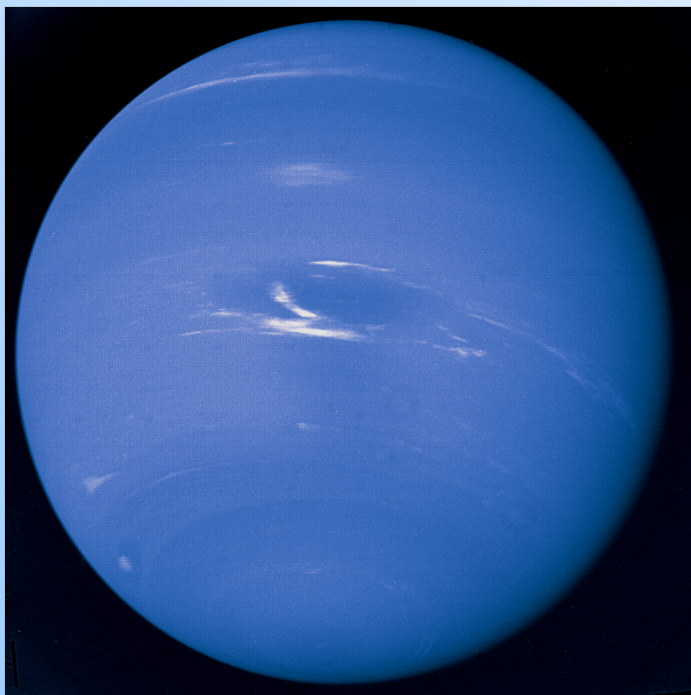




# Aerocapture Implementation of NASA's "Neptune Orbiter With Probes" Vision Mission

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Andrew P. Ingersoll

*California Institute of Technology*

Thomas R. Spilker

*Jet Propulsion Laboratory,  
California Institute of Technology*

Outer Planets Assessment Group (OPAG)

Boulder, Colorado

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# Technical Teams



## –Science Team

- Prof. Andrew P. Ingersoll, PI
  - California Institute of Technology
- Co-Investigators
  - ♦ Dr. Thomas R. Spilker, Study Lead
    - Jet Propulsion Laboratory
  - ♦ Dr. Heidi B. Hammel
    - Space Science Institute
  - ♦ Prof. William B. Hubbard
    - University of Arizona
  - ♦ Prof. Krishan K. Khurana
    - University of California, Los Angeles
  - ♦ Dr. Laurence A. Soderblom
    - US Geological Survey
  - ♦ Dr. William D. Smythe
    - Jet Propulsion Laboratory
  - ♦ Dr. Linda J. Spilker
    - Jet Propulsion Laboratory
  - ♦ Dr. Richard E. Young
    - NASA Ames Research Center

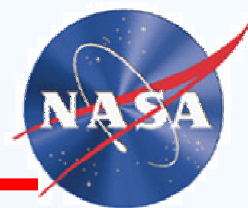
## –Implementation Team

- Dr. Thomas R. Spilker
  - ♦ Study Lead, Mission Architect; JPL
- Mr. Robert W. Bailey
  - ♦ Systems Engineer
  - ♦ Jet Propulsion Laboratory
- Dr. Robert J. Haw
  - ♦ Transfer trajectory design
  - ♦ Jet Propulsion Laboratory
- Dr. Mary K. Lockwood
  - ♦ Aerocapture systems
  - ♦ NASA Langley Research center
- Dr. Bernard Laub
  - ♦ Aerocapture thermal protection
  - ♦ NASA Ames Research Center
- Dr. Robert N. Miyake
  - ♦ Spacecraft thermal systems
  - ♦ Jet Propulsion Laboratory
- Mr. Nathan J. Strange
  - ♦ Tour trajectory design
  - ♦ Jet Propulsion Laboratory
- Dr. Ethiraj Venkatapathy
  - ♦ Entry probe thermal protection
  - ♦ NASA Ames Research Center



