (Lbs)	Total Weight (Lbs)	Mass Lumped At Location $\frac{\text{Lb}}{\text{Sec.}^2}$ In	Remark (For Attached Weight)	Mass Joint Location	Synthesized Mass Numbering	Weight* On Each Location	Attached Weight (Lbs)	Total Weight On Each Location	Mass Lumped $\frac{\text{Lbs. Sec}^2}{\text{In}}$	Remark (For Attached Weight)
1.37	.45 + .5		2.32	.00601	(1/2) Leg Hinges	2	23	1.5	.1	1.6	.00414	(1/4) Tie Points
					(1/8) Leveling Leg Assy	4	3**	2.02	.2	2.22	.00575	(1/2) Hinges
1.344	.45 + .5 + .3		2.594	.00672	(1/2) Leg Hinges	6	24	2.02	.1	2.12	.00549	(1/4) Tie Points
					(1/8) Leveling Leg Assy	8	25	4.55		4.55	.01178	
2.74	.1		2.84	.00735	+(1/2) Tie Pts.	10	26	2.02		2.02	.00523	
1.37	.2		1.57	.00406	Back Rest	12	27	2.02		2.02	.00523	
2.714	1.0		3.714	.00962	(1/2) Array Hinges	14	28	4.11		4.11	.01064	
2.714			2.714	.00703	(1/4) Leveling Leg Assy	16	16**	2.02	.2	2.02	.00523	(1/2) Hinges
2.74			2.74	.00709		18	29	2.02	.1	2.02	.00523	(1/4) Tie Points
2.688			2.688	.00696		20	30	1.94	.1	1.94	.00502	(1/4) Tie Points
2.74			2.74	.00709								
1.37			1.37	.00354								
2.714	1.0		3.714	.00962	(1/4) Leveling Leg Assy							
2.714			2.714	.00703								
1.37			1.37	.00354								
2.74			2.74	.00709								
2.688			2.688	.00696								
2.74			2.74	.00709								
1.37	.2		1.57	.00406	(1/2) Array Hinges							
2.714	1.0		3.714	.00962	(1/4) Leveling Leg Assy							
2.714			2.714	.00703								
1.37			1.37	.00354								
1.37	.375		1.745	.00452	(1/2) Pins & Name Pl							
1.344	.5 + .3		2.144	.00555	Handle & Handling							
					Socket + (1/2) Tie Pt.							
1.37	1.25		2.62	.00678	Suncompass Assy.							

$$\text{*Weight on each location} = 24.2 \times \frac{A_i}{A_t} = 24.2 \times \frac{A_i}{237.402}$$

**Mass of stations 3.16 will be added to the stations 3.16 of the Lower Array

***Mass at this location is transformed to locations 3, 16, and 21



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by making use of the fact that both the independent and dependent systems must have the same kinetic energy if all lumped masses have the same velocity in each system.

The only elements in the mass matrix affected by the constraint will involve the coordinates at the attachment points. These elements are calculated below. For the unconstrained system:

$$\text{Kinetic Energy} = \frac{1}{2} (M_{21} U_{21}^2 + M_{16} U_{16}^2 + M_3 U_3^2 + M_d U_d^2) \quad (1)$$

Where:

M_i = the mass lumped at the i th coordinate location,

U_i = the velocity of the mass lumped at the i th coordinate location

$()_d$ = the displacement or velocity of the dependent coordinate

But since the 4 mass points are constrained to move in a plane:

$$U_d = U_{21} - 1.398 U_{16} + 1.398 U_3 \quad (2)$$

Where the constants are determined by geometry of the plane.

Substituting (2) into (1) gives:

$$\begin{aligned} \text{K. E.} = \frac{1}{2} \left\{ (M_{21} + M_d) U_{21}^2 - 2.796 M_d U_{21} U_{16} \right. \\ \left. + (M_{16} + 1.9544 M_d) U_{16}^2 + 2.796 M_d U_{21} U_3 \right. \\ \left. - 3.4088 M_d U_{16} U_3 + (M_3 + 1.9544 M_d) U_3^2 \right\} \quad (3) \end{aligned}$$

Expressed in matrix form:

$$\text{K. E.} = \frac{1}{2} \begin{bmatrix} U_{21} & U_{16} & U_3 \end{bmatrix} \begin{bmatrix} (M_{21} + M_d) & -1.398 M_d & 1.398 M_d \\ -1.398 M_d & (M_{16} + 1.9544 M_d) & -1.9544 M_d \\ 1.398 M_d & -1.9544 M_d & (M_3 + 1.9544 M_d) \end{bmatrix} \begin{bmatrix} U_{21} \\ U_{16} \\ U_3 \end{bmatrix} \quad (4)$$



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By definition of a mass matrix, the elements of the 3 x 3 matrix in (4) are the non-trivial elements of the mass matrix in the independent coordinates. These are placed in the independent, $[M_q]$ matrix as shown below

$$K. E. = \frac{1}{2} [U_{21}, U_{16}, U_3] \begin{bmatrix} M_{21,21} & M_{21,16} & M_{21,3} \\ M_{16,21} & M_{16,16} & M_{16,3} \\ M_{3,21} & M_{3,16} & M_{3,3} \end{bmatrix} \begin{Bmatrix} U_{21} \\ U_{16} \\ U_3 \end{Bmatrix} \quad (5)$$

The remainder of the independent mass matrix is a diagonal matrix whose elements are the masses lumped at the appropriate points. The independent mass matrix is shown in Figure 2.1.5.

Next the two stiffness matrices calculated for the large and small arrays were used to calculate a stiffness matrix for the whole system. The stiffness for the upper array was calculated relative to a plane through its four attachments to the lower array. It is necessary to find the stiffness of the whole system relative to ground. To do this, the stiffness matrix for the upper array was transformed using the transformation matrix relating displacements relative to the plane and displacements relative to ground. This was accomplished as follows:

Let: q_i , ($i = 1-22$) = displacement of the i th lumped mass point on the lower array relative to ground

q_i , ($i = 23-30$) = displacement of the i th lumped mass point on the upper array relative to ground

u_{ri} , ($i = 23-30$) = displacement relative to ground of the i th point on the plane through any 3 of the 4 points attaching the two arrays, u_i has the same dimensions relative to the attachment points as the i th lumped mass

u_i , ($i = 23-30$) = displacement of the i th lumped mass point on the upper array relative to the above plane

u_i , ($i = 1-22$) = displacement of the i th lumped mass point on the lower array relative to ground

These coordinate definitions are shown in Figure 2.1.6.

MASS MATRIX

Figure 2.1.5

1	0.601000-02	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
2	0.0	0.735000-02	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
3	0.0	0.0	0.229400-01	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	-0.131300-01	0.0	0.0
	0.0	0.0	0.940000-02	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
4	0.0	0.0	0.0	0.962000-02	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
5	0.0	0.0	0.0	0.0	0.703000-02	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
6	0.0	0.0	0.0	0.0	0.0	0.703000-02
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
7	0.0	0.0	0.0	0.0	0.0	0.0
	0.696000-02	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
8	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.709000-02	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
9	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.354000-02	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
10	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.962000-02	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0

Fig 2.1.5 (cont.)

	0.0	0.0	0.0	0.0	0.0	0.0
11	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.703000-02	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
12	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.354000-02
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
13	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.709000-02	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
14	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.696000-02	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
15	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.709000-02	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
16	0.0	0.0	-0.131300-01	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.224200-01	0.0	0.0
	0.0	0.0	-0.940000-02	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
17	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.962000-02	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
18	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.703000-02
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
19	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.354000-02	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
20	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.452000-02	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0

Fig. 2.1.5 (cont)

21	0.0	0.0	0.94000D-02	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	-0.94000D-02	0.0	0.0
	0.0	0.0	0.12270D-01	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
22	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.67800D-02	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
23	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.41400D-02	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
24	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.54900D-02
	0.0	0.0	0.0	0.0	0.0	0.0
25	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.11780D-01	0.0	0.0	0.0	0.0	0.0
26	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.52300D-02	0.0	0.0	0.0	0.0
27	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.52300D-02	0.0	0.0	0.0
28	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.10640D-01	0.0	0.0
29	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.52300D-02	0.0
30	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.50200D-02

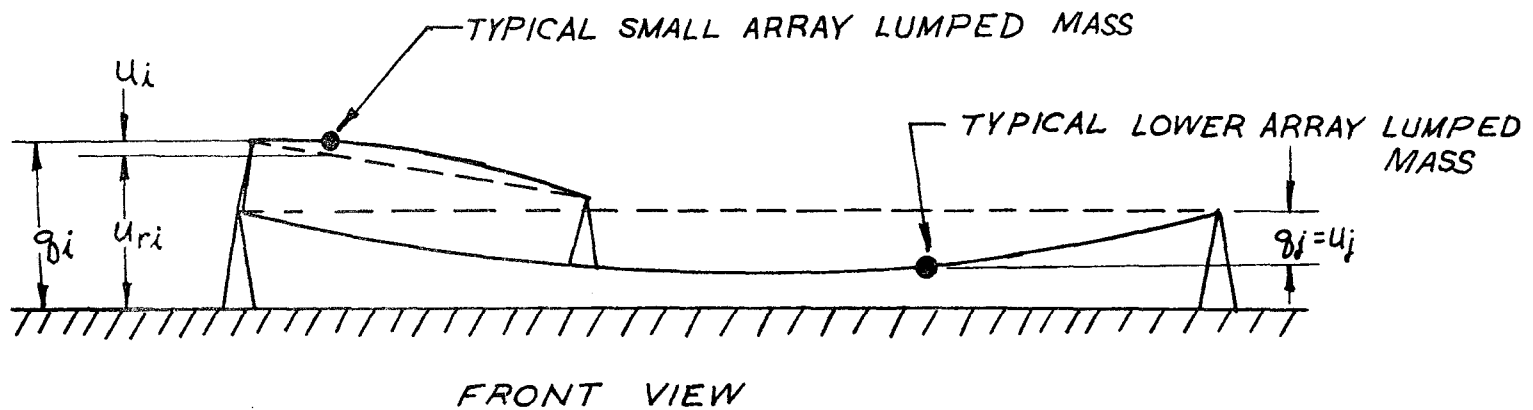


FIG. 2.1.6
DEFINITION OF COORDINATES



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But since the u_{ri} 's are displacements of points on a rigid plane, they can be expressed in terms of the absolute displacements of the 3 independent coordinates determining the plane. In Figure 2.1.1 above, q_3 , q_{16} , and q_{21} were chosen as the independent coordinates determining the plane through the attachment fittings. We can then write:

$$\begin{Bmatrix} u_{r23} \\ u_{r24} \\ \vdots \\ u_{r30} \end{Bmatrix} = [C] \begin{Bmatrix} q_{21} \\ q_{16} \\ q_3 \end{Bmatrix} \quad (6)$$

The $[C]$ matrix is determined by the geometry of the lumped mass locations on the plane.

$$\begin{Bmatrix} P_{23} \\ \vdots \\ P_{30} \end{Bmatrix} = [K_u] \begin{Bmatrix} U_{23} \\ \vdots \\ U_{30} \end{Bmatrix} \quad (7)$$

Where:

P_i , ($i = 23-30$) = the force at the i th coordinate

$[K_u]$ = the stiffness matrix calculated for the upper array relative to the plane through its attachment points.

By definition:

$$u_i = q_i - u_{ri} \quad (i = 23-30) \quad (8)$$

Substituting (8) into (7),

$$\{P\} = [K_u] \{q - u_r\} = [K_u] \{q\} - [K_u] \{u_r\} \quad (9)$$

To balance these forces, there must be equal and opposite forces on the rigid plane.



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$$\{P_r\} = [K_u] \{q\} + [K_u] \{u_r\} \quad (10)$$

Expressing these two sets as one,

$$\begin{Bmatrix} P_{23} \\ \vdots \\ P_{30} \\ \dots \\ P_{r23} \\ \vdots \\ P_{r23} \end{Bmatrix} = \begin{bmatrix} [K_u] & [-K_u] \\ [-K_u] & [K_u] \end{bmatrix} \begin{Bmatrix} q_i \\ \vdots \\ \vdots \\ \vdots \\ u_{ri} \\ \vdots \\ \vdots \end{Bmatrix} \quad (11)$$

This equation allows calculation of dynamic loads on the structure.

This set of equations is also used to combine the two stiffness matrices to calculate a stiffness matrix for the whole system.

The coordinate transform can be written using (6)

$$\begin{Bmatrix} u_1 \\ \vdots \\ u_{22} \\ \vdots \\ u_{23} \\ \vdots \\ u_{30} \\ \vdots \\ u_{r23} \\ \vdots \\ u_{r30} \end{Bmatrix} = [\beta] \begin{Bmatrix} q_1 \\ \vdots \\ \vdots \\ \vdots \\ q_{30} \end{Bmatrix} \quad (12)$$

The $[\beta]$ matrix is shown in Figure 2.1.7. It is seen that the $[\beta]$ matrix is simply a unit diagonal matrix with the $[C]$ matrix of equation (6) tacked on the bottom.

Figure 2.1.7
B MATRIX

2.1.7

ROW 1	0.1C00000E 01	C.C	C.C	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	C.C	0.0	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	C.C	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	C.C	0.0	0.0	0.0	0.0	0.0
ROW 2	0.0	0.1C00000E 01	C.C	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
ROW 3	0.0	C.C	0.1C00000E 01	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	C.C	0.0	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	C.C	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	C.C	0.0	0.0	0.0	0.0	0.0
ROW 4	0.0	C.C	0.0	0.1000000E 01	0.0	0.0	0.0	0.0	0.0
	0.0	C.C	0.0	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	C.C	0.0	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	C.C	0.0	0.0	0.0	0.0	0.0	0.0
ROW 5	0.0	0.0	C.C	0.0	0.1000000E 01	0.0	0.0	0.0	0.0
	0.0	0.0	C.C	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	C.C	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	C.C	0.0	0.0	0.0	0.0	0.0	0.0
ROW 6	0.0	C.C	C.C	0.0	0.0	0.1000000E 01	0.0	0.0	0.0
	0.0	C.C	C.C	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	C.C	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	C.C	0.0	0.0	0.0	0.0	0.0	0.0
ROW 7	0.0	C.C	0.0	0.0	0.0	0.0	0.1000000E 01	0.0	0.0
	0.0	C.C	0.0	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	C.C	0.0	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	C.C	0.0	0.0	0.0	0.0	0.0	0.0	0.0
ROW 8	0.0	0.0	C.C	0.0	0.0	0.0	0.0	0.1000000E 01	0.0
	0.0	0.0	C.C	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	C.C	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	C.C	0.0	0.0	0.0	0.0	0.0	0.0
ROW 9	0.0	C.C	C.C	0.0	0.0	0.0	0.0	0.0	0.0
	0.1C00000E 01	C.C	C.C	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	C.C	C.C	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	C.C	C.C	0.0	0.0	0.0	0.0	0.0	0.0
ROW 10	0.0	0.0	C.1CCCCCE 01	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	C.C	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	C.C	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	C.C	0.0	0.0	0.0	0.0	0.0	0.0
ROW 11	0.0	C.C	0.0	0.1CCCCCE 01	0.0	0.0	0.0	0.0	0.0
	0.0	C.C	0.0	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	C.C	0.0	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	C.C	0.0	0.0	0.0	0.0	0.0	0.0	0.0
ROW 12	0.0	0.0	C.C	0.0	0.0	0.0	0.0	0.0	0.0

Fig. 2.1.7 (cont)

	0.0	C.0	C.0	0.1000000E 01	0.0	0.0	0.0	0.0
	0.0	0.0	C.C	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	C.0	0.0	0.0	0.0	0.0	0.0
ROW 13	0.0	0.0	C.0	0.0	0.0	0.0	0.0	0.0
	0.0	C.C	C.0	0.0	0.1000000E 01	0.0	0.0	0.0
	0.0	C.0	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	C.0	C.0	0.0	0.0	0.0	0.0	0.0
ROW 14	0.0	C.0	C.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.1000000E 01	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	C.C	C.0	0.0	0.0	0.0	0.0	0.0
ROW 15	0.0	0.0	C.0	0.0	0.0	0.0	0.0	0.0
	0.0	C.C	0.0	0.0	0.0	0.0	0.1000000E 01	0.0
	0.0	C.0	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	C.0	0.0	0.0	0.0	0.0	0.0
ROW 16	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	C.0	C.0	0.0	0.0	0.0	0.0	0.1000000E 01
	0.0	C.C	C.0	0.0	0.0	0.0	0.0	0.0
	0.0	C.0	0.0	0.0	0.0	0.0	0.0	0.0
ROW 17	0.0	0.0	C.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
	C.1000000E 01	0.0	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	C.C	0.0	0.0	0.0	0.0	0.0	0.0
ROW 18	0.0	C.0	C.0	0.0	0.0	0.0	0.0	0.0
	0.0	C.0	C.0	0.0	0.0	0.0	0.0	0.0
	0.0	C.1000000E 01	C.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	C.0	0.0	0.0	0.0	0.0	0.0
ROW 19	0.0	0.0	C.C	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	C.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.1000000E 01	0.0	0.0	0.0	0.0	0.0
	0.0	C.C	C.0	0.0	0.0	0.0	0.0	0.0
ROW 20	0.0	C.C	C.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	C.0	0.0	0.1000000E 01	0.0	0.0	0.0	0.0
	0.0	C.0	C.0	0.0	0.0	0.0	0.0	0.0
ROW 21	0.0	C.C	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	C.0	C.0	0.0	0.0	0.0	0.0	0.0
	0.0	C.0	C.0	0.0	0.1000000E 01	0.0	0.0	0.0
	0.0	0.0	C.0	0.0	0.0	0.0	0.0	0.0
ROW 22	0.0	C.C	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	C.0	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.1000000E 01	0.0	0.0
	0.0	C.C	C.0	0.0	0.0	0.0	0.0	0.0
ROW 23	0.0	0.0	C.0	0.0	0.0	0.0	0.0	0.0
	0.0	C.0	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0	0.1000000E 01	0.0
	0.0	0.0	C.0	0.0	0.0	0.0	0.0	0.0
ROW 24	0.0	0.0	C.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	C.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.1000000E 01

Fig 2.1.7 (cont)

	0.0	0.0	0.0	0.0	0.0	0.0		
ROW 25	0.0	C.C	C.0	0.0	0.0	0.0	0.0	0.0
	0.0	C.0	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	C.C	C.0	0.0	0.0	0.0	0.0	0.0
	0.1000000E 01	0.0	0.0	0.0	0.0	0.0	0.0	0.0
ROW 26	0.0	0.0	C.0	0.0	0.0	0.0	0.0	0.0
	0.0	C.0	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	C.1000000E 01	0.0	0.0	0.0	0.0	0.0	0.0
ROW 27	0.0	0.0	C.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	C.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	C.C	C.1000000E 01	0.0	0.0	0.0	0.0	0.0
ROW 28	0.0	0.0	C.0	0.0	0.0	0.0	0.0	0.0
	0.0	C.0	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	C.C	0.0	0.1000000E 01	0.0	0.0	0.0	0.0
ROW 29	0.0	C.0	C.0	0.0	0.0	0.0	0.0	0.0
	0.0	C.0	C.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	C.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.C	C.0	0.0	0.1000000E 01	0.0	0.0	0.0
ROW 30	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.C	C.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	C.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.C	C.0	0.0	0.0	0.1000000E 01	0.0	0.0
ROW 31	0.0	C.C	C.12857CCE 01	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0	0.0	-0.7857000E 00
	0.0	0.0	C.0	0.0	0.5000000E 00	0.0	0.0	0.0
	0.0	0.C	C.0	0.0	0.0	0.0	0.0	0.0
ROW 32	0.0	0.0	C.1224799E 01	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	C.0	0.0	0.0	0.0	0.0	-0.1224700E 01
	0.0	C.C	C.0	0.0	0.1000000E 01	0.0	0.0	0.0
	0.C	0.0	C.0	0.0	0.0	0.0	0.0	0.0
ROW 33	0.0	0.0	C.81980CCE 00	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0	0.0	-0.3198000E 00
	0.0	0.C	C.0	0.0	0.5000000E 00	0.0	0.0	0.0
	0.0	C.C	0.0	0.0	0.0	0.0	0.0	0.0
ROW 34	0.0	0.0	C.4744559E 00	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	C.0	0.0	0.0	0.0	0.0	0.C
	0.0	0.C	C.0	0.0	0.1000000E 01	0.0	0.0	0.52550CCE 00
	0.0	C.C	C.0	0.0	0.0	0.0	0.0	0.0
ROW 35	0.0	0.C	C.69910CCE 00	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0	0.0	-0.6991000E 00
	0.0	0.0	C.0	0.0	0.1000000E 01	0.0	0.0	0.0
	0.0	C.0	0.0	0.0	0.0	0.0	0.0	0.0
ROW 36	0.0	0.0	C.35380CCE 00	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.1461999E 00
	0.0	0.C	0.0	0.0	0.5000000E 00	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0

Fig 2.1.7 (cont)

ROW 37	0.0	0.0	C.2248CCCE	OC	0.0	0.0	0.0	0.0
	0.0	0.0	0.0		0.0	0.0	0.0	-0.2248000E 00
	0.0	0.0	0.0		0.0	0.0	0.0	0.0
	0.0	0.0	0.0		0.0	0.0	0.0	0.0
					0.1000000E 01	0.0	0.0	0.0
					0.0	0.0		
ROW 38	0.0	0.0	-0.1120999E	OC	0.0	0.0	0.0	0.0
	0.0	0.0	0.0		0.0	0.0	0.0	0.6120999E 00
	0.0	0.0	0.0		0.0	0.0	0.0	0.0
	0.0	0.0	0.0		0.5000000E 00	0.0	0.0	0.0
	0.0	0.0	0.0		0.0	0.0		



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Then the stiffness matrix for the whole system can be calculated:

$$K_s = [\beta]^T \begin{bmatrix} [K_L] & 0 & [-K_u] \\ 0 & [-K_u] & [K_u] \end{bmatrix} [\beta] \quad (13)$$

The $[K_s]$ matrix is shown in Figure 2.1.8.

With the system completely described, solution of the eigenvalue problem for the system was accomplished to find the mode shapes and frequencies of the model. These are listed in Figure 2.1.9. The mode shapes and frequencies were then used to calculate the system's frequency and time response through the method of modal analysis. Viscous modal damping of 10 percent of critical damping was assumed for all modes in calculating responses. Experience with previous retro-reflector arrays has shown this assumption to give good agreement between analysis and test results.

2.2 Dynamic Model In Plane of Array

The dynamic model predicting the structure's response to inputs in the plane of the array is shown in Figure 2.2.1. In this model the arrays are assumed to be rigid in comparison to the fittings connecting the experiment to the LM and the fittings connecting the two arrays. Each array is considered a rigid body with 3 degrees of freedom as shown in Figure 2.2.1. All flexibility is attributed to the 8 fittings shown as springs in the Figure. The rear LM interface fittings are seen to have been attributed no stiffness in the X coordinate direction. This is because they are free to slide over pins in this direction.

Figure 2.2.2 shows plan views of each array defining parameters and showing pertinent dimensions. Given this, a stiffness matrix for each array can be evaluated from the definitions of the various influence coefficients assuming both arrays fixed to ground. Figure 2.2.3 shows both stiffness matrices. To find a stiffness matrix for the whole system relative to ground the same procedure was followed as was used for the out-of-plane model.

Figure 2.1.8
[K_s] Matrix

ROW 1	0.52600000	06 0.61980000	04-0.25880000	05-0.58680000	06-0.33210000	05 0.13780000	06 0.15200000	06 0.79289990	03
	-0.17920000	05 0.10860000	05-0.24920000	05 0.81250000	03-0.18270000	05-0.18350000	05 0.36820000	04 0.83970000	04
	-0.15830000	05 0.58680000	04-0.13770000	04 0.42070000	04 0.24110000	04-0.18650000	04 0.0	0.0	
	0.0	0.0	0.0	0.0	0.0	0.0			
ROW 2	0.61980000	04 0.45170000	05 0.20780000	04 0.41670000	05 0.42280000	05-0.10480000	05-0.11080000	05-0.99740000	05
	0.13560000	04-0.85420000	04 0.32540000	05 0.17330000	04 0.16610000	04-0.31110000	04 0.79470000	04-0.75600000	03
	0.20590000	04-0.76850000	04 0.23650000	05-0.71520000	03 0.96639990	03 0.16390000	04 0.0	0.0	
	0.0	0.0	0.0	0.0	0.0	0.0			
ROW 3	-0.25880000	05 0.20780000	04 0.17760000	07 0.12420000	06-0.11010000	05-0.40210000	06 0.11490000	05-0.17860000	03
	-0.30840000	06 0.12670000	06-0.33660000	04 0.18350000	03 0.53040000	05-0.19870000	05 0.15680000	04-0.60241500	06
	-0.37940000	05 0.41580000	04-0.72239990	03 0.76100000	04 0.70153590	06-0.19970000	04-0.23686290	06-0.61539130	06
	-0.47385020	05-0.98278680	05 0.26044360	06 0.93686070	04-0.99091560	05-0.11820790	05		
ROW 4	-0.58680000	06 0.41670000	05 0.12420000	06 0.13750000	07 0.15720000	06-0.65660000	06-0.71970000	06-0.41980000	04
	0.92600000	05 0.62360000	05 0.12700000	06-0.42340000	04 0.45560000	05 0.43070000	05-0.16820000	05-0.32430000	05
	0.46510000	05-0.21340000	05 0.60020000	04-0.12290000	05-0.59130000	04 0.65530000	04 0.0	0.0	
	0.0	0.0	0.0	0.0	0.0	0.0			
ROW 5	-0.33210000	05 0.42280000	05-0.11010000	05 0.15720000	06 0.22250000	07 0.10760000	05-0.93560000	06-0.78200000	06
	-0.36090000	04 0.17280000	06-0.12820000	06 0.18340000	06-0.24880000	05 0.73250000	05 0.62220000	05 0.74250000	04
	-0.31920000	05 0.78480000	05-0.72980000	05 0.10040000	05-0.10590000	05-0.20280000	05 0.0	0.0	
	0.0	0.0	0.0	0.0	0.0	0.0			
ROW 6	0.13780000	06-0.10480000	05-0.40210000	06-0.65530000	06 0.10770000	05 0.14230000	07 0.28310000	06 0.18170000	04
	-0.34980000	06-0.83310000	06-0.14840000	05 0.70870000	02 0.14000000	06 0.10900000	06-0.20990000	04 0.52300000	05
	0.58400000	05-0.16660000	05 0.16170000	04-0.17040000	05-0.11830000	05 0.55100000	04 0.0	0.0	
	0.0	0.0	0.0	0.0	0.0	0.0			
ROW 7	0.15200000	06-0.11080000	05 0.11490000	05-0.71970000	06-0.93560000	06 0.28310000	06 0.19640000	07 0.33750000	06
	-0.95240000	04-0.91970000	06-0.86790000	06-0.15430000	05 0.10990000	06 0.22960000	06 0.10130000	06-0.21180000	05
	0.62330000	05 0.53940000	05-0.17560000	05-0.15270000	05-0.29040000	05-0.12920000	05 0.0	0.0	
	0.0	0.0	0.0	0.0	0.0	0.0			
ROW 8	0.79409590	03-0.99740000	05-0.17780000	03-0.41980000	04-0.78200000	06 0.18150000	04 0.33750000	06 0.12610000	07
	0.55239990	03-0.24900000	05-0.73270000	06-0.51460000	06-0.75500000	02 0.92880000	05 0.16140000	06 0.74509990	03
	-0.12580000	05 0.33730000	05 0.12470000	06 0.38650000	04-0.80400000	04-0.11760000	05 0.0	0.0	
	0.0	0.0	0.0	0.0	0.0	0.0			
ROW 9	-0.17920000	05 0.13560000	04-0.30840000	06 0.92600000	05-0.36120000	04-0.34980000	06-0.95210000	04 0.55170000	03
	0.62430000	06 0.24870000	06-0.75560000	04 0.23980000	03-0.35280000	06-0.14000000	05-0.53679980	03-0.30770000	06
	0.10380000	06 0.12910000	04 0.21260000	03-0.15700000	05-0.14120000	05 0.23840000	04 0.0	0.0	
	0.0	0.0	0.0	0.0	0.0	0.0			
ROW 10	0.10860000	06-0.85410000	04 0.12670000	06 0.62370000	05 0.17280000	06-0.83310000	06-0.91970000	06-0.24910000	05
	0.24870000	06 0.21950000	07 0.35120000	06-0.72170000	04-0.81690000	06-0.88870000	06-0.11830000	05 0.12330000	06
	-0.63089990	03 0.12370000	06-0.41190000	04 0.98150000	05 0.65600000	05-0.34490000	05 0.0	0.0	
	0.0	0.0	0.0	0.0	0.0	0.0			
ROW 11	-0.24920000	05 0.32540000	05-0.33680000	04 0.12700000	06-0.12820000	06-0.14840000	05-0.86790000	06-0.73270000	06
	-0.75560000	04 0.35120000	06 0.22060000	07 0.28950000	06-0.12270000	05-0.89220000	06-0.82980000	06-0.41470000	04
	0.12390000	06 0.80000000	03 0.10540000	06-0.34480000	05 0.65570000	05 0.98120000	05 0.0	0.0	
	0.0	0.0	0.0	0.0	0.0	0.0			
ROW 12	0.83200000	03 0.17800000	04 0.18360000	03-0.42340000	04 0.18340000	06 0.72159990	02-0.16480000	05-0.51460000	06

Figure 2.18 (cont)

0.23820000	03-0.72200000	04 0.28950000	06 0.50700000	06-0.15820000	03-0.20270000	05-0.33100000	06 0.40009990	02
0.24880000	04 0.90400000	05-0.25630000	06 0.17470000	04-0.12050000	05-0.13080000	05 0.0	0.0	
0.0	0.0	0.0	0.0	0.0	0.0			
ROW 13-0.18270000	05 0.16610000	04 0.53040000	05 0.45560000	05-0.24790000	05 0.14000000	06 0.10990000	06-0.75099990	02
-0.35280000	06-0.81690000	06-0.12270000	05-0.15080000	03 0.14430000	07 0.31250000	06 0.93720000	04-0.40580000	06
-0.74460000	06-0.24660000	05-0.23260000	04 0.12120000	06 0.10950000	06-0.16140000	05 0.0	0.0	
0.0	0.0	0.0	0.0	0.0	0.0			
ROW 14-0.18340000	05-0.31110000	04-0.19870000	05 0.43060000	05 0.73250000	05 0.10900000	06 0.22960000	06 0.92880000	05
-0.14010000	05-0.88870000	06-0.89220000	06-0.20270000	05 0.31250000	06 0.19690000	07 0.31480000	06 0.58110000	04
-0.84060000	06-0.84060000	06 0.84900000	04 0.11640000	06 0.23800000	06 0.11640000	06 0.0	0.0	
0.0	0.0	0.0	0.0	0.0	0.0			
ROW 15 0.36850000	04 0.79480000	04 0.15670000	04-0.16830000	05 0.62220000	05-0.20960000	04 0.10130000	06 0.16140000	06
-0.53609990	03-0.11830000	05-0.82980000	06-0.33100000	06 0.93680000	04 0.31480000	06 0.14400000	07-0.20760000	04
-0.25120000	05-0.74150000	06-0.41530000	06-0.15960000	05 0.10900000	06 0.12050000	06 0.0	0.0	
0.0	0.0	0.0	0.0	0.0	0.0			
ROW 16 0.83970000	04-0.75589990	03-0.60241500	06-0.32430000	05 0.74220000	04 0.52300000	05-0.21180000	05 0.74509990	03
-0.30770000	06 0.12320000	06-0.41460000	04 0.40060000	02-0.40580000	06 0.58080000	04-0.20770000	04 0.16879480	07
0.14060000	06-0.29470000	04 0.93829980	03-0.22920000	05-0.48363740	06 0.43320000	04 0.84004110	05 0.67320610	06
-0.43845010	05-0.26721370	05-0.21984730	06-0.13362690	06 0.16373800	06-0.15928840	06		
0.0	0.0	0.0	0.0	0.0	0.0			
ROW 17-0.15830000	05 0.20580000	04-0.37940000	05 0.46520000	05-0.31930000	05 0.58400000	05 0.62340000	05-0.12570000	05
-0.10380000	06-0.64900000	03 0.12390000	06 0.24880000	04-0.74460000	06-0.84060000	06-0.25110000	05 0.14060000	06
-0.17500000	07 0.27560000	06-0.34740000	04-0.51290000	06-0.48750000	06 0.39220000	05 0.0	0.0	
0.0	0.0	0.0	0.0	0.0	0.0			
ROW 18 0.58660000	04-0.76850000	04 0.41580000	04-0.21340000	05 0.78480000	05-0.16670000	05 0.53940000	05 0.33730000	05
-0.12930000	04 0.12370000	06 0.78570000	03 0.93390000	05-0.24660000	05-0.84060000	06-0.74140000	06-0.29480000	04
-0.27560000	06 0.17490000	07 0.14650000	06 0.39200000	05-0.48740000	06-0.51270000	06 0.0	0.0	
0.0	0.0	0.0	0.0	0.0	0.0			
ROW 19-0.13780000	04 0.23660000	05-0.72179980	03 0.60040000	04-0.72980000	05 0.16140000	04-0.17570000	05 0.12470000	06
-0.21320000	03-0.41180000	04 0.10540000	06-0.25530000	06-0.23270000	04 0.84890000	04-0.41530000	06 0.93800000	03
-0.34700000	04 0.14650000	06 0.77480000	06 0.46140000	04-0.20550000	05-0.24070000	05 0.0	0.0	
0.0	0.0	0.0	0.0	0.0	0.0			
ROW 20 0.42070000	04-0.71475980	03 0.76110000	04-0.12290000	05 0.10040000	05-0.17040000	05-0.15270000	05 0.38620000	04
-0.15700000	05 0.98150000	05-0.34480000	05 0.17480000	04 0.12120000	06 0.11640000	06-0.15960000	05-0.22910000	05
-0.51290000	06 0.39200000	05 0.46160000	04 0.53650000	06-0.71540000	05 0.37100000	05 0.0	0.0	
0.0	0.0	0.0	0.0	0.0	0.0			
ROW 21 0.24120000	04 0.96659990	03 0.70153610	06-0.59150000	04-0.10590000	05-0.11630000	05-0.29040000	05-0.60420000	04
-0.14120000	05 0.65500000	05 0.65570000	05-0.12050000	05 0.10950000	06 0.23800000	06 0.10900000	06-0.48363740	06
-0.48750000	06-0.48740000	06-0.20590000	05-0.71530000	05 0.15800900	07-0.71550000	05-0.10382570	06-0.59290600	06
0.25301000	06-0.33973500	06 0.17916850	06 0.20125500	06-0.41684030	06-0.13419300	06		
0.0	0.0	0.0	0.0	0.0	0.0			
ROW 22-0.18650000	04 0.16390000	04-0.19970000	04 0.65520000	04-0.20280000	05 0.55120000	04-0.12920000	05-0.11770000	05
-0.23840000	04-0.34490000	05 0.98130000	05-0.13080000	05-0.16140000	05 0.11640000	06 0.12050000	06 0.43320000	04
0.39230000	05-0.51270000	06-0.24070000	05 0.37100000	05-0.71550000	05 0.53650000	06 0.0	0.0	
0.0	0.0	0.0	0.0	0.0	0.0			
ROW 23 0.0	0.0	-0.23686290	06 0.0	0.0	0.0	0.0	0.0	
0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.84004090	05
0.0	0.0	0.0	0.0	-0.10382560	06 0.0	0.21960000	06-0.51890000	05
-0.14310000	05 0.36610000	05 0.21410000	04 0.29690000	05 0.71059990	03-0.24720000	04		
0.0	0.0	0.0	0.0	0.0	0.0			
ROW 24 0.0	0.0	-0.61539130	06 0.0	0.0	0.0	0.0	0.0	
0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.67320610	06
0.0	0.0	0.0	0.0	-0.59290600	06 0.0	-0.51890000	05 0.79850000	06

Figure 2.1.8 (Cont)

-0.20160000 06 0.51080000 05-0.28090000 06 0.33390000 05 0.13320000 06 0.21520000 04									
ROW 25	0.0	0.0	-0.47385020	05 0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0	0.0	-0.43845010	05
	0.0	0.0	0.0	0.0	0.25301000	06 0.0	-0.14310000	05-0.20160000	06
	0.70120000	06-0.20570000	06-0.18290000	06-0.12050000	06 0.40240000	05 0.27510000	05		
ROW 26	0.0	0.0	-0.98278680	05 0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0	0.0	-0.26721370	05
	0.0	0.0	0.0	0.0	-0.33973500	06 0.0	0.36610000	05 0.51080000	05
	-0.20570000	06 0.27120000	06 0.12960000	06-0.17920000	06 0.34060000	05 0.55880000	05		
ROW 27	0.0	0.0	0.26044360	06 0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0	0.0	-0.21984730	06
	0.0	0.0	0.0	0.0	0.17916850	06 0.0	0.21410000	04-0.28090000	06
	-0.18290000	06 0.12960000	06 0.41140000	06-0.16250000	06-0.26910000	06 0.29220000	04		
ROW 28	0.0	0.0	0.93686070	04 0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0	0.0	-0.13362690	06
	0.0	0.0	0.0	0.0	0.20125500	06 0.0	0.29690000	05 0.33390000	05
	-0.12050000	06-0.17920000	06-0.16250000	06 0.71120000	06-0.19640000	06-0.13480000	05		
ROW 29	0.0	0.0	-0.99091690	05 0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.16373810	06
	0.0	0.0	0.0	0.0	-0.41684030	06 0.0	0.71070000	03 0.13320000	06
	0.40240000	05 0.34060000	05-0.26910000	06-0.19640000	06 0.61740000	06-0.41990000	05		
ROW 30	0.0	0.0	-0.11820790	05 0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0	0.0	-0.15928840	06
	0.0	0.0	0.0	0.0	-0.13419300	06 0.0	-0.24720000	04 0.21520000	04
	0.27510000	05 0.55880000	05 0.29220000	04-0.13480000	05-0.41990000	05 0.21890000	06		

Figure 2.1.9
MODE SHAPES & FREQUENCIES

MODE 1	MODE 2	MODE 3	MODE 4
EIGENVALUE= 0.196510-C5	EIGENVALUE= 0.55108D-06	EIGENVALUE= 0.28643D-06	EIGENVALUE= 0.19232D-06
FREQUENCY= 0.11353D 03	FREQUENCY= 0.21439D 03	FREQUENCY= 0.29738D 03	FREQUENCY= 0.36292D 03
EIGENVECTOR	EIGENVECTOR	EIGENVECTOR	EIGENVECTOR
-0.30581D-01	-0.17804D 00	-0.11513D 00	-0.14914D 00
0.23246D 00	0.10000D 01	-0.29798D 00	0.10000D 01
0.13529D 00	-0.18654D 00	-0.22949D 00	0.39701D 00
0.76182D-01	-0.23406D 00	-0.36736D 00	-0.14462D 00
0.11190D 00	0.80631D-01	-0.24461D 00	-0.19193D 00
0.20352D CC	-0.26744D 00	-0.53199D 00	0.26420D 00
0.20653D CC	-0.56150D-01	-0.48294D 00	-0.28270D 00
0.24826D 00	0.39896D 00	-0.37626D 00	-0.28724D 00
0.22776D CC	-0.34598D 00	-0.67376D 00	0.74952D 00
0.31962D CC	-0.13978D 00	-0.58949D 00	-0.40864D-01
0.33453D CC	0.15902D 00	-0.46738D 00	-0.44089D 00
0.26736D 00	0.53369D 00	-0.32326D 00	-0.44336D 00
0.33228D CC	-0.19440D 00	-0.60002D 00	0.29251D 00
0.48068D 00	0.25863D-01	-0.44987D 00	-0.30072D 00
0.35243D CC	0.26002D 00	-0.32838D 00	-0.50937D 00
0.98560D-01	-0.16603D 00	-0.45395D 00	0.39596D 00
0.56713D 00	-0.34741D-01	-0.30751D 00	-0.84251D-01
0.57088D 00	0.10190D 00	-0.21357D 00	-0.38579D 00
0.12266D 00	0.19046D 00	-0.14144D 00	-0.31060D 00
0.39510D 00	0.85769D-02	-0.38062D-01	-0.65683D-01
0.81226D 00	0.34505D-01	0.36173D-01	-0.21828D 00
0.39508D 00	0.32868D-01	-0.97549D-04	-0.18763D 00
0.51207D CC	-0.93823D-01	0.14589D 00	0.89583D-01
0.92032D 00	-0.61167D-02	0.62539D 00	-0.86527D-01
0.56505D CC	-0.10944D 00	0.30458D 00	0.31133D 00
0.94719D CC	-0.15973D 00	-0.38358D 00	0.30884D 00
0.10000D 01	-0.19403D-01	0.10000D 01	0.13134D 00
0.54670D 00	-0.98357D-01	0.19214D 00	0.30276D 00
0.90060D 00	0.10921D-01	0.47358D 00	-0.50921D-01
0.45936D 00	-0.62522D-01	-0.19010D 00	0.75833D-01

