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Documentation Requirements for Space Experiment System Programs

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R. Wallace ALSEP Project Engineer

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Manager, ALSEP



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PREFACE

Documentation Requirements for Space Experiment System Programs

The objective of program documentation is to provide the means for documenting an equipment design by defining performance, configuration, interfaces, operation, maintenance, etc. As such, the program documentation forms the basic media of communication between personnel engaged in research, development, and design, and the personnel responsible for procurement, production, inspection, maintenance, and related logistic functions. It is the purpose of the discussion in this technical memorandum to establish a perspective and to motivate some thought on the documentation requirements for space flight scientific experiment programs from a cost effective viewpoint.

Cost effective program documentation requirements are not easily developed for a complex program as in the example of the Apollo Lunar Surface Experiments Package (ALSEP) for the Apollo program. The documentation requirements for ALSEP are discussed and compared to ASTP program requirements in the report and are presented as the basis for the formulation of recommendations for documentation requirements for future scientific experiment programs.



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1. OBJECTIVE

The objective of this technical memorandum is to provide cost effective documentation recommendations for future NASA Space Experiment Programs based on previous ALSEP, Viking, and Rocket Payload requirements.

2. METHODOLOGY

To attain this objective, this technical memorandum presents:

- a. A review of documentation requirements and the document data produced for ALSEP Array E.
- b. A matrix of all pertinent documentation by individual program area.
- c. A functional/value effectiveness evaluation of document types.
- d. Recommendations for cost effective baseline documentation and alternate approaches for subsequent NASA Science Payloads and Experiments.
- e. A documentation guideline exhibit to satisfy space experiment program requirements for performance, reliability, and safety.
- 3. DOCUMENTATION CLASSIFICATION

The documentation required by contract for the ALSEP program was divided into categories by type:

Type I - Data requiring NASA written approval prior to implementation.

Type II - Data submitted to NASA for coordination, surveillance, information, review and/or management control.

Type III - Data retained by contractor to be made available to NASA upon request.

Other - Documentation required for internal information and reporting not specified by contract.



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4. DOCUMENTATION COST DRIVERS

Various factors tend to increase the cost of program documentation. The factors of primary concern during this study were:

- a. Large number and variety of documents for all program areas.
- b. Formality of documentation classification, i.e., Type I, Type II, or Type III.
- c. Approval requirements for generated documents.
- d. Frequency of submittals.
- e. Number of copies and resultant reviewing groups inputs and changes.
- f. Number of documentation updates.

5. DOCUMENTATION REVIEW

The documentation requirements and submission schedule for the ALSEP Array E program is presented in Appendix A. Each area of the documentation requirements was reviewed to determine the types of documents produced.

A matrix of all pertinent documentation by individual program area was generated as a result of this review and is presented in Appendix B. This matrix includes an assessment of function/cost effectiveness and recommendations for alternate approaches for subsequent NASA Science Payloads and Experiments.

5.1 PROGRAM MANAGEMENT

Management control documents are the top level basis management documents which describe the overall plan for implementing and administering the program. The ALSEP program was managed using an integrated set of Management Control Plans (MCP's) and Functional Plans combined with a Work Breakdown Structure (WBS) and multitier hardware schedules.



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The overall Management Control Plan (MCP) was generated as a proposal item and negotiated into the contract, originally and for each major revision of the ALSEP contract. For the system and each individual experiment, the primary MCP areas were:

Documentation Requirements for Space Experiment System Programs

Program Control

Engineering

Crew Engineering

Configuration Control

Reliability

Manufacturing

Testing

Quality Assurance

Logistics

Training

Field Support

Procurement

Costs controls were organized to define and control costs for each of the MCP areas, which were also divided for hardware costing to reflect the primary hardware areas of the WBS at Tier I and Tier II levels.

Configuration Management, Reliability, Quality and Safety Program plans were also prepared to further detail scope and method in accordance with planning requirements set forth in NASA program standards.

The ALSEP program used Program Directives, Engineering Directives, Manufacturing Directives, Test Directives and Quality Directives to coordinate the management of program functions between two or more separate organization groups as needed to accomplish program requirements on a timely and efficient basis. These directives replace or supplement functional program plans and may invoke or define



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procedures and processes in the same manner as functional plans. In combination with MCP task statements, directives may fully replace functional plans and be very specific and effective for each individual project or program.

