

DEVELOPMENT OF STANDARDIZED LUNAR REGOLITH SIMULANT MATERIALS. P. Carpenter¹, L. Sibille¹, and S. Wilson², ¹BAE Systems, Marshall Space Flight Center, AL 35812, paul.k.carpenter@nasa.gov, ²United States Geological Survey, MS964, Lakewood CO 80025.

Introduction: Lunar exploration requires scientific and engineering studies using standardized testing procedures that ultimately support flight certification of technologies and hardware. It is necessary to anticipate the range of source materials and environmental constraints that are expected on the Moon and Mars, and to evaluate in-situ resource utilization (ISRU) coupled with testing and development. Physical properties of the regolith dominate processes such as excavation and drilling, while chemical properties dominate processes such as elemental extraction. We describe here the development of standardized lunar regolith simulant (SLRS) materials that are traceable inter-laboratory standards for testing and technology development. These SLRS materials must simulate the lunar regolith in terms of physical, chemical, and mineralogical properties. A comprehensive simulant development program has been outlined and is in progress. A summary of these issues is contained in the 2005 Workshop on Lunar Regolith Simulant Materials [1].

Lunar Regolith Simulants: The historical use of lunar simulants focused on physical aspects of the lunar regolith for landing and transportation activities, as extended missions and ISRU activities were not under consideration. Lunar mare basalt simulants MLS-1 and JSC-1 were developed in the late 1980s. MLS-1 approximates an Apollo 11 high-Ti basalt, and was produced by milling of a holocrystalline, coarse-grained intrusive gabbro (Fig. 1). JSC-1 approximates an Apollo 14 basalt with a relatively low-Ti content, and was obtained from a glassy volcanic ash (Fig. 2). Supplies of MLS-1 and JSC-1 have been exhausted and these materials are no longer available. No highland anorthosite simulant was previously developed. Support for upcoming lunar polar missions thus requires identification, assessment, and development of both mare and highland simulants.

Root and Derivative Lunar Simulants: A lunar regolith simulant is manufactured from terrestrial components for the purpose of simulating the physical and chemical properties of the lunar regolith. Significant challenges exist in the identification of appropriate terrestrial source materials. Lunar materials formed under comparatively reducing conditions in the absence of water, and were modified by meteorite impact events. Terrestrial materials typically formed under more oxidizing conditions with significantly greater access to water, and were modified by a wide range of weathering processes. A *root* simulant represents an end-member in terms of simulant properties such as

bulk chemistry, and a *derivative* simulant is formed from a root by modification or addition of material [2]. The *fidelity* of the simulant represents the degree of duplication of soil characteristics compared to lunar soils.

The composition space of lunar materials can be modeled by mixing programs utilizing a low-Ti basalt, ilmenite, KREEP component, high-Ca anorthosite, and meteoritic components. This approach has been used for genetic studies of lunar samples via chemical and

that must be retained in order to match the target lunar regolith. As the sample size is reduced, at some point it is no longer representative. The deviation in properties can be monitored by chemical analysis using progressively smaller sample sizes, in comparison with analyses performed on bulk material. This problem has been observed in analyses of MLS-1, where a ~10% variation in SiO₂ is observed, compared to a 160% variation for Cr [1]. This chemical contrast illustrates small differences in silicate minerals for the major element Si, but large differences in minerals for the trace element Cr since it is concentrated in chromite. Similarly, geotechnical properties may be dominated by a large difference in mineral hardness, and rogue grains such as quartz could stand out in tests using too small a quantity of material. The variability of simulant material thus is an inherent property but must be taken into account for both quality control and for simulant use by the scientific community.

Simulant Production and Quality Control: The workshop recommended redeployment of JSC-1 to serve as a testing material for immediate needs while root simulants are developed. Simulant JSC-1A, and a dust size fraction of JSC-1A (for studies on human toxicology and mechanical abrasion) are being produced under the direction of Marshall Space Flight Center (MSFC).

Root simulant production requires: (1) establishment of simulant requirements, based on analysis of lunar regolith properties and planned activities on the lunar surface; (2) screening and selection of appropriate terrestrial source materials; (3) milling and other processing necessary to produce the required physical attributes such as grain size distribution; and (4) characterization of chemistry, mineralogy, and geotechnical properties in order to monitor the adherence to simulant requirements and homogeneity during production. Simulant must be characterized during critical stages of production and packaging, and this quality control includes traceability using geochemical reference standards. A supplied certificate provides compositional data and other information specific to the SLRS material for the end user. Development of derivative simulants will include synthetic agglutinate production using plasma fusion melting to produce glassy microspheres similar in bulk composition to the root simulant. These synthetic agglutinates may also be processed to contain nanophase iron inclusions and vesicles in order to increase the fidelity of the simulant. Ion implantation can be utilized to simulate the exposure to the solar wind.

Fig. 1. Backscattered-electron (BSE) image of lunar simulant MLS-1 (polished mount, electron-probe microanalysis) illustrating coarse grain size, and absence of glass. MLS-1 has monomineralic grains of the same size as lithic fragments in JSC-1. Compared to Apollo 11 soils, MLS-1 represents a good match for composition, but a poor match for texture.

Fig. 2. BSE image of lunar simulant JSC-1 (polished mount, electron-probe microanalysis). Grain size distribution of JSC-1 is similar to Apollo soils. Glass content and fine grain size of JSC-1 is desirable in a mare lunar simulant, but no equivalent of lunar agglutinates is present and must be added to form a derivative simulant.

References: [1] Sibille L. et al., (2005) *Lunar Regolith Simulant Materials: Recommendations for Standardization, Production, and Usage*. NASA TP (*In preparation*, <http://est.msfc.nasa.gov/workshops/lrsm2005.html>). [2] Carter J., et al. (2004) Space Resources Roundtable VI, 15. [3] Taylor L. et al. (2004) Space Resources Roundtable VI, 46.