Fig. 2.1.9 (con't)

MODE 5

EIGENVALUE= 0.12965D-06
FREQUENCY= 0.44201D 03

EIGENVECTOR

0.27675D 00
-0.24522D-01
0.17291D 00
0.40416D 00
0.24383D 00
0.31730D 00
0.38361D 00
0.40441D 00
0.16195D 00
0.22327D 00
0.34296D 00
0.49699D 00
-0.56750D-02

MODE 6

EIGENVALUE= 0.10408D-06
FREQUENCY= 0.49333D 03

EIGENVECTOR

0.14138D 00
0.24049D 00
0.46092D 00
0.20747D 00
0.11447D-01
0.32868D 00
0.84459D-01
-0.29375D-01
0.12343D 00
0.10351D 00
-0.22033D-02
-0.12570D 00
-0.11329D 00

MODE 7

EIGENVALUE= 0.81465D-07
FREQUENCY= 0.55761D 03

EIGENVECTOR

-0.44726D-01
-0.40280D 00
0.29299D 00
-0.19001D 00
-0.29074D-01
-0.42027D-01
-0.30120D 00
0.33751D 00
0.34566D 00
-0.38718D 00
-0.55031D-01
0.10000D 01
-0.15234D 00

MODE 8

EIGENVALUE= 0.57610D-07
FREQUENCY= 0.66309D 03

EIGENVECTOR

-0.25280D 00
-0.21050D 00
0.26078D 00
-0.28797D 00
-0.10487D 00
0.66089D-01
-0.30514D 00
0.10022D 00
0.58417D 00
-0.14599D 00
-0.14555D 00
0.53355D 00
0.22079D 00

0.21889D-01
0.28010D 00
-0.14472D-01
-0.33984D 00
-0.21189D 00
0.25216D 00
-0.40521D 00
-0.73918D 00
-0.35542D 00
-0.11818D-01
0.94122D-01
0.41042D 00
-0.92185D 00
0.10000D 01
0.31079D 00
0.53485D-01
-0.31056D 00

0.29386D-02
-0.37508D-01
-0.35701D 00
-0.11150D 00
-0.86028D-01
-0.83283D-01
-0.12525D 00
-0.17711D 00
-0.83580D-01
0.10000D 01
0.61270D 00
0.27924D-01
-0.34353D 00
-0.41595D 00
-0.47813D 00
-0.47099D 00
-0.47236D 00

-0.35397D 00
0.41892D 00
0.14454D 00
-0.32014D 00
-0.49263D-01
0.55039D 00
-0.15264D 00
-0.10735D 00
-0.16362D-01
0.83573D-01
-0.44843D-01
0.31087D 00
0.81090D 00
-0.40047D 00
0.15928D 00
-0.29568D 00
-0.32158D 00

-0.16034D 00
0.17605D 00
0.31627D 00
0.43683D-01
0.42024D-02
0.29382D 00
0.83474D-01
0.10319D 00
0.55635D-01
0.56623D 00
0.68651D-01
-0.35190D 00
-0.85753D 00
0.21741D 00
-0.20047D 00
0.21271D 00
0.10000D 01

Fig 2.1.9 (cont)

MODE 9

EIGENVALUE= 0.40305D-C7
FREQUENCY= 0.79276D C3

EIGENVECTOR

0.10000D 01
0.27486D 00
-0.65856D-01
0.67274D 00
-0.70864D-01
0.12293D 00
0.28220D-02
-0.95205D-01
-0.15303D-01
-0.25516D 00
-0.18592D 00
0.10457D 00
-0.29082D 00
-0.33480D 00
0.40450D-C1
0.75004D-01
-0.35274D 00
0.12004D 00
0.83940D-01
-0.18614D 00
0.25080D 00
0.36888D 00
-0.50192D-01
-0.50610D-01
-0.18019D 00
0.14334D 00
-0.43809D-01
0.28157D-01
0.22908D 00
0.54995D 00

MODE 10

EIGENVALUE= 0.28655D-07
FREQUENCY= 0.94020D 03

EIGENVECTOR

-0.44886D 00
0.22405D-01
0.82913D-01
-0.18847D 00
-0.98485D-01
0.11095D 00
0.69464D-02
-0.25668D 00
0.13608D 00
0.12930D 00
0.74927D-01
-0.38240D 00
-0.76432D-01
0.18622D 00
0.17526D 00
0.69159D-01
-0.45482D 00
0.62641D 00
-0.11951D 00
-0.83549D 00
-0.13511D 00
0.10000D 01
0.22733D-01
-0.54671D-01
0.67075D-01
-0.12460D 00
0.29286D-01
0.17331D-01
-0.10136D 00
-0.11329D 00

MODE 11

EIGENVALUE= 0.20255D-07
FREQUENCY= 0.11183D 04

EIGENVECTOR

-0.26620D 00
-0.54929D-01
0.84989D-01
-0.93814D-01
0.37654D 00
-0.18566D 00
0.36522D 00
0.27997D 00
-0.22099D 00
-0.10882D 00
0.42198D 00
-0.54904D 00
-0.35473D 00
0.23164D-01
-0.13715D 00
-0.11738D-01
-0.32151D 00
-0.10921D 00
-0.39439D 00
-0.29629D 00
-0.68903D-01
-0.36388D 00
0.29860D 00
-0.30187D-01
0.13512D 00
0.25851D 00
-0.25170D 00
0.11466D 00
-0.89182D-01
0.10000D 01

MODE 12

EIGENVALUE= 0.16168D-07
FREQUENCY= 0.12517D 04

EIGENVECTOR

0.62424D 00
0.45396D-01
-0.15511D 00
-0.36222D-01
-0.11686D 00
-0.43502D 00
-0.20839D 00
-C.12117D 00
-C.13476D 00
-0.26126D 00
0.10536D 00
-C.21260D 00
0.69182D-01
C.23492D 00
C.16077D 00
0.22422D 00
0.33832D 00
0.28290D 00
-0.65759D-01
0.52156D 00
-0.10117D 00
C.27436D 00
0.36508D 00
0.16664D 00
C.10000D 01
-0.18439D 00
0.41822D-01
-0.71911D 00
-0.88705D 00
0.47407D 00

Fig 2.1.9 (cont)

ODE 13 MODE 14 MODE 15 MODE 16
 EIGENVALUE= 0.15975D-07 EIGENVALUE= 0.13276D-07 EIGENVALUE= 0.12680D-07 EIGENVALUE= 0.10913D-07

REQUENCY= 0.12592D 04 FREQUENCY= 0.13813D 04 FREQUENCY= 0.14134D 04 FREQUENCY= 0.15235D 04

EIGENVECTOR	EIGENVECTOR	EIGENVECTOR	EIGENVECTOR
0.21618D 00	-0.29000D 00	0.41173D 00	0.34548D 00
-0.32651D-02	0.80116D-01	0.16336D 00	-0.84774D-01
-0.79311D-02	-0.34792D 00	0.43362D 00	0.15792D 00
-0.40029D-01	0.15025D 00	-0.14177D 00	-0.92391D-01
0.92122D-01	-0.37455D 00	-0.58730D 00	0.40489D-01
-0.21552D 00	0.12490D 00	-0.37957D-01	-0.11151D 00
0.55296D-02	0.19699D-01	-0.40673D 00	-0.22874D 00
0.87197D-01	-0.54362D 00	-0.89217D 00	0.42709D 00
0.43406D-01	-0.73892D 00	0.19324D 00	-0.46818D 00
-0.19076D 00	0.45240D 00	-0.30410D 00	0.12366D 00
0.16732D 00	-0.36185D 00	0.43991D 00	-0.23217D 00
-0.25185D 00	0.50139D 00	-0.67458D 00	0.42927D 00
-0.33338D-01	0.14469D-01	-0.35988D 00	0.88401D-01
0.56123D-01	0.35178D-01	0.64053D 00	0.17094D 00
0.58383D-02	0.20490D 00	0.10000D 01	-0.50694D 00
0.16866D 00	-0.21703D 00	-0.71866D-01	-0.39336D 00
0.44907D-01	-0.94290D-02	0.80400D-01	0.37845D 00
0.92628D-01	-0.85318D-01	0.62702D 00	-0.16968D 00
-0.17530D 00	0.58428D 00	0.25350D 00	-0.20505D 00
0.15642D 00	-0.31256D 00	0.27685D 00	0.37501D 00
0.88498D-02	-0.61425D-01	-0.37893D 00	-0.50753D 00
0.55737D-01	-0.27357D 00	-0.89032D 00	0.86947D 00
0.10000D 01	0.10000D 01	-0.56663D 00	-0.58100D-01
-0.26164D 00	-0.23640D 00	-0.83446D-01	-0.15628D 00
-0.18871D 00	0.12519D 00	-0.12424D 00	0.13832D 00
0.17719D-01	0.20921D 00	-0.37855D 00	-0.66533D 00
0.28572D-01	-0.18386D 00	-0.30371D 00	-0.72195D 00
0.20810D 00	0.16305D 00	0.59959D 00	0.10000D 01
0.15651D 00	-0.36358D-01	-0.14828D 00	-0.53291D 00
-0.46664D 00	0.36352D 00	0.17558D 00	0.42361D 00

Fig 2.1.9 (cont)

MODE 17

EIGENVALUE= 0.94248D-C8
FREQUENCY= 0.16394D 04

EIGENVECTOR

-0.55100D 00
0.42116D-01
0.19603D-01
0.24289D 00
0.90211D-01
0.24965D 00
0.30506D 00
-0.18367D 00
0.24245D-01
-0.11152D 00
0.25039D-01
-0.12552D 00
-0.33701D 00
-0.29476D 00
0.16678D 00
0.45773D-01
-0.74431D-01
-0.31437D-01
0.18258D 00
0.10000D 01

-0.14474D-01
0.25838D 00
0.38233D-C2
-0.78325D-01
0.63552D-C1
-0.12715D 00
-0.53059D-C1
0.57776D-01
-0.57622D-01
-0.30688D-01

MODE 18

EIGENVALUE= 0.77534D-08
FREQUENCY= 0.18075D 04

EIGENVECTOR

-0.11011D-02
-0.22670D-01
0.53657D-01
-0.71259D-03
-0.68101D-01
0.17710D 00
-0.16245D 00
0.12227D 00
0.45125D-01
-0.30084D-02
-0.79110D-03
0.49415D-01
-0.13110D 00
0.12325D 00
-0.75130D-01
-0.82594D-01
0.48201D-03
0.57614D-01
-0.16437D 00
0.15250D-01

-0.10438D-01
-0.43922D-01
-0.10758D-01
-0.72486D 00
0.41966D 00
-0.27297D 00
-0.42770D 00
-0.21578D 00
0.10000D 01
-0.73850D-01

MODE 19

EIGENVALUE= 0.70912D-08
FREQUENCY= 0.18900D 04

EIGENVECTOR

0.25170D 00
0.11802D 00
0.85610D-01
-0.19959D 00
0.52453D 00
-0.99020D 00
0.89885D 00
-0.64253D 00
0.16149D 00
-0.40569D-01
0.65727D-01
-0.44047D 00
0.90881D 00
-0.75711D 00
0.44897D 00
-0.62302D-02
0.80874D-01
-0.37689D 00
0.10000D 01
-0.40670D 00

0.17637D 00
0.20973D 00
-0.15702D 00
0.83387D-02
0.25446D 00
-0.46467D 00
-0.59059D 00
0.12289D 00
0.43866D 00
-0.13100D 00

MODE 20

EIGENVALUE= 0.57876D-08
FREQUENCY= 0.20920D 04

EIGENVECTOR

-0.39331D 00
-0.30989D-01
-0.80498D 00
0.34288D 00
-0.48270D-01
0.31655D 00
-0.23538D 00
0.31165D 00
0.69407D 00
-0.25273D 00
-0.53000D-01
-0.11549D 00
0.24163D 00
-0.28210D 00
0.72356D-C1
0.91698D-01
-0.66306D-01
0.44740D 00
-0.59596D 00
-0.38968D 00

0.10000D 01
-0.50077D 00
0.60326D-01
0.83312D 00
0.29229D 00
-0.79738D 00
-0.96141D 00
0.45220D 00
-0.15749D 00
-0.27218D 00

Fig 2.1.9 (cont)

MODE 21

EIGENVALUE= 0.45785D-C8
FREQUENCY= 0.22556D C4

EIGENVECTOR

0.29633D 00
-0.14395D-01
-0.62335D 00
-0.33120D 00
0.83962D-01
0.50065D 00
-0.32786D 00
0.20546D 00
0.10000D 01 ✓
0.40343D-01
0.23454D 00
-0.59190D 00
0.55728D 00
-0.46871D 00
0.49633D 00
-0.98886D 00
-0.19033D 00
-0.87373D-01
-0.13170D 00
0.35988D 00
-0.41554D 00
0.13327D 00
0.23279D 00
-0.42233D 00
-0.45819D-01
0.36570D 00
0.31577D 00
-0.36002D-01
-0.14649D 00
0.31151D 00

MODE 22

EIGENVALUE= 0.43724D-08
FREQUENCY= 0.24069D 04

EIGENVECTOR

-0.89953D 00
0.20868D-01
0.14067D 00
0.10000D 01 ✓
-0.96449D-01
-0.24325D 00
-0.17261D 00
0.10164D 00
-0.91934D-01
-0.86398D 00
0.49503D-01
-0.13673D 00
0.38232D 00
-0.13657D 00
0.18402D 00
-0.24665D 00
0.83028D 00
0.14571D-01
-0.25731D 00
-0.96069D 00
-0.11130D 00
0.47818D-01
0.10739D-01
-0.93059D 00
0.12709D-01
0.54131D-01
0.60703D 00
-0.17694D-01
-0.50612D 00
0.83061D-01

MODE 23

EIGENVALUE= 0.40889D-08
FREQUENCY= 0.24890D 04

EIGENVECTOR

0.19821D-01
0.27657D-01
-0.25663D 00
-0.62207D-01
0.20998D 00
0.92668D-01
0.15647D 00
-0.20655D 00
0.10000D 01 ✓
-0.22547D 00
0.55821D-01
0.63990D-01
-0.52884D 00
0.50086D 00
-0.41591D 00
-0.16324D 00
0.24730D 00
-0.22740D 00
0.94641D 00
-0.30088D 00
0.66220D-01
0.10271D 00
0.50709D-01
0.61336D-02
0.15471D-01
-0.67381D-03
-0.18370D-01
0.23108D-01
-0.42232D-01
0.21641D-01

MODE 24

EIGENVALUE= 0.38624D-08
FREQUENCY= 0.25609D 04

EIGENVECTOR

0.29810D-01
-0.39924D-02
-0.95755D-01
-0.15640D-01
-0.12252D 00
-0.60744D 00
0.55728D 00
-0.30854D 00
0.86164D 00
0.14747D 00
-0.23421D 00
0.10000D 01 ✓
0.52107D-01
0.13621D 00
-0.31906D 00
-0.16020D 00
-0.39650D 00
0.49309D 00
-0.79572D 00
0.36168D 00
0.28898D 00
-0.26591D 00
0.10872D-01
-0.63003D 00
0.61869D-01
-0.99529D-01
0.34535D 00
0.48296D-01
-0.44046D 00
0.13796D-02

Fig 8.1.9 (cont)

ODE 25

EIGENVALUE= 0.37384D-C8
FREQUENCY= 0.26030D 04

EIGENVECTOR

0.32361D 00
-0.79121D-02
-0.28139D-02
-0.42551D 00

MODE 26

EIGENVALUE= 0.30801D-08
FREQUENCY= 0.28677D 04

EIGENVECTOR

-0.20762D 00
-0.10200D-02
0.69473D-01
0.33373D 00

MODE 27

EIGENVALUE= 0.28635D-08
FREQUENCY= 0.29742D 04

EIGENVECTOR

0.79828D-01
0.70909D-02
0.76997D-01
-0.15698D 00

MODE 28

EIGENVALUE= 0.23246D-08
FREQUENCY= 0.33010D 04

EIGENVECTOR

-0.98732D-01
-0.26011D-01
0.52828D-01
0.28124D 00

0.59825D 00
0.58404D 00
-0.13791D 00
0.22481D 00
-0.64632D 00
0.23326D 00
-0.34503D 00
-0.64508D 00
-0.26244D 00
-0.30458D 00
0.47570D-02
0.35546D 00
0.42682D-01
0.29012D 00
0.78965D 00
-0.16924D 00
0.10000D 01
-0.25151D 00
-0.38978D-01
-0.96656D 00
0.14938D 00
-0.37510D 00
0.50508D 00
0.11806D 00
-0.82715D 00
-0.15572D 00

0.22976D 00
-0.35632D 00
-0.30267D 00
0.33801D 00
-0.14844D-01
0.53425D-02
-0.35178D 00
-0.48735D 00
0.40295D 00
0.26656D 00
-0.37185D 00
-0.11186D 00
-0.37687D 00
0.58599D 00
0.10000D 01
0.36876D 00
-0.27739D 00
-0.21633D 00
0.10598D-02
0.78740D-01
-0.27585D-01
0.74346D-01
-0.30177D-01
-0.23260D-01
0.10497D 00
0.37991D-01

0.10000D 01
0.27126D 00
0.18388D 00
-0.26305D 00
-0.19655D-01
-0.17403D 00
-0.79546D 00
0.31334D 00
-0.18265D 00
-0.14645D 00
0.21940D 00
-0.15011D 00
0.40326D 00
0.62870D 00
-0.56131D 00
-0.16791D 00
-0.56416D 00
-0.18718D 00
0.16383D-01
0.23120D 00
-0.53250D-01
0.14005D 00
-0.97192D-01
-0.43914D-01
0.22210D 00
0.62908D-01

-0.18404D 00
-0.91066D 00
-0.19976D 00
0.65309D 00
0.10000D 01
0.87141D 00
-0.82944D 00
-0.83847D 00
-0.91935D 00
-0.14454D 00
0.80692D 00
0.53981D-01
0.30351D 00
-0.31723D 00
-0.21630D 00
-0.13919D 00
-0.20373D-02
0.70430D-01
-0.51779D-02
0.37236D-02
-0.11986D-02
-0.27958D-02
-0.15171D-03
-0.18296D-02
0.33034D-02
-0.47605D-02

Fig 2.1.9 (cont)

MODE 29

MODE 30

EIGENVALUE= 0.19952D-C8
FREQUENCY= 0.35631D C4

EIGENVALUE= 0.14337D-08
FREQUENCY= 0.42032D 04

EIGENVECTOR

EIGENVECTOR

-0.13472D 00
0.29190D-01
-0.70724D-01
0.26747D 00
0.10000D 01
-0.14864D 00
-0.80140D 00
-0.48837D 00
-0.43554D-C1
0.48694D-01
-0.25322D-01
0.26053D 00
0.21286D 00
0.74745D 00
0.42405D 00
0.92738D-01
-0.42938D 00
-0.75788D C0
-0.47768D C0
0.17281D 00
0.35097D 00
0.17092D 00
-0.55597D-02
-0.63253D-C1
0.19707D-01
-0.53688D-01
0.21551D-01

-0.71511D-01
0.22505D-01
0.59460D-01
0.20888D 00
0.45062D 00
-0.33214D 00
-0.87414D 00
-0.47696D 00
0.24007D 00
0.60611D 00
0.10000D 01
0.42398D 00
-0.35814D 00
-0.90804D 00
-0.52015D 00
-0.10903D-01
0.27214D 00
0.45621D 00
0.11676D 00
-0.81480D-01
-0.13948D 00
-0.72688D-01
-0.32406D-03
0.14487D-01
-0.54739D-02
0.13305D-01
-0.38730D-02

0.15853D-01
-0.70547D-01
-0.26472D-01

-0.45130D-02
0.17915D-01
0.59894D-02

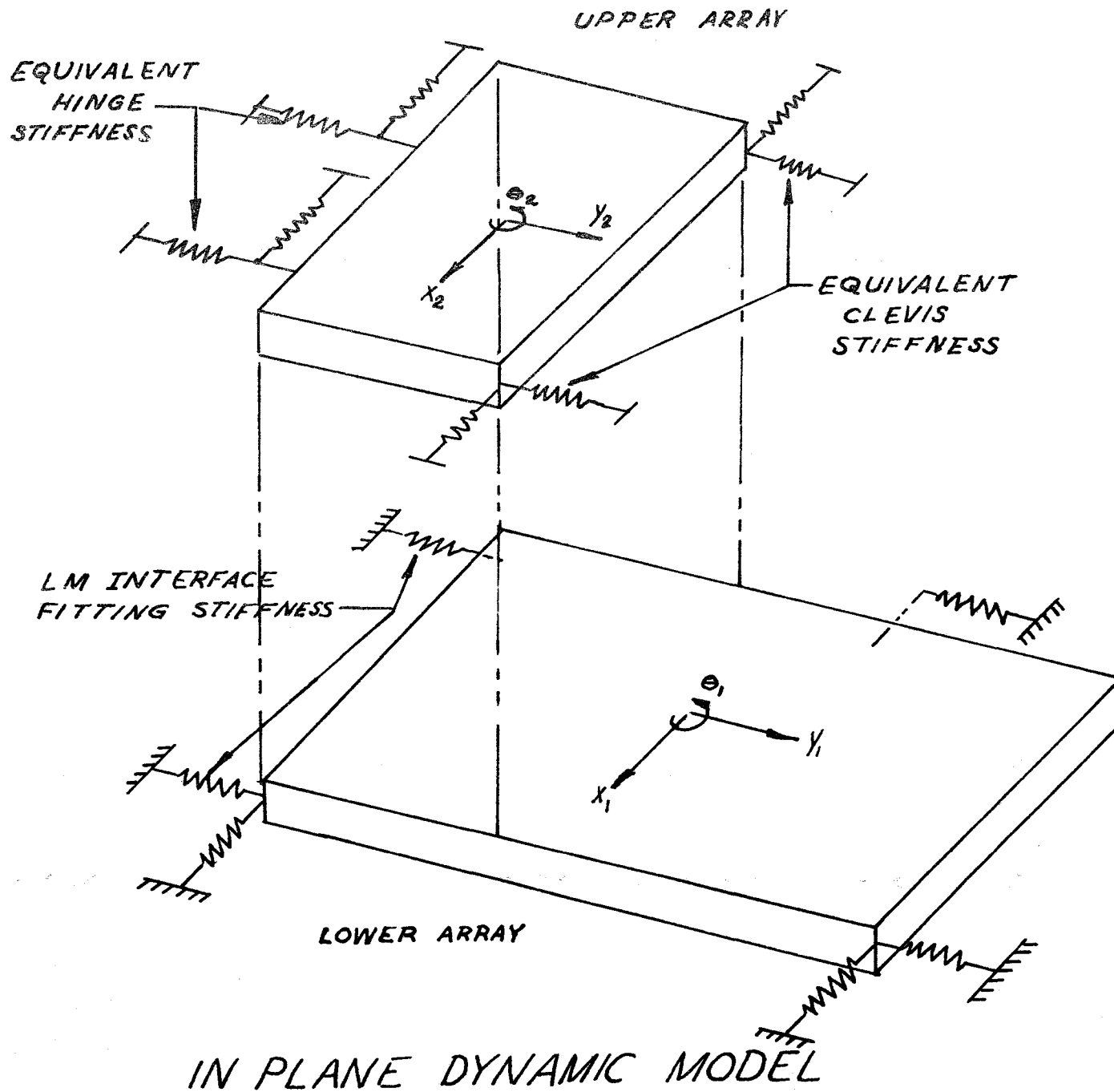
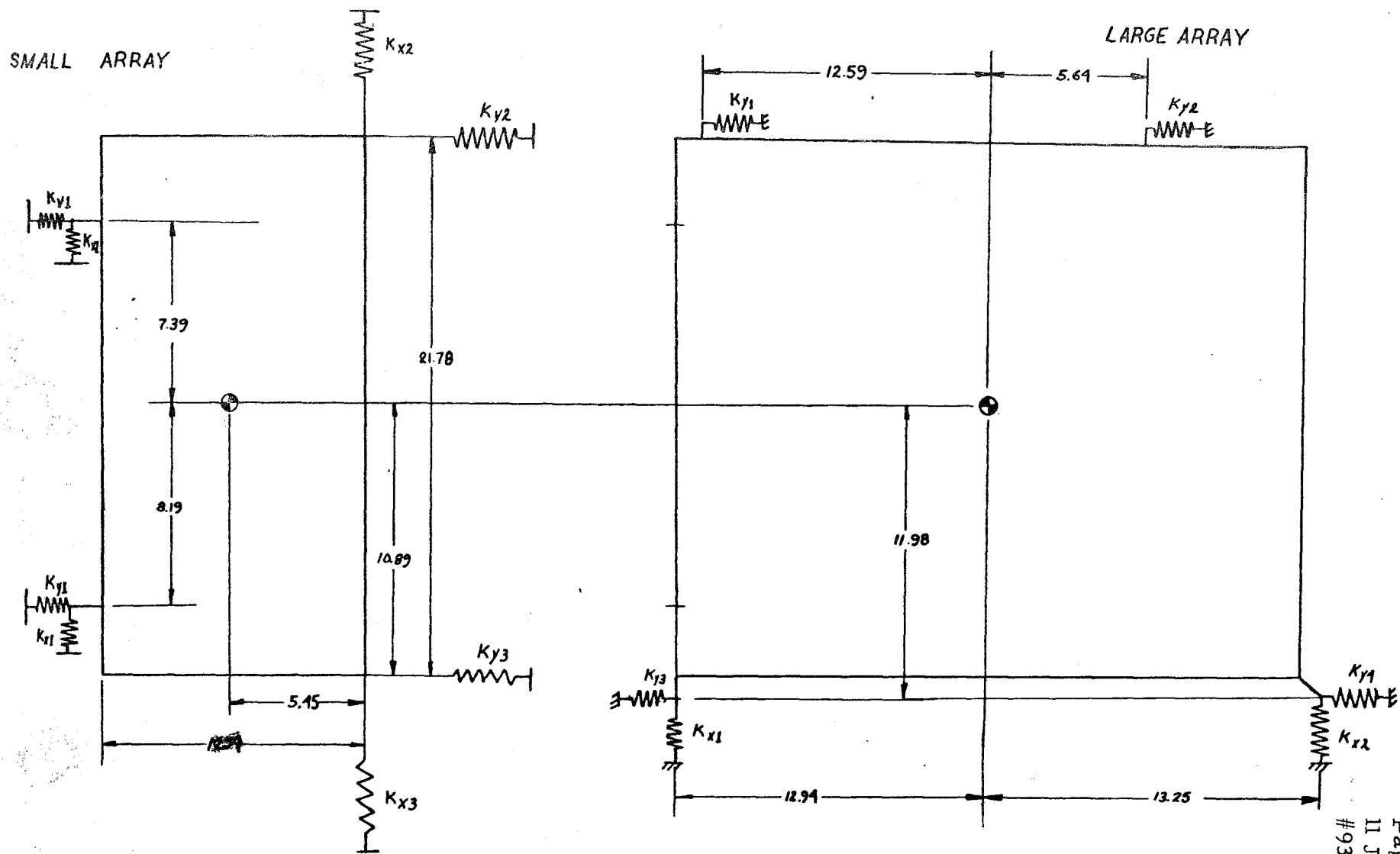


Figure 2.2.1



TOP VIEWS OF ARRAYS

Figure 2.2.2

FIGURE 2.23
STIFFNESS MATRICES
IN PLANE MODEL

Lower Array

$$[K_L] = \begin{bmatrix} k_{11} & k_{12} & k_{13} \\ k_{21} & k_{22} & k_{23} \\ k_{31} & k_{32} & k_{33} \end{bmatrix}$$

$$\begin{aligned} K_{X1} &= 8.00 \times 10^5 & K_{Y2} &= 8.00 \times 10^5 \\ K_{X2} &= 8.77 \times 10^5 & K_{Y3} &= 9.08 \times 10^5 \text{ (see Fig. 2.2.2)} \\ K_{Y1} &= 8.85 \times 10^5 & K_{Y4} &= 9.53 \times 10^5 \end{aligned}$$

$$k_{11} = K_{X1} + K_{X2}$$

$$k_{12} = k_{21} = 0$$

$$k_{31} = k_{13} = -12.94^2 K_{X1} + 13.25^2 K_{X2}$$

$$k_{22} = K_{Y1} + K_{Y2} + K_{Y3} + K_{Y4}$$

$$k_{23} = k_{32} = 10.89^2 (K_{Y1} + K_{Y2}) - 11.98^2 (K_{Y3} + K_{Y4})$$

$$k_{33} = 11.98^2 (K_{Y3} + K_{Y4}) + 10.89^2 (K_{Y1} + K_{Y2}) + 12.94^2 K_{X1} + 13.25^2 K_{X2}$$

$$[K_L] = \begin{bmatrix} 1.68 \times 10^6 & 0 & -1.99 \times 10^7 \\ 0 & 3.55 \times 10^6 & -6.65 \times 10^7 \\ -1.99 \times 10^7 & -6.65 \times 10^7 & 7.53 \times 10^5 \end{bmatrix}$$

Upper Array

$$[K_U] = \begin{bmatrix} k_{11} & k_{12} & k_{13} \\ k_{21} & k_{22} & k_{23} \\ k_{31} & k_{32} & k_{33} \end{bmatrix}$$

$$\begin{aligned} K_{X1} &= 1.52 \times 10^4 & K_{Y1} &= 1.76 \times 10^3 \\ K_{X2} &= 3.22 \times 10^2 & K_{Y2} &= 6.17 \times 10^4 \text{ (see Fig. 2.2.2)} \\ K_{X3} &= 1.71 \times 10^4 & K_{Y3} &= 3.61 \times 10^4 \end{aligned}$$

$$k_{11} = 2K_{X1} + K_{X2} + K_{X3}$$

$$k_{21} = k_{12} = 0$$

$$k_{22} = 2K_{Y1} + K_{Y2} + K_{Y3}$$

$$k_{32} = k_{23} = 7.39^2 K_{Y1} + 10.89^2 K_{Y2} - 8.19^2 K_{Y1} - 10.9 K_{Y3}$$

$$k_{33} = 10.89^2 (K_{Y2} + K_{Y3}) + K_{Y1} (7.39^2 + 8.19^2) + 5.45^2 (2K_{X1} + K_{X2} + K_{X3})$$

$$k_{31} = k_{13} = 5.45^2 (K_{X2} + K_{X3}) - 5.45^2 (2 \times K_{X1})$$

$$[K_U] = \begin{bmatrix} 4.78 \times 10^4 & 0 & -3.88 \times 10^5 \\ 0 & 1.01 \times 10^5 & 3.02 \times 10^6 \\ -3.88 \times 10^5 & 3.02 \times 10^6 & 1.18 \times 10^7 \end{bmatrix}$$



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The coordinate transformation employed is:

$$\begin{Bmatrix} X_L \\ Y_L \\ \theta_L \\ X_{ru} \\ Y_{ru} \\ \theta_{ru} \end{Bmatrix} = [\beta] \begin{Bmatrix} X_1 \\ Y_1 \\ \theta_1 \\ X_2 \\ Y_2 \\ \theta_2 \end{Bmatrix}$$

$$\beta = \begin{bmatrix} 1 & 0 & 0 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 & 0 & 0 \\ 0 & 0 & 1 & 0 & 0 & 0 \\ -1 & 0 & -7.49 & 1 & 0 & 0 \\ 0 & -1 & -7.49 & 0 & 1 & 0 \\ 0 & 0 & -1 & 0 & 0 & 1 \end{bmatrix}$$

Where

X_L, Y_L, θ_L = Displacements of the lower array relative to ground

$X_{ru}, Y_{ru}, \theta_{ru}$ = Displacements of the upper array relative to the lower array

X_1, Y_1, θ_1 = Generalized coordinates of the lower array relative to ground

X_2, Y_2, θ_2 = Generalized coordinates of the upper array relative to ground

The mass matrix is diagonal; made up of the mass or moment of inertia of the appropriate array.



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$$[M] = \begin{bmatrix} M_L & & & & & & & \\ & M_L & & & & & & \\ & & I_L & & & & & \\ & & & M_u & & & & \\ & & & & M_u & & & \\ & & & & & I_u & & \\ & & & & & & & \\ & .1555 & & & & & & \\ & & .1555 & & & & & \\ & & & 15.0 & & & & \\ & & & & .0726 & & & \\ & & & & & .0726 & & \\ & & & & & & 3.27 & \end{bmatrix}$$

Where

M_u = Mass of the upper array in lb sec²

M_L = Mass of the lower array in lb sec²

I_L = Polar moment of inertia of the lower array,
lb sec² in.

I_u = Polar moment of inertia of the upper array,
lb sec² in.

The resulting eigenvalue problem could then be solved giving the mode shapes and natural frequencies of the system. These are shown in Figure 2.2.4. Modal analysis was used to calculate response of the system in the frequency domain. In calculating responses, 10% of critical viscous damping was used on all but mode 4 which was assigned 20% of critical damping. The value of 10% has been shown by experiment to be typical of such structures. Mode 4 was assigned 20% of critical damping since this mode will cause the rear LM interface fittings to slide on their pins and introduce more damping to the structure.

Figure 2.2.4

MODE 1	MODE 2	MODE 3	MODE 4
EIGENVALUE= 0.14891D-05	EIGENVALUE= 0.61479D-06	EIGENVALUE= 0.15536D-06	EIGENVALUE= 0.83034D-07
FREQUENCY= 0.13042D 03	FREQUENCY= 0.20298D 03	FREQUENCY= 0.40379D 03	FREQUENCY= 0.55232D 03
EIGENVECTOR	EIGENVECTOR	EIGENVECTOR	EIGENVECTOR
0.14240D-01	0.29105D 00	0.23427D 00	0.10000D 01
-0.27280D-01	0.18588D-01	0.50405D 00	-0.26126D 00
-0.15979D-02	-0.31334D-02	0.57902D-01	0.43344D-02
0.10000D 01	-0.27939D 00	-0.46694D-01	-0.55388D-02
0.10822D 00	0.10000D 01	0.10000D 01	-0.30928D 00
-0.74693D-02	-0.42419D 00	0.52526D-01	0.13915D-01
MODE 5	MODE 6		
EIGENVALUE= 0.36088D-C7	EIGENVALUE= 0.97458D-08		
FREQUENCY= 0.83780D 03	FREQUENCY= 0.16122D 04		
EIGENVECTOR	EIGENVECTOR		
0.32363D-01	0.27979D 00		
-0.45351D 00	0.10000D 01		
-0.55684D-01	-0.92726D-01		
-0.13514D 00	0.35371D-01		
0.10000D 01	-0.12617D 00		
0.49501D-01	0.63589D-02		



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2.3 Out-of-Plane Model of Off-Loaded Condition

To model the experiment with the upper array removed, a lumped mass model was again employed. Masses were lumped at 23 locations as shown in Figure 2.3.1. These are the same locations as used for the model of the lower array with the arrays connected. The stiffness matrix calculated for the lower array alone could then be employed directly. Since the attachments are no longer constrained to move in a plane by the upper array, the degree of freedom eliminated by the constraint can no longer be eliminated. The mass matrix becomes a diagonal matrix whose elements are simply the mass lumped at each node. Mass and stiffness matrices for this model are shown in Figure 2.3.2.

Solving the eigenvalue problem again gave mode shapes and frequencies which were used to calculate the frequency response of the system. The mode shapes and frequencies are shown in Figure 2.3.3. Again 10% of critical damping was used in calculating the system's response.

2.4 Vibration Environment

The qualification and acceptance vibration environments to be imposed on the LRRR 300 corner array in tests is that of the LM quadrant III. This then is the environment that was input to the base of the three models. The environment was taken from Exhibit B-1, Design and Performance Specification for the Laser Ranging Retro-Reflector Experiment (300 Corner Array), November 1, 1970. This environment is listed in Table 2.4.1 and the qualification levels shown in Figure 2.4.1.

The responses that were calculated here are responses to the qualification environment. Since the acceptance environment is less severe, the qualification environment then determines the dynamic loads on the structure.

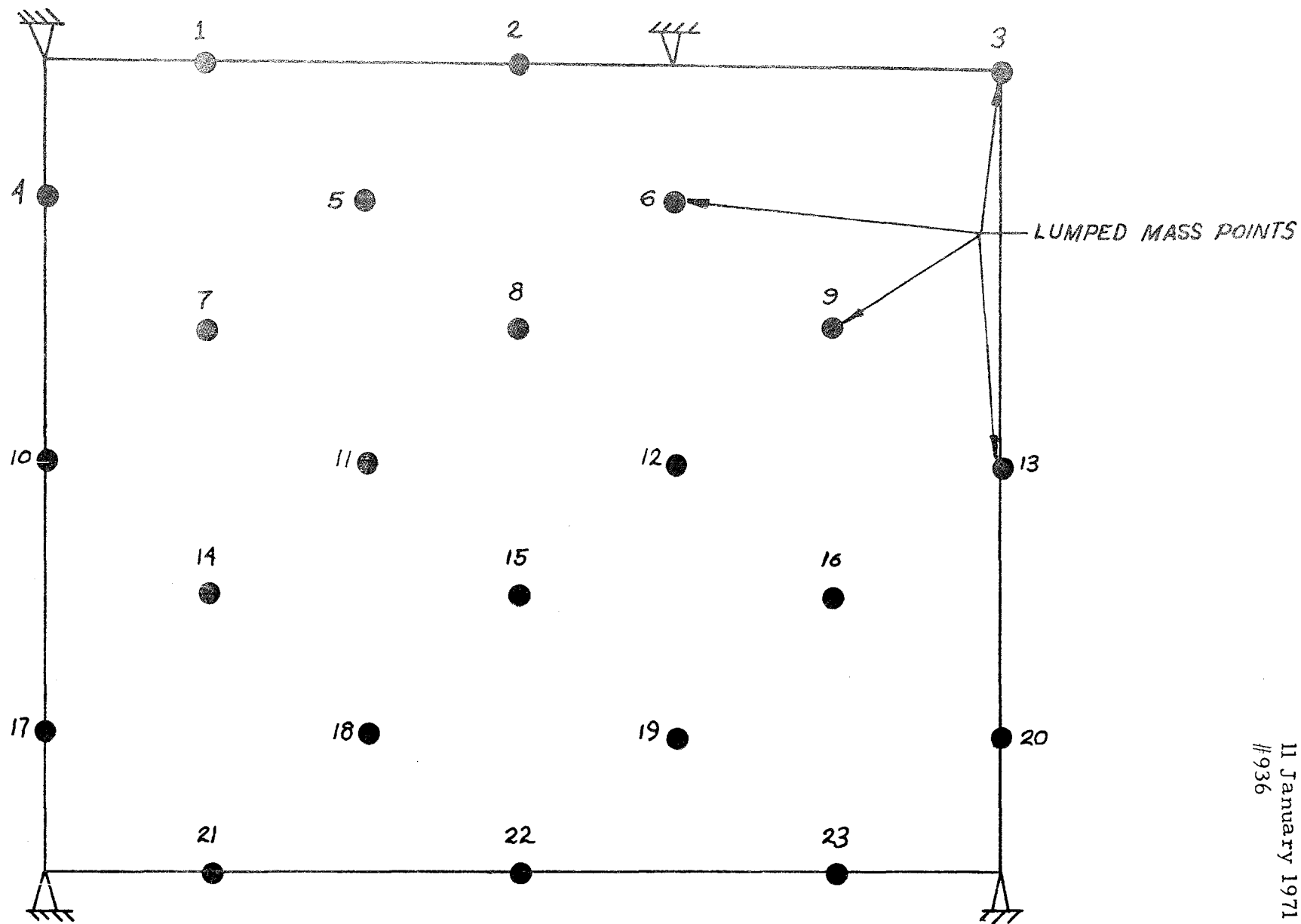


Figure 2.3.1
 MODEL OF LR^3 IN OFF LOADED CONDITION

MASS MATRIX -OFF LOADED CONDITION

Figure 2.3.2-

1	0.60100D-02	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
2	0.0	0.67200D-02	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
3	0.0	0.0	0.73500D-02	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
4	0.0	0.0	0.0	0.40600D-02	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
5	0.0	0.0	0.0	0.0	0.96200D-02	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
6	0.0	0.0	0.0	0.0	0.0	0.70300D-02
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
7	0.0	0.0	0.0	0.0	0.0	0.0
	0.70900D-02	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
8	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.69600D-02	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
9	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.70900D-02	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
10	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.35400D-02	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
11	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.96200D-02	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
12	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.70300D-02
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0

Figure 2.3.2 (cont.)