5.2 SYSTEM AND DESIGN ENGINEERING DOCUMENTATION

System and design engineering documentation provide the means for documenting an engineering design by defining performance, configuration, interfaces, operation, etc. These documents form the basic media of communication between personnel engaged in research, development, and design and personnel responsible for procurement, production, inspection, maintenance, and related logistic support functions. Specific documents are listed in the System and Design Engineering section of the Documentation Matrix in Appendix B.

Figure 5.1 illustrates over 90 specifications which were delineated at the start of the Array A Program. The LRRR and Array E experiments added another dozen equipment and interface specifications. Therefore, over 100 equipment and interface specifications were maintained during the program.

Content and format of the subject specifications adhered to the requirements set forth in NASA publication NPC 500-1 (replaced later by the equivalent NASA publication NHB 8040.2). In addition to the flight and GSE hardware specifications for ALSEP, the contractor was responsible for documenting all functional, procedural and hardware interfaces, e.g.,

The interfaces between subsystems,

Test equipment interfaces,

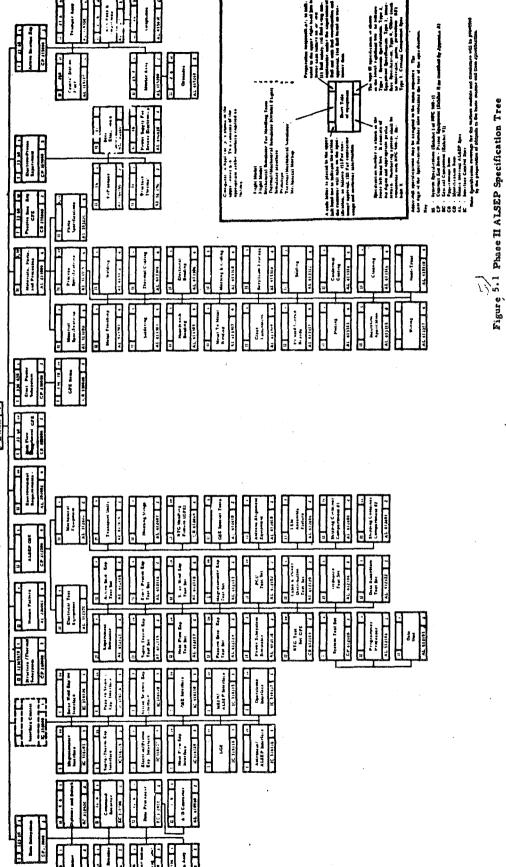
Manned space flight network (MSFN),

The GFE supplied to contractor, and

The crew interfaces.

For specifications, MIL-STD-490 format guidelines for a simple 6 part specification should be used in lieu of NPC 500-1, NHB 8040.2 or contractor procedures (e.g. failure reporting, MRB activity, fault tree analysis, etc.).

ALSEP SPECIFICATION TREE



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Bendix

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Space	Experime	ent Sy	stem	Program	ıs

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At the piece parts level, the contractor was required to use NASA preferred parts wherever feasible based on a reliability-oriented "Order of Precedence." When new specifications were needed to add part screening, burn-in or other controls to satisfy space program requirements, the contractor was required to write, negotiate and issue piece part specifications equivalent to those in the NASA preferred parts lists. There were more than 200 new usercontrolled specs (SCD's) required for ALSEP; 122 of these were used on Array E and they are listed in "Table 5.1." A good number of the new part specifications were required for the LSG and other experiments where the PI could not allow part type substitutions of similar parts from the NASA or other Hi-Rel preferred part lists.

Specification/Source Control Drawings (or documents) need not follow general government standards if "User Specifications" may be jointly negotiated with suppliers to limit key physical, functional, test and traceability requirements of piece parts or components (rather than to document the typical spectrums of requirements applicable to piece part standards).

There were from 12,000 to 14,000 drawings released for each ALSEP array. All ALSEP drawings were prepared to meet the contractor's drawing system requirements for deliverable hardware, i.e., Class B drawings which conform to high but cost effective professional standards. Equivalent standards were imposed on subcontractors and suppliers having design responsibility.

ALSEP Technical Memoranda (ATM's) were defined by ALSEP program directive as the primary means for documenting engineering analyses and reports not explicitly covered by other NASA requirements. Over 1100 such documents were generated during the program (153 of which were prepared for Array E alone). As shown in Table 5.2 the Array E ATM's covered weight and power budgets, stress analysis performed in response to design review action items, scheduled and unscheduled reliability analysis, requirement analysis studies, etc.