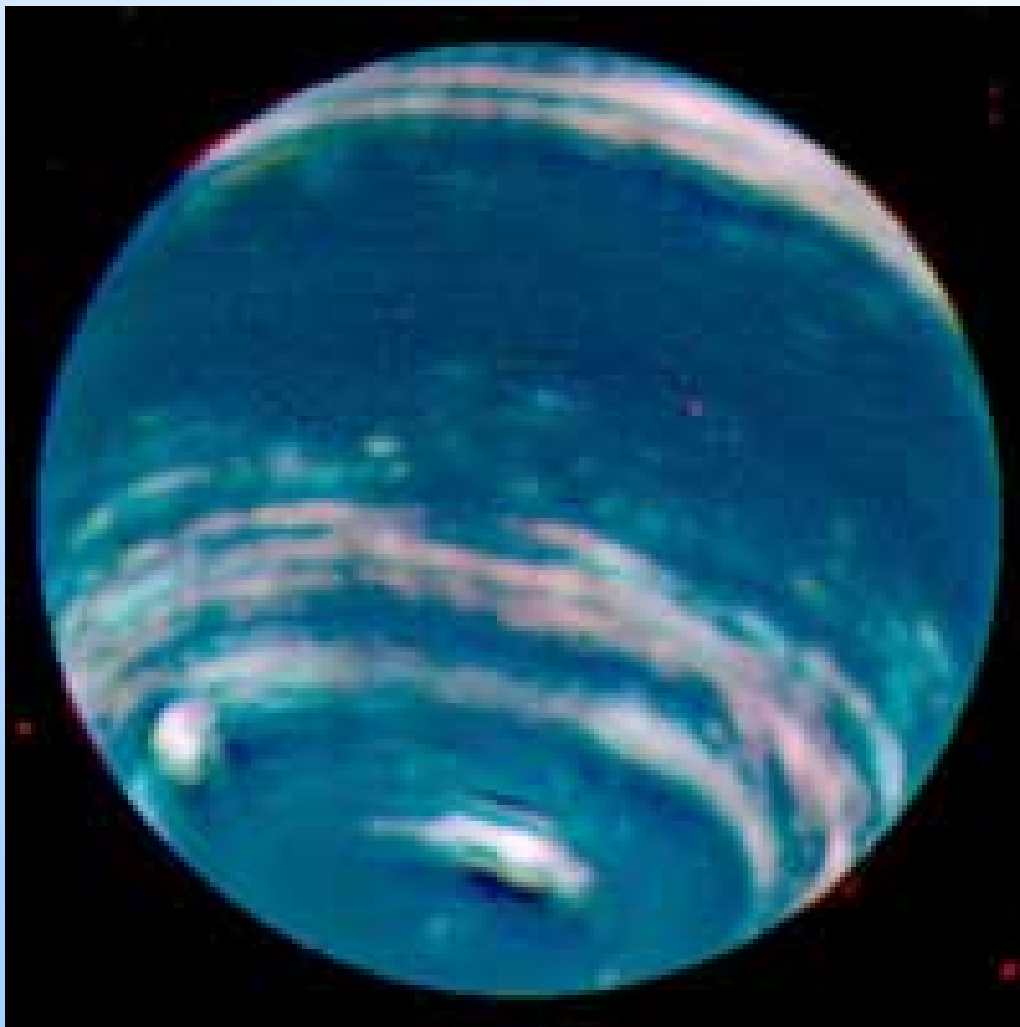
# Science Objectives: Why Study the Neptune System?

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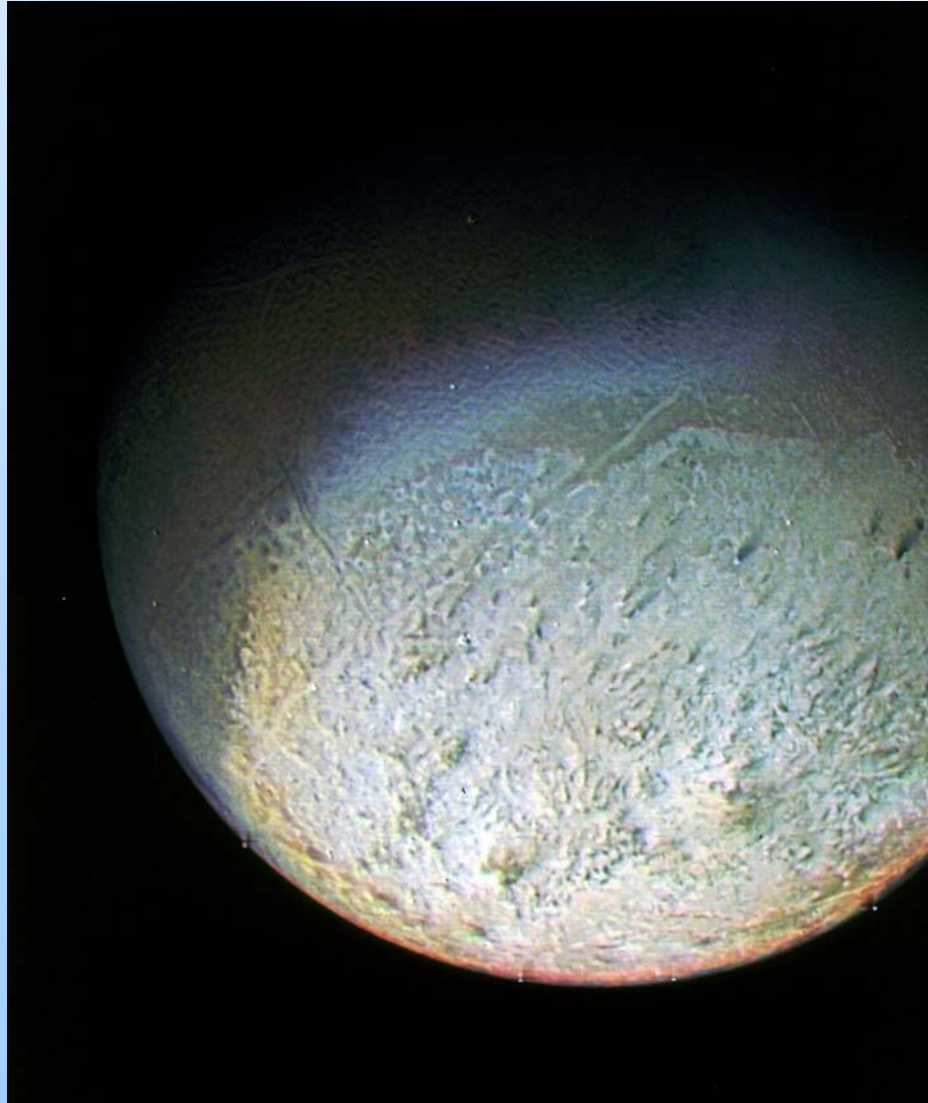