	0.0	0.0	0.0	0.0	0.0	0.0
	0.354000-02	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
14	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.709000-02	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
15	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.696000-02	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
16	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.709000-02	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
17	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.406000-02	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
18	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.962000-02
	0.0	0.0	0.0	0.0	0.0	0.0
19	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.703000-02	0.0	0.0	0.0	0.0	0.0
20	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.354000-02	0.0	0.0	0.0	0.0
21	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.452000-02	0.0	0.0	0.0
22	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.555000-02	0.0	0.0
23	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.678000-02	0.0

Figure 2.3.2 (cont)

ROW 1	0.5445000E	06-C.1206000E	06-C.5851000E	04-0.2205000E	05-0.5017000E	06-0.2842000E	05 0.1176000E	06 0.1301000E	06
	0.5930999E	03-0.1526000E	05 0.9282000E	05-0.2132000E	05 0.7447998E	03-0.1563000E	05-0.1568000E	05 0.3144000E	04
	0.7167000E	04-0.1353000E	05 0.5021000E	04-0.1187000E	04 0.3596000E	04 0.2061000E	04-0.1595000E	04	
ROW 2	-0.1206000E	06 0.7842000E	06 0.7838000E	05-0.2488000E	05-0.5536000E	06-0.3121000E	05 0.1313000E	06 0.1422000E	06
	0.1300000E	04-0.1730000E	05 0.1024000E	06-0.2342000E	05 0.5702998E	03-0.1719000E	05-0.1732000E	05 0.3497000E	04
	0.8006000E	04-C.1495000E	05 0.5515000E	04-0.1238000E	04 0.3970000E	04 0.2281000E	04-0.1755000E	04	
ROW 3	-0.5852000E	04 C.7838000E	05 0.5300000E	05-0.4087998E	03-0.1366000E	05 0.3916000E	05 0.2646000E	04 0.3131000E	04
	-0.9961000E	05-0.3732998E	03 0.1695000E	04 0.3019000E	05 0.1837000E	04-0.5665999E	02-0.4842000E	04 0.8297000E	04
	0.4417000E	02 0.5647998E	03-C.7134000E	04 0.2353000E	05-0.3182998E	03 0.1194000E	04 0.1464000E	04	
ROW 4	-0.2205000E	05-0.2488000E	05-0.4087000E	03 0.7975000E	06 0.1417000E	06-0.1002000E	05-0.4063000E	06 0.6982000E	04
	-0.2198000E	03-0.3078000E	06 0.1235000E	06-0.2623000E	04 0.1654000E	03 0.5358000E	05-0.1932000E	05 0.1458000E	04
	0.1350000E	06-0.3747000E	05 C.3983000E	04-0.6830999E	03 0.7484000E	04 0.5795000E	04-0.1941000E	04	
ROW 5	-0.5017000E	06-C.5536000E	06-0.1367000E	05 0.1417000E	06 0.1766000E	07 0.1793000E	06-0.7495000E	06-0.8201000E	06
	-0.5115000E	04 0.1048000E	06-0.9949000E	04 0.1435000E	06-0.4637000E	04 0.5770000E	05 0.5530000E	05-0.1929000E	05
	-0.3808000E	05 0.5707000E	05-0.2524000E	05 0.6875000E	04-0.1509000E	05-0.7523000E	04 0.7793000E	04	
ROW 6	-0.2842000E	05-C.3121000E	05 C.3916000E	05-0.1002000E	05 0.1793000E	06 0.2226000E	07 C.5540000E	04-0.9413000E	06
	-0.7820000E	06-0.2921000E	04 0.1687000E	06-0.1272000E	06 0.1834000E	06-0.2411000E	05 0.7394000E	05 0.6208000E	05
	0.7106000E	04-0.3123000E	05 0.7826000E	05-0.7293000E	05 0.9883000E	04-0.1068000E	05-0.2021000E	05	
ROW 7	0.1176000E	06 0.1313000E	06 0.2646000E	04-0.4063000E	06-0.7495000E	06 0.5545000E	04 0.1445000E	07 0.3069000E	06
	0.2034000E	04-0.3527000E	06-0.8160000E	06-0.1876000E	05 0.1664000E	03 0.1371000E	06 0.1061000E	06-0.1514000E	04
	0.5364000E	05 C.5589000E	05-0.1574000E	05 0.1410000E	04-0.1638000E	05-0.1145000E	05 0.5217000E	04	
ROW 8	0.1301000E	06 0.1422000E	06 C.3131000E	04 0.6981000E	04-0.8201000E	06-0.9413000E	06 0.3069000E	06 0.1989000E	07
	0.3377000E	06-0.1266000E	05-0.9011000E	06-0.8722000E	06-0.1638000E	05 0.1068000E	06 0.2265000E	06 0.1020000E	06
	-0.1973000E	05 0.5962000E	05 0.5494000E	05-0.1779000E	05-0.1455000E	05-0.2862000E	05-0.1324000E	05	
ROW 9	0.5940000E	03 C.1301000E	04-C.9961000E	05-0.2191000E	03-0.5116000E	04-0.7820000E	06 0.2033000E	04 0.3377000E	06
	0.1261000E	07 0.5237000E	03-0.2473000E	05-0.7327000E	06-0.5146000E	06-0.1038000E	03 0.9286000E	05 0.1614000E	06
	0.7582998E	03-C.1260000E	05 0.3374000E	05 0.1247000E	06 0.3872000E	04-0.8036000E	04-C.1177000E	05	
ROW 10	-0.1526000E	05-0.1730000E	05-C.3732998E	03-0.3078000E	06 0.1048000E	06-0.2923000E	04-0.3527000E	06-0.1266000E	05
	0.5230999E	03 C.6246000E	06 C.2464000E	06-0.7040000E	04 0.2272000E	03-C.3524000E	06-0.1362000E	05-0.6138999E	03
	-0.3079000E	06 0.1042000E	06 C.1169000E	04 0.2399000E	03-0.1579000E	05-0.1417000E	05 0.2423000E	04	
ROW 11	0.9281000E	05 0.1024000E	06 0.1695000E	04 0.1235000E	06-0.9936000E	04 0.1687000E	06-0.8160000E	06-0.9011000E	06
	-0.2474000E	05 0.2464000E	06 0.2208000E	07 0.3481000E	06-0.7145000E	04-0.8191000E	06-C.8910000E	06-0.1137000E	05
	0.1243000E	06-C.2584000E	04 0.1244000E	06-0.4280000E	04 0.9867000E	05 0.6590000E	05-0.3472000E	05	
ROW 12	-0.2132000E	05-0.2342000E	05 0.3019000E	05-0.2625000E	04 0.1436000E	06-0.1272000E	06-0.1877000E	05-0.8722000E	06
	-0.7327000E	06-0.7040000E	04 0.3481000E	06 0.2206000E	07 0.2895000E	06-0.1175000E	05-0.8917000E	06-C.8299000E	06
	-0.4386000E	04 0.1243000E	06 0.6353999E	03 0.1055000E	06-0.3460000E	05 0.6550000E	05 0.9818000E	05	
ROW 13	0.7445000E	03 C.5697000E	03 0.1837000E	04 0.1655000E	03-0.4636000E	04 0.1834000E	06 0.1676000E	03-0.1638000E	05
	-0.5146000E	06 0.2257000E	03-0.7146000E	04 0.2895000E	06 0.5070000E	06-0.1709000E	03-0.2029000E	05-0.3310000E	06
	0.4587000E	02 0.2478000E	04 0.9040000E	05-0.2563000E	06 0.1750000E	04-0.1205000E	05-0.1308000E	05	
ROW 14	-0.1563000E	05-0.1719000E	05-0.5639000E	02 0.5358000E	05 0.5769000E	05-0.2411000E	05 0.1371000E	06 0.1068000E	06
	-0.1037000E	03-0.3524000E	06-0.8191000E	06-0.1176000E	05-0.1732000E	03 0.1444000E	07 0.3128000E	06 0.9295000E	04
	-0.4060000E	06-C.7442000E	06-0.2478000E	05-0.2299000E	04 0.1211000E	06 0.1095000E	06-0.1610000E	05	

Figure 2.3.2 (cont)

OW 15-	0.1568000E	05-	0.1731000E	05-	0.4841000E	04-	0.1932000E	05	0.5529000E	05	0.7394000E	05	0.1061000E	06	0.2265000E	06
	0.9285000E	05-	0.1362000E	05-	0.8909000E	06-	0.8917000E	06-	0.2028000E	05	0.3128000E	06	0.1970000E	07	0.3147000E	08
	0.5634000E	04-	0.8403000E	06-	0.8407000E	06	0.8517000E	04	0.1164000E	06	0.2380000E	06	0.1164000E	06		
OW 16	0.3146000E	04	0.2501000E	04	0.8298000E	04	0.1456000E	04-	0.1930000E	05	0.6208000E	05-	0.1510000E	04	0.1020000E	06
	0.1614000E	06-	0.6132998E	03-	0.1137000E	05-	0.8299000E	06-	0.3310000E	06	0.9292000E	04	0.3147000E	06	0.1440000E	07
	-0.2040000E	04-	0.2519000E	05-	0.7414000E	06-	0.4153000E	06-	0.1594000E	05	0.1091000E	06	0.1205000E	06		
OW 17	0.7166000E	04	0.8004000E	04	0.4414000E	02	0.1350000E	06-	0.3808000E	05	0.7104000E	04	0.5364000E	05-	0.1973000E	05
	0.7583999E	03-	0.3079000E	06	0.1243000E	06-	0.4385000E	04	0.4587999E	02-	0.4060000E	06	0.5631000E	04-	0.2041000E	04
	0.7974000E	06	0.1405000E	06-	0.2891000E	04	0.9259999E	03-	0.2288000E	05-	0.1962000E	05	0.4314000E	04		
OW 18-	0.1353000E	05-	0.1495000E	05	0.5637000E	03-	0.3747000E	05	0.5707000E	05-	0.3134000E	05	0.5590000E	05	0.5963000E	05
	-0.1260000E	05	0.1042000E	06-	0.2602000E	04	0.1243000E	06	0.2477000E	04-	0.7443000E	06-	0.8403000E	06-	0.2518000E	05
	0.1405000E	06	0.1750000E	07	0.2755000E	06-	0.3451000E	04-	0.5130000E	06-	0.4875000E	06	0.3926000E	05		
OW 19	0.5019000E	04	0.5511000E	04-	0.7135000E	04	0.3984000E	04-	0.2523000E	05	0.7826000E	05-	0.1574000E	05	0.5494000E	05
	0.3374000E	05	0.1172000E	04	0.1244000E	06	0.6213999E	03	0.9040000E	05-	0.2478000E	05-	0.8407000E	06-	0.7414000E	06
	-0.2892000E	04	0.2755000E	06	0.1749000E	07	0.1465000E	06	0.3923000E	05-	0.4874000E	06-	0.5127000E	06		
OW 20-	0.1188000E	04-	0.1240000E	04	0.2353000E	05-	0.6825000E	03	0.6879000E	04-	0.7293000E	05	0.1407000E	04-	0.1779000E	05
	0.1247000E	06	0.2406000E	03-	0.4280000E	04	0.1055000E	06-	0.2563000E	06-	0.2300000E	04	0.8516000E	04-	0.4153000E	06
	0.9253999E	03-	0.3447000E	04	0.1465000E	06	0.7748000E	06	0.4608000E	04-	0.2056000E	05-	0.2406000E	05		
OW 21	0.3596000E	04	0.3971000E	04-	0.3180000E	03	0.7485000E	04-	0.1509000E	05	0.9884000E	04-	0.1638000E	05-	0.1455000E	05
	0.3869000E	04-	0.1579000E	05	0.9867000E	05-	0.3460000E	05	0.1751000E	04	0.1211000E	06	0.1163000E	06-	0.1594000E	05
	-0.2287000E	05-	0.5130000E	06	0.3923000E	05	0.4609000E	04	0.5365000E	06-	0.7153000E	05	0.3709000E	05		
OW 22	0.2061000E	04	0.2283000E	04	0.1195000E	04	0.5795000E	04-	0.7526000E	04-	0.1068000E	05-	0.1145000E	05-	0.2863000E	05
	-0.8038000E	04-	0.1418000E	05	0.6590000E	05	0.6551000E	05-	0.1203000E	05	0.1095000E	06	0.2380000E	06	0.1091000E	06
	-0.1962000E	05-	0.4875000E	06-	0.4874000E	06-	0.2056000E	05-	0.7152000E	05	0.5179000E	06-	0.7155000E	05		
OW 23-	0.1595000E	04-	0.1755000E	04	0.1464000E	04-	0.1942000E	04	0.7791000E	04-	0.2021000E	05	0.5218000E	04-	0.1324000E	05
	-0.1177000E	05	0.2423000E	04-	0.3472000E	05	0.9818000E	05-	0.1308000E	05-	0.1611000E	05	0.1164000E	06	0.1205000E	06
	0.4314000E	04	0.3926000E	05-	0.5127000E	06-	0.2406000E	05	0.3709000E	05-	0.7155000E	05	0.5365000E	06		

Figure 2.3.3

MODE SHAPES & FREQUENCIES

MODE 1	MODE 2	MODE 3	MODE 4
EIGENVALUE= 0.83674D-06	EIGENVALUE= 0.51348D-06	EIGENVALUE= 0.16389D-06	EIGENVALUE= 0.13964D-06
FREQUENCY= 0.17399D 03	FREQUENCY= 0.22219D 03	FREQUENCY= 0.39314D 03	FREQUENCY= 0.42590D 03
EIGENVECTOR	EIGENVECTOR	EIGENVECTOR	EIGENVECTOR
-0.90204D-01	-0.20004D 00	0.31109D-03	-0.40735D 00
-0.13762D 00	-0.25066D 00	-0.91758D-01	-0.33294D 00
0.71584D 00	0.10000D 01	0.94892D 00	0.17524D 00
0.95291D-01	-0.20896D 00	0.50234D 00	-0.48512D-01
0.66816D-01	-0.30848D 00	0.16656D 00	-0.56015D 00
0.22629D 00	0.18263D-01	-0.52331D-01	-0.39533D 00
0.21788D 00	-0.35448D 00	0.59223D 00	-0.30404D 00
0.33713D 00	-0.17372D 00	0.52449D-01	-0.58221D 00
0.55169D 00	0.29576D 00	-0.17952D 00	-0.69318D 00
0.21328D 00	-0.40920D 00	0.10000D 01	-0.33638D-01
0.45160D 00	-0.28715D 00	0.32410D 00	-0.28049D 00
0.58776D 00	0.24122D-02	-0.20082D 00	-0.59662D 00
0.62038D 00	0.43156D 00	-0.45753D 00	-0.90200D 00
0.42798D 00	-0.31837D 00	0.54820D 00	0.63216D-01
0.70539D 00	-0.16507D 00	-0.67280D-01	-0.79672D-01
0.62235D 00	0.11517D 00	-0.42832D 00	-0.55596D 00
0.11809D 00	-0.20055D 00	0.50021D 00	0.14805D-01
0.74594D 00	-0.22242D 00	0.89943D-02	0.46829D 00
0.79965D 00	-0.92635D-01	-0.36043D 00	0.20677D 00
0.26468D 00	0.13665D 00	-0.31494D 00	-0.47065D 00
0.49909D 00	-0.10611D 00	-0.12352D 00	0.55143D 00
0.10000D 01	-0.18991D 00	-0.39133D 00	0.10000D 01
0.50639D 00	-0.84837D-01	-0.27247D 00	0.45548D 00

ODE 5

MODE 6

Figure 2.3.3 (cont)

MODE 7

MODE 8

 EIGENVALUE= 0.61094D-07
 FREQUENCY= 0.64391D 03

 EIGENVALUE= 0.44504D-07
 FREQUENCY= 0.75443D 03

 EIGENVALUE= 0.28968D-07
 FREQUENCY= 0.93511D 03

 EIGENVALUE= 0.14667D-07
 FREQUENCY= 0.13142D 04

EIGENVECTOR

EIGENVECTOR

EIGENVECTOR

EIGENVECTOR

 -0.36612D 00
 -0.23432D 00
 -0.41674D 00
 0.47744D 00
 -0.41894D 00
 -0.13481D 00
 0.15106D 00
 -0.46156D 00
 0.23713D 00
 0.99220D 00
 -0.25569D 00
 -0.20705D 00
 0.10000D 01
 0.31115D 00
 -0.32683D 00
 0.33842D 00
 -0.52192D 00
 -0.59786D-01
 -0.64324D-01
 0.55054D 00

 0.10000D 01
 0.70609D 00
 0.36683D 00
 0.22702D 00
 0.57569D 00
 -0.21555D 00
 0.19344D 00
 -0.28885D 00
 -0.11832D 00
 0.28125D 00
 -0.42126D 00
 -0.36438D 00
 0.36168D 00
 -0.25976D 00
 -0.42480D 00
 0.11062D 00
 0.57195D-01
 -0.23562D 00
 0.22730D 00
 0.21797D 00

 0.42401D 00
 0.30457D 00
 0.11114D-02
 -0.18937D 00
 0.10396D 00
 0.66965D-01
 -3.24147D 00
 -0.82306D-01
 0.24528D 00
 -0.16557D 00
 -0.16803D 00
 -0.87336D-01
 0.38483D 00
 0.15988D 00
 -0.13662D 00
 -0.16658D 00
 0.21293D-01
 0.60388D 00
 -0.58048D 00
 0.11707D 00

 -0.24221D 00
 -0.16309D 00
 0.10651D 00
 -0.22748D 00
 0.23996D 00
 -0.74162D 00
 0.47813D 00
 -0.24351D 00
 -0.84562D 00
 -0.91814D 00
 0.70650D 00
 -0.71487D 00
 0.10000D 01
 0.36759D-01
 0.12754D 00
 0.31573D 00
 -0.72466D 00
 0.21326D 00
 -0.81339D-01
 0.95990D 00

 0.19452D-01
 0.25998D-01
 0.67508D-02

 0.13072D-01
 0.53210D 00
 0.55492D 00

 0.10000D 01
 0.20662D 00
 -0.95796D 00

 -0.92317D-01
 -0.21629D 00
 -0.16143D 00

Figure 2.3.3(cont)

DE 9

MODE 10

MODE 11

MODE 12

GENVALUE= 0.13178D-07
EQUENCY= 0.13864D 04

EIGENVALUE= 0.12370D-07
FREQUENCY= 0.14313D 04

EIGENVALUE= 0.94401D-08
FREQUENCY= 0.16381D 04

EIGENVALUE= 0.82991D-08
FREQUENCY= 0.17470D 04

EIGENVECTOR

EIGENVECTOR

EIGENVECTOR

EIGENVECTOR

-0.36470D 00
-0.19283D 00
-0.16549D-01
0.10000D 01
0.77293D-01
0.12037D-01
0.69896D 00
0.31427D-02
-0.14574D-01
0.21645D 00
-0.18790D 00
0.20888D 00
-0.28011D 00
-0.84326D 00
0.47337D-01
0.13608D 00
-0.77434D 00
-0.12462D 00
0.91333D-01
-0.11779D 00
0.48775D 00
0.34001D 00
-0.51520D 00

0.55885D 00
0.23876D 00
0.17493D 00
0.44703D-01
-0.27691D 00
-0.53252D 00
-0.40584D 00
-0.42650D 00
-0.82736D 00
0.44843D 00
-0.33194D 00
0.50627D 00
-0.77150D 00
0.96536D-01
0.64064D 00
0.10000D 01
0.42450D 00
0.51658D-01
0.64176D 00
0.18277D 00
-0.12301D 00
-0.53148D 00
-0.81419D 00

-0.85620D 00
-0.13134D 00
0.14345D 00
-0.80343D 00
0.45617D 00
0.26777D 00
-0.19321D 00
0.10000D 01
-0.74636D 00
0.11349D 00
-0.96323D-01
0.57109D-01
-0.39232D 00
0.83941D-01
-0.90840D 00
0.60816D 00
0.92474D 00
-0.46578D 00
-0.58334D-01
0.72324D 00
0.22759D 00
0.94212D 00
-0.46316D 00

-0.19706D 00
0.11280D 00
0.40350D-01
0.39864D-01
0.50780D-01
0.21403D-01
-0.11123D-01
0.11392D 00
-0.15542D 00
0.53844D-01
-0.86560D-01
0.80084D-01
-0.12647D 00
-0.51824D-01
-0.82624D-01
0.16027D 00
0.18662D-01
0.53485D-01
-0.85352D-01
0.15503D 00
0.10000D 01
-0.72185D 00
0.55078D 00

MODE 13

MODE 14

Figure 2.33 (cont)

MODE 15

MODE 16

EIGENVALUE= 0.80027D-08
FREQUENCY= 0.17791D 04EIGENVALUE= 0.55243D-08
FREQUENCY= 0.21413D 04EIGENVALUE= 0.42705D-08
FREQUENCY= 0.24355D 04EIGENVALUE= 0.30931D-08
FREQUENCY= 0.25508D 04

EIGENVECTOR

EIGENVECTOR

EIGENVECTOR

EIGENVECTOR

-0.78458D 00
0.10000D 01
0.75056D-01
0.16149D-02
0.28308D-01
-0.71201D-01
-0.24007D-01
-0.15722D 00
0.94113D-01
-0.65267D-02
-0.25345D-01
-0.83915D-02
0.47449D-01
0.26606D-01
0.12847D 00
-0.62432D-01
-0.86262D-02
0.47749D-01

0.17569D 00
0.29950D 00
0.86292D-01
0.10000D 01
-0.36131D 00
0.47175D 00
-0.40049D 00
0.44427D 00
-0.35471D 00
0.11064D 00
-0.52849D-01
0.48938D-01
-0.31532D 00
0.23283D 00
-0.23410D 00
0.11532D 00
-0.86856D 00
0.36201D 00

-0.16626D-01
-0.21247D-01
0.29343D-01
0.22765D 00
0.27546D-01
-0.32421D-01
-0.52442D 00
0.58780D 00
-0.50017D 00
0.71663D 00
-0.58075D-01
-0.88201D-01
0.99890D 00
-0.87734D 00
0.10000D 01
-0.88218D 00
0.79118D 00
-0.28562D-01

-0.51834D 00
-0.63172D 00
-0.31797D-01
-0.20960D 00
0.96892D 00
-0.13072D 00
0.17630D 00
-0.17643D 00
0.43999D-01
-0.39567D-01
-0.98889D 00
0.20446D 00
-0.18333D 00
-0.10276D 00
0.11295D 00
0.69786D-01
0.16676D 00
0.10000D 01

0.57294D-01
-0.13340D 00
-0.20521D 00
0.68097D-02
-0.76903D-01

-0.42925D 00
0.95350D 00
-0.59608D 00
0.83093D-01
0.22397D 00

-0.51153D-01
0.63170D 00
0.13997D-01
0.21571D-01
0.13746D-01

-0.29692D 00
0.15954D 00
-0.99331D 00
-0.36049D 00
0.22613D 00

MODE 17

MODE 18

Figure 2.3.3 (cont)

MODE 19

MODE 20

EIGENVALUE= 0.36477D-06
FREQUENCY= 0.26352D 04EIGENVALUE= 0.31192D-06
FREQUENCY= 0.28497D 04EIGENVALUE= 0.30352D-06
FREQUENCY= 0.28869D 04EIGENVALUE= 0.27647D-06
FREQUENCY= 0.30269D 04

EIGENVECTOR

EIGENVECTOR

EIGENVECTOR

EIGENVECTOR

0.77714D-01
0.94995D-01
0.10972D-01
-0.43661D 00
-0.16524D 00
0.19264D 00
0.38983D 00
-0.32633D 00
0.19558D 00
-0.62514D-01
0.69309D-01
0.55155D-01
-0.62007D 00
-0.22202D 00
0.16115D 00
-0.10039D 00
0.30040D 00
-0.13370D-01
-0.17335D 00
0.10000D 01
-0.92712D-02
0.80570D-01
0.60604D-01

-0.23342D-02
-0.26749D-02
0.16663D-02
-0.84935D 00
-0.35024D-03
0.27617D-01
-0.32292D-02
0.36361D-01
-0.36439D-01
0.10000D 01
-0.43316D-01
-0.56466D-02
0.61587D-01
0.19190D-01
0.39230D-01
-0.53322D-01
-0.86958D 00
-0.15180D-01
0.42415D-01
0.16663D-01
0.12044D-01
-0.11754D-01
-0.14523D-01

-0.13269D 00
-0.15058D 00
-0.11998D-01
0.25521D 00
0.29764D 00
0.71438D 00
-0.19872D 00
-0.90518D-01
0.18349D 00
-0.11964D 00
-0.15925D 00
-0.79753D 00
-0.23452D 00
0.29945D 00
0.80452D-01
-0.26171D 00
-0.17214D 00
-0.68752D-01
0.10000D 01
0.56965D 00
0.17461D 00
-0.37351D 00
-0.35809D 00

0.21265D 00
0.23774D 00
0.29593D-01
-0.51046D 00
-0.53555D 00
0.10000D 01
0.64787D 00
0.20931D 00
-0.46119D 00
-0.82018D-02
-0.26687D-01
-0.57070D 00
0.47685D 00
-0.63324D 00
-0.19935D 00
0.47429D 00
0.50851D 00
0.57031D 00
0.12768D 00
-0.97666D 00
-0.33939D 00
-0.29025D 00
-0.23219D-01

Figure 2.3.3 (cont)

MODE 21

EIGENVALUE= 0.23109D-08
FREQUENCY= 0.33108D 04

EIGENVECTOR

0.10070D 00
0.11105D 00
0.39997D-01
-0.27335D 00
-0.37092D 00
0.29010D 00
0.10000D 01
0.17368D 00
-0.75976D 00
-0.93238D 00
-0.89465D 00
0.90102D 00
0.94050D 00
0.97257D 00
0.16759D 00
-0.82711D 00

-0.25350D 00
-0.34347D 00
0.28806D 00
0.16600D 00
0.15633D 00
0.36260D-01
-0.62806D-01

MODE 22

EIGENVALUE= 0.19588D-08
FREQUENCY= 0.35961D 04

EIGENVECTOR

-0.14726D 00
-0.15537D 00
0.15564D-01
0.12975D 00
0.41279D 00
0.10000D 01
-0.26809D 00
-0.85038D 00
-0.45636D 00
-0.88845D-02
0.30751D-01
-0.64442D-01
0.21733D 00
0.28454D 00
0.82327D 00
0.45943D 00

-0.13548D 00
-0.47379D 00
-0.80550D 00
-0.47110D 00
0.19256D 00
0.38960D 00
0.17752D 00

MODE 23

EIGENVALUE= 0.14260D-08
FREQUENCY= 0.42147D 04

EIGENVECTOR

-0.67070D-01
-0.68997D-01
0.16648D-01
0.84349D-01
0.25619D 00
0.46435D 00
-0.35161D 00
-0.90356D 00
-0.47849D 00
0.22839D 00
0.60463D 00
0.10000D 01
0.42208D 00
-0.36503D 00
-0.89993D 00
-0.51574D 00

0.92331D-01
0.27483D 00
0.45059D 00
0.11486D 00
-0.82346D-01
-0.15803D 00
-0.69862D-01

Table 2.4.1

Quad III LM-10 Vibration Environment Qualification Levels

1. Launch and Boost Phases

a. Random (all axes - duration 1.0 min/axis)

20-40 hz	+12 db/oct
40-85	0.030 g^2/hz
85-110	+6 db/oct
110-400	0.050 g^2/hz
400-450	-6 db/oct
450-1100	0.040 g^2/hz
1100-2000	-12 db/oct

b. Sinusoidal Sweep (all axes)

5-12	.2 in. da
12-100	1.4 g. peak

Sweep: 5-100-5 hz
Rate: 3 oct/min

c. Sinusoidal Dwell (all axes-duration 10 sec)
6 hz 1.5 g-peak

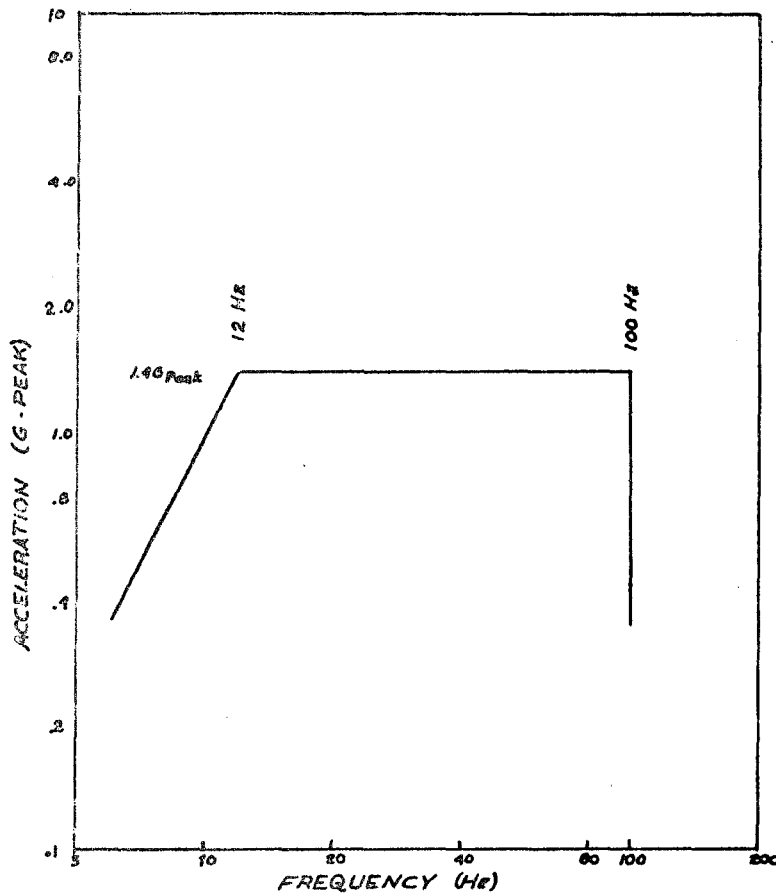
2. Lunar Descent Phase

a. Random (all axes-duration 12.5 min)

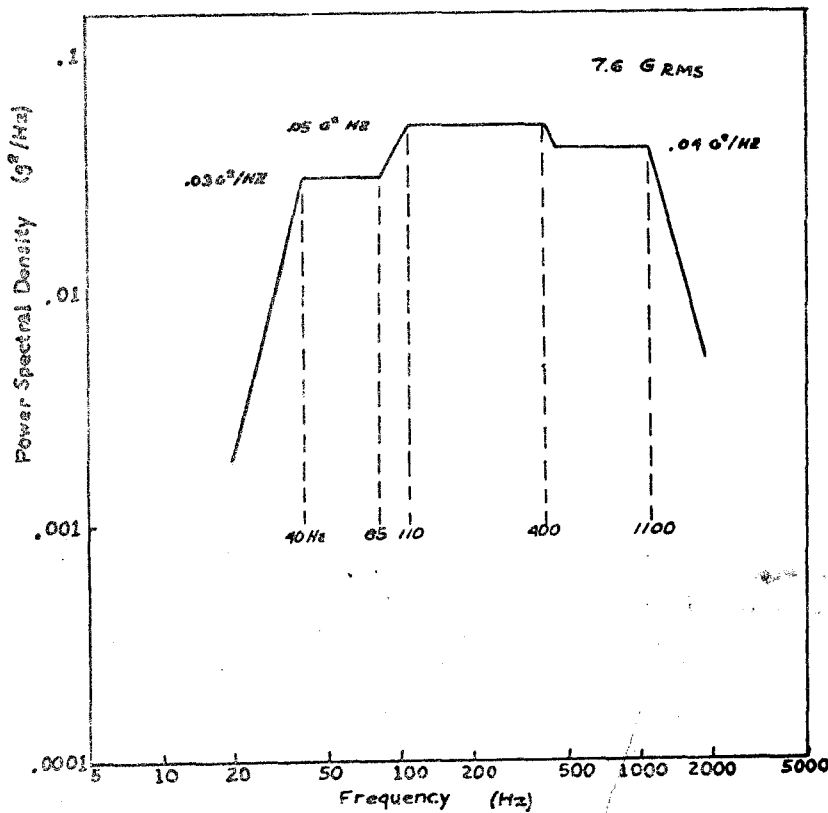
20-2000	0.005 g^2/hz
---------	----------------

NOTE: Levels specified are to the GAC interface points in the LM.

Figure 2.4.1
QUAD III SINUSOIDAL ENVIRONMENT



QUAD III L&B RANDOM ENVIRONMENT





Space
Systems Division

LRRR 300 Corner Array Dynamic Analysis

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3.0 Computed Responses

3.1 Responses of Out of Plane Model of Both Arrays

The response at each lumped mass location of the out-of-plane model of both arrays was calculated to the sine environment and to the random environment. Transmissibility, sine response, and random response at five location on the experiment are shown in Figure 3.1.1 through Figure 3.1.5. The random responses shown are responses to the "Launch & Boost Environment". The "Lunar Descent" random environment gave responses that were in all cases less severe than responses to the L&B environment. Out-of-plane responses are summarized in Table 3.1.1.

The most severe response was found to be at the center of the right edge of the small array, lumped mass location 27 of Figure 2.1.2. The response to the L&B random excitation at this location is shown in Figure 3.1.6. This response has a root mean square value of 15.87 g, giving a 3-sigma peak value of 47.6 g. Thus, the greatest acceleration perpendicular to the plane of the arrays likely to be seen by a corner reflector mount is 47.6 g. The greatest power spectral density of this response is in the frequency range between 100 and 400 Hz. If the corner mount design can survive sinusoidal vibration well in excess of 47 g peak perpendicular to the array face, then the corner mount design is of sufficient strength in this direction.

Response at the 3 independent attachment points between the arrays determine the vibration levels seen by the upper array. An envelope enclosing the responses at all the attachments could then be used for testing the upper array by itself. The three responses were superimposed and a suitable envelope, reduced, by a factor of 1.69 to acceptance level, is shown in Figure 3.1.7.

3.2 Response of the In Plane Model

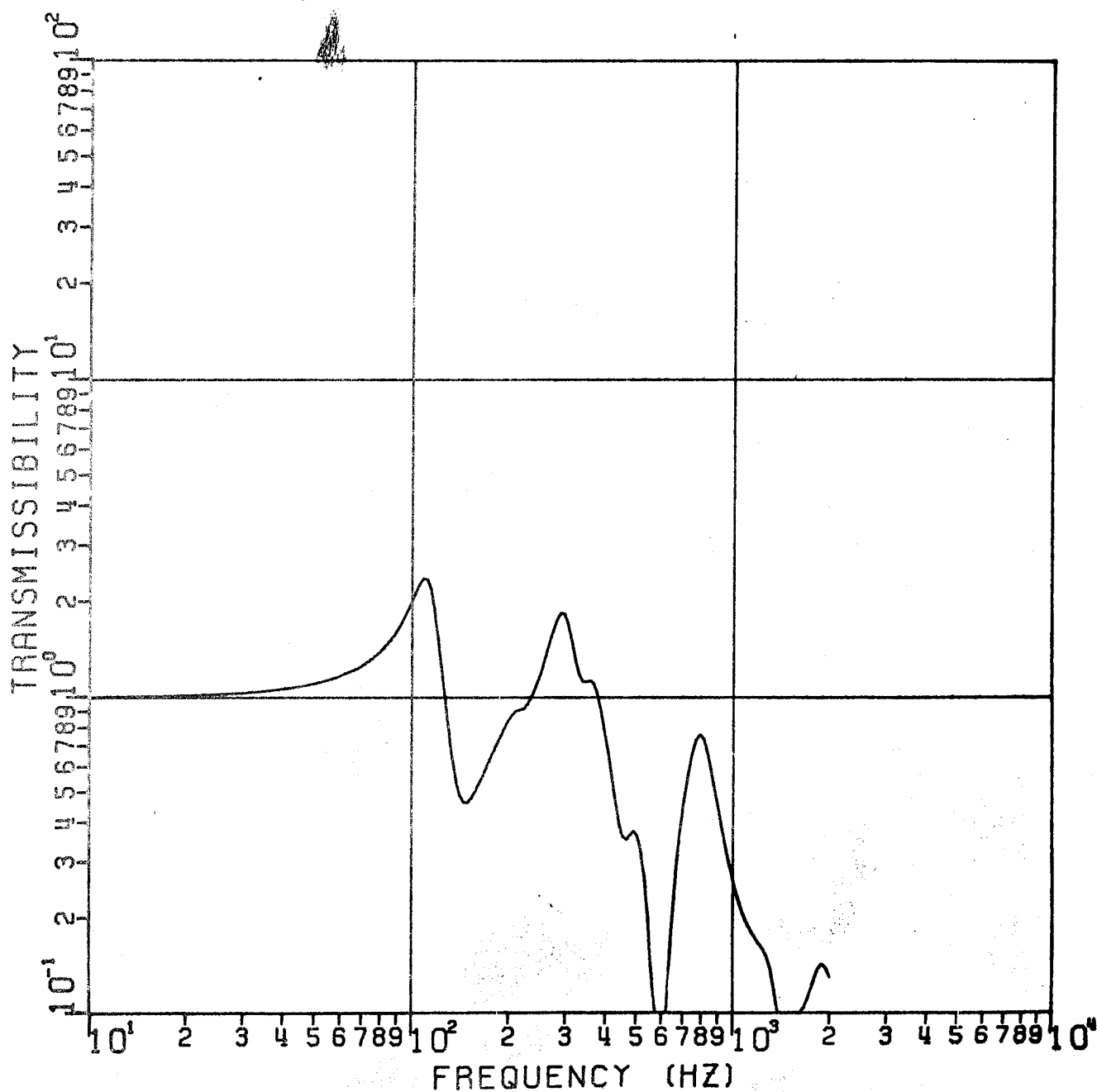
Responses in the six coordinate directions of the in plane model (coordinate directions shown in Figure 2.2.1) are shown in Figure 3.2.1 through 3.2.6. Response of the angular coordinates θ_1 and θ_2 , are given in "pseudo g" units of $\frac{\text{rad}}{\text{sec}^2} \cdot \frac{1}{386}$. These responses are summarized in Table 3.2.1.

The worst vibration level seen in-plane by a corner mount will occur along the edge of one of the two arrays. The response will be the sum of translational response of the center of gravity plus the rotational response about the c.g. times the distance to the edge from the c.g.. The maximum responses at corner locations on both arrays to excitation in the X and Y directions are shown in Table 3.2.2.

