

Aerospace

tems Division

Thermal Vacuum Contingency Plan

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ACCEPTANCE TESTING

THERMAL VACUUM CONTINGENCY PLAN

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QUAL SA, FLIGHT 1 & 2

Prepared by: Duthat

Approved by: <u>L. Lewis</u> L. Lewis



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1.0 THERMAL VACUUM WATCHKEEPING GUIDELINES

- 1. The watchkeeping mode of operation is that which occurs after initial ALSEP turn on and between functional tests while achieving thermal soaking or transitions. This specifically excludes RTG start up.
- 2. The recommended condition of the ALSEP during these periods is:

Central Station Timer - OFF Central Station - ON DATA RATE - NORMAL TRANSMITTER A Data Processor Y ADDRESS - A CENTRAL STATION BACK UP HEATERS - ON **PCU #1** DUMP LOAD #1 - OFF DUMP LOAD #2 - OFF EXPERIMENT #1 (PSE) - ON EXPERIMENT #2 (LSM) - ON EXPERIMENT #3 (SWS) - ON* EXPERIMENT #4 (SIDE) - ON* EXPERIMENT #5 (COMMANDABLE HEATERS) - STANDBY OFF** **PSE STATUS - PRESET** LSM STATUS - PRESET ***SIDE STATUS - PRESET** SIDE -3.5 Kv - OFF SIDE 4.5 Kv - OFF SWS -- low voltage range

3. The recommended program loading for these periods is:

Executive Operating System - 2335245 CMALT Thermal Vacuum - 2335240 LSM Temperature MONITOR - 2335233 SWS Temperature MONITOR - 2335243 SIDE Temperature MONITOR - 2335205 A1-A2 Integrated Decommutation - 2335210

*****SWS and SIDE will be turned on during lunar morning IST until that time they will be in standby.

******Except for the lunar night calibration; Exp #5 shall be in the Power On Mode.

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- 4. Manual Plotting is required on several parameters in order to maintain a constant and readily accessible record useful in determining trends. Graphical forms are provided for this purpose. The parameters to be plotted are shown in the attached table. This plotting is to be done at intervals indicated in the table. Note that these intervals vary depending on whether active testing or watchkeeping mode is in effect.
- 5. Emergency guidelines are provided to give "first aid" direction to test personnel whenever OT conditions occur. It is important to keep in mind that on a system as complex as ALSEP, the most obvious emergency procedure (e.g. turn-off) is not always correct or even safe. For this reason the Emergency Guidelines are given to be used whenever an OT condition occurs on the STS printout.
- Reference should be made to CMALT listing and ATM-704 for the values of normal limits (contained in programs) and danger limits.
- 2.0 STRUCTURAL/THERMAL TEMPERATURES

The following are actions to be taken if these temperatures exceed their operating limits and are approaching or passed their danger limits.

Corrective Action

Channel
Assignment

- 2.1 Sunshield Temperatures
 - 1) Contact engineering
 - 2) Verify temperature OT with DAS Temp. = HK #
 to confirm data. #1 = 27
 #2 = 42
 - 3) Verify thermal plate temperatures are within limits.
 - 4) Verify external heat sources are functioning properly.
 - 5) Continue to monitor thermal plate temperatures to insure they are not approaching limits if 3) and 4) are normal.





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Corrective Action

Thermal Plate Temperature

are not approaching limits.

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1) Contact engineering 2) Verify temperature OT with DAS #1 = 4for confirmation of data. #2 = 28#3 = 43#4 = 58 3) If 1) verifies and temperature is too #5 = 71low, turn on commandable heater Exp #5 if too high, turn off commandable heater. Reduce or increase external heat sources 4) to obtain a within tolerance output. 5) If unable to control temperature, hold meeting with engineering, NASA prior to turn off of ALSEP. 2.3 Vertical Structures, Bottom Structure, Outer Multilayer Insulation 1) Contact engineering. Verify temperature OT with DAS 2) Temp. #1 = 59 for confirmation. #2 = 87 #3 = 15#4 = 88Outer multilayer = 723) Verify thermal plate temperatures are within limits. 4) Verify external heat sources are functioning properly. 5) If 3) and 4) are normal, continue to monitor thermal plate temperatures to insure they