There is no doubt that the ATM served as a useful and flexible means for issuing all sorts of technical data using an organized distribution system. The extensive contractor and government distribution list was costly to implement, where essentially all documents on all matters

	TAB	ιε.5.1	
SPEC	CONTROL	DRAWING	INDEX

ISSUE:			

SCD NUMBER	REV	DESCRIPTION		VENDOR	VOL	USAGE B 3 4 5	SCD NUMBER	REV	DESCRIPTION	TYPE	VENDOR	VOL	USAGE B 3 4 5
2330151	C3	Relay		Teledyne	1	x	2340382	٨	Transistor Pr.	2N3811	National	1	+
2331210	E	Connector		Scheldahl	1		2340386	B	Resistor	2505L			x
2334459	A	Transistor	2N3263	ҚСА	11	X	2340383	Ă	SCR		Sage		X
2335190	RCI	Transistor		Т.1.	1	x	2340389	Â		2N1777A	Solid State		X
2335191	RCI	Transistor	2N3045	Т. І.	11	x'x '	2340391		Transistor	2N3866	RCA	[]]	X
2335293	A .	Switch		Haydon Sw	11	X	2340392	•	Capacitor	100B-471	A. T. C.	1	X
2335661	[-	Thermistor	GB42MM62	Fenwal	1	XXX		-	Diode	1N4716-	Dickson	1 1	
2335662	-	Thermistor	FG108N	Tylan	1 1	X.	2340395	1	Transistor	2N2484	Teledyne	1	X
2338162	A	Resistor	PZ-A	Precision	1	x	2340397	B	Resistor	MH51	Welwyn		X
2340305		Microckt	DM7800H	National	1		2340398	B2	Resistor	SPEC	Sage	11	X
2340307	-	Microckt	LMIOZE	National	li.	x x	2340399	B	Capacitor	SPEC	Aerovox	11	L X
2340311	A	Microckt		National	1i -	xx	2346200	BZ	Fuse .	262-XXX	Little Fuse	2	x
2340312		Microckt	LM107	National	li -	XX .	2346201	C	Microckt	54L-XX	T.I.	2	XXXX
2340313	A3	Crystal	Special	Monitor	1 i -	x x	2346202	С	Transformer	SPCL	O. Schott	2	X
2340319		Coil	1025	Delevan	11	x	2346203	A	Choke	SPCL	O. Schott	2	
2340328	A	Microckt	UL02C	AMI	li.	x	2346204	С	Choke	SPCL	O. Schott	2	x
2340329	A	Microckt	MX02D	AMI	li	X X	2346206	•	Choke	SPCL	O. Schott	2	l x
2340330	C	Microckt	SPOIC	AMI	1.	X.	2346207	E2	Microcket	54XX	T. I.	2	lx x x >
2340338	c	Diode	IN4568A	Dickson		x x x	2346208	-	Capacitor	WQM2DM	Elmenco	2	
2340351	14	Heater	SPCL	Minco	11	2 x	2346209	A	Choke	SPCL	O. Schott	2	x I
2340353	A	Registor	CEA	IRC	1:	T Tx	2346211	A	Choke	SPCL	O. Schott	2	x
2340354	A3	Connector	SPCL	Omni Spectra	1:	x î	2346217	A,	Connector	WST, WSW	Hughes .	Ž	xx
2340355	A	Diode	1N4370A	Dickson	li	x x	2346220	•	Cutter	1SE166	Atlas	2	LEAM
2340356		Transformer	SPEC	O. Schott		x x	2346222	A	Inductor	SPCL	O. Schott	2	xx
2340361	A	Connector	02025-0001	Transitron	1:		2346223	A	Inductor	SPCL	O. Schott	2	x x
2340362	RCI				11		2346224 .	A	Choke	SPCL	O. Schott	2	Îxî
6	RU	Connector	MDA/MDM	ITT Cannor		2	2346225	A	Choke	SPCL	O. Schott		
2340363	1	Diode	SEL-1N4568A	Dickson	11	X	2346226	В	Transformer	SPCL	O. Schott		X
2340365	В	Transformer	SPEC	O. Schett	11	x	2346227	B	Transformer	SPCL		2	x
2340366	A	Transistor	2N4416A	T.I.	1	xi ·	2346230		Capacitor	CKR05	O. Schott	2	x
2340367	A	Resistor	мн 681 - Х	Caddock	1	X	2346231		Capacitor .	CKR06	Aerovox	~2 ~2	XXXX
2340368	AZ	Microckt,	NHOO19 SPEC	National	11	X	2346233		Capacitor	CKRII	Aerovox		XXXX
2340369	AZ	Connector	MCDBI	Microdot	1		2346234		Capacitor	CKR12	Aerovox	2	XXXX
2340370	-	Diode		TRW	11	x	2346235	-	Capacitor	CKR12 CKR14	Aerovox	2	XXXX
2340371	RCI	Microckt	NH0001 AF-SEL	National	1	X ·	2346236	-	Capacitor	CKR14 CKR15	Aerovox	2	хх
2346372	A	Cap. Filt.	1250-700	Erie	1	X	2346237	-	Capacitor	CKR15 CKR16	Aerovox	2	X.
2340376	В	Diode	IN5483	Unitrode	11	X	2346238	Ā	Thermistor		Aerovox	2.	1.
2340379	С	Diode	5082-3006	H. P. Assoc.	11	X,	2346239			44032,	YSI	2	2 X
			6 (1		23-16240	Ā	Relay	431-12 CSR13	Teledyne	2	X
					-	- Lineare -	2346241		Capacitor		Sprague	2	xxxx
B - Basic-Ce	ntral Si	ation	4 - LMS				2346242	•	Capacitor	CSR13	Kernet	2	1 X X X

