

Impact-generated signals can be estimated through impact rate modeling, integrating both the impactor flux [5], atmospheric ablation effects [6] and seismic calibration on the Moon [7]. Modeling predicts that about 20 events with a Signal to Noise Ratio > 3 (with respect to the requirement instrument noise) should be detected during the nominal mission of InSight, including 5 with $SNR > 9$ (Figure 4).

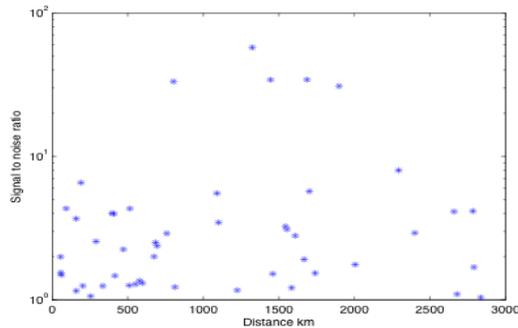


Figure 4 Modeling results of impact SNR with a Monte-Carlo simulation of impactors and seismic amplitude impulse/distance estimates calibrated on the Moon and corrected with the a priori attenuation differences between Mars and the Moon. Large events correspond to impacts of about 1 ton. The InSight L1 requirement is for the detection of an impact of 10^6 Ns impulse at distances less than 400 km with a $SNR > 3$.

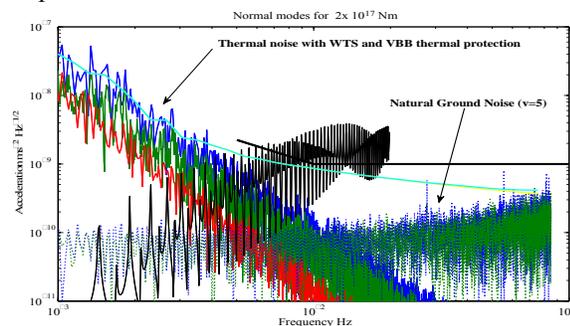


Figure 5: Amplitude of a 3×10^{17} Nm marsquake free oscillation signal compared to instrument noise (black curve is SEIS requirement, cyan is expected capability) and environment noise (left, temperature, right, pressure, for day, sol, night in blue, green, red respectively). One or two such quakes are expected to occur during the two years of InSight operation and will provide a seismic “grail” of information, comparable to those used on Earth for elaboration of the standard PREM model [9]. Temperature noise and ground pressure noise are shown, the latter for a non-consolidated subsurface with 1 km/s V_p and 500 m/s V_s . The eigenmode frequencies will strongly constrain the lithosphere mean shear velocity.

Environmental noise modeling: Thermal and ground pressure noise estimates have been performed

using Pathfinder temperature and pressure data together with the performance of the WTS and VBB Sphere thermal protection. In addition, natural pressure-induced ground acceleration has been modeled by using ground static loading theory [8]. Modeling results for normal modes and surface waves are shown in Figure 5.

Mantle and crust seismic inversion: As only one seismic station is available, structure inversion will be performed using:

- Secondary seismic data which do not depend on the event location: e.g., free oscillation frequencies for the largest quakes constraining the interior down to 200 km and receiver functions constraining the crust-mantle discontinuity below the landing site (see [10] for lunar data);
- Seismic impact data from impacts post-located by a Mars orbiter [11];
- Seismic data associated with events with more than 3 different wave arrival time determinations (for V_s inversion with constant V_p/V_s) or more than 4 (for full V_p, V_s inversions).

Seismic activity levels and wave amplitudes have been used to estimate the number of events with multiple arrivals. We estimate that about 35 events will be detected with both P and S waves, and about 10 with P, S and R1 surface waves and core phases (e.g., PcP, ScP). For about half of the latter, the R2 surface wave will be also be detected, enabling an epicentral distance determination contaminated only by lateral variability, which can be corrected with 3D modeling [12]. These events and associated seismic data set will allow the determination of seismic velocities down to 600 km to within ± 0.25 km/sec, enabling the first seismic model of another planet than Earth and exciting constrains in term of planetary formation and evolution.

References: [1] Anderson et al. (1977) *J. Geophys. Res.*, 82, 4524-4546; [2] Lognonné et al. (1998) *Planet. Space Sci.* 46, 739-747; [3] Phillips (1991) LPI Tech. Rept. 91-02, 35-38; [4] Golombek et al (1992) *Science* 258, 979-81; [5] Le Feuvre and Wieczorek (2011) *Icarus* 214, 1-20; [6] Lognonné et al. (1996) *Planet. Space Sci.* 44, 1237-1249; [6] Lognonné (2005) *Ann. Rev. Earth Planet. Sci.* 33 :19.1-19.34; [7] Lognonne et al, (2009) *J. Geophys. Res.* 114, E12003; [8] Sorrels et al. (1971) *Geophys. J. RAS* 26, 83-98; [9] Dziewonski and Anderson (1981) *PEPI* 25, 297-356; [10] Vinnik et al. (2001) *Geophys. Res. Lett.* 28, 3031-3034; [11] Daubar et al. (2011) *LPSC XXXXII*, abs. #1608, p.2232; [12] Larmat et al. (2008) *Icarus* 196, 78-89.