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			Co	rrective Action		Channel A	ssignment		
	2.4	Inne	er M	ultilayer Insulation Temperatu	ure				
		1)	Co	stact engineering.					
		2)	Ve	ify temperatures are within li	imits.	Inner Mu	ltilayer = 60)	
		3)		ify external heat sources are ctioning correctly.					
		4)	Cer) and 3) are normal, monitor tral Station to insure tempera not approaching critical limit					
3.0	ELE	CTRC	ONIC	TEMPERATURES					
	3.1	Rec	eive	r Local Oscillator					
\bigcirc		1)	HK · leve	ck Receiver Pre-Limiting leve 21, Receiver Local Oscillator 21 HK-36. Determine if a tren utput exists.	r	Local Osc. Local Osc.			
		2)		ermine if switchover from A to occurred i.e. is temperature		?			
		3)	rect com	ermine if uplink is functioning ly by sending the PSE filter in mand twice, check for proper tout.	1/out				
		-	Noti noti:	olink is functioning, continue to fy Engineering. If uplink has by appropriate Test Conductor ineering to determine appropri	failed and	-			
:	3.2	Tran	nsmi	tter, Heat Sink and Crystal Te	emper	ature			
		1)	Cont	act engineering					
				fy thermal plate temperatures in limits.	s are	A Crystal = B Crystal = A Heat Sink B Heat Sink	31 = 19		



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- 3) Switch transmitters and verify a within tolerance output from the alternate transmitter.
- 4) Monitor temperatures of alternate transmitter to determine trend.

3.3 Analog and Digital Processor Temperatures

- 1) Verify thermal plate temperatures are within limits.
- Analog Base = 33Analog Internal = 34 Digital Base = 46Digital Internal = 47
- 2) Switch data processors and determine temperature trend.
- 3) If Step 2 fails to control temperature, verify correct operation of processors from the printout of analog channels (PSE) Status, etc.)
- Contact Engineering if temperature cannot 4) be controlled or processor does not operate correctly.

3.4 **Command Decoder Temperatures**

1) Verify Thermal Plate temperatures are Base = 48 within limits. Internal = 49

Mod. VCO = 61

2) Test uplink by sending PSE filters in/out command twice, verify correct CV printout.



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Corrective Action

Channel Assignment

 3) If there is no response to commands, switch to appropriate alternate address (1A to 1B etc.), retest uplink and determine temperature trend.

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4) Contact appropriate test conductor and Engineering if temperature cannot be controlled.

3.5 Power Distribution Temperatures

- Verify thermal plate temperatures Base = 62 are within limits. Internal = 63
- 2) Contact appropriate test conductor and Engineering.

3.6 PCU, Power Oscillator and Regulator Temperatures

1}	Verify thermal plate temperatures	Osc. $#1 = 64$
	are within limits.	Osc. #2 = 74
		Reg. $#1 = 77$
		Reg. $#2 = 78$
21		-

- 2) Switch over PCU's monitor temperature to determine trend.
- 3) If unable to control temperature, contact appropriate Test conductor and Engineering.

4.0 CENTRAL STATION ELECTRICAL

4.1 Analog Converter Calibration

- 1) Contact Engineering.
- Verify PCU output voltages are within ADC Cal 0.25 = 2 tolerance. ADC Cal 4.75 = 3
- 3) Switch processors.