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3 - LSP

SCD NUMBER	REV	DESCRIPTION	TYPE	VENDOR	VOL	USAGE B'3 ¹ 4 5
2346244 2346247 2346247 2346250 2346252 2346255 2346255 2346257 2346257 2346257 2346259 2346260 2346261 2346261 2346263 2346265 2346265 2346265 2346265 2346265 2346265 2346270 2346271 2346271 2346275 2346275 2346275 2346275 2346275 2346275 2346275 2346275 2346275 2346275 2346275 2346275 2346275 2346275 2346275 2346275 2346275 2346275 2346283 2346283 2346283 2346283 2346283	- C2 B2 C2 C2 A A A C A B B C2 A C2 A C2 A	Grystal Capacitor Gapacitor Gapacitor Transformer Transformer Transformer Grystal, Filt Mosfat Wire Wire H. V. Transformer Relay Relay Choke Relay Switch Relay Transformer Transformer Transformer Microckt Microckt Microckt Microckt Microckt Microckt Microckt Microckt Microckt Microckt Transformer Transformer Transformer Transformer Transformer Capacitor Transitor Gepacitor Resistor Crystal Resistor	2.0352MH2 337D HL-5 SPCL 2N5071 2N4857 SPCL SPCL SPCL SPCL SPCL BR-17 SPCL BR-17 SPCL BR20-AX- SPCL BR20-AX- SPCL MI001AF LM108AF&H 1402-02 LM103-XX- CDA2-1&2-2 GT80, 8T90, NH00f9F 2N930A D0-T37 FHA-15 JMC52XX 2N3019 MC605 CEC SPCL S102	Monitor Prod Sprague Marshall Ind. O. Schott RCA/Solitron Teledyne O. Schott McCoy Motorola Gore O. Schott Electrol Babcock O. Schott Teledyne T. I. Babcock O. Schott O. Schott O. Schott O. Schott National Philbrick National	2 2 2 2 3 3 3 3 3 3 3 3 3 3 3 3 3 3 3 3	B 3 4 5 X X X X X X X X X X X X X X X
2346290 2346291 2346291 2346294 2346295 2346296	RCI A A C	Resistor Resis, -Network Diode Connector Diode Microckt		Vishay Vishay Motorola Microdot Unitrode Motorola	3 3 3 3 3 3	X X X X X

TABLE 5.1 (CONT'D)

SCD NUMBER	REV	DESCRIPTION	туре	VENDOR	VOL	USAGE B 3 4 5
2346297 2349497 2349498	A3 A B	Microckt Connector Connector	931.11 1 -202845 - X 202844 - X	Fairchild Amp. Inc. Amp. Inc.	3 3 3	x x x x x x
2363402	۸	Meter	SPCL	int inst	3	X,
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ALSEP

ARRAY "E"

VOLUME: 1

ISSUE: B

SEPT. 20, 1971

PREPARED BY:

TO: 2340399

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R. Dallaire S. Ellison P. Sondeen R. Roukas R. Place SPECIFICATION CONTROL DRAWINGS R. Kovac H. Brown R. Howell J. Staats V. Schiller FROM: 2330151 B. Cole R. Muggee J. Hendrickson R. Fatka D. Cook H. Spencer N. Hendrick (THIS ISSUE "B" SUPERSEDES SCD VOL 1, 11, 111 J. Mansour AND SUPPLEMENT 1 DATED JUNE 14, 1971) K. Andreasen J. Patel E. Ford E. Lavers J. Dennis ALSEP RELIABILITY DEPARTMENT P. Steinmeyer J. Staley

N. Veit

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Table 5.2

SYSTEM ATMS - ALSEP ARRAY E

ATM No.	Date	Author	Title				
241 E (1)	Sept. '71	P. Sondeen	ALSEP Acceptable Parts List	ATM No.	Date	Author	Title
511 2 (1)			for Array E, Amendment 1.	1004	5/7/71	C. Jensen	EMI Investigation for Array E
242 E (1)	Sept. '71	P. Sondeen	Approved Materials List for ALSEP Equipment.	1015	6/10/71	R. Dallaire	Array E Uplink Redundancy Method
268AF ·	2/25/71	T. Swann	ALSEP Weight Report.			D.J. Thomas	Justification
			ALSEP Weight Report.	1023	6/17/71	R. Dallaire	Array E Time/Cycle Sensitive List.
268 AG 268 AH	Sept. '71 4/3/72	R. Foster	ALSEP Weight Report	1023B 1023C 1023D	10/4/72 10/30/72 11/2/72	•	· · · · ·
605A	10/26/70	R. Dallaire	Failure Rate Data for ALSEP.	1033	7/20/71	R, West	
780B	10/29/71	L. Moskowitz	Qualification Status List Fuel Cask and Structure Assembly ALSEP	1033A	9/27/72		ALSEP "EEE" Composite Parts List.
930 930A	12/9/70 2/11/71	O. Neau D. Thomas	Array E, Flight 6. Command List (Array E)	1034	7/26/71	B. Lavin	System Safety Progress Report ALSEP Array E.
930B	6/2/71	D. Inomas		1037	8/3/71	G. Cripps	Schjeldahl Dale_Connectors.
930 C 930 D	11/2/71 3/8/72			1045 1045A	8/11/71 9/21/72	P. Sondeen	ALSEP Composite Non-Metalic Materials List.
935	1/11/71	J. Jones	System Safety Program Plan for ALSEP Array E.	. 1048	8/11/71	R. Lavin	July System Safety Progress Report - ALSEP Array E.
953	1/22/71	R. Dallaire	ALSEP Flight System 6 (Array E)	1054 1054 A	9/1/71 Oct. 1971	J. Hendrickson	Monthly Array E Qualification versus Flight System Differences
953 A		L. Moskowitz	System Level Failure Mode Effects:. and Criticality Analysis.	1054B	11/1/71		Report.
953B	10/4/72			1054C 1054D 1054 E	12/9/71 1/3/72 2/1/72		1054G 4/1/72 1054H 5/10/72 1054I 7/10/73
962	2/2/71	R. Wallace	Array E System Grounding Philosophy.	1054 F	3/1/72		10541 7/10/72 10543 10/4/72
988 .	3/8/71	A. Romans	ALSEP Array E Multi-layer Printed Circuit Source Qual Test Plan.	1067	11/4/71	; J. Jones	October System Safety Progress Report ALSEP Array E.
989	3/10/71	A. Wadleigh	Array Subpackage 1 Dynamic	1069	11/9/71	R. Sporzynski	Array E Software Description
	•	Dr. Min	Analysis.	1072	11/17/71	J. Kasser	Array E System Description
990	3/12/71	R. Wallaco	Array E LEAM Digital Interface - 54L Versus Amelco Logic.	1073	11/18/71	R. Redick	Apollo 17 Array E Lunar Surface Deployment Procedures

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SYSTEM ATMS - ALSEP ARRAY E (CONT.)