FREQUENCY RESPONSE OF LR3 300, 30 D.F. MODEL

FIGURE 3.1.1A TRANSMISSIBILITY

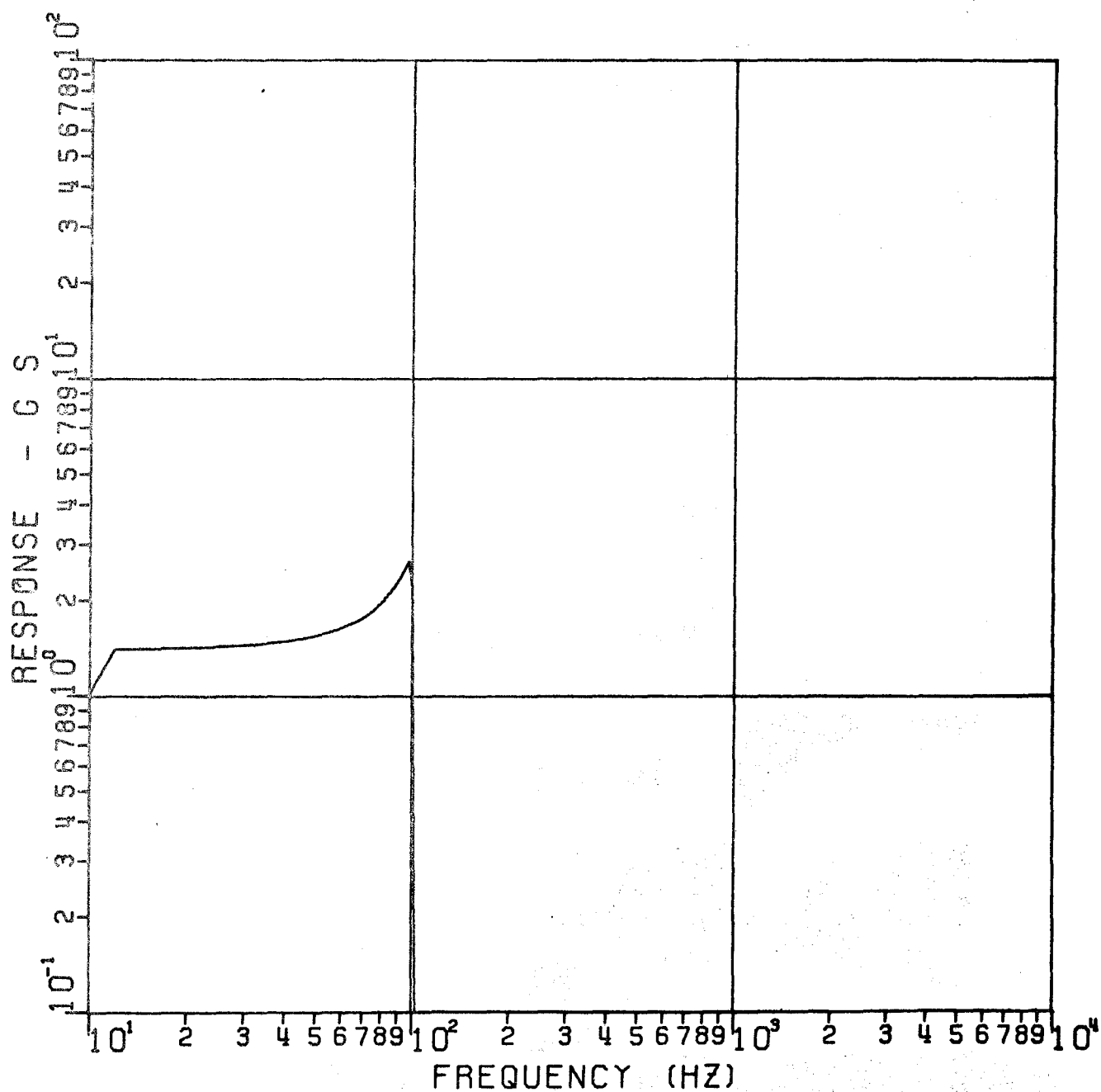
LOCATION 2



FREQUENCY RESPONSE OF LR3 300, 30 D.F. MODEL

FIGURE 3.1.1 *B* SINE RESPONSE

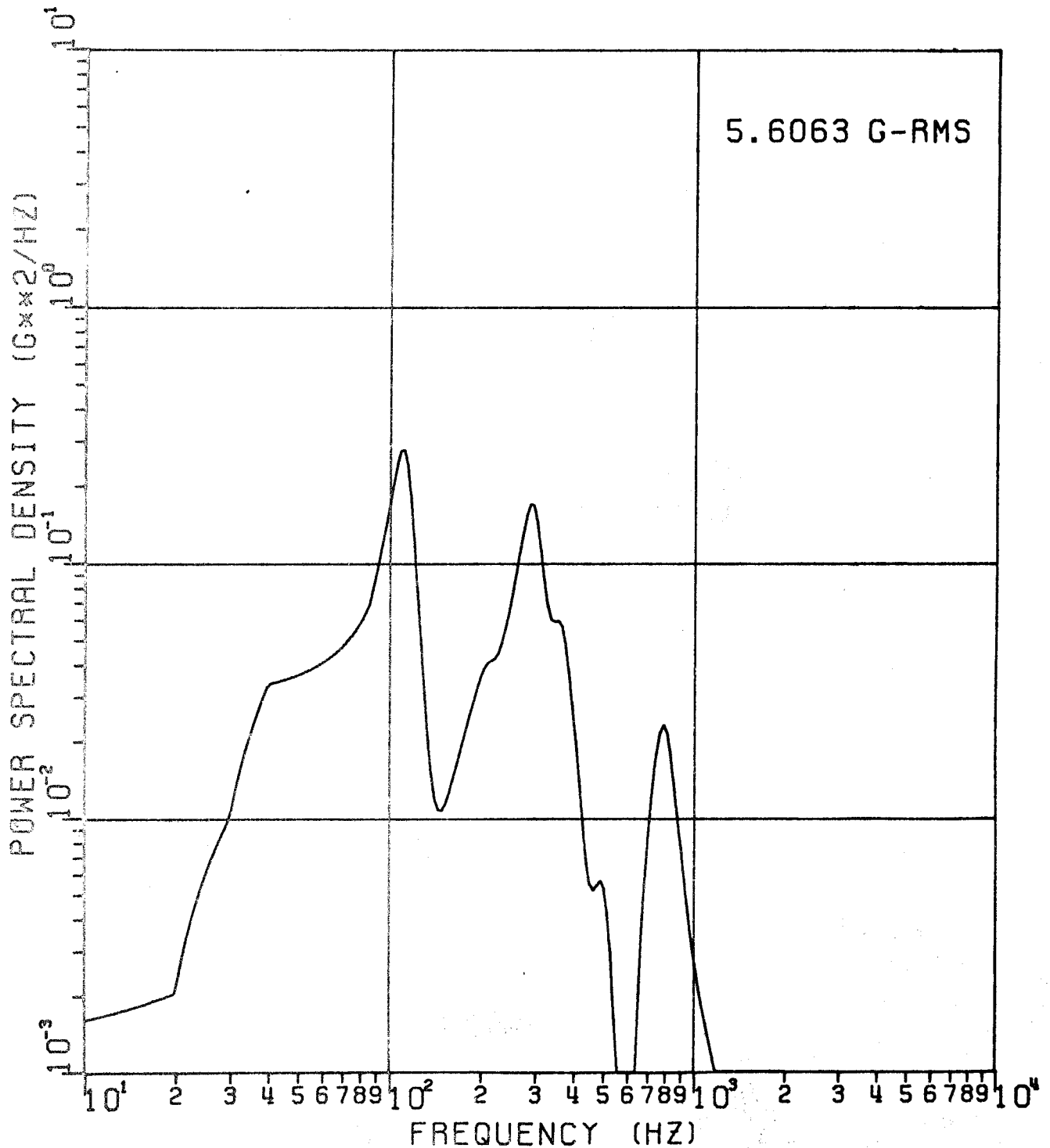
LOCATION 2



FREQUENCY RESPONSE OF LR3 300, 30 D.F. MODEL
#936

FIGURE 3.1.1C RANDOM VIBRATION SPECTRUM

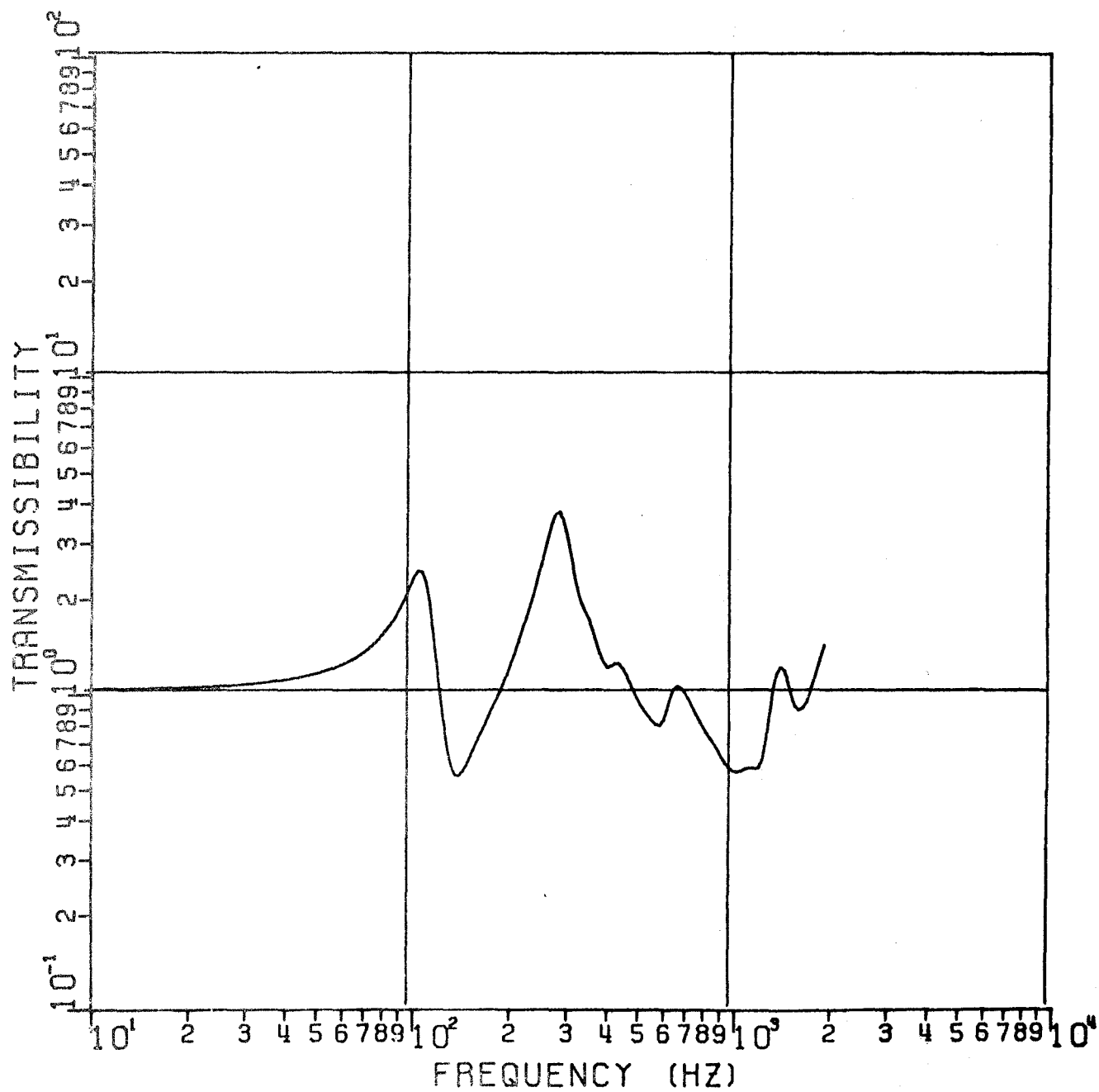
LOCATION 2



FREQUENCY RESPONSE OF LR3 300, 30 D.F. MODEL

FIGURE 3.1.2 A TRANSMISSIBILITY

LOCATION 9

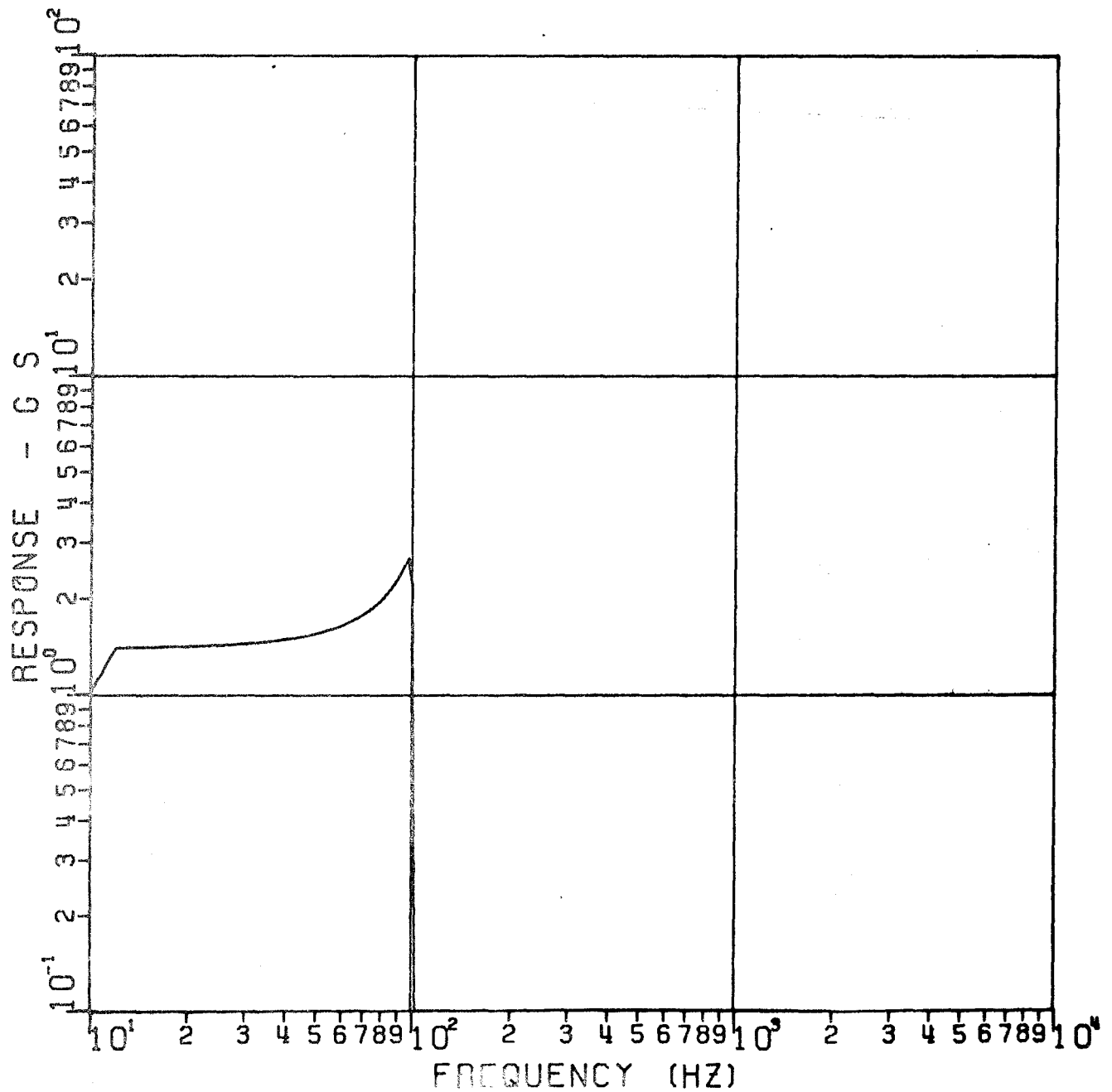


FREQUENCY RESPONSE OF LR3 300, 30 D.F. MODEL

FIGURE 3.1.2 B SINE RESPONSE

LOCATION

9

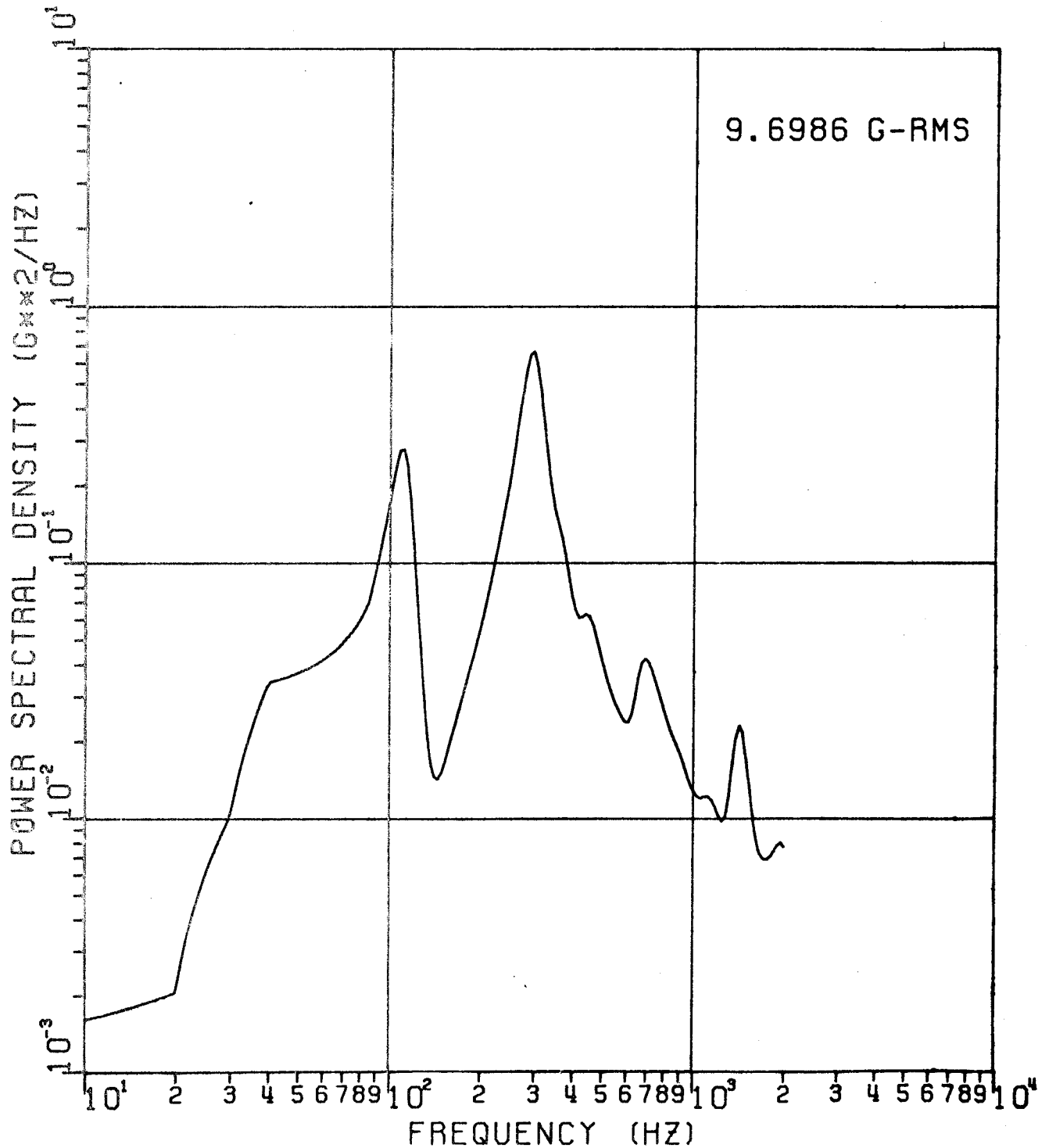


FREQUENCY RESPONSE OF LR3 300, 30 D.F. MODEL

FIGURE 3.1.2 C RANDOM VIBRATION SPECTRUM

LOCATION

9

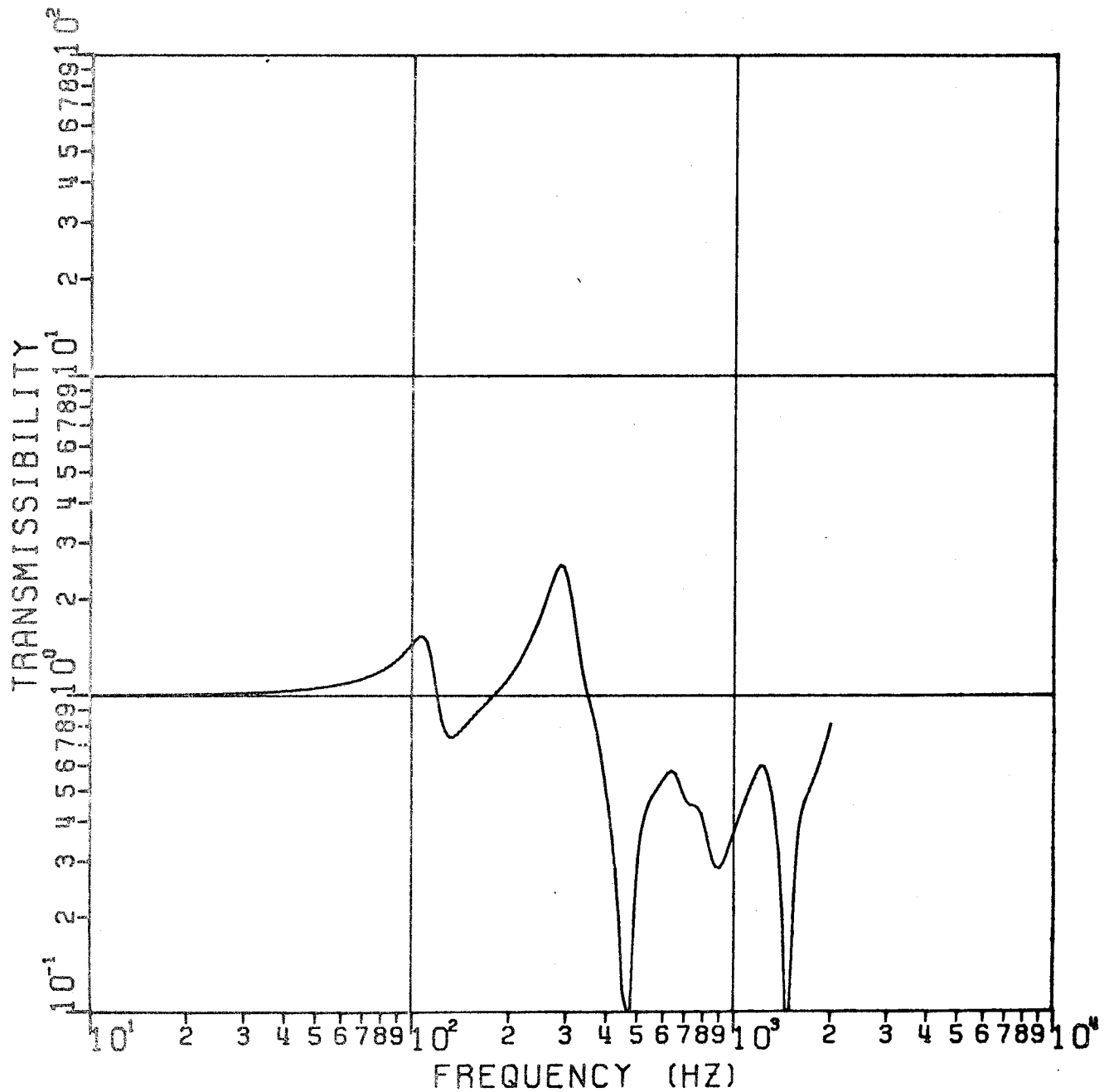


FREQUENCY RESPONSE OF LR3 300, 30 D.F. MODEL

FIGURE 3.3A TRANSMISSIBILITY

LOCATION

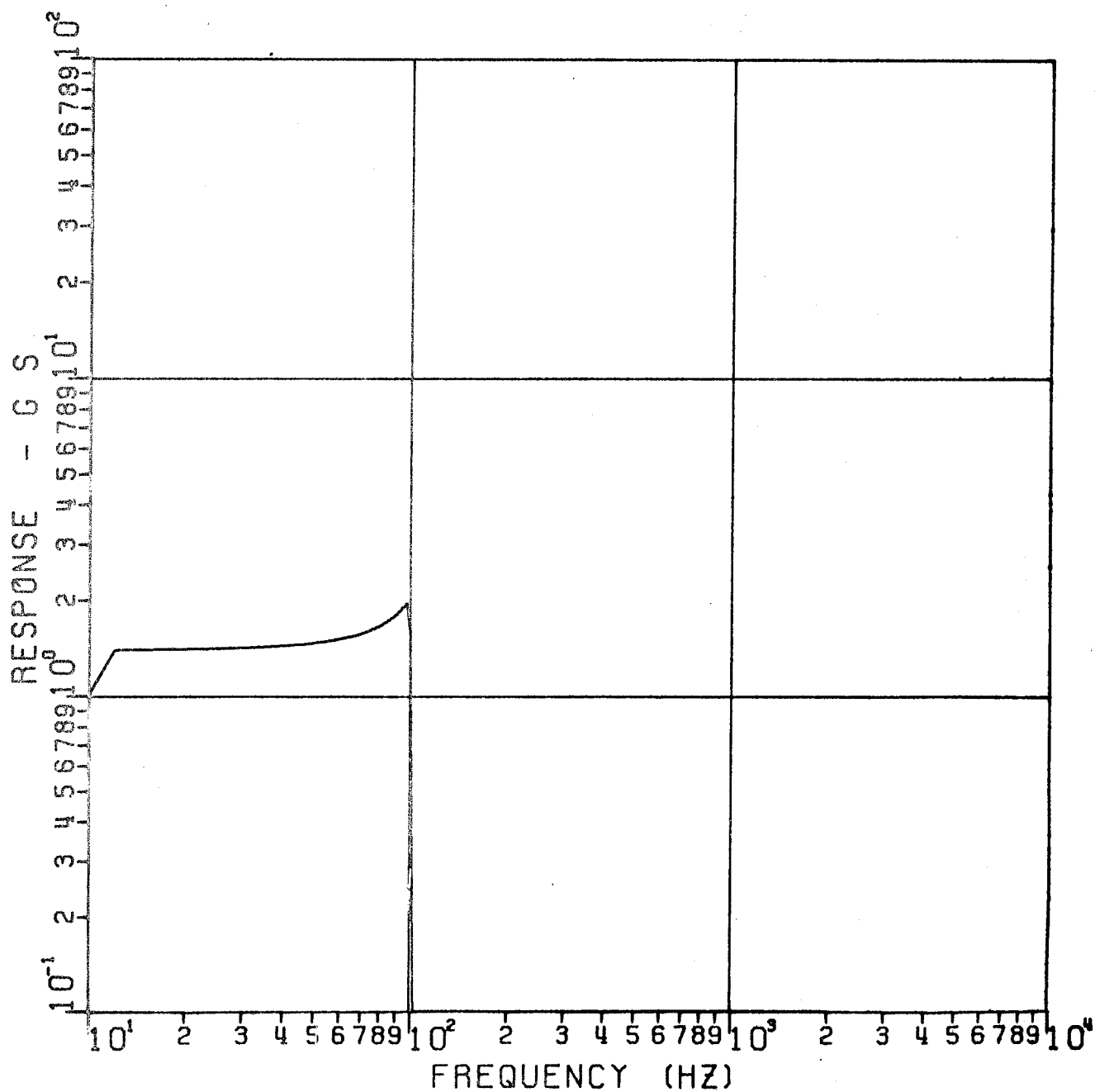
16



FREQUENCY RESPONSE OF LR3 300, 30 D.F. MODEL

FIGURE 3.13 B SINE RESPONSE

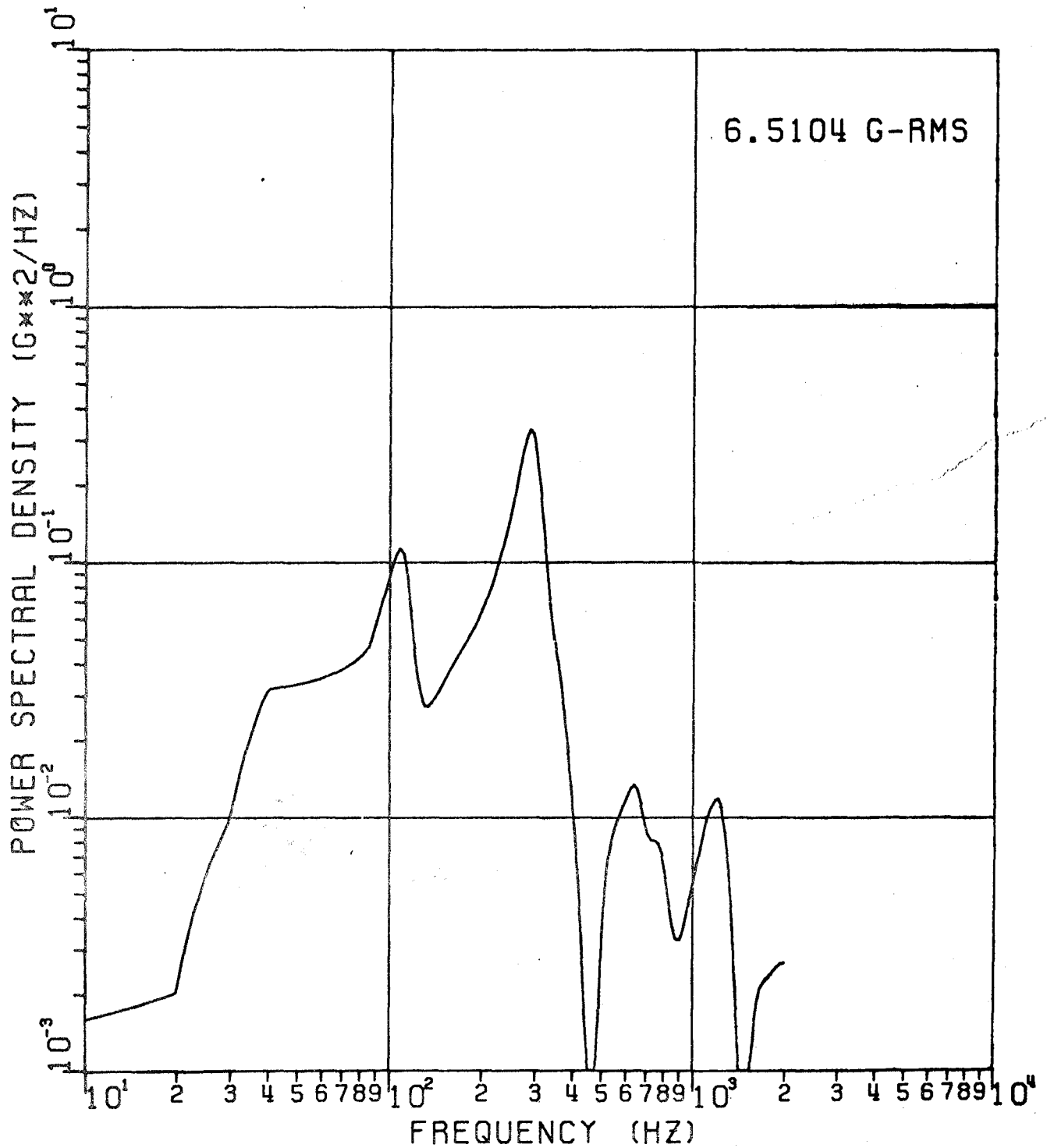
LOCATION 16



FREQUENCY RESPONSE OF LR3 300, 30 D.F. MODEL

FIGURE 3.1.3C RANDOM VIBRATION SPECTRUM

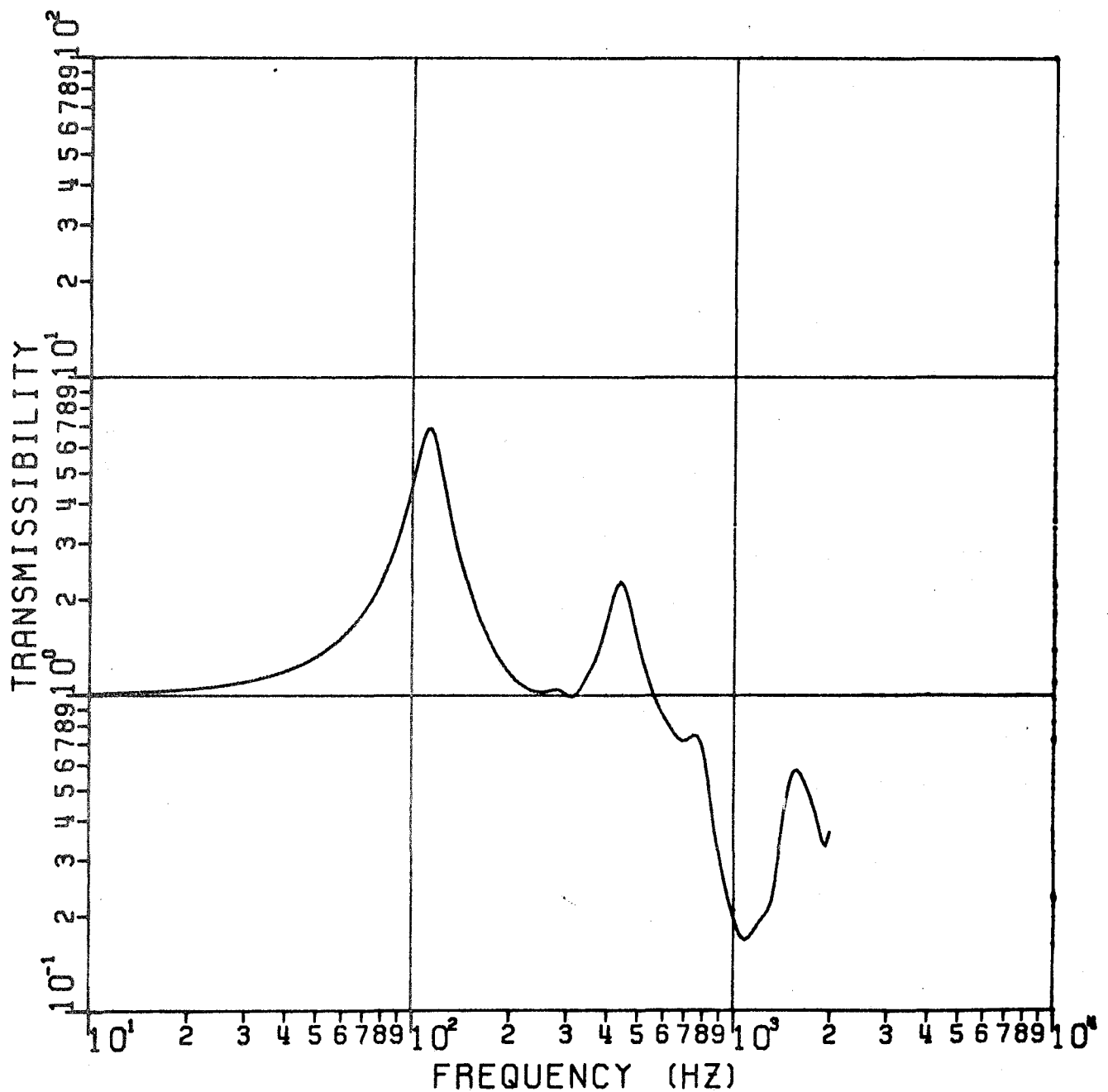
LOCATION 16



FREQUENCY RESPONSE OF LR3 300, 30 D.F. MODEL

FIGURE 3.1.4A TRANSMISSIBILITY

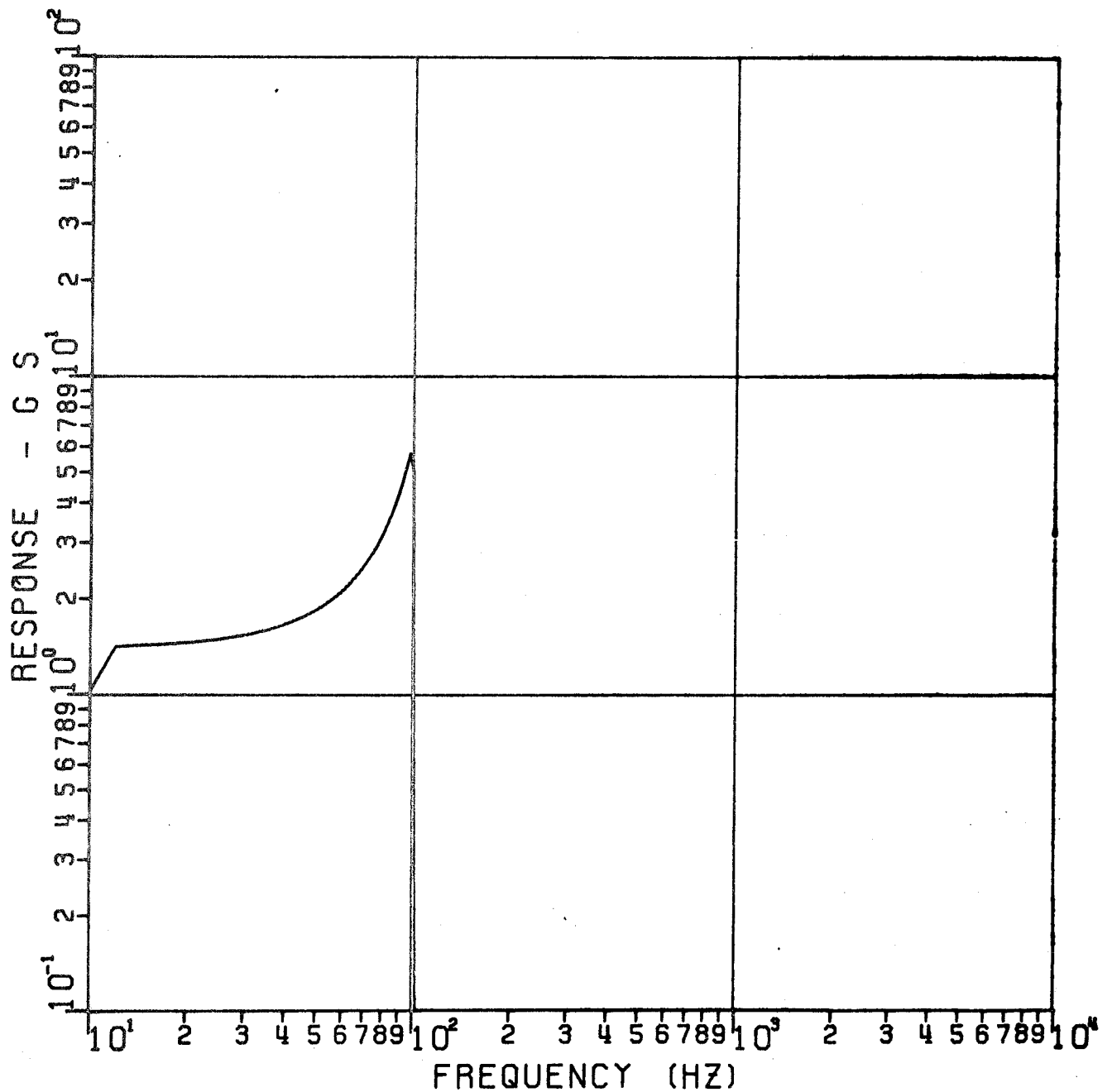
LOCATION 21



FREQUENCY RESPONSE OF LR3 300, 30 D.F. MODEL

FIGURE 3.1.4 B SINE RESPONSE

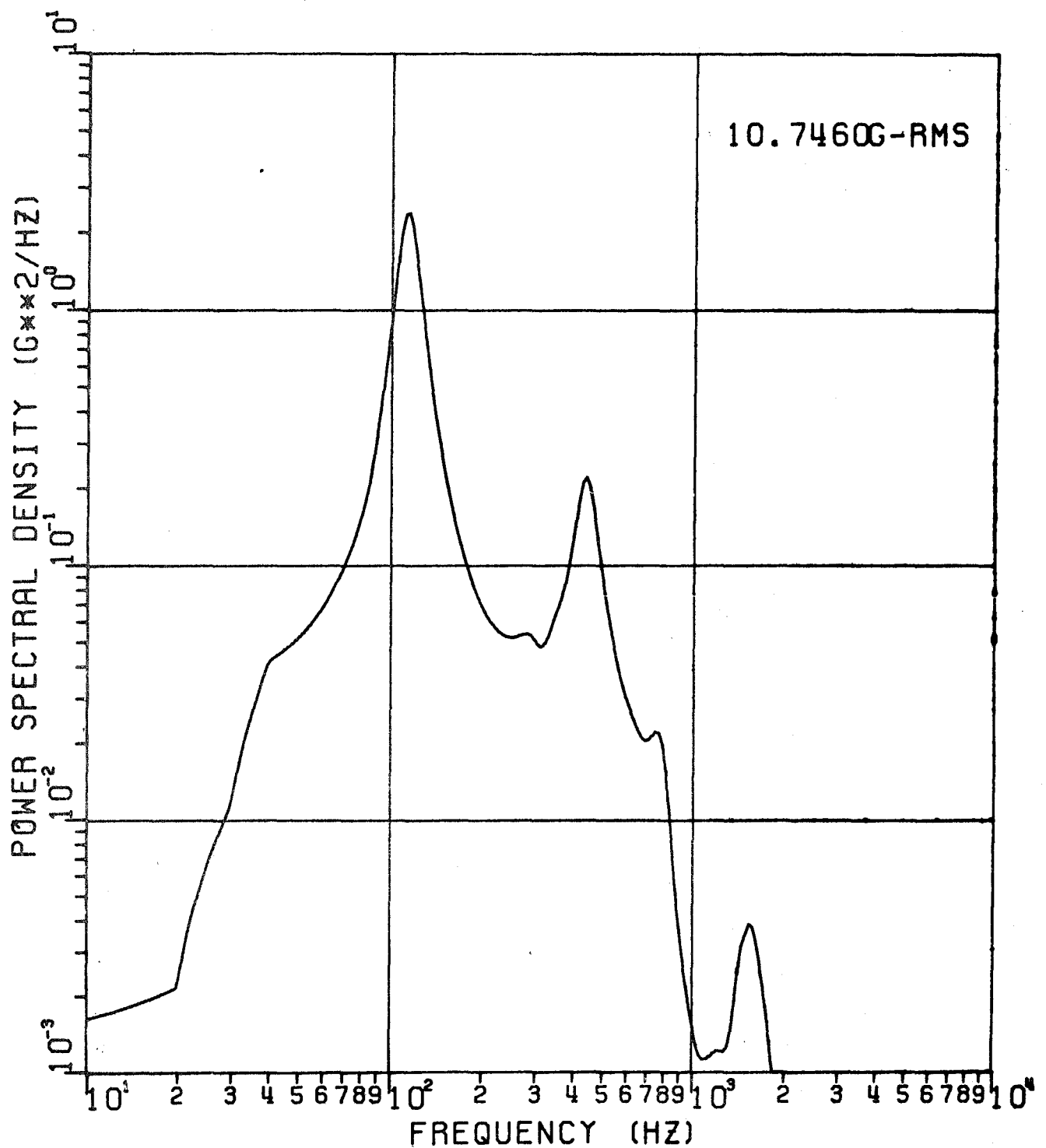
LOCATION 21



FREQUENCY RESPONSE OF LR3 300, 30 D.F. MODEL

FIGURE 3.14C RANDOM VIBRATION SPECTRUM