- Study the interior structure, composition, and atmospheric dynamics of an “ice giant,” with probes to 100 bars and orbiter remote sensing over 3-year orbital tour
- Investigate the surface, interior, and atmosphere of Triton, which may be a captured Kuiper Belt Object, with  $> 40$  flybys (altitude  $< 1000$  km) and possibly a Triton lander
- Observe the response of the magnetosphere to daily “pole on” orientation to the solar wind, with 3-year orbital tour
- Study Neptune’s unique ring arcs and associated moons over 3 year mission, with 3-year orbital tour

# Neptune in 2003: Keck Telescope

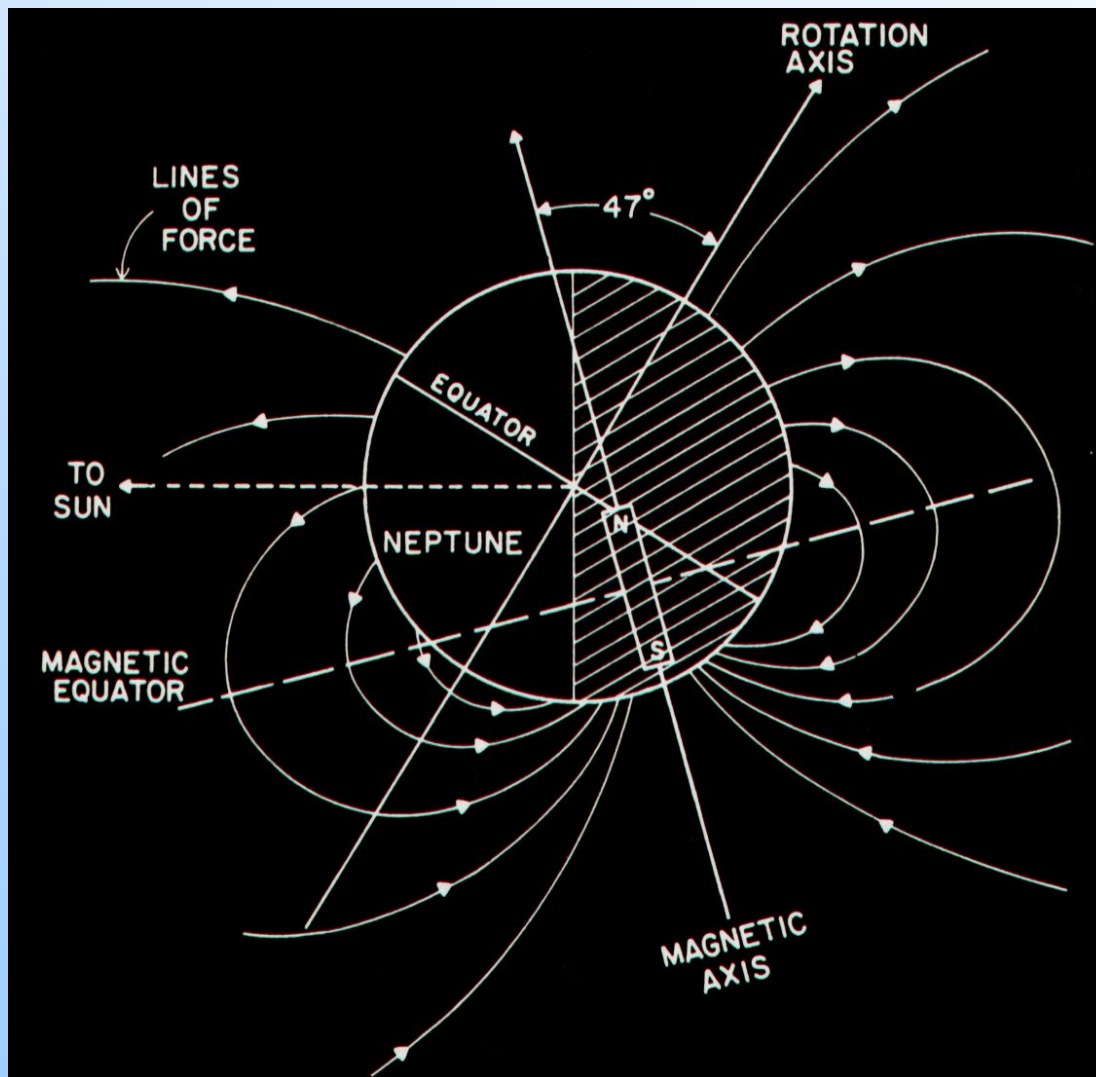


# Voyager 2 View of Triton

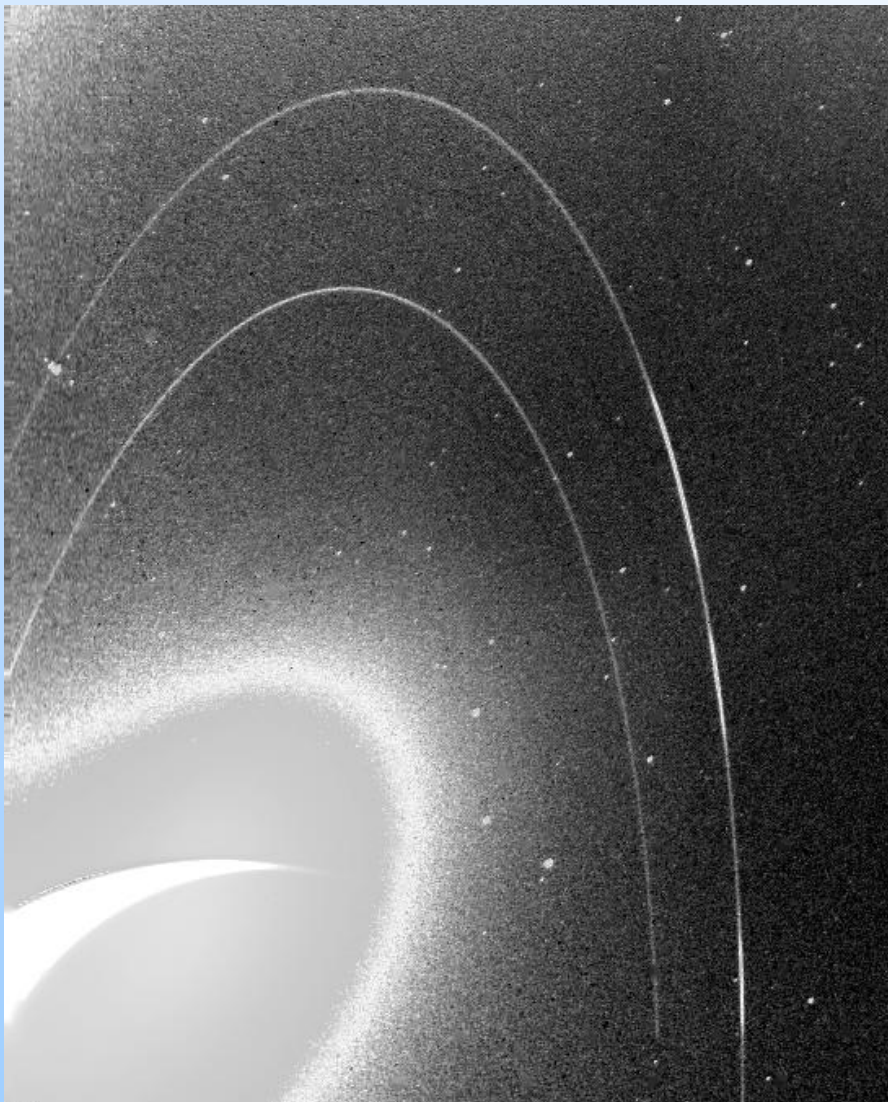




# Asymmetric Magnetic Field



# Rings and Ring Arcs



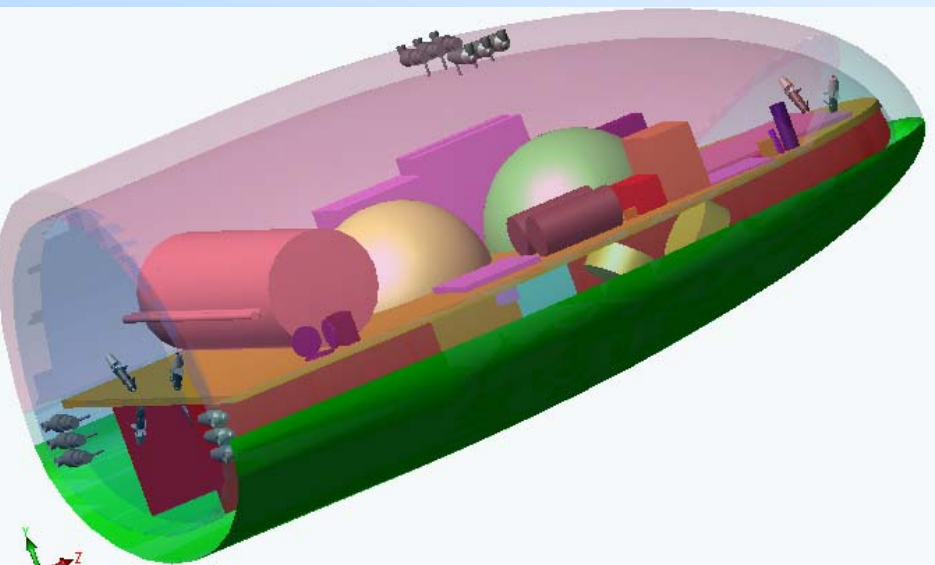


# Mission Architecture

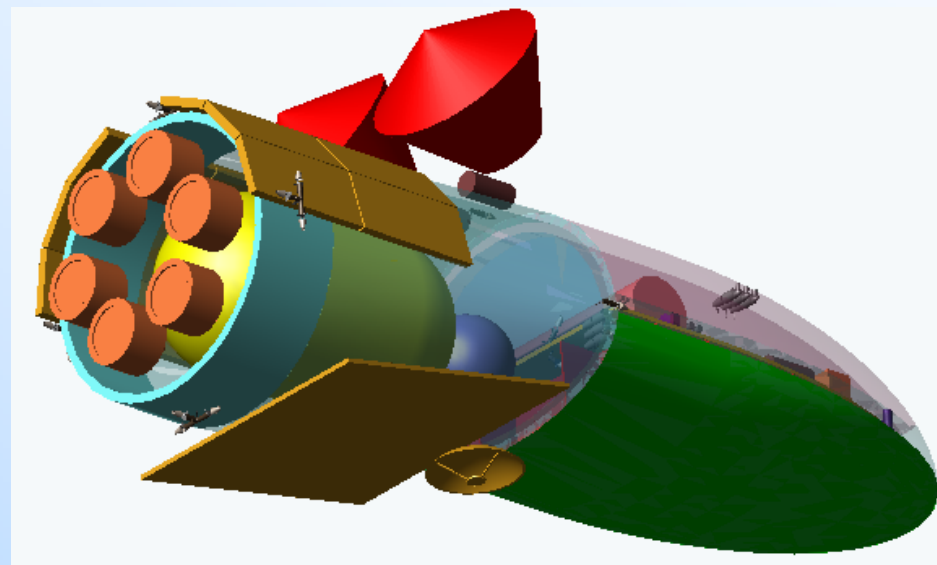


- Single launch with Delta IV-Heavy-class LV
- SEP or chemical propulsion, + gravity assists, to Jupiter; JGA to Neptune; total transfer time ~12 years or less
- Aerocapture insertion into Neptune orbit
  - Entry speeds from 23 to 32 km/s available
- At least 2 Neptune entry probes to at least 100 bar levels, deployed and supported (i.e., data relay) pre-aerocapture
- 3-year prime mission in Neptune orbit, powered by RTGs; extended mission possible
- At least 40 close (<1000 km) Triton flybys during prime mission for Triton science and orbit evolution; Triton lander is possible.
- Ka-band radio for primary downlink
  - 10 terabits data volume with Next Gen DSN





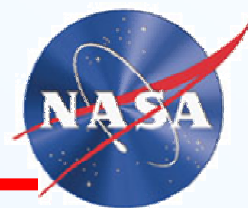
Aerocapture System  
Analysis Team orbiter  
and “Ellipsled” design



Ellipsled and SEP stage,  
with cruise HGA, entry  
probes, and probe data  
relay antenna



