

B-14951

Calibration Information for Experiments  
71-063A-10 and 72-031A-09

January 1973

This document consists of a temperature correction  
procedure supplied by Dr. Paul Bjockholm of  
American Science and Engineering, Inc., Cambridge, Massachusetts

### TEMPERATURE CORRECTIONS

In order to overlay data from different detectors and temperatures, it is necessary to compensate for gain and bias differences. An alpha particle of energy  $E$  stopping in the  $i^{\text{th}}$  detector produces a count in channel number  $N$  according to the formula

$$N = (E - B(i, T)) / G(i, T) - 256$$

where the gain of the  $i^{\text{th}}$  detector at temperature  $T$  is fitted to the quadratic form

$$G(i, T) = G_0(i) + G_1(i) T + G_2(i) T^2$$

and the bias is

$$B(i, T) = B_0(i) + B_1(i) T + B_2(i) T^2$$

The coefficients  $G_0$ ,  $G_1$ , etc. were calculated from data obtained in laboratory tests using  $\text{Am}^{241}$  ( $E = 5.486$  Mev) and  $\text{Cm}^{242}$  ( $E = 6.115$ ) test sources. The values obtained are given in Table III. The temperature, measured in  $^{\circ}\text{C}$ , was monitored by averaging the readings of two sensors attached to the cases of detectors 5 and 6. There is also available a temperature measurement at the low voltage power supply.

TABLE III Temperature Correction Coefficients

A. Apollo 15

DET	GO	G1	G2
1	.016511	1.11682E-5	1.20094E-8
2	1.63657E-2	3.6829E-6	8.69821E-7
3	1.68476E-2	1.52701E-5	-1.94532E-7
4	.0162362	6.8508E-6	8.04508E-7
5	1.65868E-2	8.1722E-6	4.30857E-7
6	1.67374E-2	1.38262E-5	-9.95569E-8
7	1.66394E-2	8.02561E-6	5.14025E-7
8	1.69597E-2	1.1784E-5	1.29059E-8
9	.0160482	6.30958E-6	1.3558E-6
10	.016149	5.07239E-6	1.4812E-6

DET	BO	B1	B2
1	.454456	-2.41821E-3	9.97633E-5
2	.490183	-6.31298E-4	-1.51483E-4
3	.366534	-3.30801E-3	1.44618E-4
4	.52963	-6.33949E-4	-1.45574E-4
5	.449883	-1.09982E-3	-5.301E-5
6	.405481	-1.70306E-3	8.27464E-5
7	.422976	-1.63199E-3	-3.81668E-5
8	.344155	-3.29596E-3	8.34875E-5
9	.593791	-9.17572E-4	-2.99902E-4
10	.579663	-1.23281E-3	-3.34881E-4

B. Apollo 16

B0(1) = 3.23599E-01	B0(2) = 4.84999E-01	B0(3) = 4.02099E-01
B0(6) = 2.09999E-03	B0(7) = 3.20999E-01	B0(8) = 3.89899E-01
B1(1) = 8.89399E-03	B1(2) = 2.13199E-03	B1(3) = -9.74999E-04
B1(6) = -8.87499E-03	B1(7) = 8.69499E-03	B1(8) = -2.26199E-03
B2(1) = 0.00000E+00	B2(2) = -2.71499E-04	B2(3) = -1.53099E-05
B2(6) = 5.86799E-04	B2(7) = 0.00000E+00	B2(8) = -6.74099E-05
G0(1) = 1.67799E-02	G0(2) = 1.63499E-02	G0(3) = 1.66299E-02
G0(6) = 1.78699E-02	G0(7) = 1.70899E-02	G0(8) = 1.66499E-02
G1(1) = -1.42899E-05	G1(2) = -5.79999E-06	G1(3) = 6.61199E-06
G1(6) = 3.17799E-05	G1(7) = -1.86499E-05	G1(8) = 9.49499E-06
G2(1) = 0.00000E+00	G2(2) = 1.17399E-06	G2(3) = 3.47299E-07
G2(6) = -1.63799E-06	G2(7) = 0.00000E+00	G2(8) = 4.97799E-07

B0(4) = 2.97099E-01	B0(5) = 3.4799E-01
B0(9) = 4.10599E-01	B0(10) = 2.98599E-01
B1(4) = -2.27099E-03	B1(5) = -1.53999E-03
B1(9) = -4.85999E-04	B1(10) = -3.21599E-03
B2(4) = 1.48999E-04	B2(5) = -2.94999E-06
B2(9) = -1.44499E-05	B2(10) = 6.36599E-05
G0(4) = 1.69599E-02	G0(5) = 1.68099E-02
G0(9) = 1.66499E-02	G0(10) = 1.70099E-02
G1(4) = 1.30999E-05	G1(5) = 9.64799E-06
G1(9) = 4.61899E-06	G1(10) = 1.28899E-05
G2(4) = -2.51299E-07	G2(5) = 2.53399E-07
G2(9) = 3.23799E-07	G2(10) = 7.73999E-08;