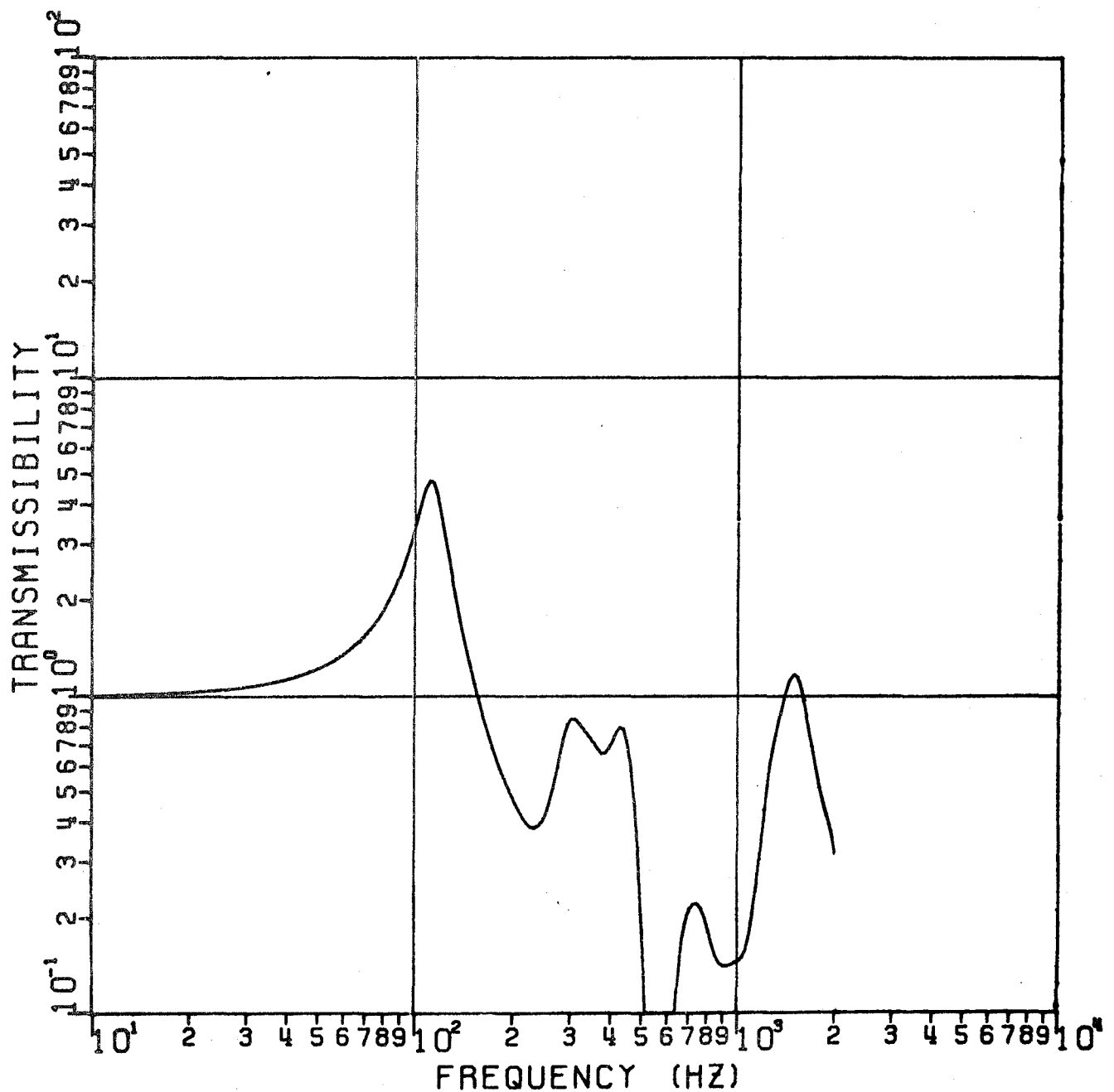
LOCATION 21



FREQUENCY RESPONSE OF LR3 300, 30 D.F. MODEL

FIGURE 3.15A TRANSMISSIBILITY

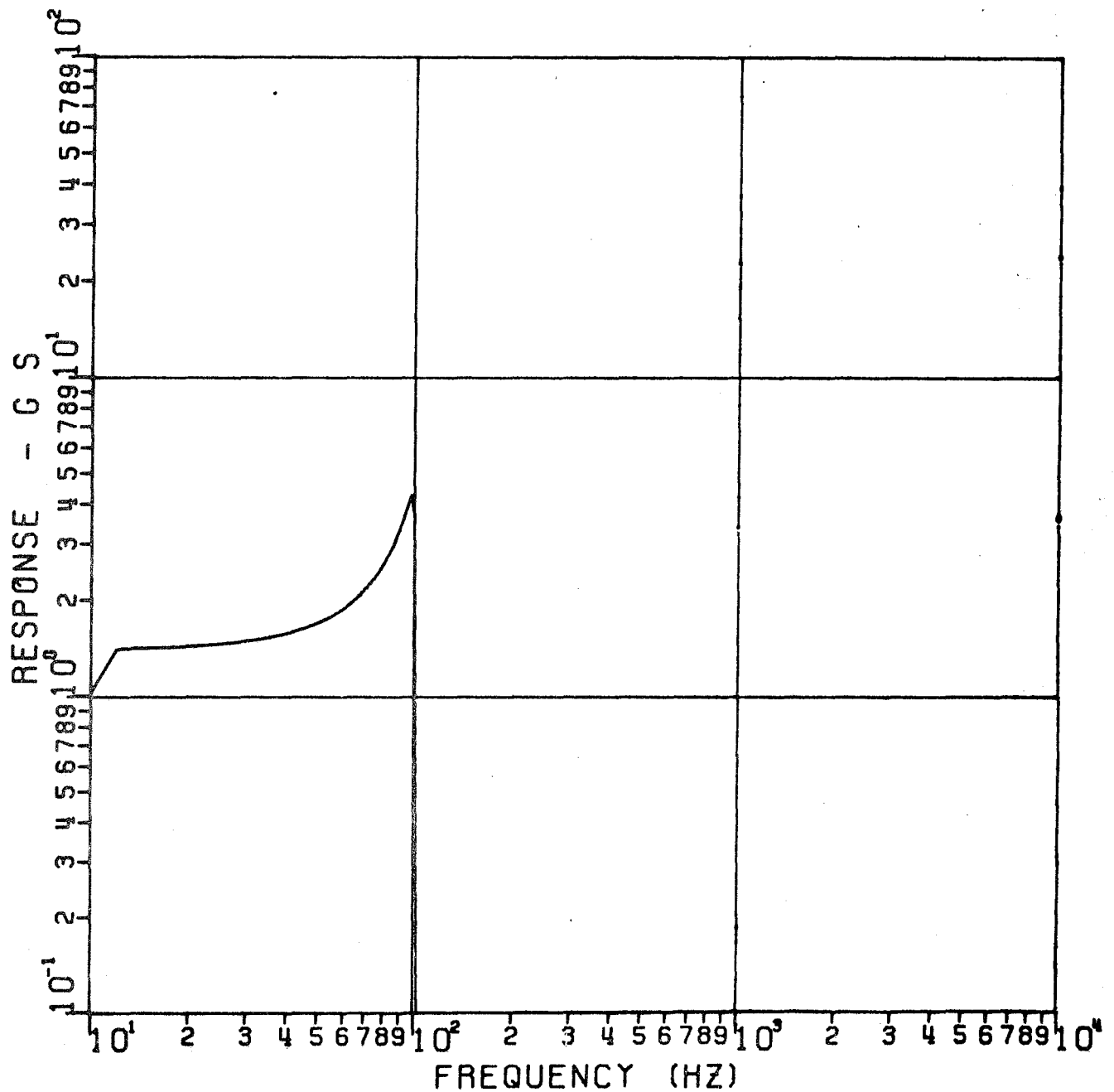
LOCATION 28



FREQUENCY RESPONSE OF LR3 300, 30 D.F. MODEL

FIGURE 3.1.5 B SINE RESPONSE

LOCATION 28



FREQUENCY RESPONSE OF LR3 300, 30 D.F. MODEL

FIGURE 3.5C RANDOM VIBRATION SPECTRUM

LOCATION

28

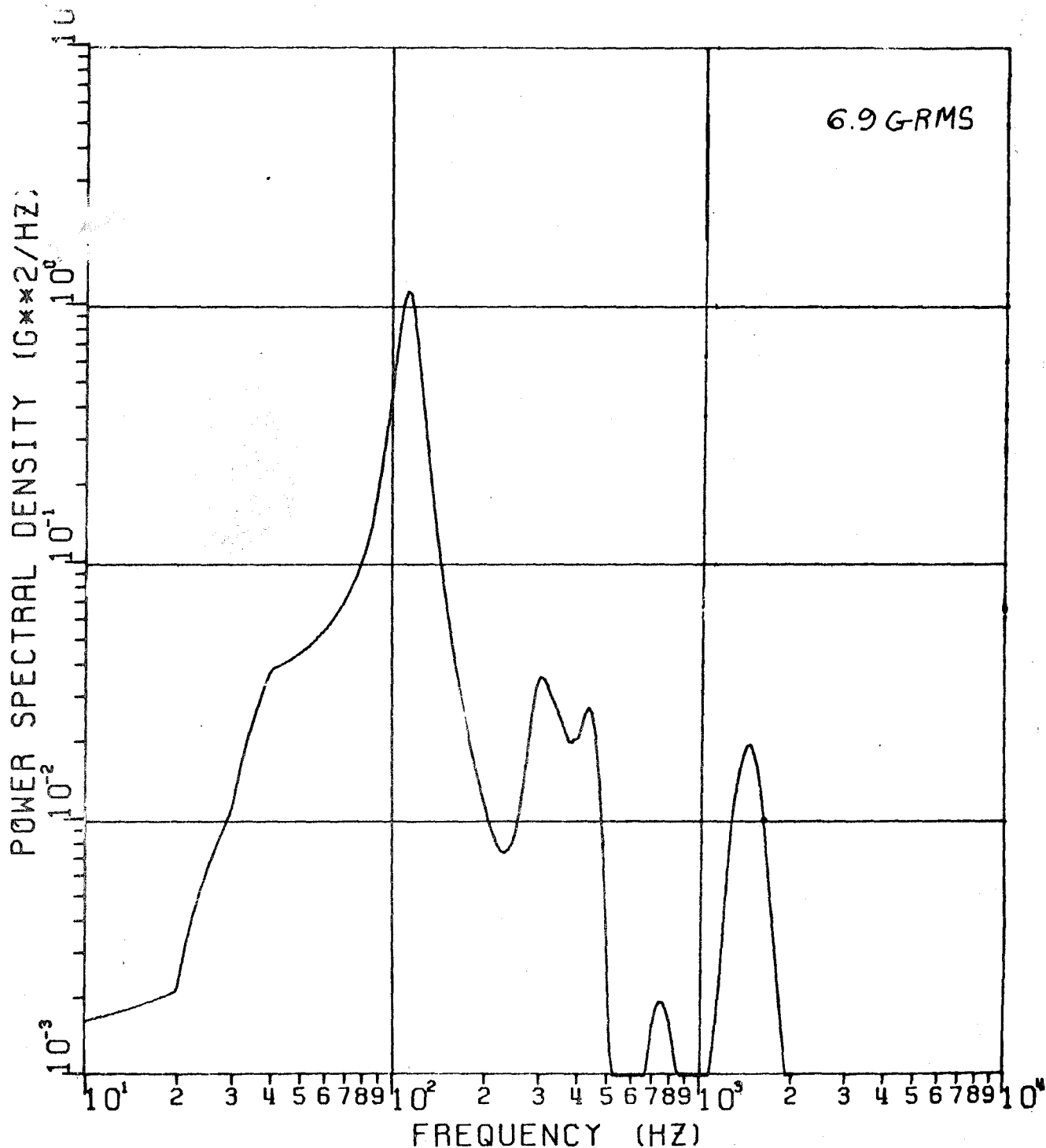


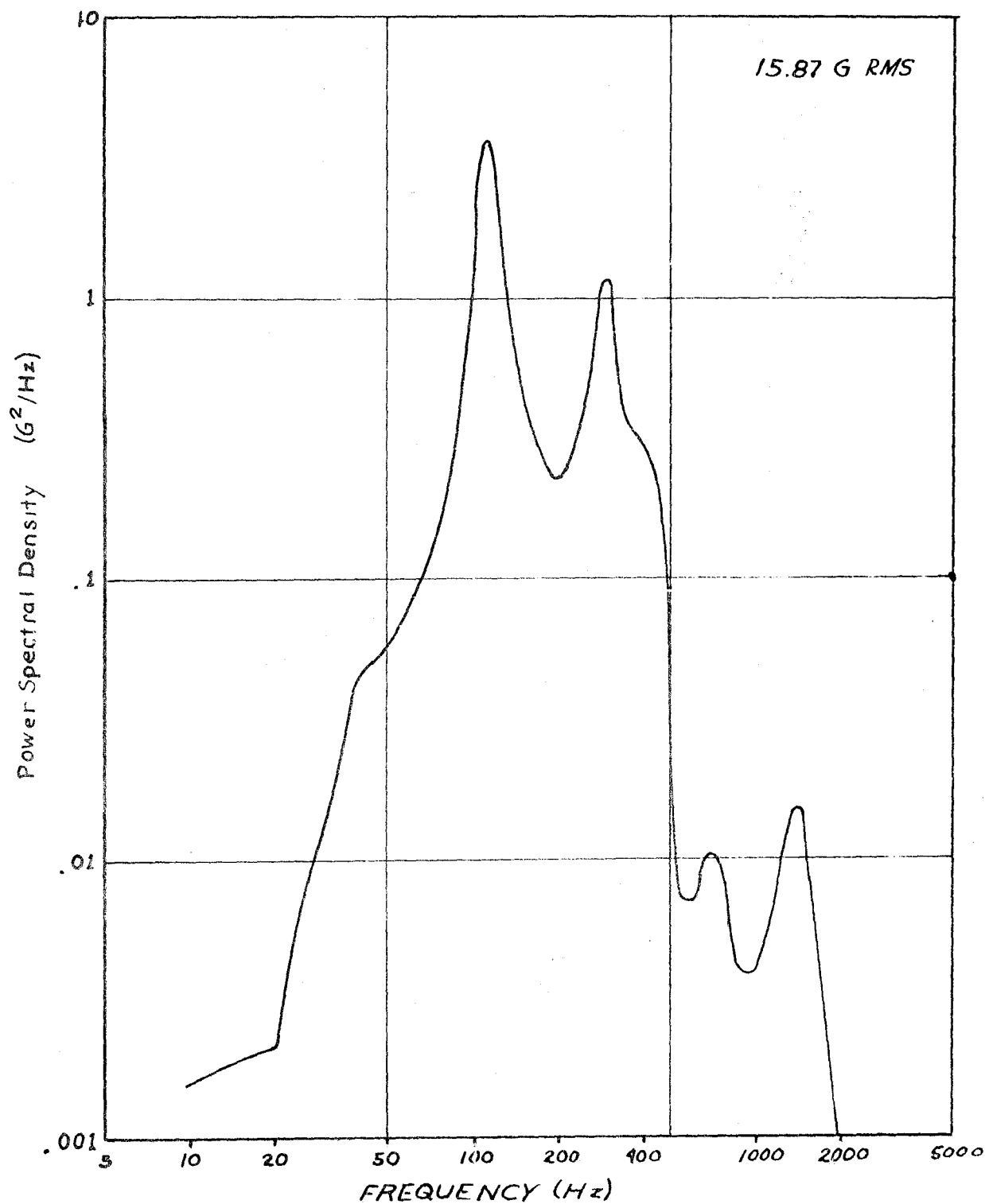
TABLE 3.1.1
SUMMARY OF OUT OF PLANE RESPONSES

Location (See Fig. 2.1.1 & 2.1.2)	Maximum Sine Resp. g's peak	Random Resp. g's RMS
1	1.4	10.9
2	2.6	5.6
3	2.0	5.7
4	1.85	8.2
5	2.0	6.1
6	2.5	8.3
7	2.5	6.8
8	2.8	6.4
9	2.7	9.7
10	3.2	8.4
11	3.2	7.0
12	2.9	7.2
13	3.2	7.9
14	4.0	7.7
15	3.3	5.8
16	2.0	6.5
17	4.4	8.0
18	4.5	6.8
19	2.0	6.3
20	3.5	9.7
21	5.7	10.8
22	3.5	8.4
23	4.1	7.1
24	6.3	11.9
25	4.4	7.8
26	6.5	14.0
27	6.7	15.9
28	4.3	6.9
29	6.2	11.4
30	3.8	8.4

Figure 3.1.6

WORST RESPONSE AT A CORNER LOCATION

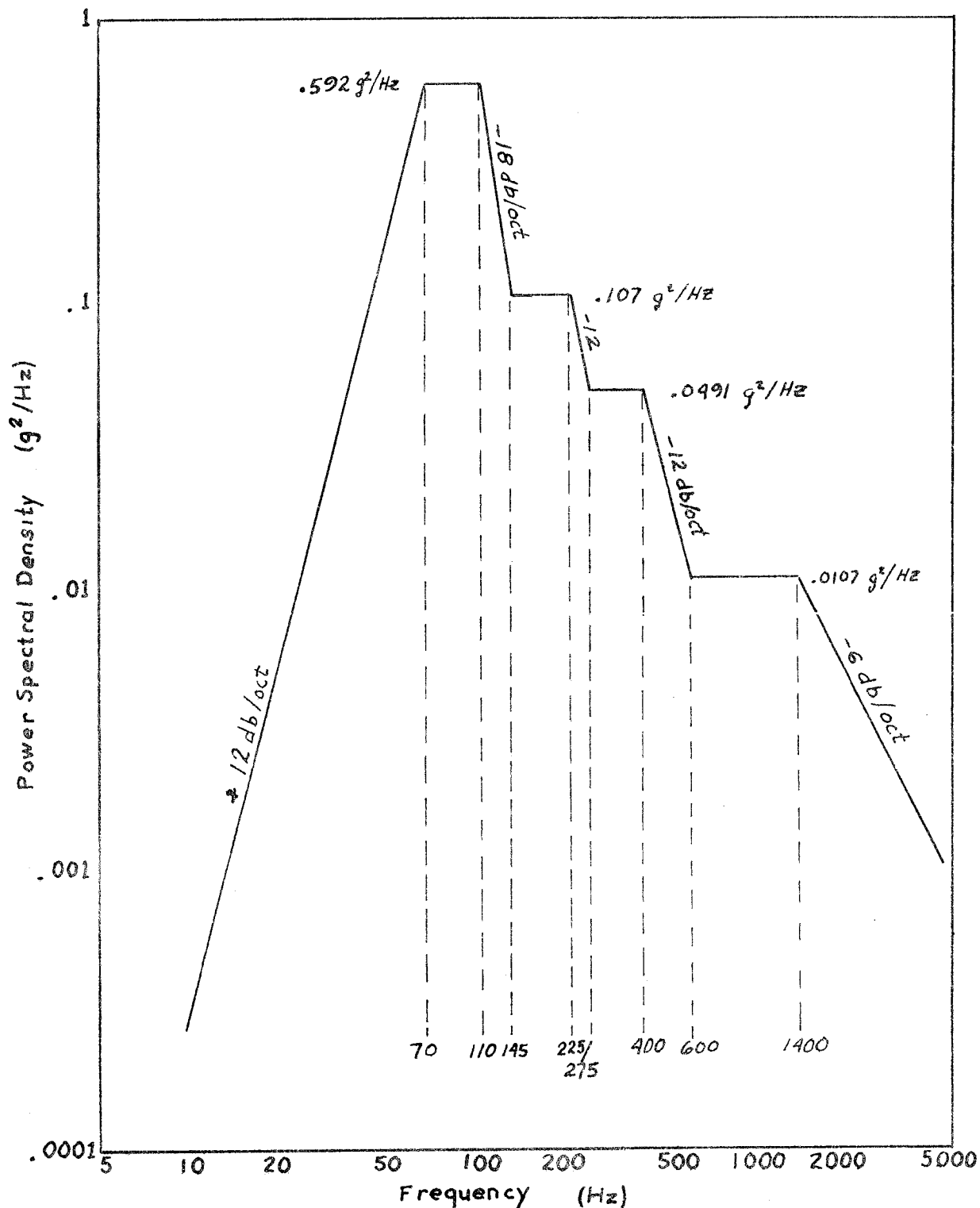
LOCATION 27
PERPENDICULAR TO ARRAY



UPPER ARRAY ACCEPTANCE RANDOM VIBRATION LEVELS

Figure 3.1.7

Z AXIS - PERPENDICULAR TO ARRAY

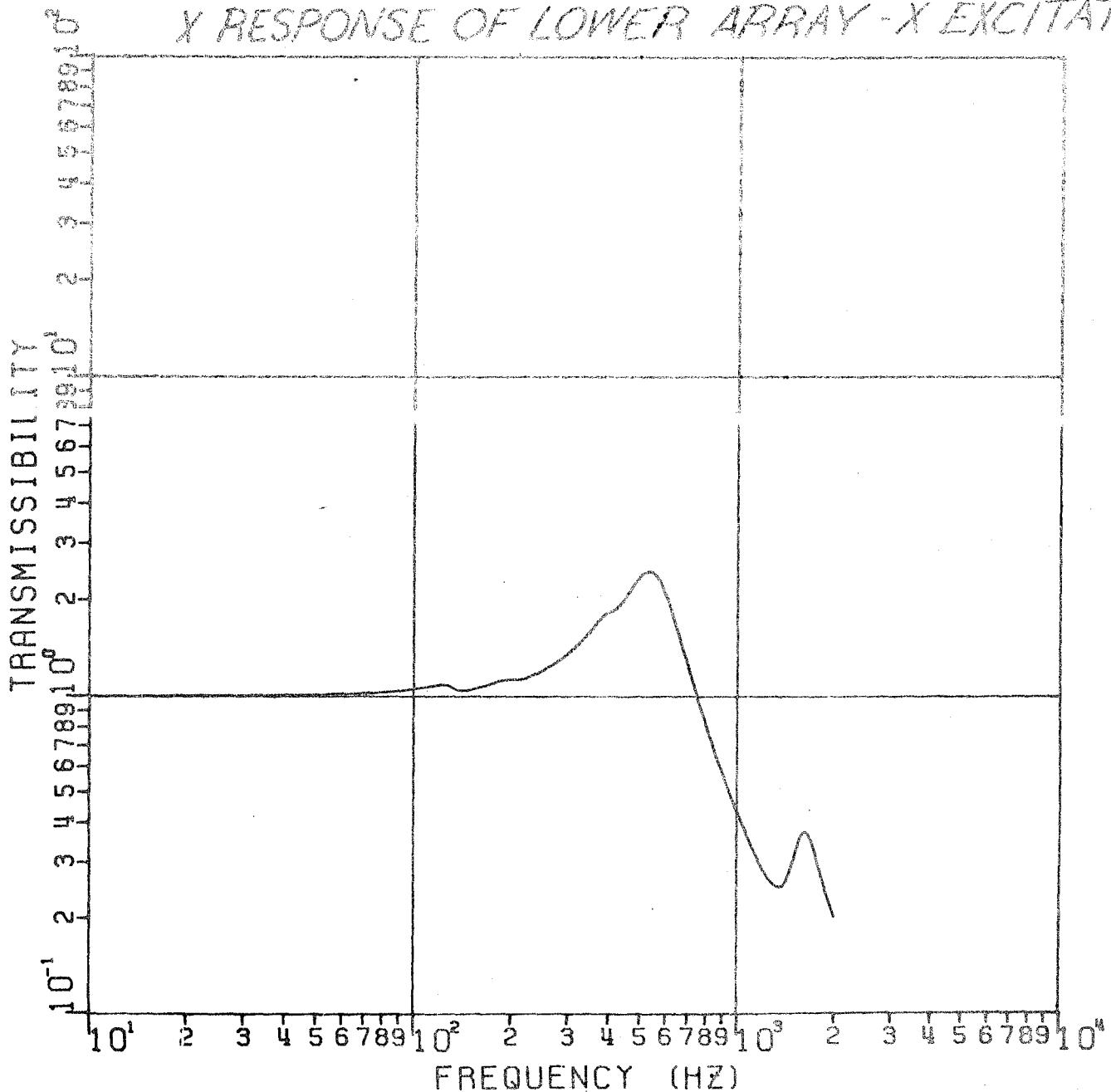


FREQUENCY RESPONSE FOR LR3 300,60.F. MODEL

FIGURE 3.2.1A TRANSMISSIBILITY

LOCATION 1

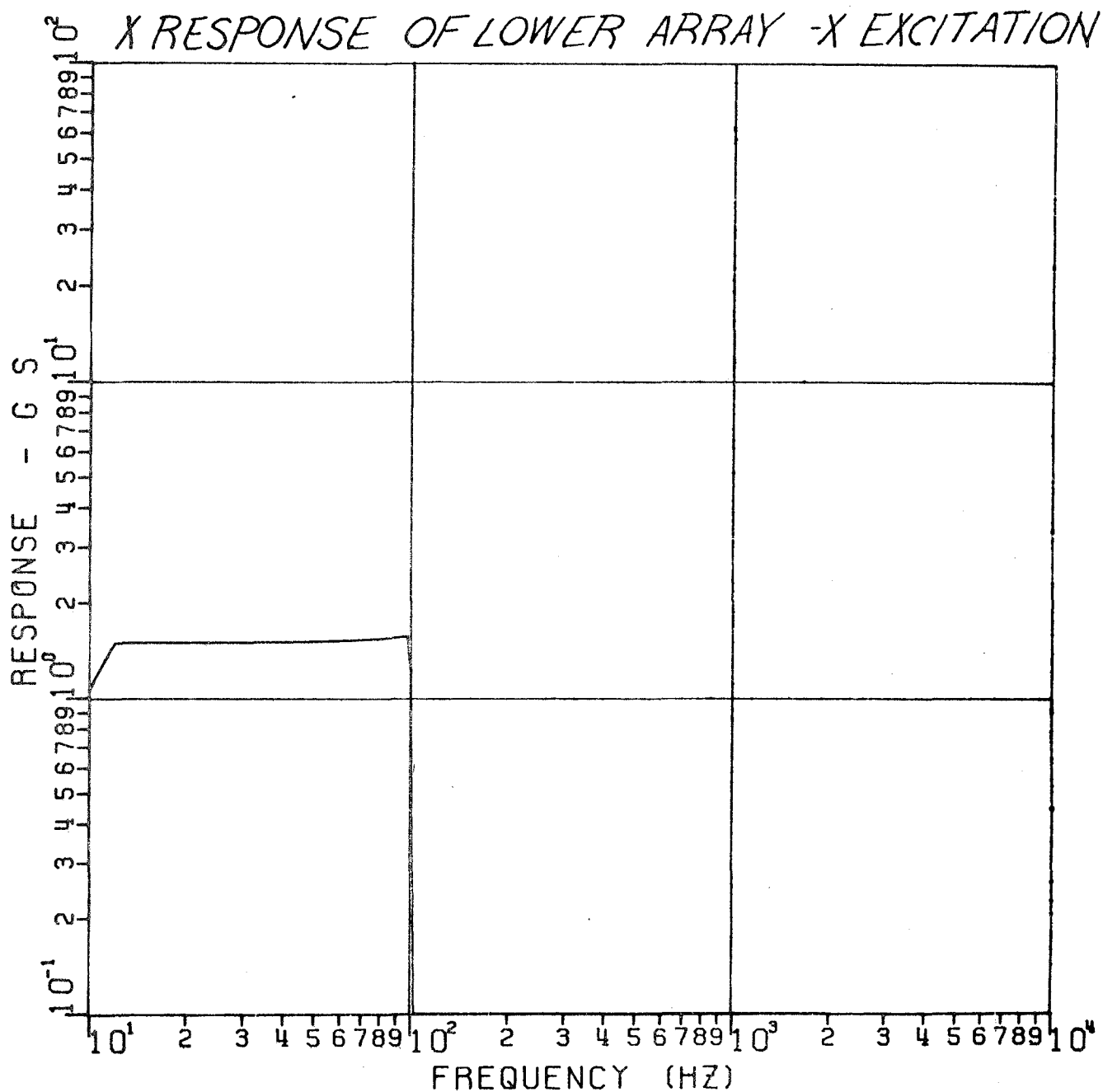
X RESPONSE OF LOWER ARRAY - X EXCITATION



FREQUENCY RESPONSE FOR LR3 300,60.F. MODEL

FIGURE 3.2.1B SINE RESPONSE

LOCATION 1



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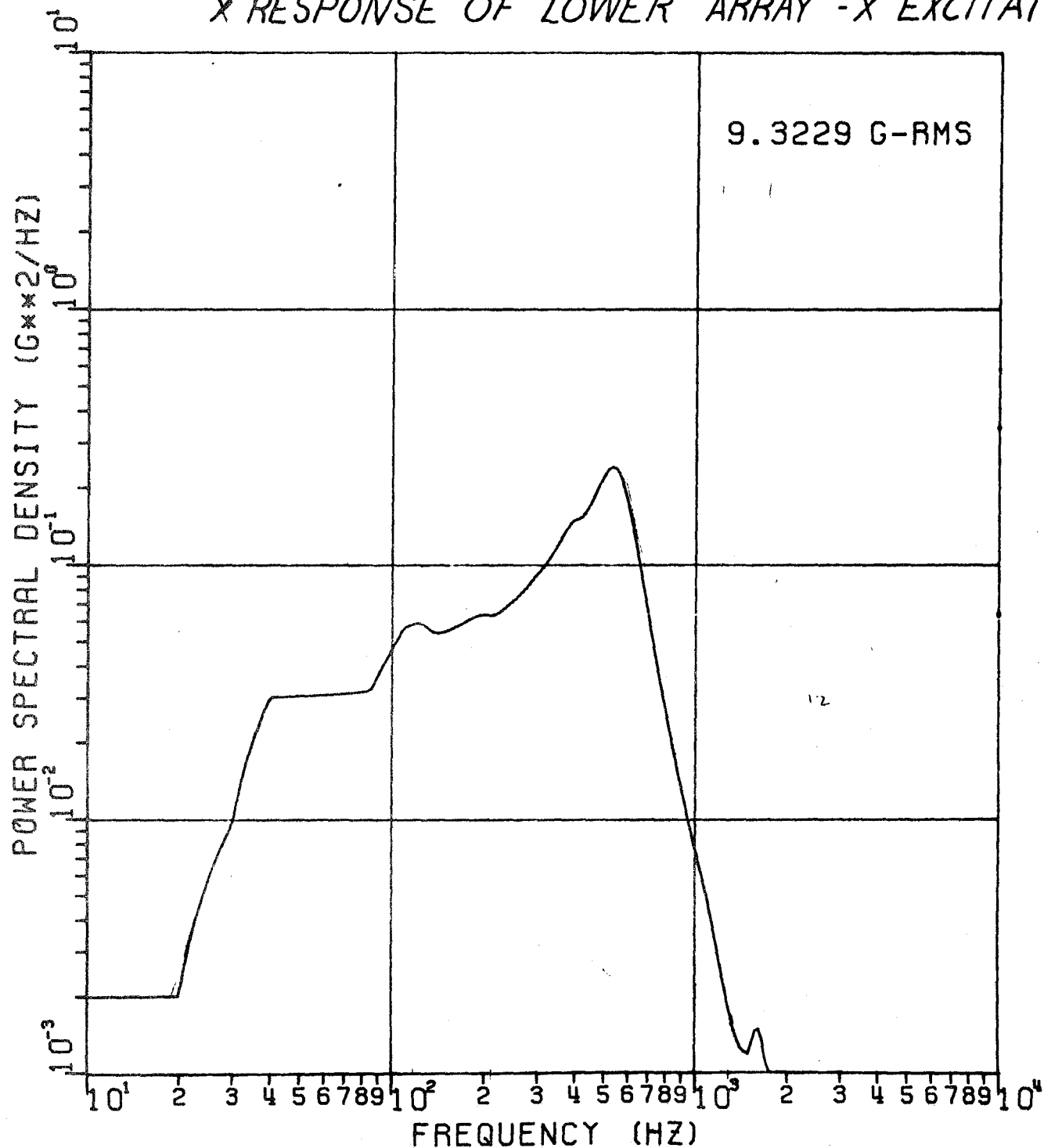
#936