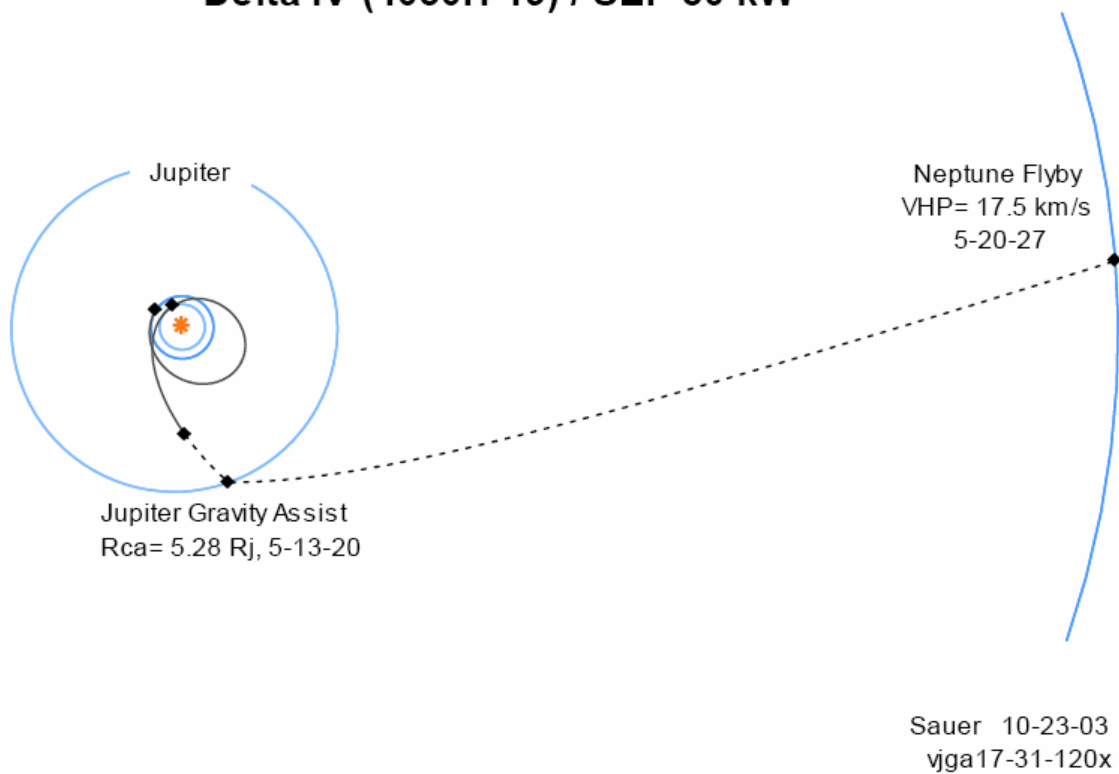
# Mission Design: Transfer to Neptune



- *Example* SEP launch option
  - Launch Feb 2017 on a Delta 4050H-19,  $C_3 = 12.5$
  - S/C launch mass, 6250 kg, yields 800 kg launch margin
  - Arr Aug 2029; 1900 kg (dry) delivered to Neptune orbit
- *Example* Chem launch option
  - Launch Dec 2015 on a Delta 4050H-19,  $C_3 = 28$
  - S/C launch mass, 4780 kg, yields 420 kg launch margin
  - Arr Aug 2029; 1900 kg (dry) delivered to Neptune orbit
- SEP system 30-40 kW, 5-6 NEXT thrusters
  - Dry mass slightly over 1300 kg
- Small (260 kg) carrier stage for chem option
  - 10 terabits data volume with Next Gen DSN

# Example Transfer to Neptune

## 10.25 Year VJGA Neptune Flyby Delta IV (4050H-19) / SEP 30 kW



- 10.25-year transfer uses SEP and a Venus gravity assist
- Entry speed 26.7 km/s prograde, 31.7 km/s retrograde