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ATM No.	Date	Author Titl	e	ATM No.	Date	<u> </u>	Title
1076	11/29/71		SEP Array E Power Budget	1100	5/12/72	C, Jensen	Invertigation of Array E Experiment EMI Test Data Volidity
1081	1/3/72 3/30/72	R. Dalaire/T. Fox	TTL-54L	1101	5/12/72	B, Lavin	April System Safety Progress Repor Array E
1081A 1082	2/11/72	t Matthie /D Breecke	Recommendations for Minimizing				
1495	6/11/16	J. Matune, D. Divetic	Green Crud.	1102		H. Geisa }	ALSEP Contingency Procedures for Apollo 17 (ALSEP-MP-08).
		* /	ALSEP Array E Engineering Model	1102 1102A	9/27/72	· · · ·	
1083	2/17/72	R. Thomas	SP-1 with PSE - Design Limit Vibration Test Results.	1103	6/30/7Z	B. Lavin	Handling, Packaging, Transportatio & Storage of ALSEP Array E Flight
			vibration lest Kesuits.				Hardware & Support Equipment
1084	2/21/72	B. Lavin	Dec Jan, System Safety Progress Report, Array E.	1105	6/19/72	D.J. Thomas/ C. Jensen	EMI Test Results & Margin of Com patability for ALSEP Array E
1085	2/23/72	W. Gilham/D. Fithian	ALSEP Array E Design Verifica-	1106 ,1106A	8/9/72 10/4/72	T. Fox	System Lovel Qualification Status Lovel Array E
1087	3/3/72	B.J. Thomas	Investigation into the Scrambling of Array E Qual Model PDU	1107	8/9/72	T. Fox	System Analysis of Two Year Life Capability
		•	Relays at Turn-On.	1108	8/10/72	D. Steinmeyer	Thermistors used as linearized temperature sensors.
1089	3/14/72	B. Lavin	Feb. System Safety Progress Report Array E.	1112	9/27/72	H. Van Hoorde	•
1090	3/22/72	R. Thomas	ALSEP Array E Engineering Model SP-2 Design Limit Vibra-	1114	10/16/72	. J. Jones	Crew/Mission Operational Hazard Analysis.
•			tion Test Results.	1119	10/31/72	J. Massatics	Qual SE (SP-1 & 2) Design Limit
1091	3/23/72	R. Thomas	ALSEP Array E Engineering Model SP-1 with LSG - Design Limit Vibration Test Results.		-		Vibration Test Results.
1092	3/25/72	C. Jensen	SEP/ALSEP EMI Interface.				
1093	4/7/72	C. Jenson	Array E S-Band Compatibility Test Results Analysis.			•	•
1095	4/11/72	A. Bedford	Array E Calibration Curves.			•	·
1096	4/18/72		urious Status Changes in Fray E				
1098	5/1/72		ray E Action Item 604 Ripple on felt Line.		·		· .

CENTRAL STATION ATMS (ARRAY E)

ATM No.	Date	Autoor	Titie	ATM No.	Date	Author	Title
New Nurther				_			
947	1/21/71	J.E. Staley	Array E Power Conditioning Unit Automatic Power Medica -	981	2/23/71	T. Fox	Reliability Prediction - Array E Redundant Command Receiver.
			ment Circuit.	. 982	2/23/71	T. Fox	Single Point Failure Summary - Array E Redundant Command
949	1/22/71	M. Papaioan	Array E Command Decoder Failure Mode Effects & Criticality Analysis.	983	2/23/71	T. Fox	Receiver. Parts Application Analysis - Array E Redundant Command
950	1/22/71	J.G. Smith	Array E Data Processor Failure Mode Effects & Criticality	984	2/23/71	T. Fox	Receiver, Failure Mode Effects & Criticality
951	1/22/71	A. Romans	Analysis. Array E PDU Failure Mode Effects				Analysis - Array E Redundant Command Receiver.
951 A	Sept. 1971		and Criticality Analysis.	985 .	2/24/71	V. C. Kemp	ALSEP Array E Command
952 952A	1/22/71 6/1/71	A. Micskowitz	Array E PCU Failure Mode Effects and Criticality Analysis.	998	4/1/71	John G. Smith	Decoder Breadboard Test Report.
954	1/22/71	M. Papaloan	Awary & Command Decoder Parts Application Analysis,	776		John G. Jintel	ALSEP Array E Parts Application Analysis of Signal Conditioning Circuits.
955	1/22/71	J.G. Smith	Array E Data Processor Parts Application Analysis.	999	4/1/71	John G. Smith	ALSEP Array E Signal Conditioning Circuits Reliability & Pailure Mode Effects Critical Analysis.
956 956 A	1/22/71 Sept. 1971	A. Romans	Array E PDU Parts Application Analysis.	1005 1005 A	5/11/71 12/15/71	A. Romans	ALSEP Array E PSK Transmitter - Failure Modes Effects & Criticality Analysis
957	1/22/71	A. Moskowits	Array E PCU Parts Application Analysis.	1006 1006 A	5/11/71 12/1/71	A. Romans	ALSEP Array E PSK Transmitter -
958	1/26/71	B. Mc Leod	Command Decoder for ALSEP Array E.				
963	2/3/71	D. Thomas	Central Station Subsystem Description for ALSEP Array E.	1028	7/2/71	H. Geiss	Crew Engineering Test Plan for Evaluation of Array E Antenna Aiming Mechanism.
964	2/3/71	J. Maszatics	ALSEP Array E Component Non- Operating Vibration Specifications.	· 1031	7/9/71	R. Sigler	ALSEP Array E Antenna Aiming Mechanism Design Verification Test Results.
971	2/5/71	R. Ziesmer	Crosstalk & Ground Differentials in the Central Station.	1032	7/15/71	N, Hadwick	Theoretical Modeling & Analysis of
974	2/12/71	B. McLeod	ALSEP Command Decoder Preliminary Functions	1051	8/17/71	T, Kuchenmeis	PCU/PDU Output Voltages.
•		,	Description.	1113	10/6/72	D. Butte	Array E Antenna Aiming Mech. AISEP Array E C/S Thermal Design, Analysis/Test Final Report.