FREQUENCY RESPONSE FOR LR3 300.6D.F. MODEL

FIGURE 3.2.1C RANDOM VIBRATION SPECTRUM

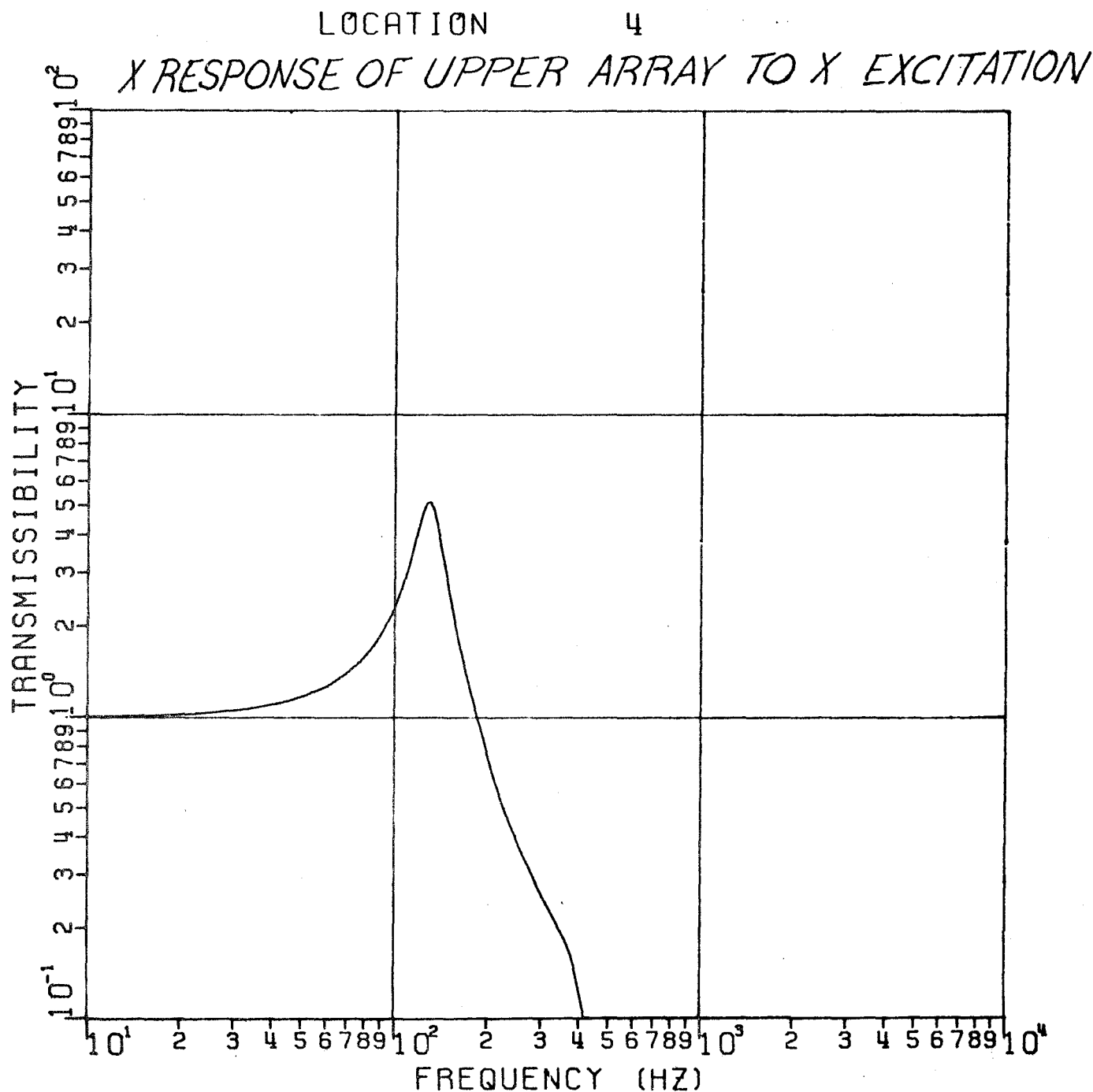
LOCATION

1

X RESPONSE OF LOWER ARRAY - X EXCITATION

FREQUENCY RESPONSE FOR LR3 300,60.F. MODEL

FIGURE 3.2.2 A TRANSMISSIBILITY



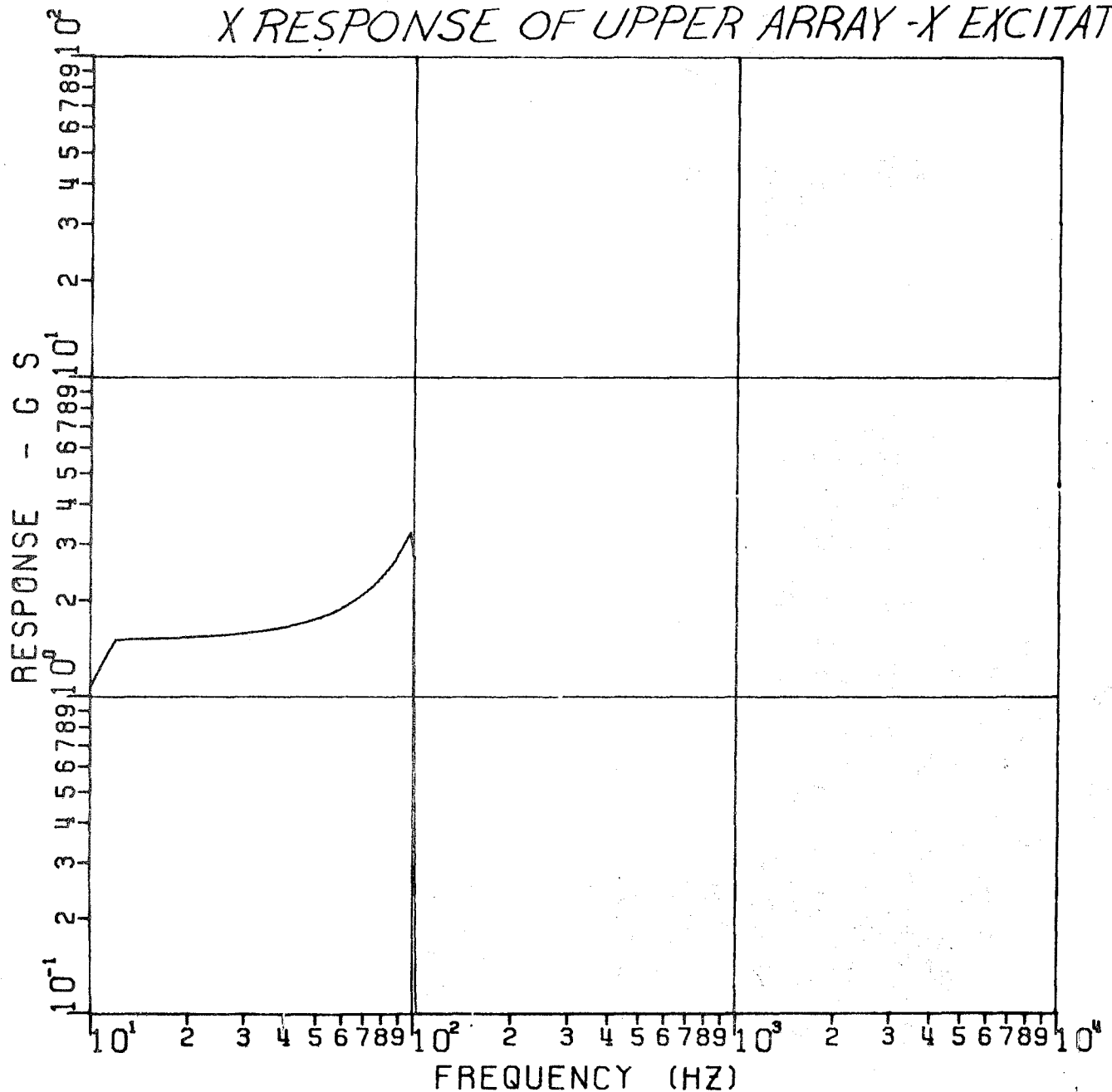
X 0.001

FREQUENCY RESPONSE FOR LR3 300,6D.F. MODEL

FIGURE 3.2.2.B SINE RESPONSE

LOCATION 4

X RESPONSE OF UPPER ARRAY - X EXCITATION



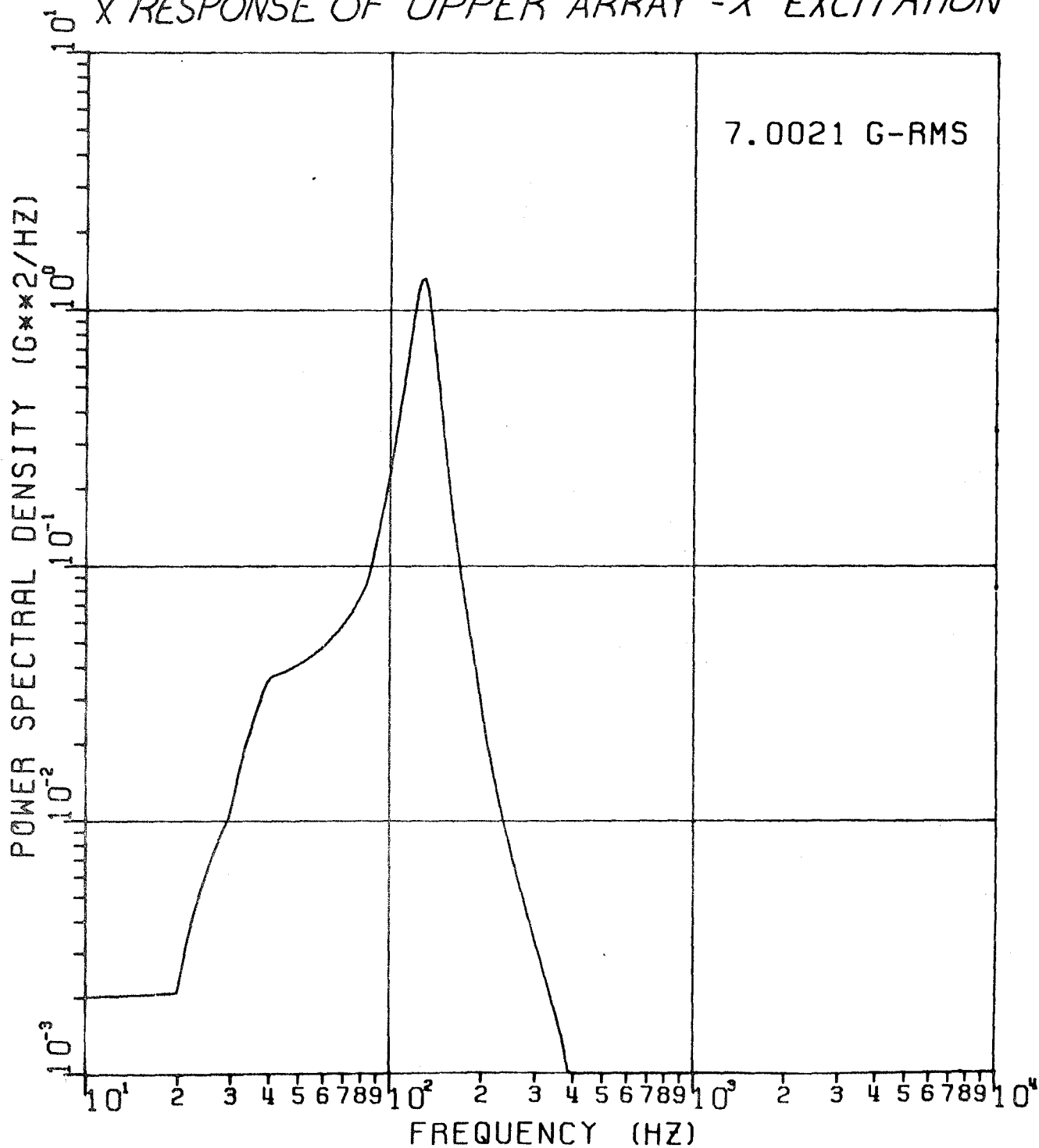
FREQUENCY RESPONSE FOR LR3 300,6D.F. MODEL

FIGURE 3.2.2C RANDOM VIBRATION SPECTRUM

LOCATION

4

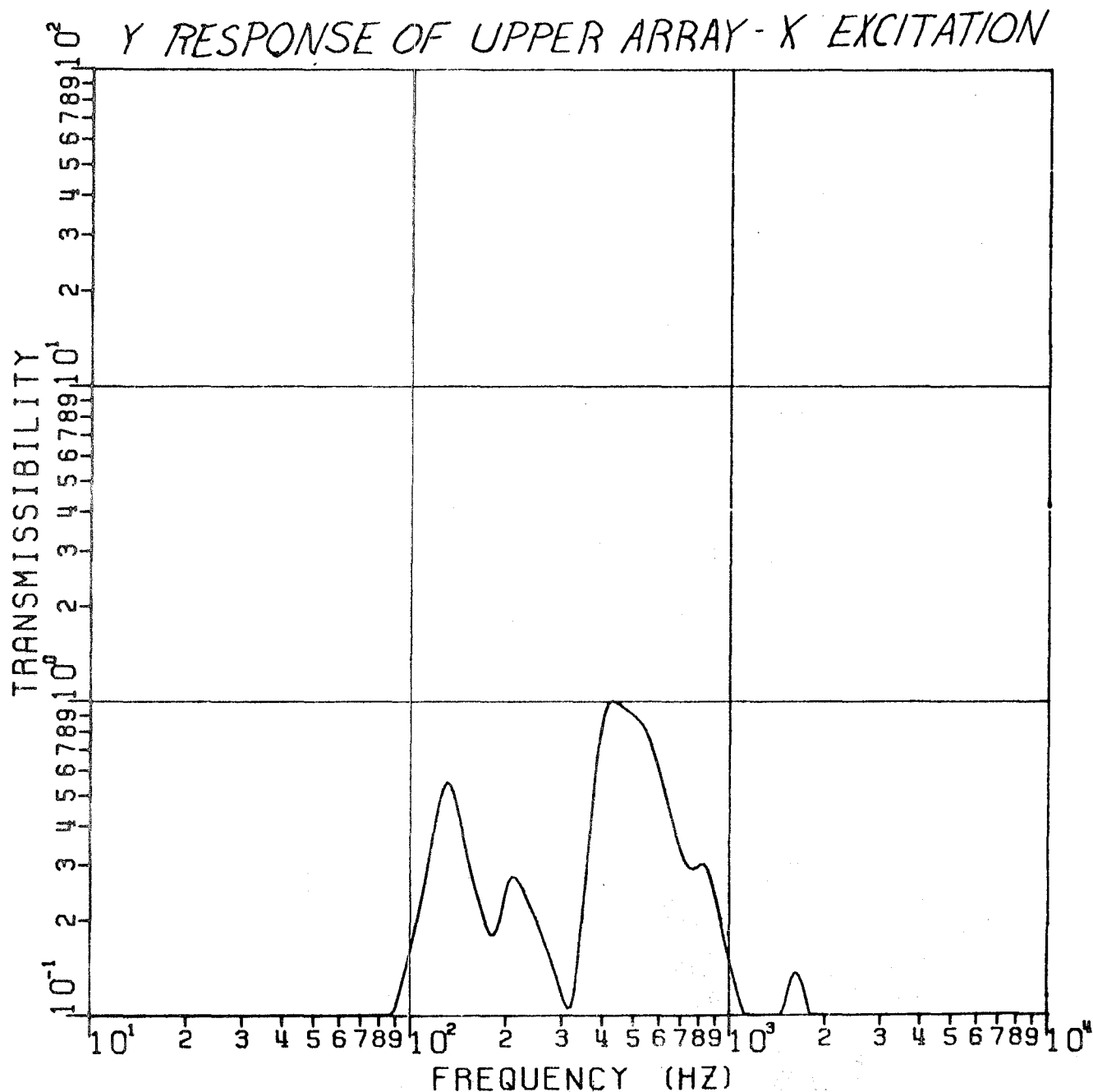
X RESPONSE OF UPPER ARRAY - X EXCITATION



FREQUENCY RESPONSE FOR LR3 300,6D.F. MODEL

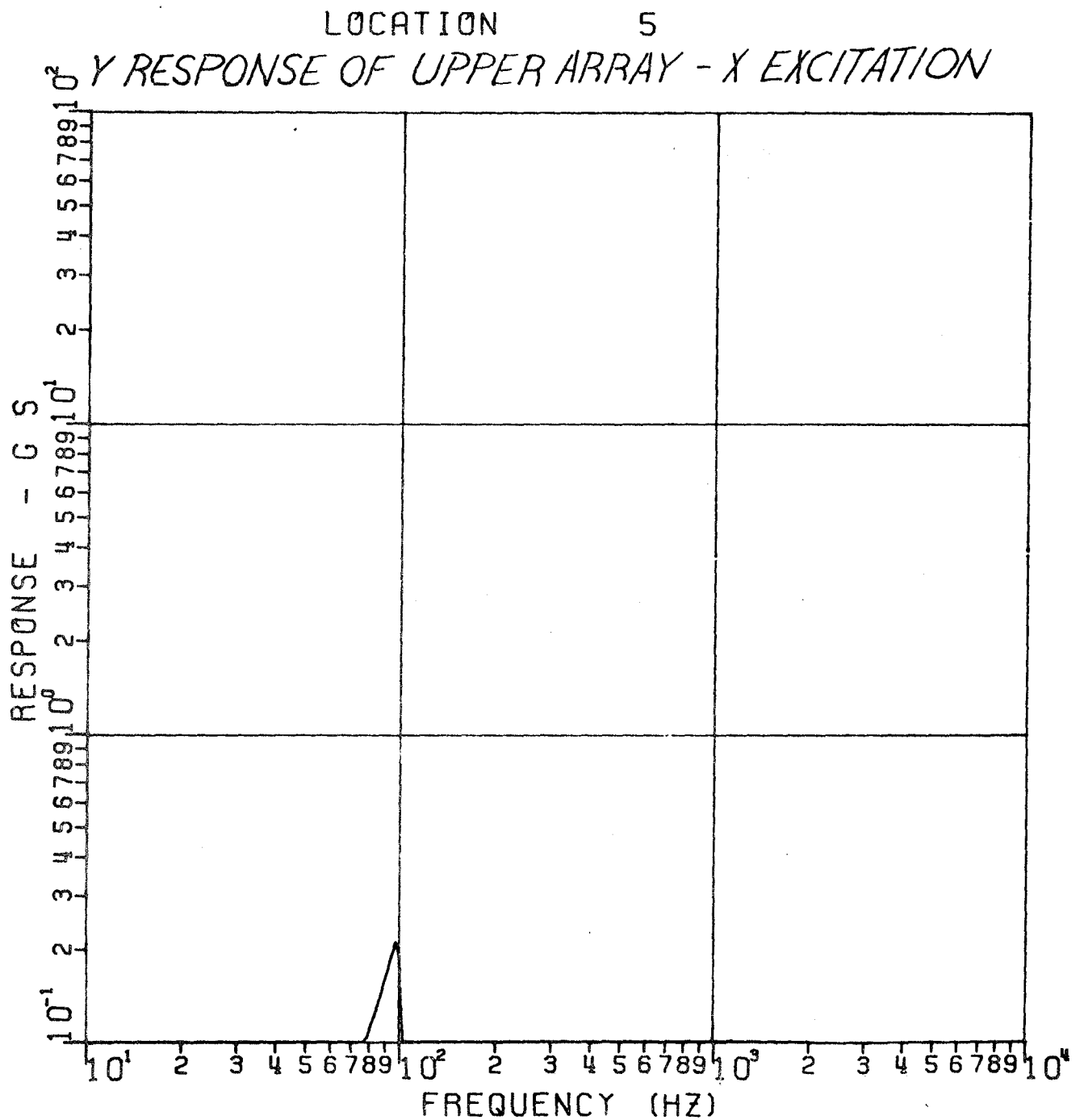
FIGURE 3.2.3 A TRANSMISSIBILITY

LOCATION 5



FREQUENCY RESPONSE FOR LR3 300.6D.F. MODEL

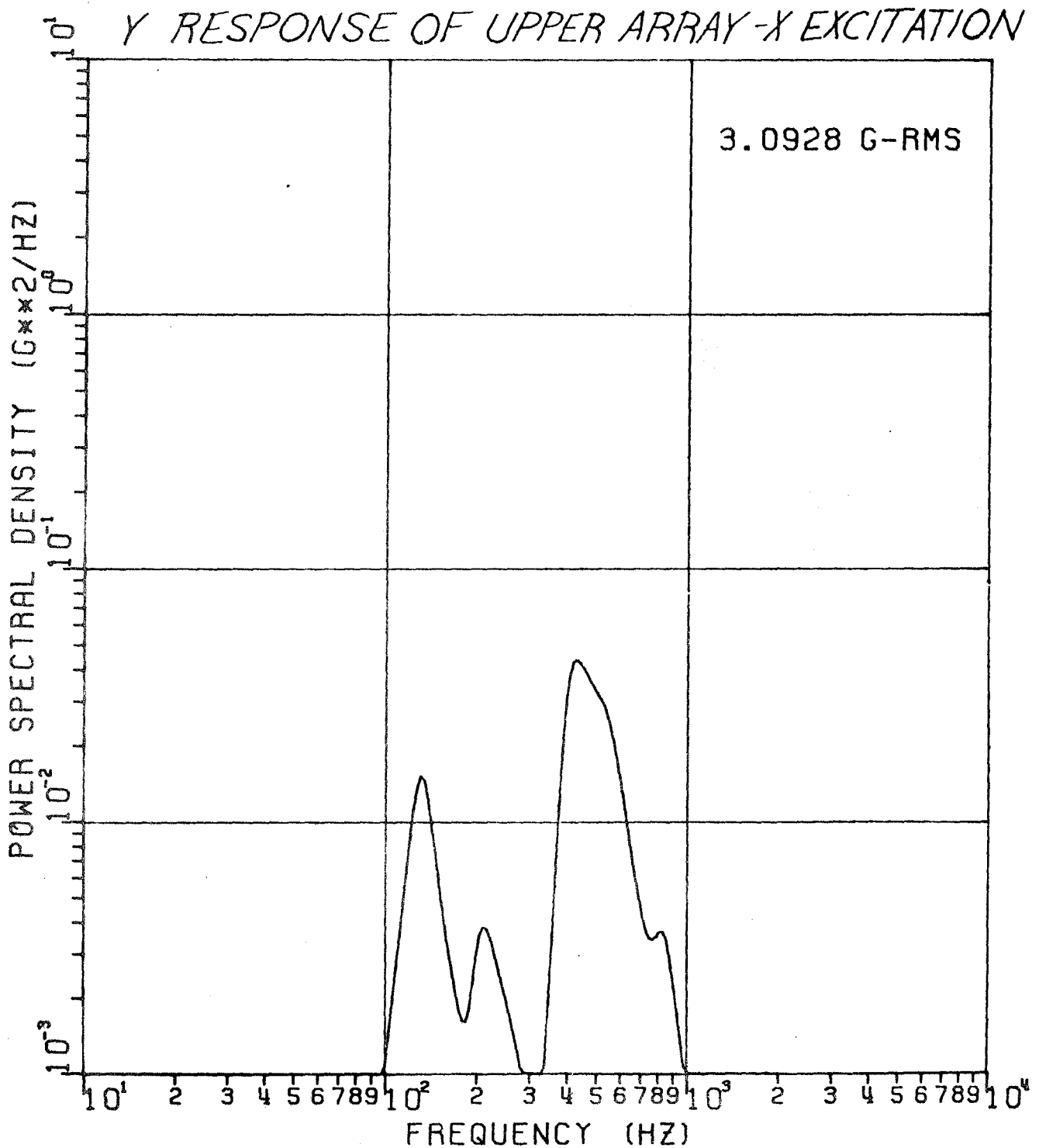
FIGURE 3.2.3 B SINE RESPONSE



FREQUENCY RESPONSE FOR LR3 300.60.F. MODE1

FIGURE 3.2.3C RANDOM VIBRATION SPECTRUM

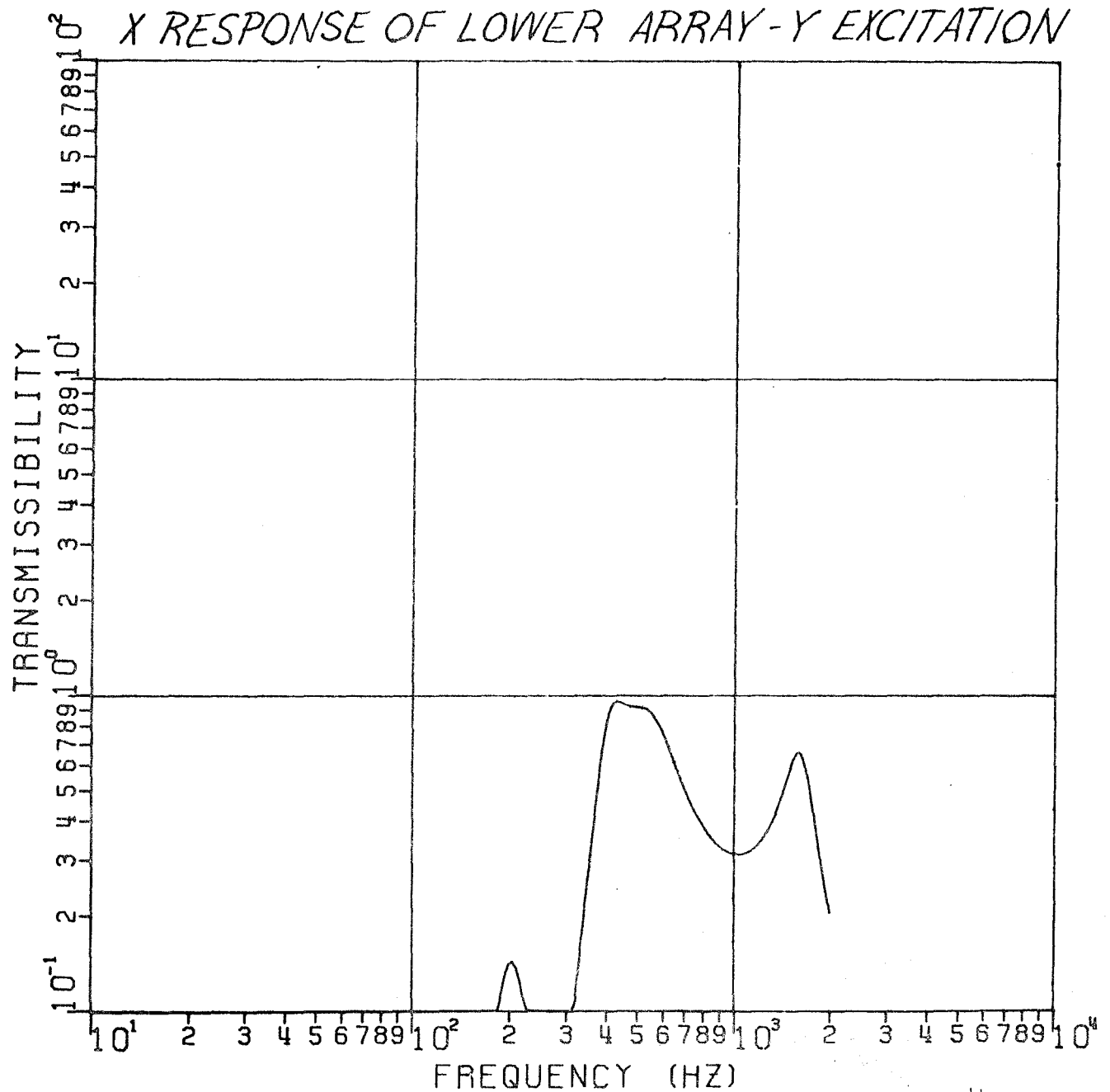
LOCATION 5



FREQUENCY RESPONSE FOR LR3 300,6D.F. MODEL

FIGURE 3.2.4A TRANSMISSIBILITY

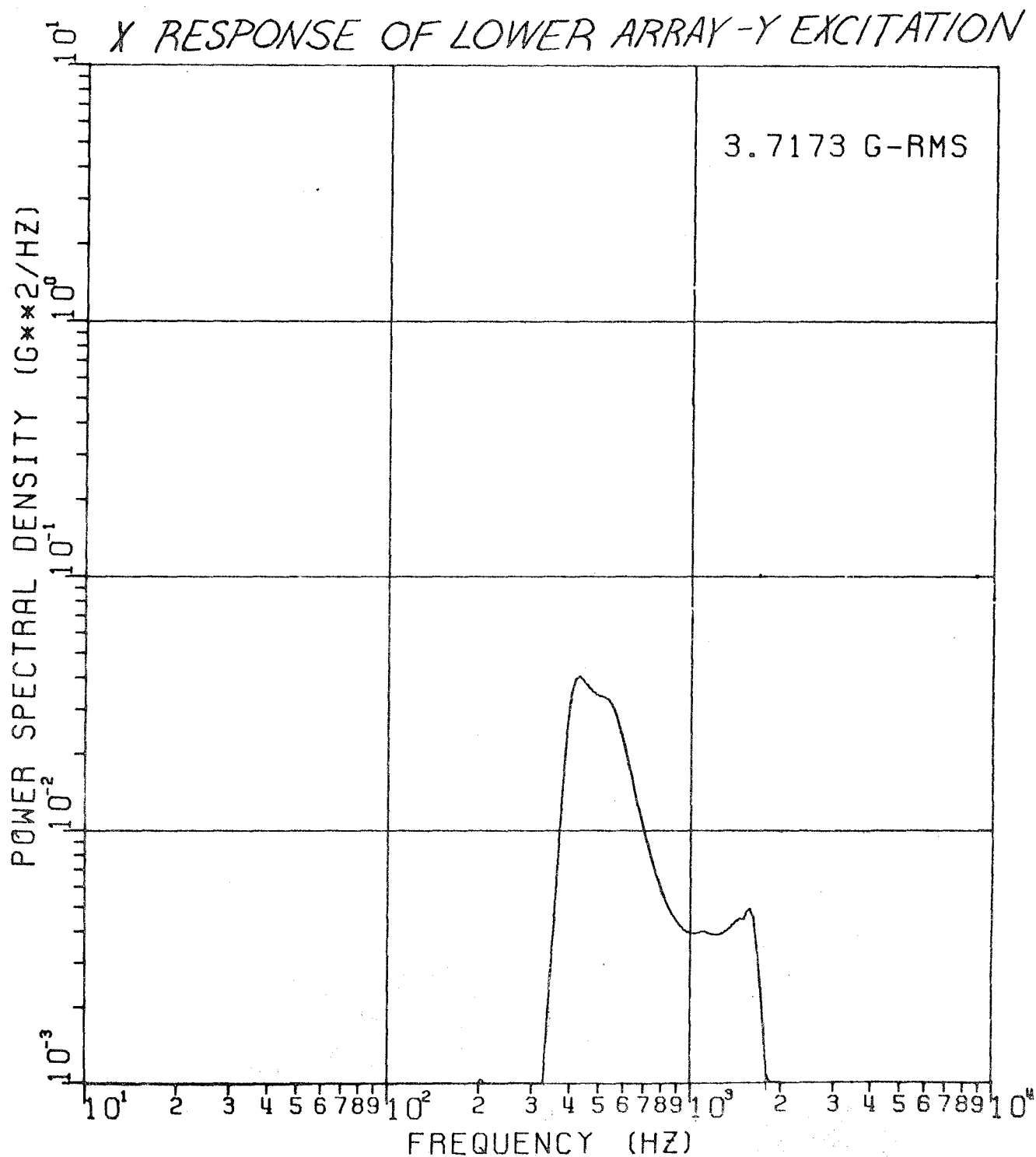
LOCATION 1



FREQUENCY RESPONSE FOR LR3 300,6D.F. MODEL

FIGURE 3.2.4 B RANDOM VIBRATION SPECTRUM

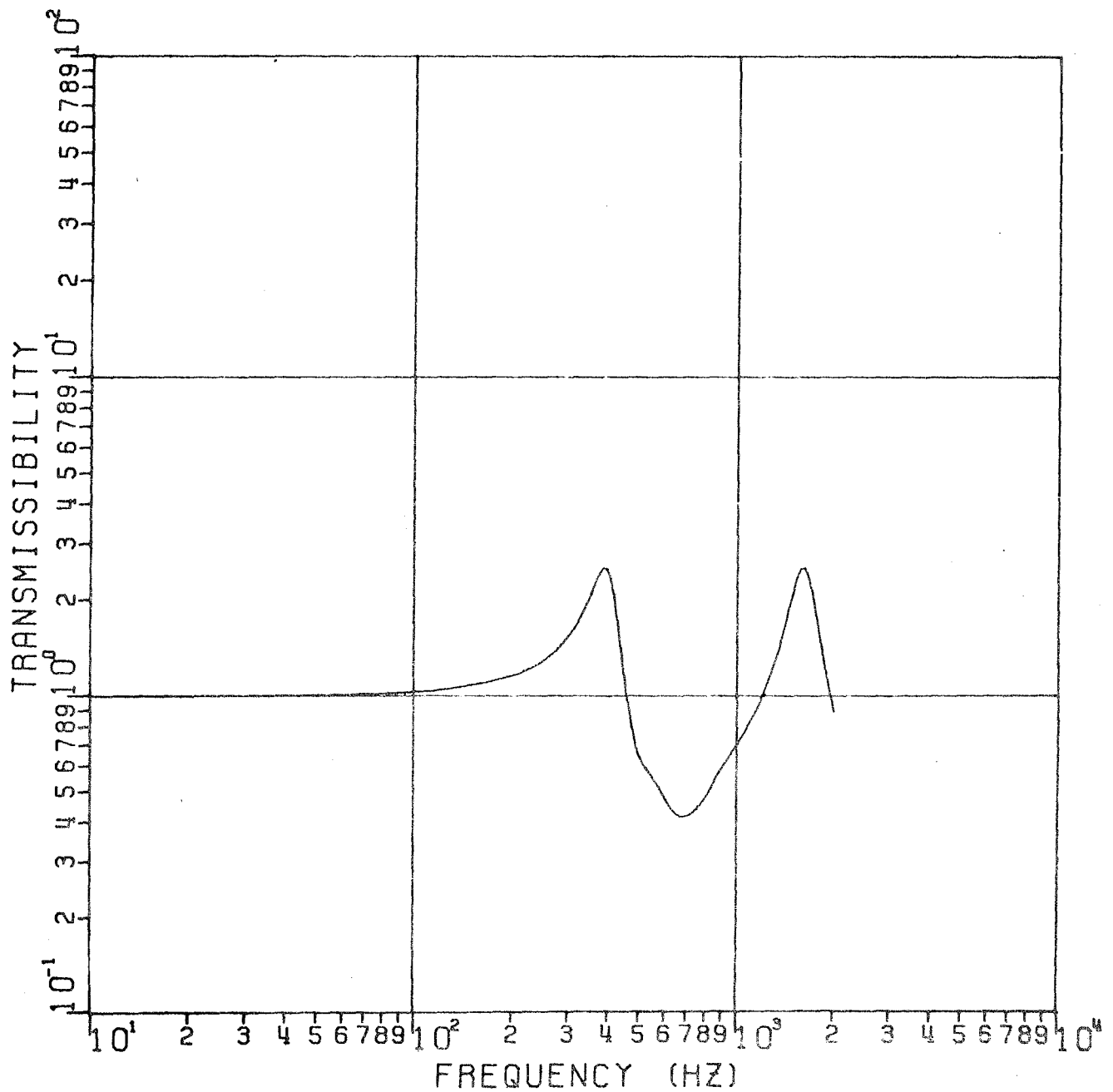
LOCATION 1



FREQUENCY RESPONSE FOR LR3 300,60.F. MODEL

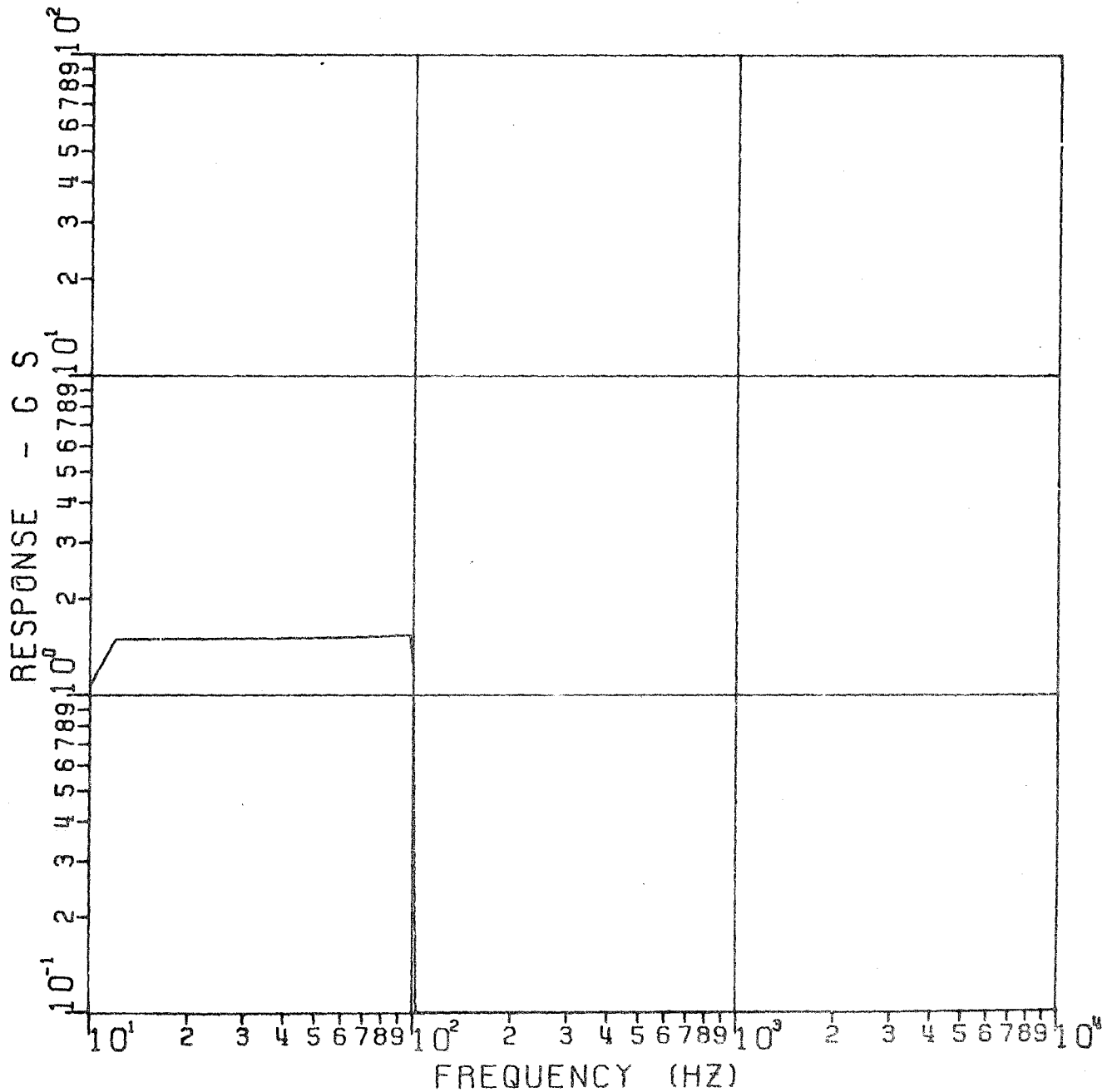
FIGURE 3.2.5 A TRANSMISSIBILITY

LOCATION 2



FREQUENCY RESPONSE FOR LR3 300,6D.F. MODEL, SEPT
FIGURE 3.2.5 B SINE RESPONSE

LOCATION 2

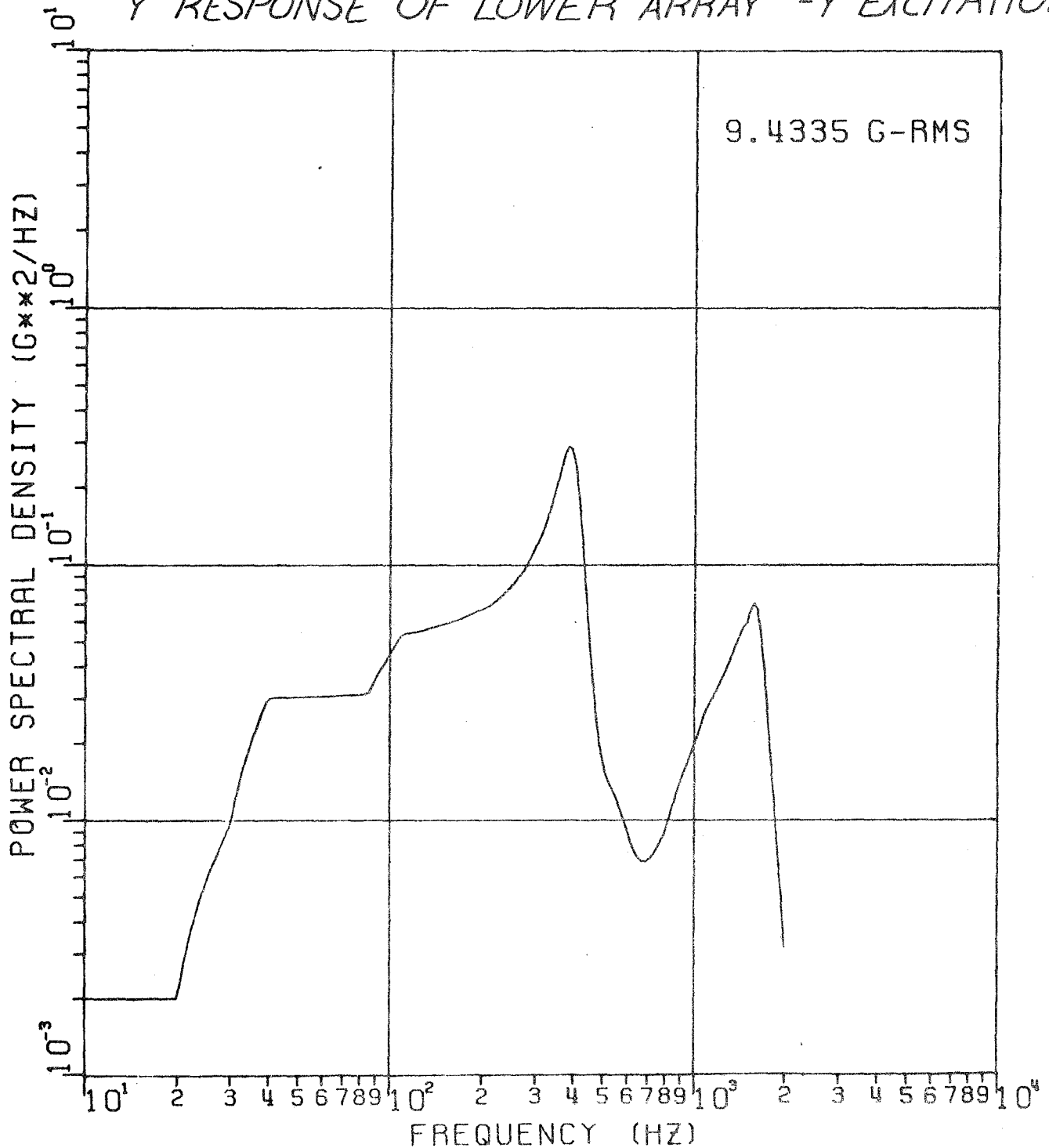


FREQUENCY RESPONSE FOR LR3 300,6D.F. MODEL

FIGURE 3.2.5 A RANDOM VIBRATION SPECTRUM

LOCATION 2

Y RESPONSE OF LOWER ARRAY -Y EXCITATION

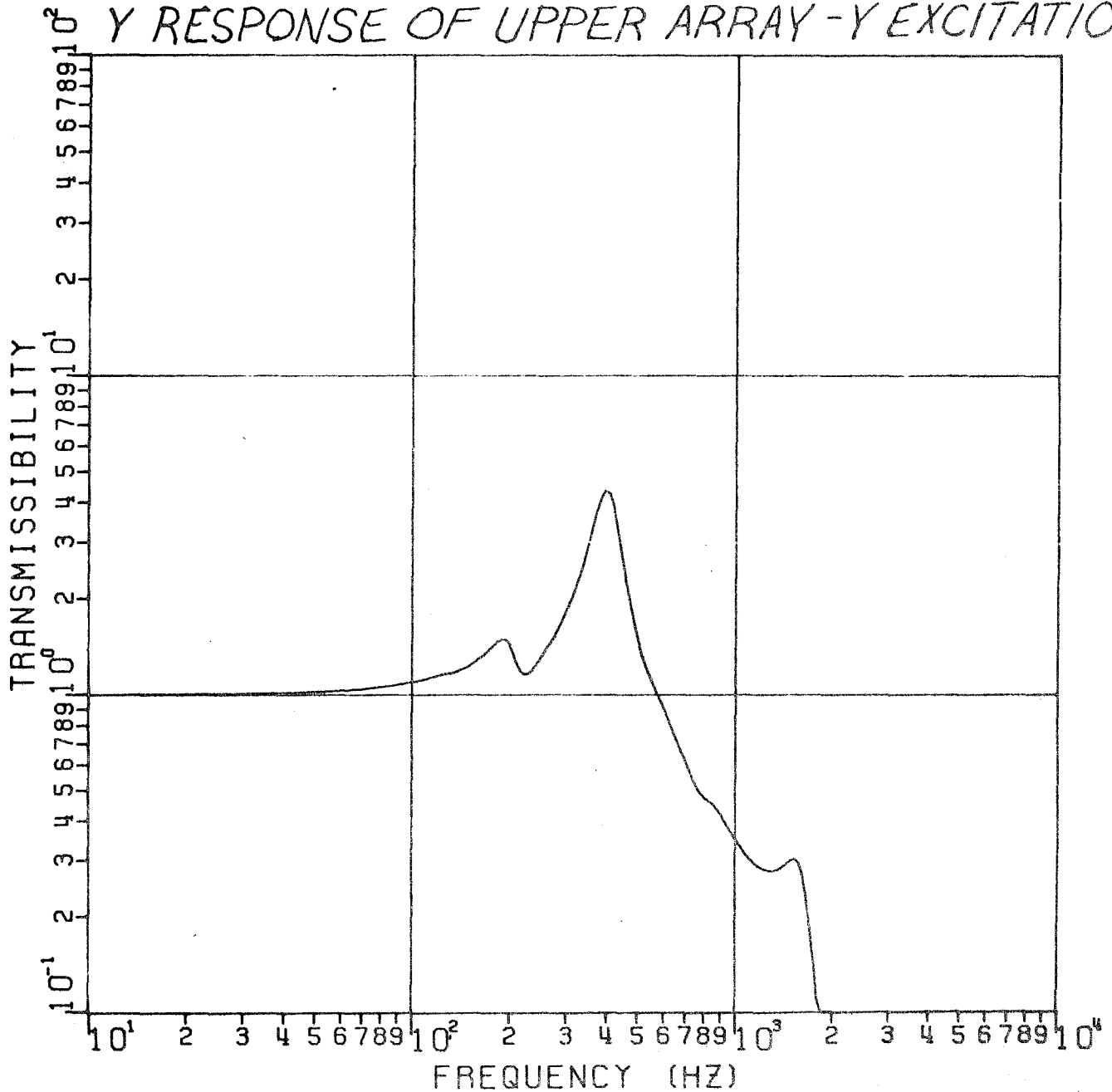


FREQUENCY RESPONSE FOR LR3 300,6D.F. MODEL

FIGURE 3.2.6 A TRANSMISSIBILITY

LOCATION 5

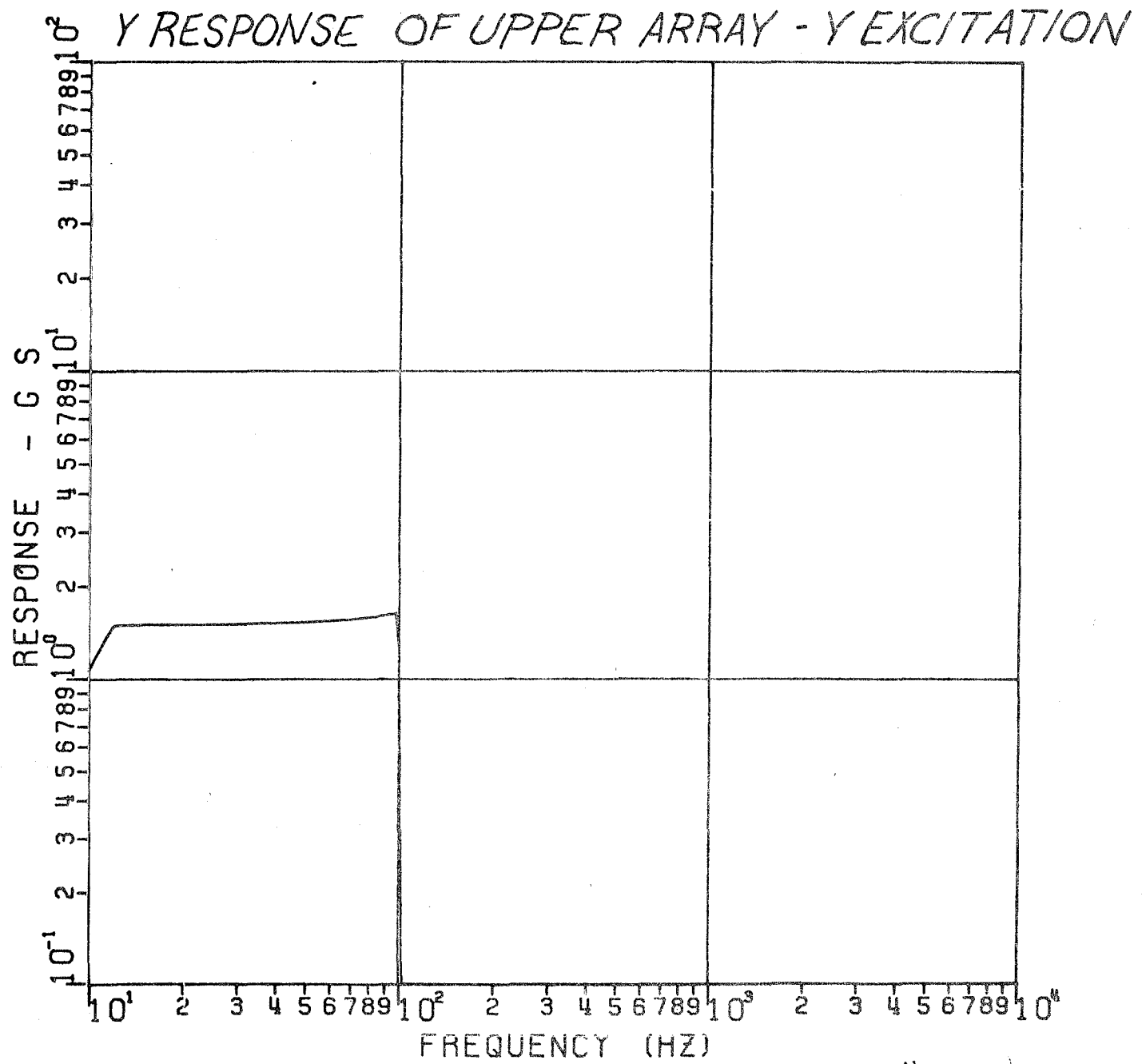
Y RESPONSE OF UPPER ARRAY - Y EXCITATION



FREQUENCY RESPONSE FOR LR3 300,6D.F. MODEL 5FP

FIGURE 3.2.6 B SINE RESPONSE

LOCATION 5



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FREQUENCY RESPONSE FOR LR3 300,6D.F. MODEL