# Mission Design: Neptune System Tour

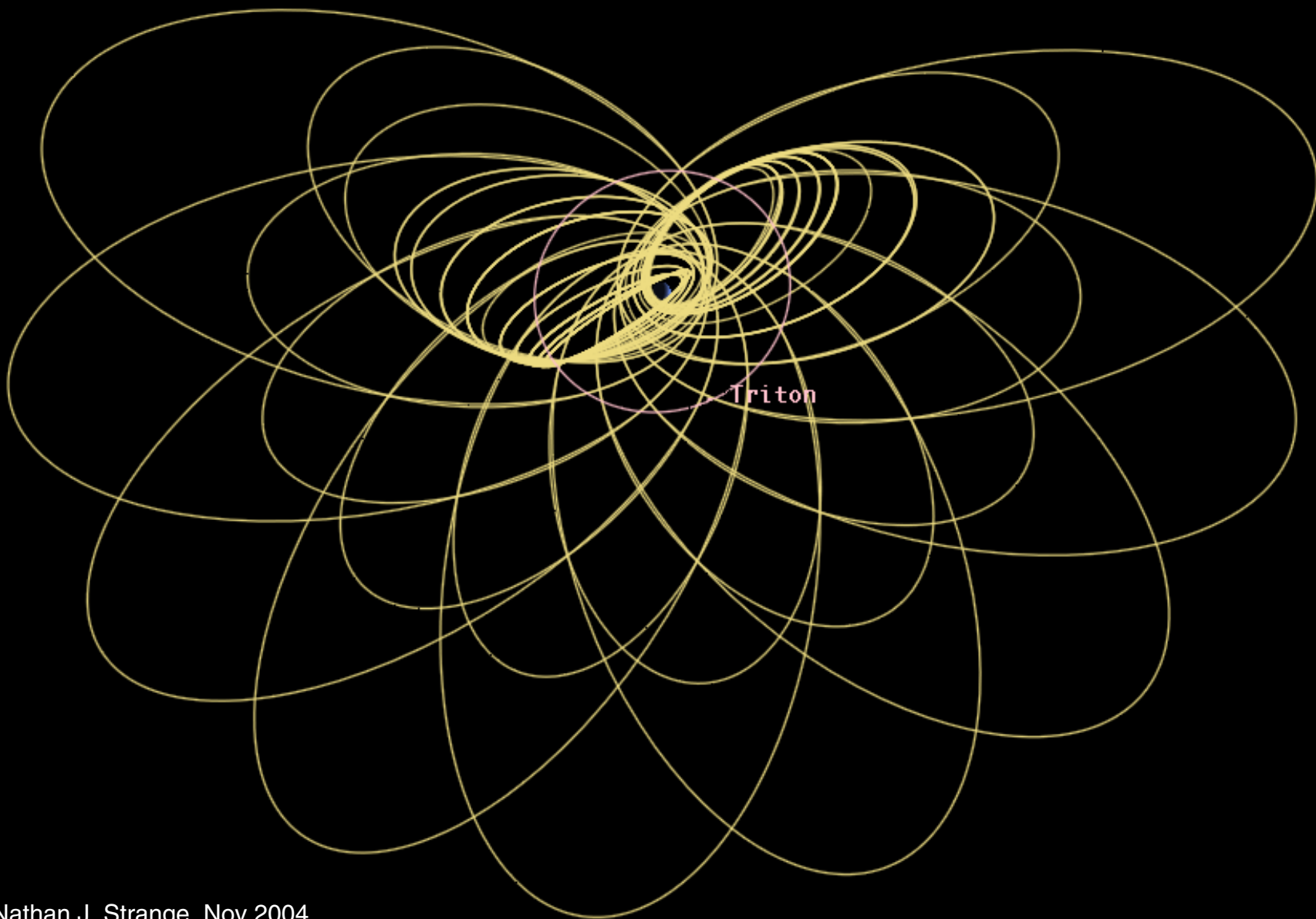
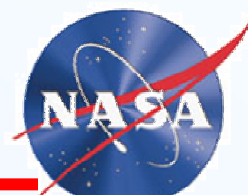
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- Tour begins after aerocapture exit cleanup maneuver
- Science investigations require observing Neptune, its rings, and its magnetosphere from a wide range of radial distances, solar phase angles, longitudes, latitudes, and inclinations
- Close flybys (<1000 km altitude) of Triton to study atmosphere, surface, interior (magnetic field and moment of inertia); flybys provide gravity assists to alter the spacecraft's orbit around Neptune
- *Fully integrated* example tour design demonstrates ability to fulfill science requirements within 2 years
- 3-year prime mission yields flexibility in the tour design
- Cassini mission is demonstrating the operations processes needed (e.g., quick post-flyby nav solutions and maneuver design)



# Mission Design: Neptune System Tour (Sun is to the right)







# Enabling Technologies



- Aerocapture systems and software: thermal protection system (TPS) materials; autonomous guidance, navigation, and control; aeroshell simulation and ground test facilities; sensors and actuators
- Aerocapture flight demonstration - next opportunity is New Millennium Program's ST-9
- Entry probe TPS, pressure vessels, and thermal insulation
- Advanced radioisotope thermoelectric generators (RTG)

## *Greatly Enhancing Technologies*

- Large, multi-engine solar electric propulsion (SEP) systems
- Advanced DSN ground stations for high data volumes



# SSE Decadal Survey 2003: Primary Recommendations on Infrastructure (1)

- ◆ We recommend that NASA commit to significant new investment in **advanced technology** in order that future high-priority flight missions can succeed..