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LSP ATMS (ARRAY E)

ATM No.	Date	Author	Title	ATM No.	Date	Author	Title
927 A	3/17/71	P. Ireton	Lunar Seismic Profiling Experiment Dynamic Analysis.	1053 1053A 1053B	8/30/71 12/9/71 1/12/72	R. Brown J. Jones J. Jones	LSP Operational Hazard Analysis.
928	12/3/70) J. Zimme	r LSPE Integrated Test Plan, ALSEP-TM-659	1056 1056A 1056B	9/14/71 12/20/71 1/12/72	J. Jones	LSP Ground Operations and Safety Plan.
948	1/22/71	E. Weidner	Deletion of Geophone Temper- ature Sensor.	[.] 1079	12/15/71	Dr. Min/ Dr. Dewhirst	LSPE Explosive Package Fragmen tation and Cratering Related to
959	2/1/71	R. Deppe	First Crew Engineering Eval- uation of Array E - LSPEt Geophone Cable Reel.	1	12/15/71	D. Toelle	Striking Probability Investigation
975	2/12/71	J. Staaté	LSPE Parts Application Analysis.	1086	2/25/72	T.W.Weir	LSPE Thermal Battery Test.
976	2/12/71	J. Staats	LSPE Failure Modes, Effect Analysis.	1088	3/7/72	L. Lewis	LSPE Safe Arm Slide Failure Evaluation Report,
1002	4/22/71 I). Toelle	LSPE Explosive Package Stowage Thermal Constraints.	1094	4/7/72	R. Worchester	LSPE Housing & Charge Assy Foam Test Report.
1035	7/26/71	J. Staats	LSPE Timer Control Module Seal Analysis.	1099	5/8/72	Dr. Min	Preliminary Test Evaluation on LSPI2 Harard Analysis
1036 1036A	7/29/71 11/10/71	J. Owens	LSPE Transmitting Antenna Stability Investigation.	1104	6/12/72	B. Lavin	Comparative Safety Analysis - LSP Timers
1038	8/3/71	J. Jones.	LSP Timer Overbanking on the Lunar Surface.	İ109	9/15/72	D. Toelle	Lunar Seismic Profiling Experi- ment Design Verification Thermal
1039	8/6/71	J. Staats	"EEE" Parts List for LSP.	1110	9/15/72	D. Toelle	Vacuum Test LSPE Qualification & Flight
1040 ·	8/6/71	J. Staats	Non-Metalic Materials List for LSP.		,,,.		Acceptance Thermal Vacuum Test Summary & Thermal Design Final
1043	8/6/71	J. Staats	Time/Cycle Sensitive Components List for LSP.	1115	10/16/72	J. Jones	Report. LSP Final Safety Report.
1046	8/11/71	Dr. G. Min	LSP Explosive Package Fragmentation Study.				
1049	8/11/71	R. Brown	LSP Detailed System Hazard Analysis,				

LEAM ATMS (ARRAY E)