FIGURE 3.2.6 C RANDOM VIBRATION SPECTRUM

LOCATION

5

Y RESPONSE OF UPPER ARRAY - Y EXCITATION

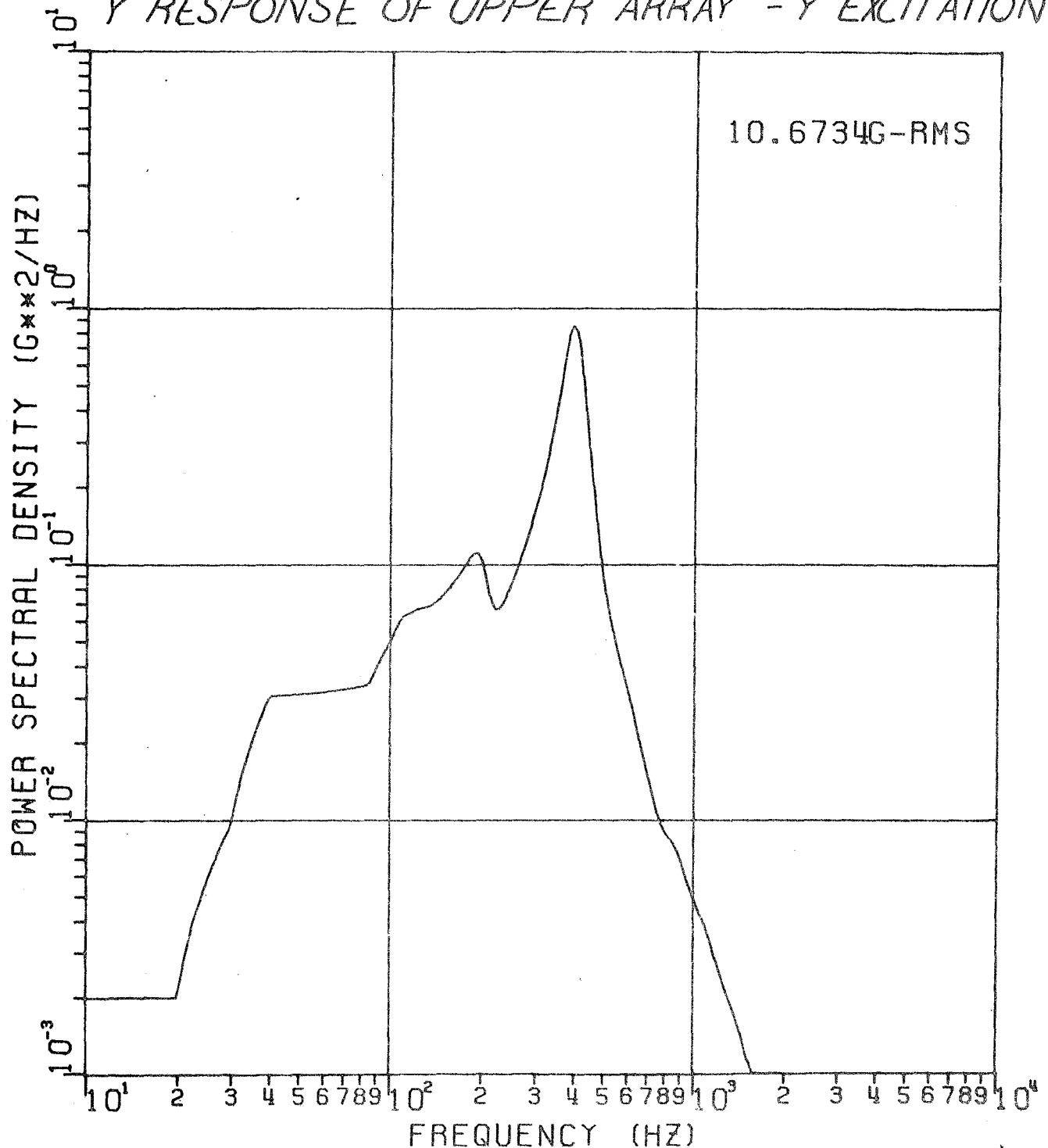


TABLE 3.2.1
SUMMARY OF IN PLANE RESPONSES

Excitation Direction	Array	Coordinate Direction	Max Sine Resp. (g's peak)	Random Resp. (g's RMS)
X	Upper	X	3.28	7.00
		Y	.215	3.09
		Θ	1.84×10^{-2} pseudo g's	.235 pseudo g's
	Lower	X	1.57	9.32
		Y	.512	2.84
		Θ	1.91×10^{-3} pseudo g's	.247 pseudo g's
Y	Upper	X	7.31×10^{-2}	.653
		Y	1.64	10.67
		Θ	1.56×10^{-2} pseudo g's	.611 pseudo g's
	Lower	X	1.55×10^{-2}	3.71
		Y	1.54	9.43
		Θ	3.86×10^{-3} pseudo g's	.859 pseudo g's

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TABLE 3.2.2

Excitation Direction	Array	Translational Response (grms)	Rotational Response $\frac{\text{rad}}{\text{sec}^2}$ 386	Distance To Edge (in)	Total* Resp. (grms)
X	Upper	7.00	.235	6.5	7.16
	Lower	9.32	.248	12.9	9.82
Y	Upper	10.67	.611	10.9	12.5
	Lower	9.43	.859	10.9	13.3

$$* \text{Total Response} = \sqrt{\text{Translational Response}^2 + (\text{Rotational Resp.} \times \text{Dist. to Edge})^2}$$



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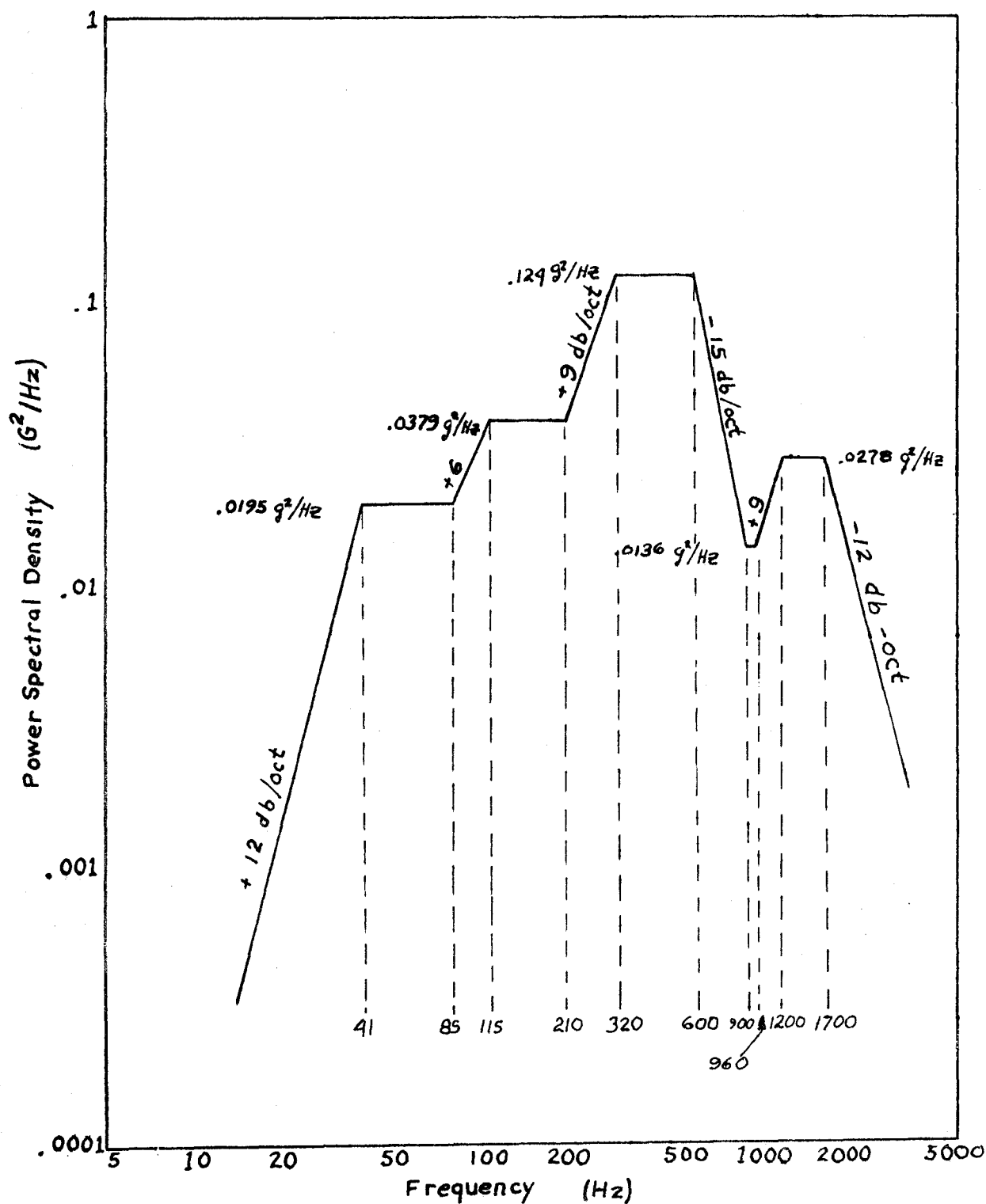
This table assumes that maximum responses in translation occur at the same frequency as maximum rotation responses, a conservative assumption. The table indicates that the maximum responses at a corner occurs along the front and rear edges of the large array. The root mean square response along these edges is 13.3 grms giving a 3 sigma peak value of 39.9 g. The peak power spectral density of the response occurs at about 400 Hz. Thus a corner mount design that will survive a sinusoidal vibration environment well in excess of 40 g peak in this frequency range is an adequate design.

In order to test the upper array alone, vibration levels in the plane of the arrays at the attachments between the two arrays were calculated. Since the test set up required the upper array to be tested with translational vibration only, the vibration spectrum on the lower array under the c. g. of the upper array was found. Due to the proximity of the centers of gravity of the two arrays and the relatively low levels of the rotational responses at the lower array c. g. , the response under the c. g. of the small array due to rotation of the large array could be neglected. In order to simplify the test set up, the responses due to X and Y excitation were superimposed to give one set of test vibration levels for both the X and Y axes. The test envelope levels, reduced by a factor of 1.69 to acceptance levels, are shown in Figure 3.2.7.

Figure 3.2.7

UPPER ARRAY ACCEPTANCE RANDOM VIBRATION LEVELS

X & Y AXES IN PLANE OF ARRAY





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3.3 Responses In Off-Loaded Condition

Response to the LM Quad III vibration environment for the LRRR in the "off-loaded condition" were calculated perpendicular to the plane of the array using the model of section 2.3. Responses of several locations on this model are shown in Figure 3.3.1 - 3.3.5. The responses for the experiment in this condition are seen to be slightly reduced compared to out-of-plane responses of the lower array with the upper array in place.

In-plane responses of the lower array alone were not calculated. This is justified since in the plane of the arrays the upper array adds no stiffness to the lower array but does mass load the lower array. Since the random excitation spectrum is almost flat in the region of the natural frequencies of both cases, larger loads on the LM interface fittings will be experienced with the upper array connected.

4.0 Dynamic Loads & Deflections

4.1 Method of Calculation

Dynamic loads on the structure were calculated using the out-of-plane model of the two connected arrays. The in-plane model gives dynamic loads on the interface fittings, hinges, and clevis fittings. The loads used in designing these fittings were so conservative that any conceivable dynamic load produced by the dynamic environment will be easily survived. Dynamic loads with the upper array removed will be lower than with it attached since mass loading induced by the small array is large compared to the amount of stiffness the small array contributes to the lower array.

The acceleration response at each lumped mass location, the stiffness matrices for each array, and the transformation between coordinates local to each array and coordinates of the whole system determine the dynamic loads on the system.

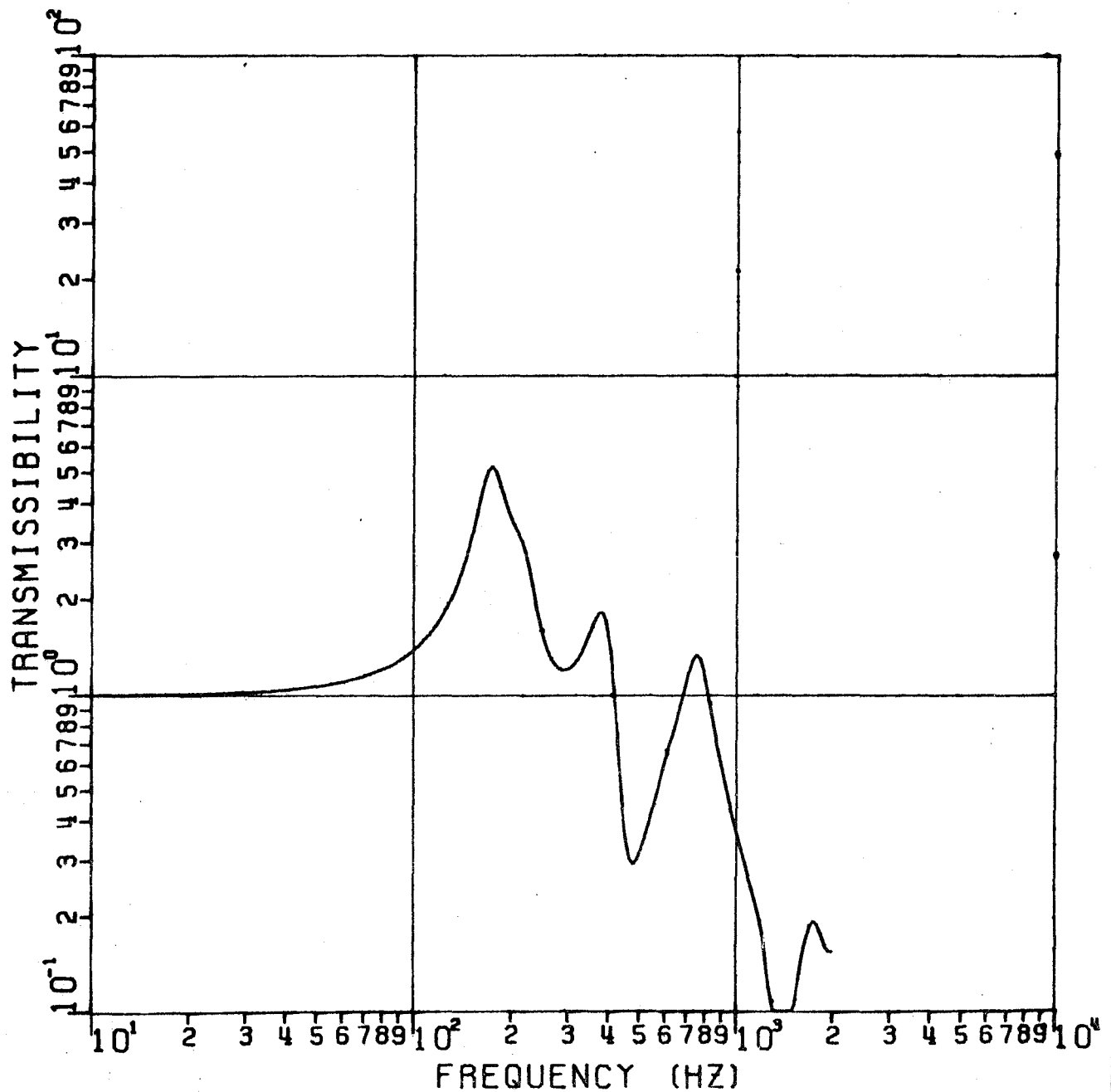
$$\text{Let } [K] = \begin{bmatrix} [K_L] & 0 & 0 \\ 0 & [K_u] & [-K_u] \\ 0 & [-K_u] & [K_u] \end{bmatrix}$$

Where $[K_L]$ & $[K_u]$ are as previously defined

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FIGURE 3.3./A TRANSMISSIBILITY