- Power: Advanced RTGs
- Power: In-space Nuclear power source
- Propulsion: Nuclear--powered electric propulsion
- Propulsion: Advanced electric engines
- Propulsion: Aerocapture
- Communications: Ka band
- Communications: Optical
- Architecture: Autonomy
- Avionics: Advanced packaging and miniaturization
- Instrumentation: Miniaturization
- Entry to landing: Autonomous entry, precision landing
- In-situ ops: Sample gathering, handling and analysis
- In-situ ops: Instrumentation
- Mobility: Autonomy
- Contamination: Forward-contamination avoidance
- Earth return: Ascent vehicles



# Backup Slides



# API's Opinions on Broader Issues

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Agree on a message - the SSE Decadal Survey

If you repeat it, they will come

Support Discovery, New Frontiers, consensus-building toward  
Flagships - the Great Observatories model

Beware of entitlements, creeping science requirements,  
special deals on spacecraft, lobbying, earmarking

If Congress thinks we are pork, we will be eaten



# SSE Decadal Survey: Mission Priority by Panel

Panel	Mission Concept Name	Cost Class
Inner Planets	<b>Venus In-Situ Explorer</b>	Medium
	<b>South Pole-Aitken Basin Sample Return</b>	Medium
	Terrestrial Planet Geophysical Network	Medium
	Venus Sample Return	Large
	Mercury Sample Return	Large
	<b>Discovery Missions</b>	Small
Primitive Bodies	<b>Kuiper Belt-Pluto Explorer</b>	Medium
	<b>Comet Surface Sample Return</b>	Medium
	Trojan/Centaur Reconnaissance Flyby	Medium
	Asteroid Rover/Sample Return	Medium
	Comet Cryogenic Sample Return	Large
	<b>Discovery Missions</b>	Small
Giant Planets	<b>Cassini Extended Mission</b>	Small
	<b>Jupiter Polar Orbiter with Probes</b>	Medium
	<b>Neptune Orbiter with Probes</b>	Large
	Saturn Ring Observer	Large
	Uranus Orbiter with Probes	Large
	<b>Discovery Missions</b>	Small
Large Satellites	<b>Europa Geophysical Explorer</b>	Large
	<b>Europa Lander</b>	Large
	<b>Titan Explorer</b>	Large
	Neptune Orbiter/Triton Explorer	Large
	Io Observer	Medium
	Ganymede Orbiter	Medium
	<b>Discovery Missions</b>	Small
Mars	<b>Mars Sample Return</b>	Large
	<b>Mars Smart Lander</b>	Medium
	<b>Mars Long-Lived Lander Network</b>	Medium
	<b>Mars Upper Atmosphere Orbiter</b>	Small
	<b>Mars Scouts</b>	Small

- ♦ Missions listed in Priority Order
- ♦ Missions in bold face were selected by the Steering Group for overall prioritization



# SRM 3 - The SSE Strategic Roadmap

## Roadmap Requirements - Technology

**Note: Protection systems  
includes thermal  
protection systems  
during hypervelocity  
entry**

### Technology and Advanced Development Area

#### Deep Space Power Generation

*Radioisotope Power (thermoelectric)  
Radioisotope Power (Stirling, SRG)  
Solar Power Generation*

#### Deep Space Transportation

*Solar Electric Propulsion -  
Aerocapture  
Advanced Chemical*

#### Deep Space Telecommunications

*Direct to Earth  
Proximity (Relay) Communications*

#### Extreme Environments

*Protection Systems  
Component hardening  
Operational Resilience*

#### Planetary Protection

*Forward Protection  
Backward Protection  
Systems Analysis*

#### Science Instruments

*Remote Sensing  
In situ Sensing*



# SSE Decadal Survey 2003: Mission Priorities

- ♦ *Small Class (<\$325M)*
  1. Discovery missions at one launch every 18 months
  2. Cassini Extended mission (CASx)
- ♦ *Medium Class (<\$650M) – New Frontiers*
  1. Kuiper Belt/Pluto (KBP)
  2. South Pole Aitken Basin Sample Return (SPA-SR)
  3. Jupiter Polar Orbiter with Probes (JPOP)
  4. Venus In-situ Explorer (VISE)
  5. Comet Surface Sample Return (CSSR)
- ♦ *Large Class (>\$650M)*
  1. Europa Geophysical Explorer (EGE)



# Priorities for New Ground-Based Activities

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Enter an equal partnership with NSF to build and operate a

- Large-aperture Synoptic Survey Telescope (LSST)