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			Corrective Action	C <u>H</u> ar	nnel Assignr	nent		
	4.2	PC	U Input - Output Currents and Voltages					
		1)	Contact engineering					
		2)	Verify Central Station is operating within regulation via Reserve Power Indication on the analog recorder.	-		-		
		3)	Verify station status has not changed i.e. dump loads on, should be off, etc.	-	V = 20	p13		
		4)	Verify correct output of RTG via IPU Test Console or RTG Simulator.	+12V = 50 +5V = 65 -12V = 79 -6V = 80				
		5)	If Steps 2-4 are normal, switch over PCU's.					
		6)	If unable to obtain correct output Test Conductor and Engineering will determine appropriate action.					
	4.3	Rec	eiver, Prelimiting and Local Oscillation L	evel				
		1)	Contact engineering					
		2)	Verify STS transmitter output is within limits.		limit = 21). = 36			
		3)	Determine if uplink is functioning cor- rectly by sending <u>PSE filter</u> in/out twice.					
		4)	If uplink is operative continue test.					
		5)	If uplink fails contact appropriate Test Conductor and Engineering.					
	4.4	Trai	ansmitter AGC Voltage, Power Doubler Current					
		1)	Switch transmitters.		CA = 51			
		2)	If unable to obtain correct limits, Verify downlink is functioning via STS printer.	Pow	CB = 66 er Doub. A er Doub. B			

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3)	Contact appropriate Test Conductor and Engineering.			
4.5 <u>Re</u>	ceiver, 1 KHz Subcarrier			
1)	Verify if uplink is functioning by sending PSE filter in/out command twice	1 K	Hz Sub = 9	
2)	If uplink has failed, check STS transmitter and cables to chamber.			
3)	If unable to rectify problem, con- tact appropriate Test Conductor and Engineering.			
4.6 <u>AI</u>	SEP Experiment Power Distribution:	-	• #1 & 2 =	
1)	Verify correct status of experiment via STS Printer or by command switching and PCU shunt regulator current.	тхр	. #3, 4, &	5 = 14
2)	Contact Engineering for an out- of-tolerance condition.			
5.0 RTG TE	MPERATURES*			
1)	Confirm data by comparing with external RTG temperature measure- ments.	Hot Hot	Fr. $#1 = 6$ Fr. $#2 = 37$ Fr. $#3 = 52$ d Fr. $#1 = 7$	2
2)	If temperatures exceed 1130°F max. for the hot frame and 575°F max. for the cold frame switch the load to the RTG Simulator and begin emer- gency turn off procedure for the RTG.	Cold	1 Fr. #2 = 6 1 Fr. #3 = 8	57

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3)	Contact the appropriate Test Conductor and PE to determine further action.
6.0 DUST	ACCRETION
1)	Contact Engineering.
2)	Command Dust Detector off if meaningful data is not being obtained.
7.0 PASSIV	E SEISMIC EXPERIMENT
1)	Send commands to obtain cor- rect response.
2)	Contact Engineering.
3)	If thermal status exceeds critical limit, determine temperature trend of PSE sensor. Advise appropriate Test Conductor and PE if temperature exceeds limit.
4)	If uncage status indicates uncaging, verify by a pressure measurement via PSE exciter. Nofify PSE PE immediately and appropriate Test Conductor.
8.0 SUPRAT	THERMAL ION DETECTOR EXPERIMENT
1)	Verify a shift in the "normal" output of the science data. SIDE words 4-5 and 9 - 10, has occurred.

NOTE: The HK is a logrithmic conversion of the science data rate.

SIDE Analog Channels LEDCR = HK-70 HEDCR = HK-85

Channel Assignment

Cell Temp. #1 = 83 Cell Temp. #2 = 30 Cell Temp. #3 = 56 Cell Volt.#1 = 84 Cell Volt.#2 = 26 Cell Volt.#3 = 41

LP AMP GAIN X, Y = 23LP AMP GAIN Z = 38 LEV. DIR. & SP. = 53 SP. AMP. GAIN = 68 LEV MOD & COARSE =24 THERMAL CONT. = 39 CAL LP & SP = 54 UNCAGE STATUS = 69

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2) If the science data appears abnormal switch the experiment to Standby. Contact the appropriate Test Conductor and the PE.

9.0 FAILURE OF DOWNLINK (LOSS OF MF LOCK)

- 1) Verify or correct if necessary phase locking of STS receiver.
- 2) Switch transmitters and return.
- 3) Switch data processors.
- 4) Reload DPS 2000 programs.
- 5) Verify proper NRZ and clock from STS demodulator to DPS 2000 with a scope.
- 6) Load a second STS with identical software and play back a portion of the Ampex magnetic tape just recorded.
- 7) Contact Systems Engineering before stopping test.