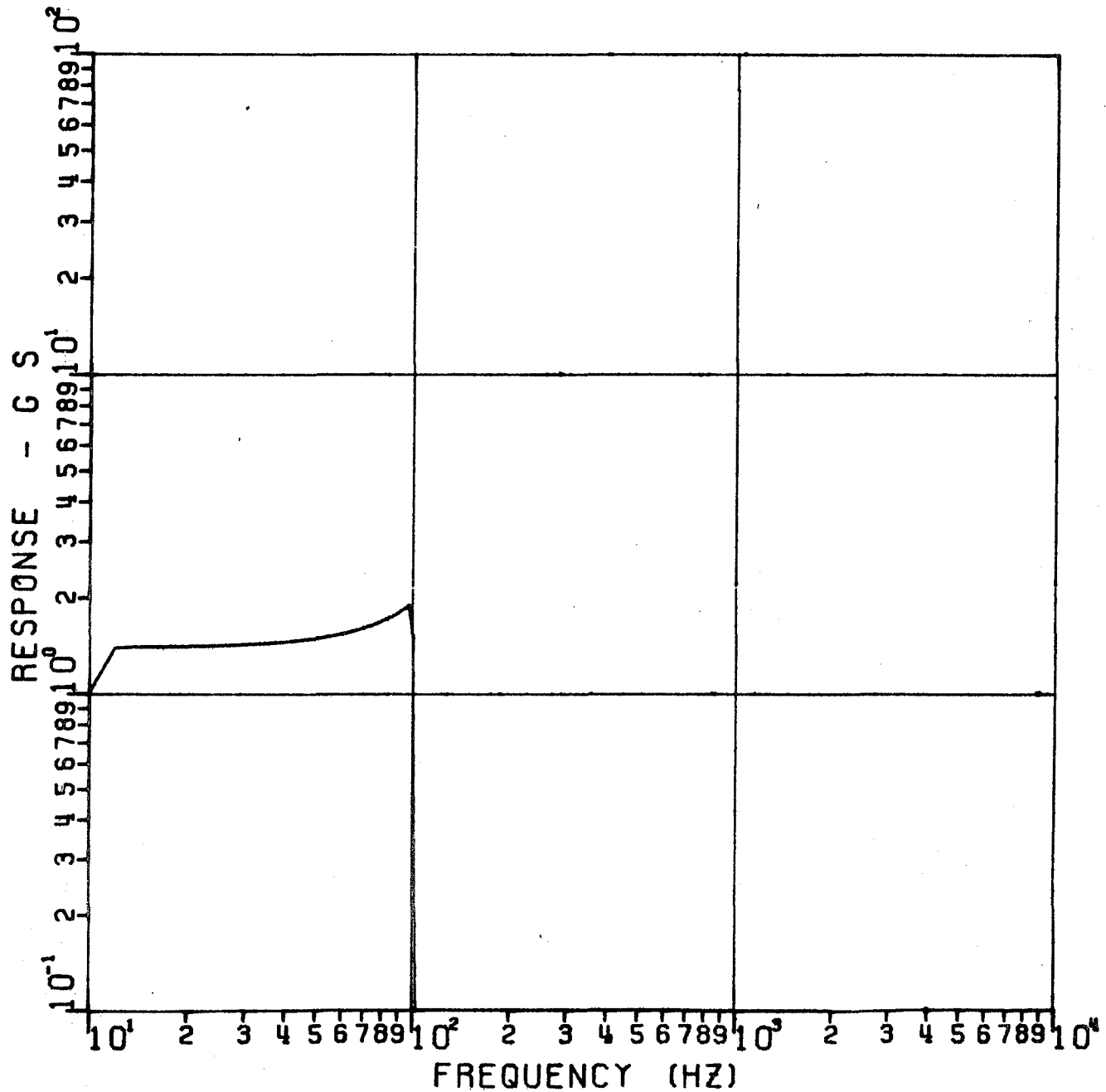
LOCATION 3



LRRR 300 IN THE OFF LOADED CONDITION 23 D.I

FIGURE 3.3.1 B SINE RESPONSE

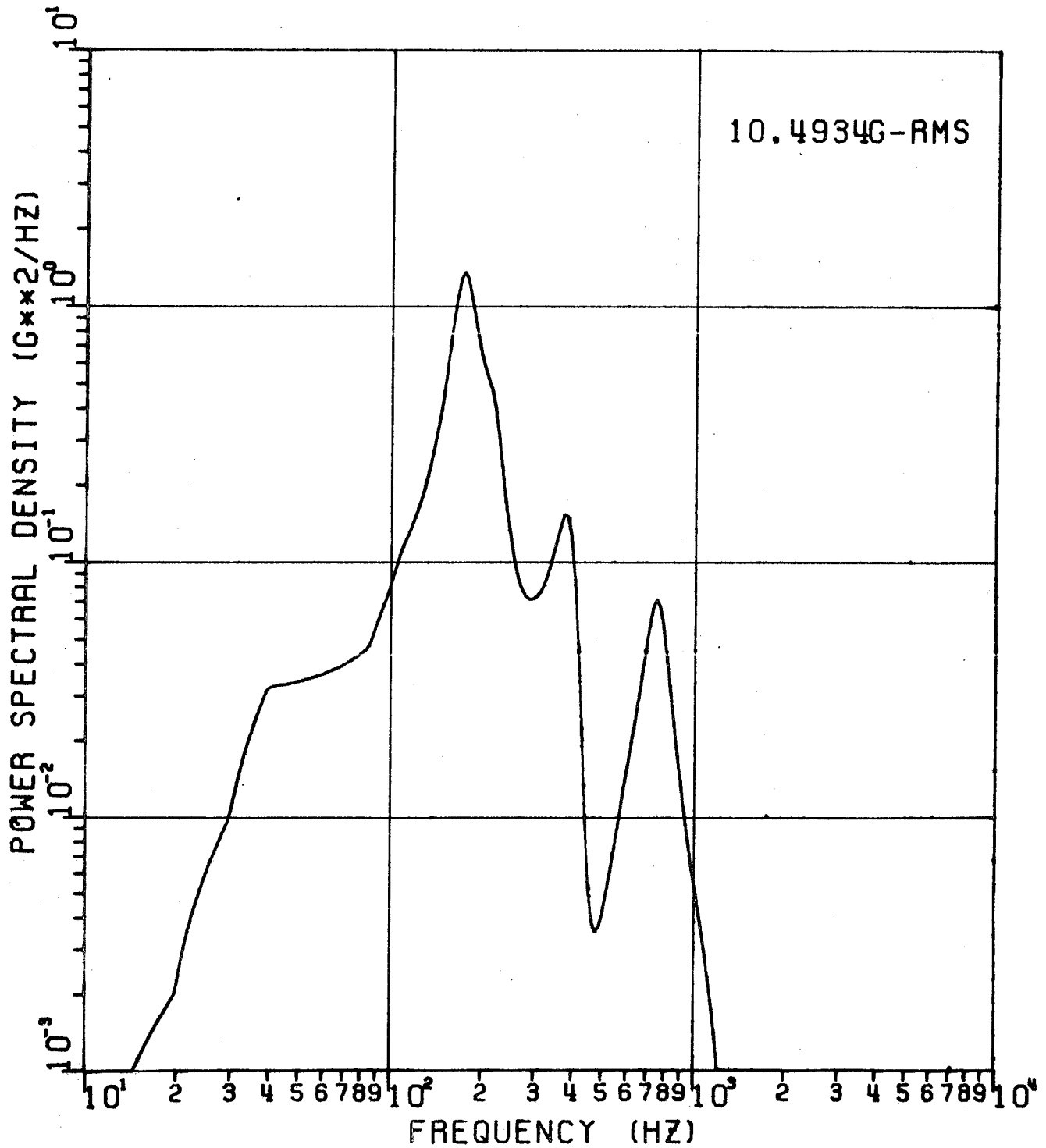
LOCATION 3



LRRR 300 IN THE OFF LOADED CONDITION 23 D.F

FIGURE 3.3./C RANDOM VIBRATION SPECTRUM

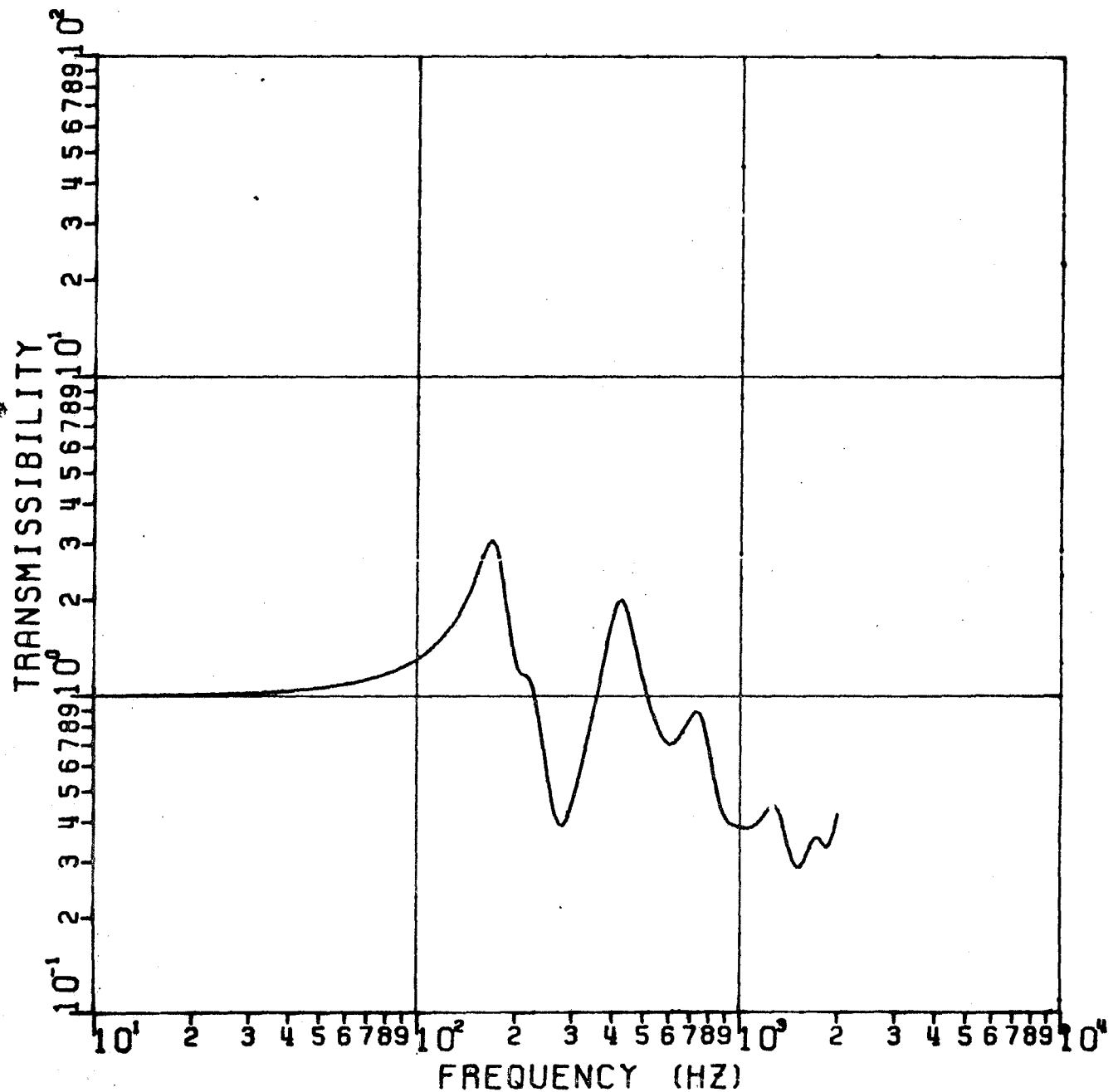
LOCATION 3



LRRR 300 IN THE OFF LOADED CONDITION 23 D.F

FIGURE 3.3.2 A TRANSMISSIBILITY

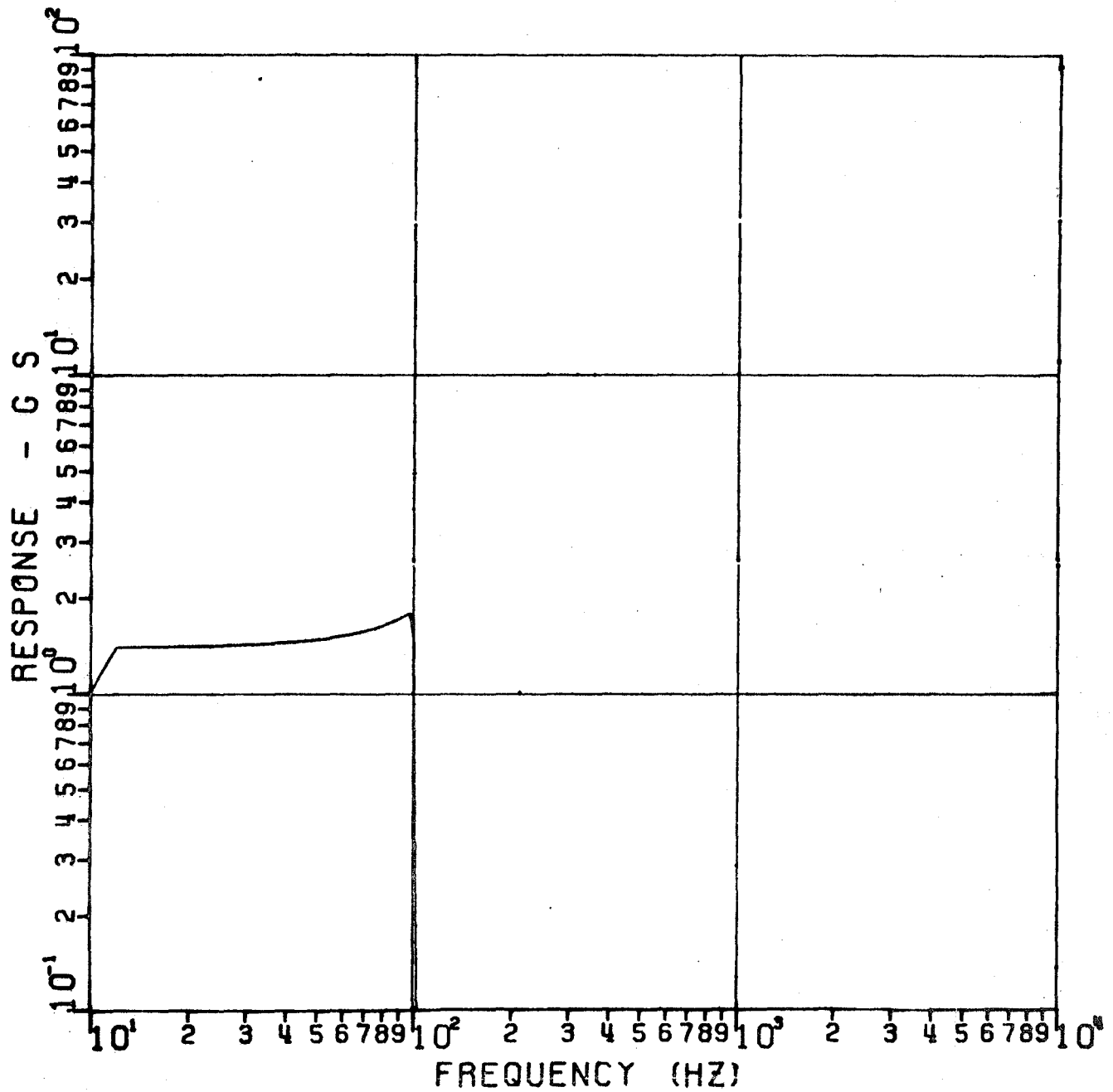
LOCATION 8



LRRR 300 IN THE OFF LOADED CONDITION 23 D.F

FIGURE 3.3.2 B SINE RESPONSE

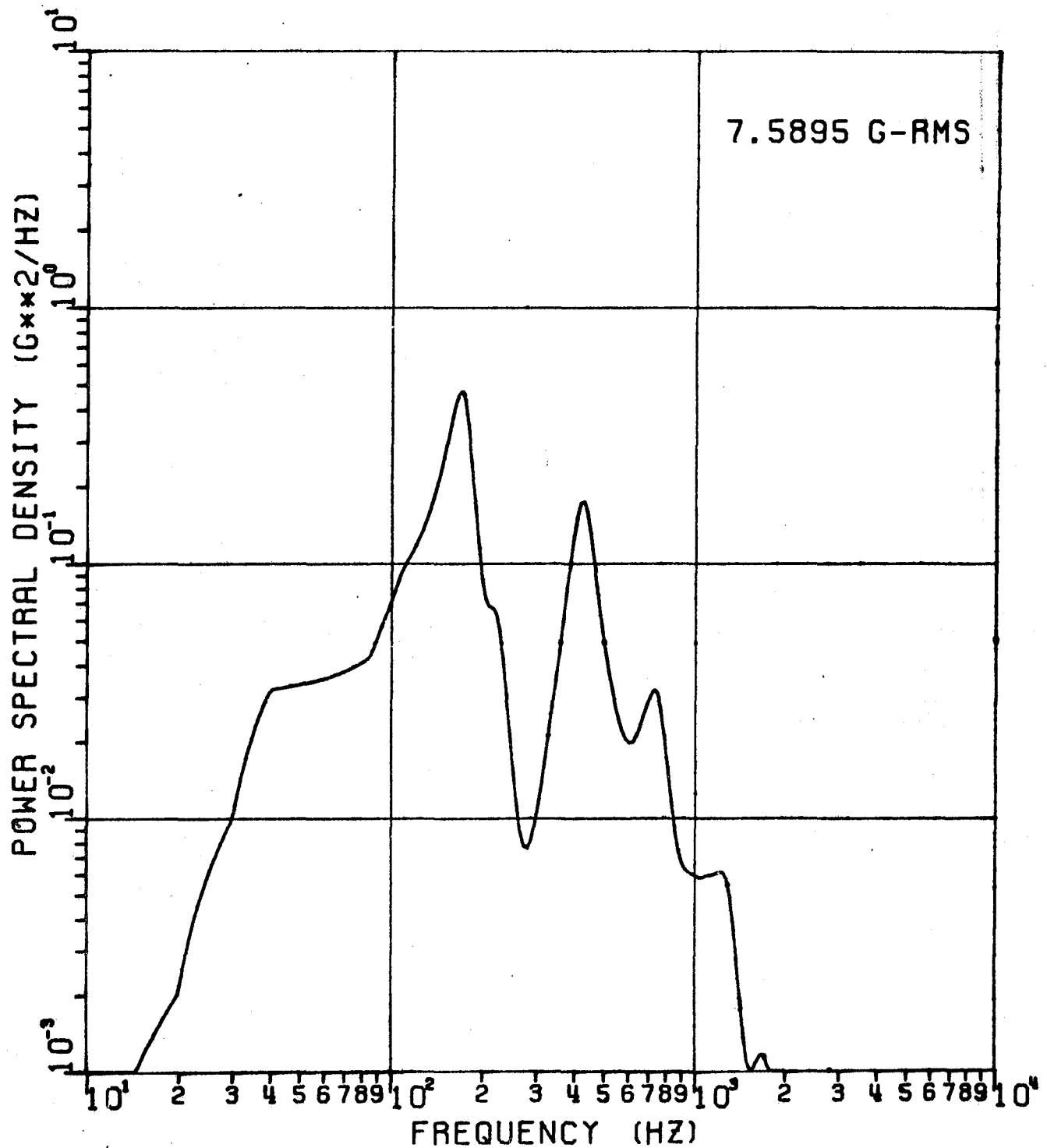
LOCATION 8



LRRR 300 IN THE OFF LOADED CONDITION 23 D.F

FIGURE 3.3.2C RANDOM VIBRATION SPECTRUM

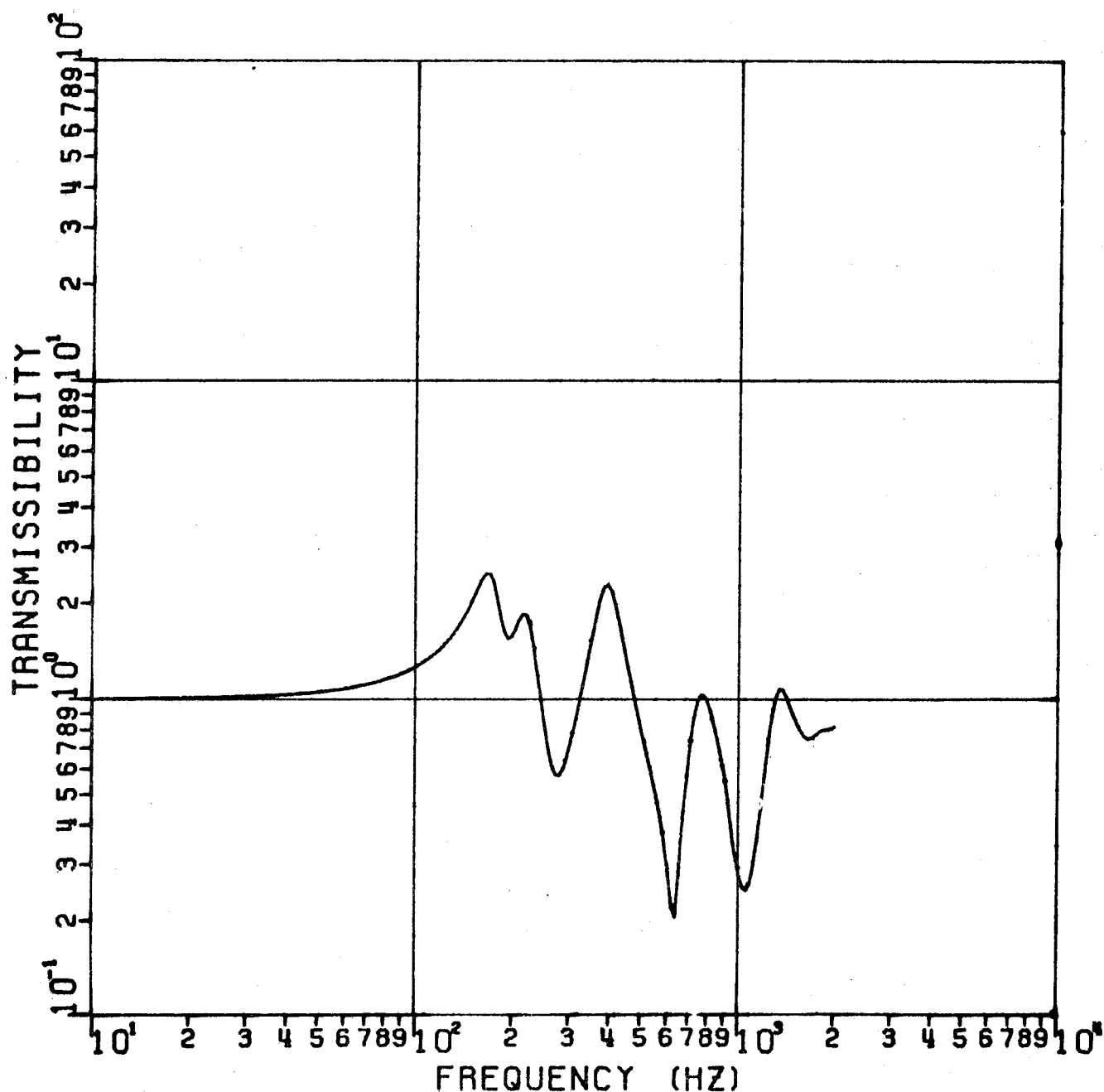
LOCATION 8



LRRR 300 IN THE OFF LOADED CONDITION 23 D.F

FIGURE 33.3A TRANSMISSIBILITY

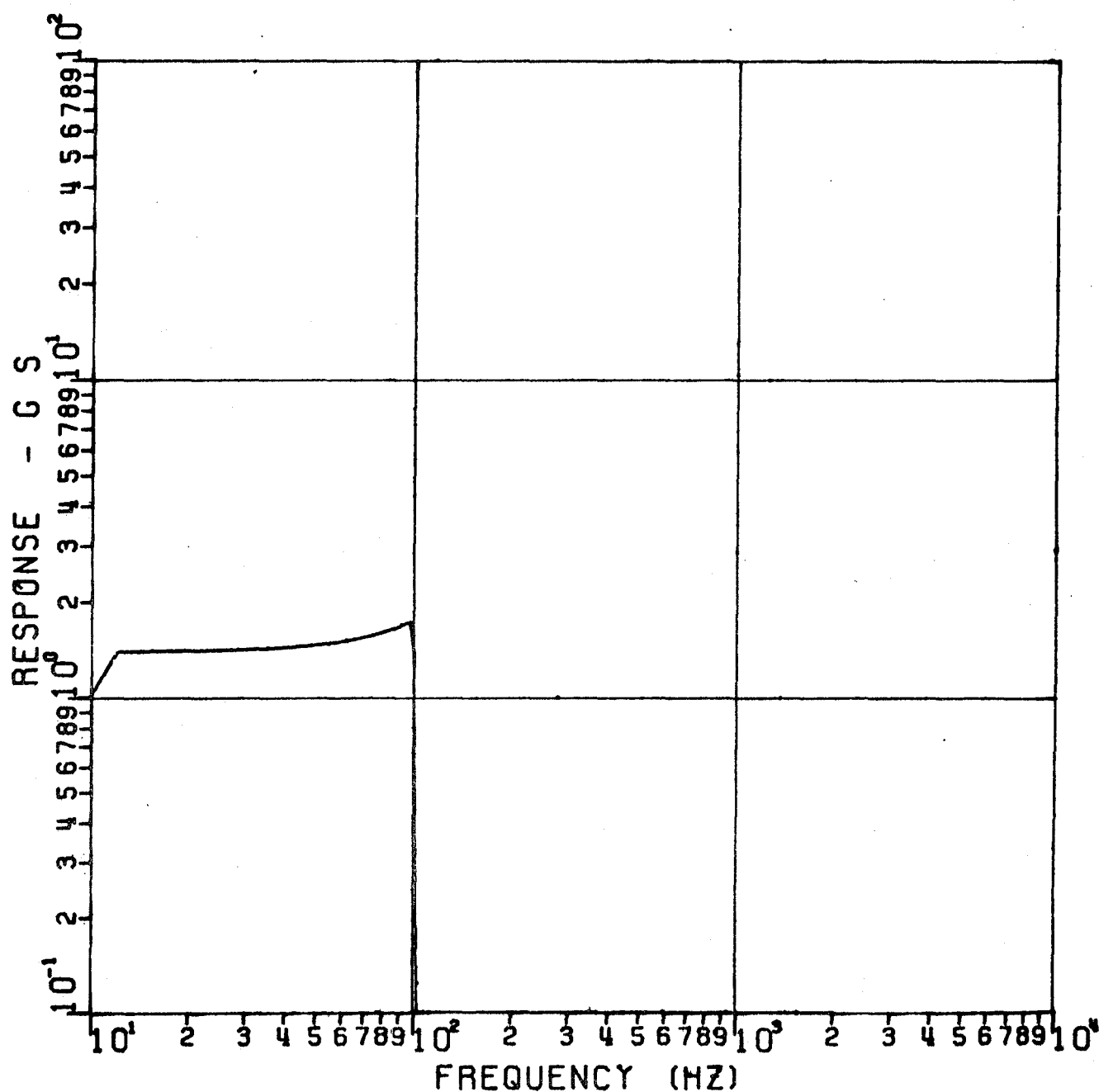
LOCATION 10



LRRR 300 IN THE OFF LOADED CONDITION 23 D.F

FIGURE 3.3.3 B SINE RESPONSE

LOCATION 10



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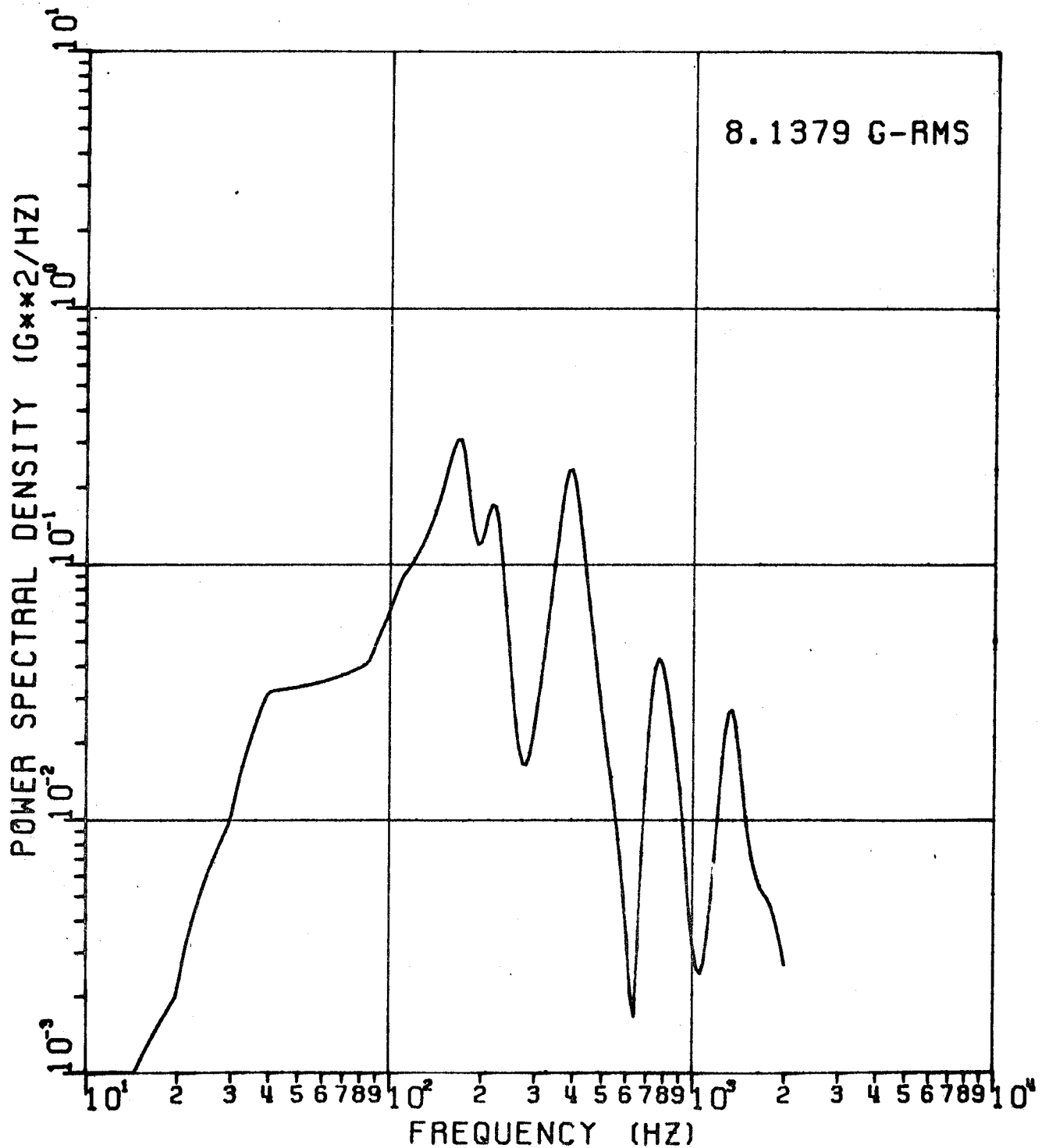
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LRRR 300 IN THE OFF LOADED CONDITION 23 D.F

FIGURE 3.3.3C RANDOM VIBRATION SPECTRUM

LOCATION

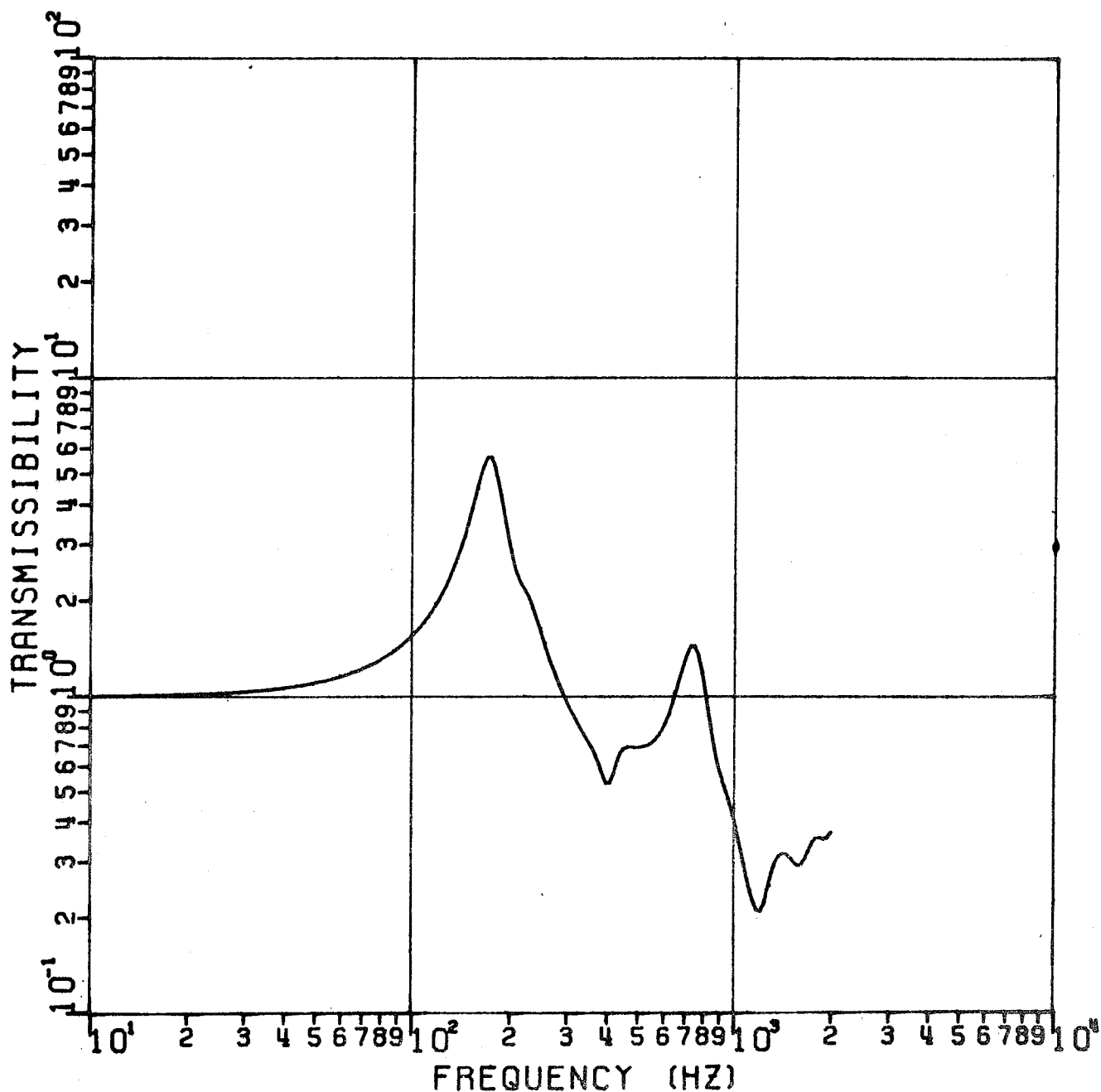
10



LRRR 300 IN THE OFF LOADED CONDITION 23 D.F

FIGURE 3.3.4 A TRANSMISSIBILITY

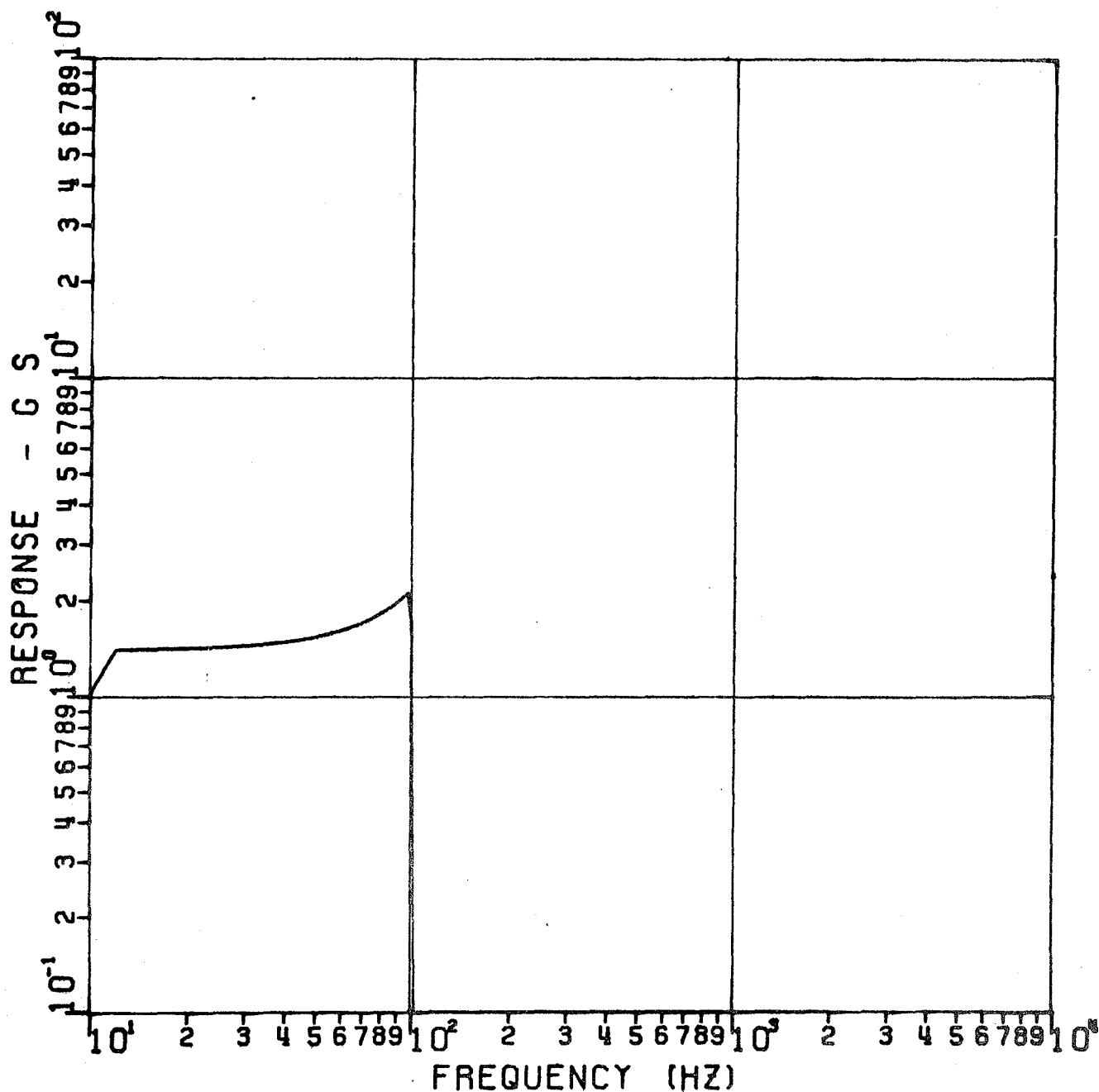
LOCATION 15



LRRR 300 IN THE OFF LOADED CONDITION 23 D.F

FIGURE 3.3.4 B SINE RESPONSE

LOCATION 15

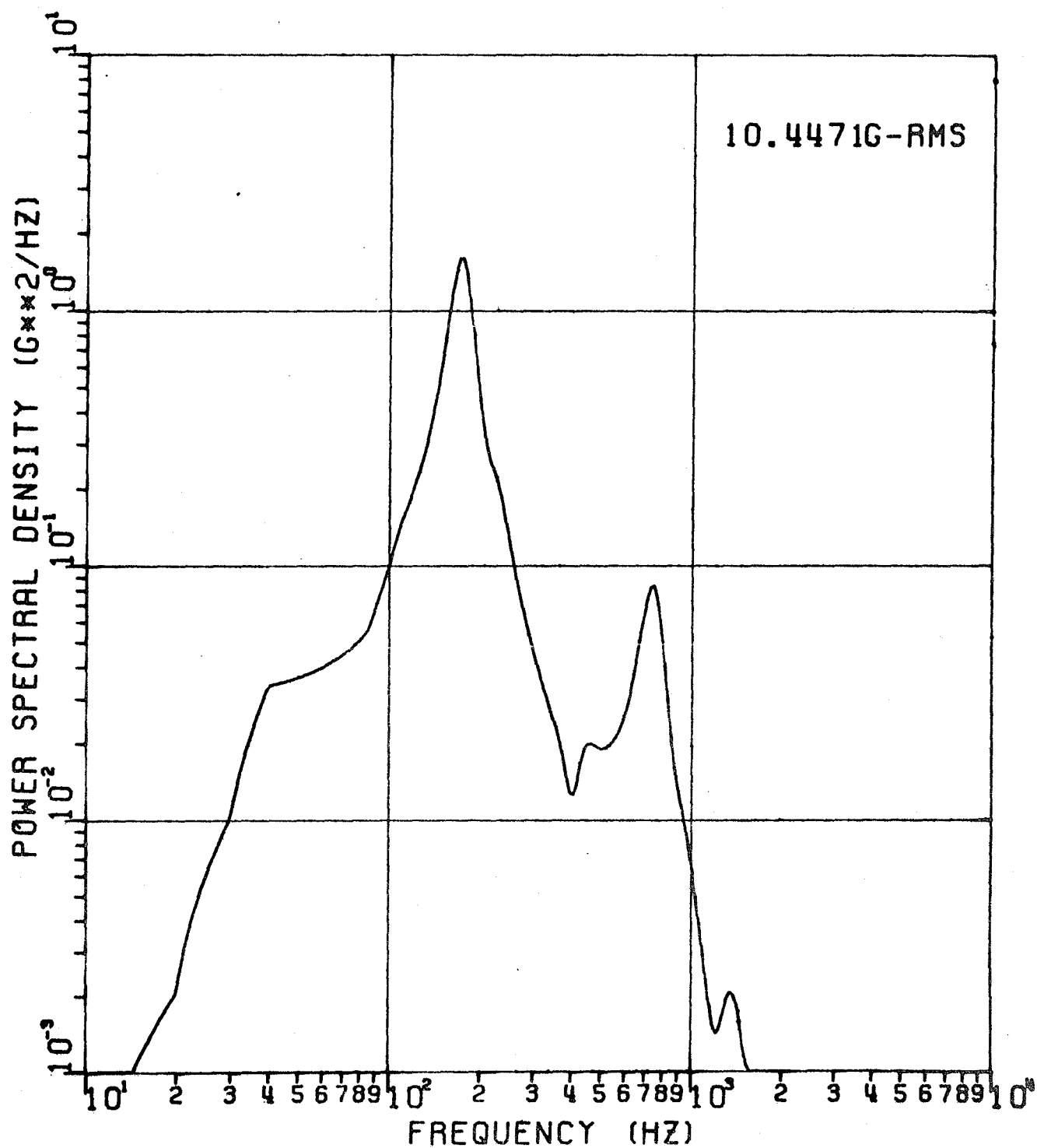


LRRR 300 IN THE OFF LOADED CONDITION 23 D.1

FIGURE 3.3.4C RANDOM VIBRATION SPECTRUM

LOCATION

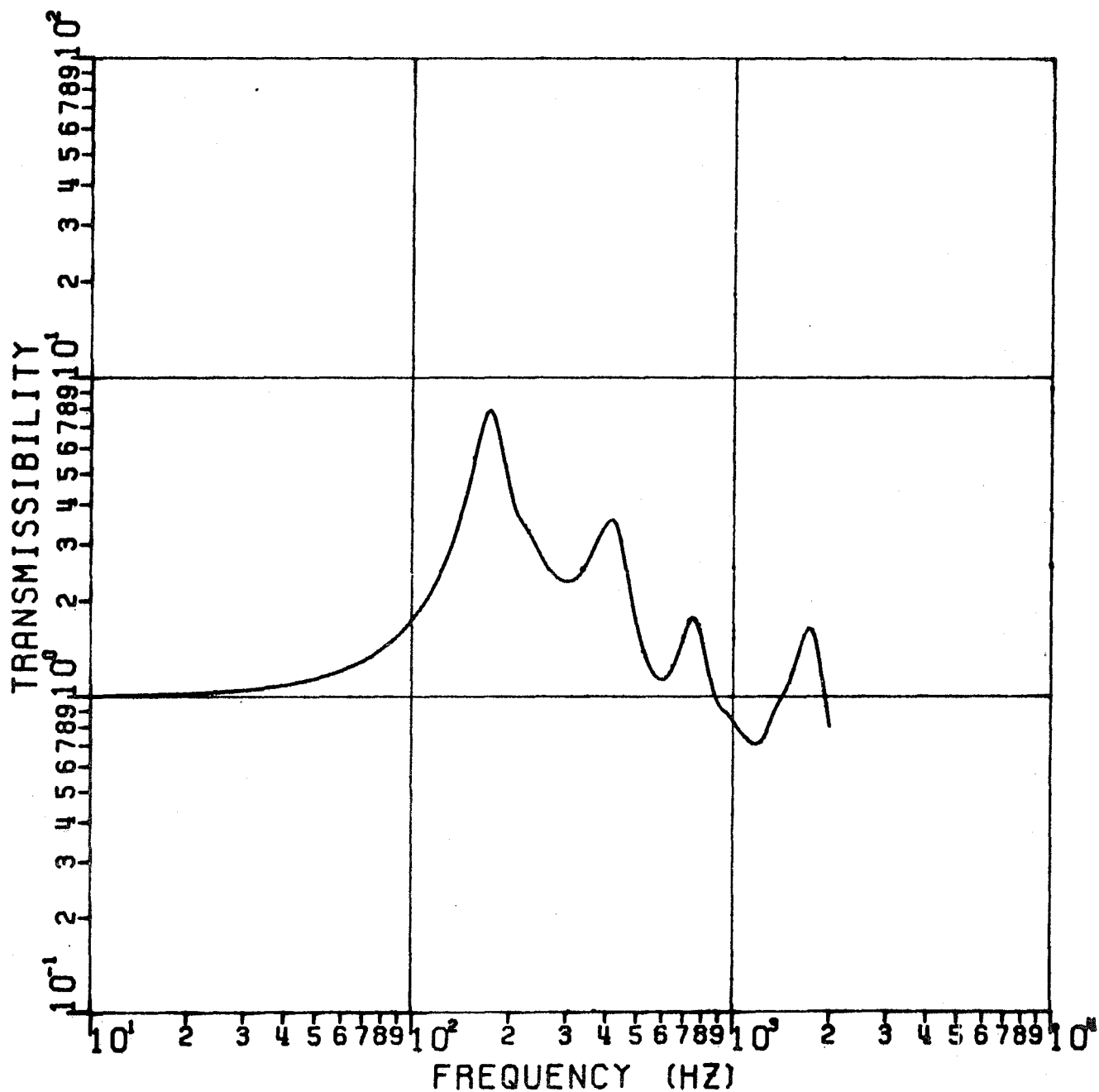
15



LRRR 300 IN THE OFF LOADED CONDITION 23 D.F

FIGURE 3.3.5 A TRANSMISSIBILITY

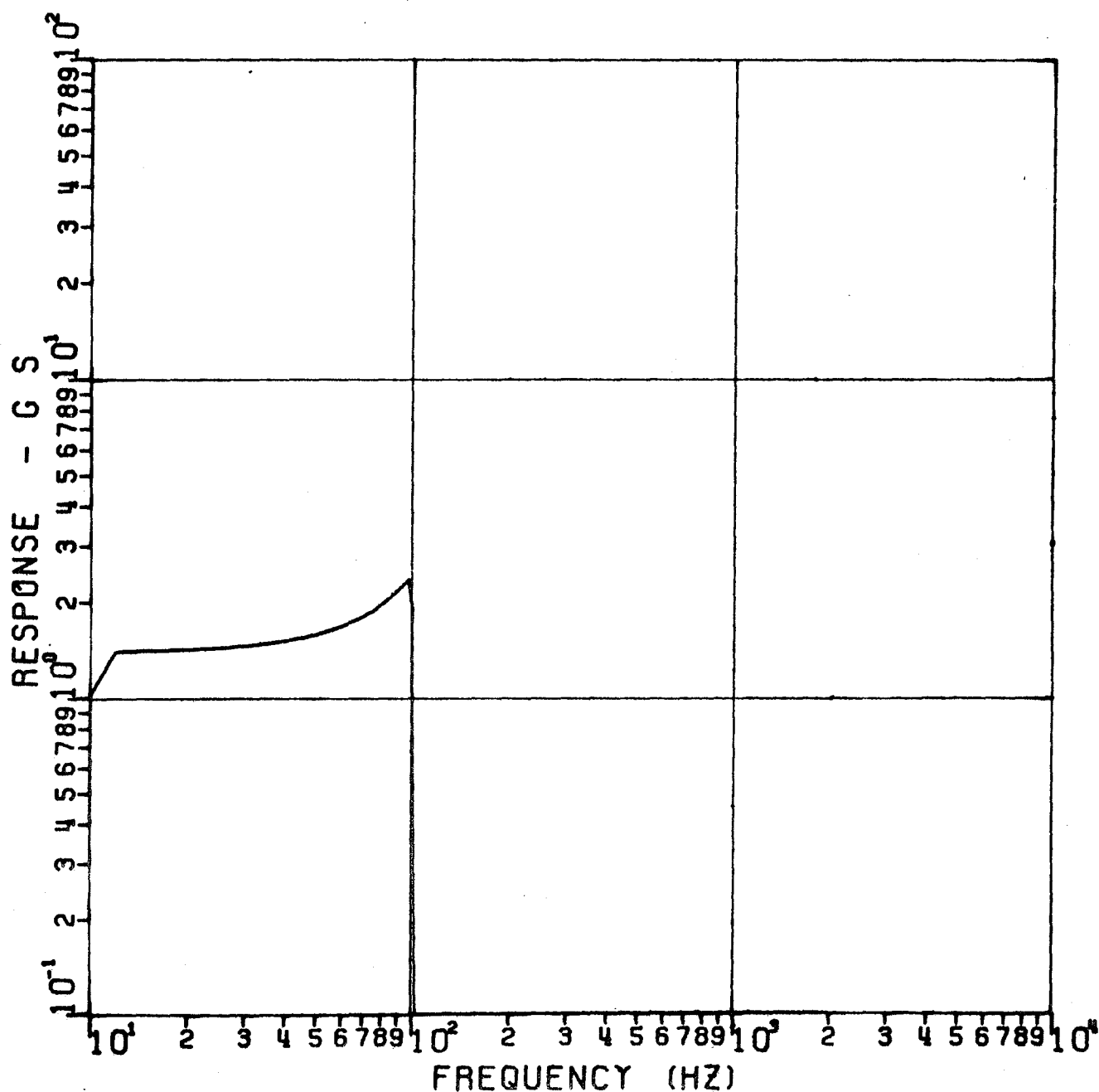
LOCATION 22



LRRR 300 IN THE OFF LOADED CONDITION 23 D.F

FIGURE 33.5 B SINE RESPONSE

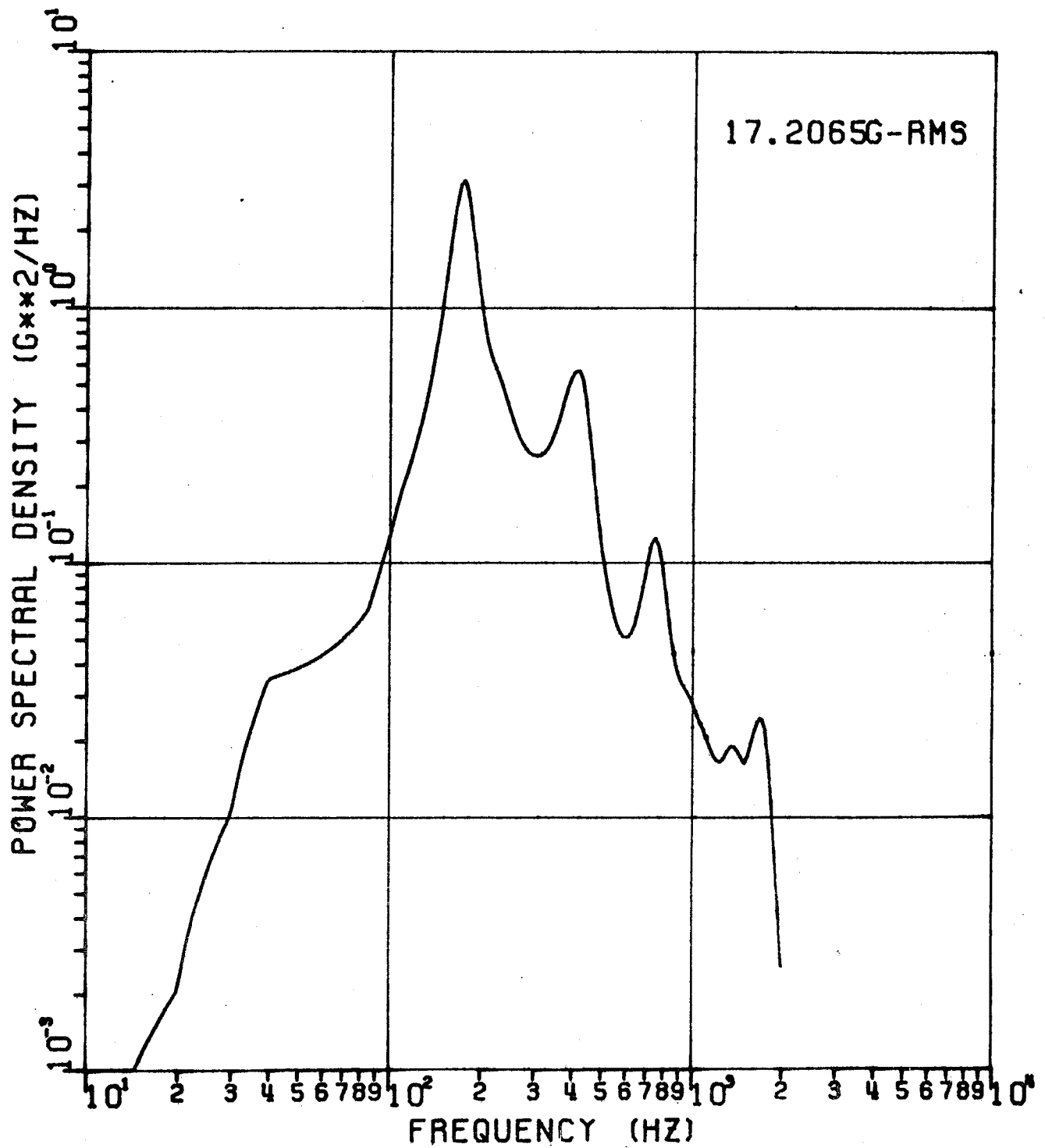
LOCATION 22



LRRR 300 IN THE OFF LOADED CONDITION 23 D.F

FIGURE 3.3.5C RANDOM VIBRATION SPECTRUM

LOCATION 22





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The K matrix is 38×38 . As explained in Section 2.1 the K matrix is reduced in size to correspond to the selected 30 coordinates by the transformation

$$K_s = [\beta]^T [K] [\beta]$$

K_s is shown in Figure 2.1.8.

$[\beta]$ = The coordinate transformation matrix as previously defined

The internal loads on the structure and associated displacements can be determined from the following relations:

$$\{P\} = [K] [\beta] [K_s]^{-1} \{Q\} \quad (2)$$

$$\{S\} = [\beta] [K_s]^{-1} \{Q\} \quad (3)$$

Where $\{P\}$ is the internal load vector, $\{S\}$ is the displacement vector

The next step is to compute the load vectors, $\{Q\}$, resulting from the dynamic environment.

4.2 Load Vectors

The expression for the load vectors, Q , for the random environment is

$$[Q]^T = [M] [A]$$

$[M]$ = Mass matrix (Figure 2.1.5)

$[A]$ = A diagonal matrix composed of local rms accelerations as previously computed



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Random Loads

When the $\{Q\}$ column vectors have been determined, they are substituted into equations (2) and (3) to compute the internal loads and displacements on the arrays. For the random case the rms internal loads and displacements are computed for each mode giving 30 sets of rms values at the 38 coordinate locations. Then for each location the 30 modal rms values are combined by taking the square root of the sum of the squares of the 30 values, thus reducing the number to 30 load and 38 displacement values.

Sine Loading

The load vectors, $\{Q\}$, are again formed by equation (4), but real and imaginary steps are computed separately and combined into one load vector for each frequency where a relative maximum sine response at any location occurs.

For the sinusoidal environment case, there is only one $\{Q\}$ vector per frequency selected and each one will produce a set of internal loads and displacements.

4.3 Discussion

Application of these procedures to the out-of-plane model results in load and deflection sets due to the random excitation shown in Table 4.3.1. Loads due to sinusoidal environment are all less than the 3-sigma random values obtained by multiplying the Table loads by 3, and are therefore not shown.

To calculate reactions at the LM interface fittings, the dynamic loads at each mass location on the lower array were applied to the "rigid grid" model of the structure as a static load case. The reactions at the fittings were calculated by the program. The largest reactions were obtained for the random excitation case. These reactions are listed in Table 4.3.2.

Table 4.3.1

ROOT MEAN SQUARE--COMBINED LOADS

DEFLECTIONS

INTERNAL LOADS

1	C.27607160-C3	0.25247300 02
2	0.11935460-C2	0.15922000 02
3	0.51153250-C3	0.81521040 02
4	C.40565560-C3	0.30545710 02
5	C.36631860-C3	0.16666230 02
6	C.74393230-C3	0.22411680 02
7	C.68735420-C3	0.18428850 02
8	C.67206700-C3	0.17476890 02
9	C.86627740-C3	0.13266280 02
10	0.10713320-02	0.31106310 02
11	0.11128810-C2	0.18938630 02
12	0.99473290-C3	0.98845470 01
13	0.11166530-02	0.21616090 02
14	C.16376150-02	0.20797030 02
15	C.12043430-02	0.15917130 02
16	0.41454650-C3	0.82857100 02
17	C.15827920-02	0.29761560 02
18	C.20082250-C2	0.18552850 02
19	C.44120680-03	0.85848210 01
20	0.14225380-02	0.17051580 02
21	0.25624080-C2	0.96221020 02
22	0.14264170-02	0.22036280 02
23	0.18765040-C2	0.11307160 02
24	C.33962230-02	0.25289120 02
25	0.15959940-C2	0.35556460 02
26	C.33653600-C2	0.28279210 02
27	C.36216110-C2	0.32072380 02
28	0.18990190-C2	0.28391610 02
29	C.32361170-C2	0.23075630 02
30	0.15903350-C2	0.16250340 02
31	C.18582240-02	0.11307160 02
32	C.32163980-02	0.25289120 02
33	0.17558460-02	0.35556460 02
34	0.32864940-C2	0.28279210 02
35	0.31052410-C2	0.32072380 02
36	C.16659400-02	0.28391610 02
37	C.30076760-C2	0.23075630 02
38	0.15773470-02	0.16250340 02

TABLE 4.3.2
VERTICAL REACTIONS
AT LM INTERFACE FITTINGS

Fitting	RMS Load (lb)	Peak Load (lb)
Rt. Front	142	426
Rt. Rear	156	468
L. Front	226	678
L. Rear	144	432