10.0 FAILURE OF UPLINK (TOTAL)

- 1) Switch Addresses
- 2) Check STS transmitter power output and cables to chamber (RF attenuation)
- 3) Check STS modulation (1000 cps attenuation, data attenuation, 1000 cps phase)
- 4) Contact Engineering.

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Channel Assignment

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11.0 FAILUI	RE OF UPLINK (PARTIAL)			
1)	Contact Engineering.			
2)	Perform steps of 9.0 and test failed commands.			
3)	Modify procedures to omit affected commands and continue test.			
12.0 LUNAR	SURFACE MAGNETOMETER (LSM)			
12.1 <u>Ana</u>	alog Engineering Data (Word 5) Temperatur	re		
1)	Contact Systems Engineering, and PE.	Temp. $#1 = 1,9$		
2)	Verify temperature OT with DAS to confirm data.	#2 = 2, 10 #3 = 3, 11 #4 = 4, 12 #5 = 5, 12		
3)	Verify external heat sources are functioning correctly.	#5 = 5,13		
4)	Reduce or increase external heat sources to return to within tolerance.			
12.2 Lev	vel Sensors			
1)	Verify by visual inspection that LSM is still in an upright condition.	L.S. #1 = 6,14 #2 = 7,15		
2)	If tipped, contact System Engineering and PE. Monitor temperatures; if critical limits approached bring the lunar surface under LSM to room temperature as quickly as possible.			
12.3 Sup	ply Voltage			
1)	Determine trend to see if danger limits have been reached or are being ap- proached.	SV = 8,16		

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2)	If at, or approaching danger limit, command LSM to standby and im- mediately institute action to keep LSM within safe temperatures (as sensed via DAS) by using the chamber environmental controls.					
3)	Contact Systems Engineering and PE.					
12.4 Sta	tus Bits (Change Only)					
1)	Note occurrence, include time, old and new values.	A11 1	Fram	nes		
2)	Restore original status by command(s).					
3)	Contact Systems Engineering and PE.					
13.0 SUPRA	THERMAL ION DETECTOR EXPERIMENTS	(SID	E)			
13.1 <u>Ter</u>	nperatures					
1)	Verify temperature OT with DAS to confirm data.	SIDE Word		SIDE Frame		
2)	Verify external heat sources are functioning correctly.	#2 =	2	2,34,6 4,36,68 6,38,7	8,100	
3)	Reduce or increase external heat to return within tolerance.		2	11,43,	-	
4)	Contact Systems Engineering and PE.		-			
cou	h Voltages (any indication of arcing such as nter skiping in its sequence or noisy science E WORDS 4,5 and 9,10)	-				
1)	Command off the OT high voltage.					
2)	Contact Systems Engineering and PE.					
14.0 SOLAR W	IND SPECTROMETER (SWS)					

14.1 Temperatures

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	1)	Verify temperature with environmental DAS to confirm data.	Temp. M	fod 100 200 300	2	
	2)	Verify external heat sources are functioning correctly.		Cups		
	3)	Reduce or increase external heat sources to return within tolerance.				
	4)	Contact Systems Engineering and PE.				
14.	2 <u>Hig</u>	h Voltage Arching On Turn ON				
	1)	Monitor reserve power profile at initial turn on. If any indication of not following regular sequence is seen Turn SWS to standby.				
and the second s	2)	Allow SWS to outgas further prior to turn	on.			
14.3	B Hig	h Voltage Arcing During Operation (erratic	science da	ita)		
	1)	If any indication of arcing during operatio is seen immediately turn SWS to standby.	n			
	2)	Contact Systems Engineering and PE.				
14.4	sws	Loss of Lock				
	1)	If SWS fails to regain lock after 75 second Switch Exp. #3 to standby ON. Analyze da determine if SWE was sequence correctly via the Reserve Power Monitor. Contact I	ata to			

Manufacture