ATM No.	Date	Author	Title	ATM No.	Date	Author	Title
977	2/15/71	J. Cooper	LEAM Failure Mode Effect & + Criticality Analysis.	1075 1118	11/23/71 10/26/72	L. Mills G. Pearos	LEAM Thermal Design Report
978	2/15/71	J. Cooper	LEAM Reliability Prediction. 🛩		10/20/12	G. FREID	LEAM Thermal Design Analysis/ Test Final Report.
980	2/23/71	J. Cooper	A Trade-Off Study of Various Methods of Releasing the LEAM Dust Covers.				
995	3/24/71	R. Sims L. Mills	LEAM Film Development Test Report.				
996	3/29/71	T. Kuechen- meister	Crew Engineering Evaluation of the Array E LEAM Experiment - Crew Engineering Model.			· · ·	
1010	6/7/71	D. Perkins	LEAM Film Development Report		•		
1011	6/7/71	P. Pilon	LEAM Film Vibration Report	•			• · · · · · · · · · · · · · · · · · · ·
1012	6/7/71	P. Pilon	LEAM Mechanical Tests				
1013	6/7/71 .	G. VanHoorde	LEAM Reliability Numerical Analysis, Reliability Mathematical Model FMECA & Single Point Failures.			· . ·	
1014	6/8/71	G, VanHoorde	LEAM CDR Parts Application Analysis				
1016	6/10/71	M. Calarese	Gross Hazard Analysis Report - LEAM Experiment				
1019	6/11/71	L. Mille	LEAM DVT Thermal Test Report.			•	· · · · ·
1022 ·	6/14/71	L. Kaliniec	LEAM Dynamics Analysis (DVT).				-
1025	6/17/71	G. VanHoorde	Time Sensitive Cycle Items - LEAM.		•		
1 1027	6/21/71	G. VanHoorde	Parts & Materials List for LEAM Experiment.				
1030	7/9/73	G. VanHoorde	LEAM Reliability Numerical Analysis,				•
1065	10/19/71	J. H. Owens	Structural Analysis Report LEAM				
1066	10/22/71	K. Wadlegh	LEAM Dynamic Analysis Flight Model.				•

LMS ATMS (ARRAY E)

ATM No.	Date	Author	Title	ATM No.	Date	Author	Title
966	2/2/7:	X. 20070	Grew Engineering Evaluation of Array E - LMS Experiment Crew Engineering Model,	929	12/4/70	G. VanHoord	s Single Point Failure Analysis Summary, 45%,
965	2/4/71	F. Howell	LMS Reliability - Reliability	929B	5/18/71		,
965A	6/9/71		Prediction	937	1/12/71	G. VanHoord	e Preliminary LSG Numerical
966 966A	2/4/71 6/3/71	F. Howell J. Hendrickson	LMS Reliability - Parts Application Analysis.		-, -=,		Reliability Analysis.
966 B 967 967 Addendi	3/9/72 2/4/71 um 1 10/2 9/7 1	F. Howell	LMS Reliability - EEE Part List for UTD and Bendix.	979	2/18/71	G. VanHoorde	Preliminary Parts Application Analysis LSGE.
967A	4/14/72			1008	6/2/71	G. VanHoorde	LSG Reliability Mathematical Model
968	2/4/71	F. Howell	LMS Reliability - Non Metalic Material List.	1			Reliability Numerical Analysis & FMECA
969 . 969 A	2/4/71 7/27/7 2	F. Howell	LMS Reliability - Time/Cycle Sensàtive Part List.	1009	6/2/71	G. VanHoorde	LSG CDR Parts Application Analysis
				1009 A	Aug. 1971	l	
970 970a 970 b	2/4/71 5/15/71 3/21/72	F. Howell	LMS Reliability - FMECA Ł Single Point Failure Summary.	1017	6/10/71		Gross Hazard Analysis Report - LSC Experiment
1018	5/10/71 C		ross Hazard Analysis Report - LMS speriment	1026	6/21/71	G. VanHoorde	Parts & Materials List for LSG Experiment.
1020	6/11/71	A. Tente	LMS Mechanical Test Reports.	1043	8/ 9/71	G.R. VanHoor	de "EEE" Parta List for LSG.
1029	7/8/71		LMS Thermal Vacuum Tests Reports.	1044	8/10/71	G.R. VanHoor	de Non-Metalic Materials List for LSG.
1042	8/9/71	J. Owens .	LMS Structural Analysis Report.	1057	9/15/71	B. Lavin	LSG Boydbolt Release Tests Rep
1071	11/15/71	L. Duesterbe	rg Array E ALSEP LMS High Voltage Power Supply Capacit Problem Analysis & Corrective	1058	9/16/71	M. Dela Crux	LSG Flight Sensor Closed Loop Performance Computer Analysis
			Action	1024	6/17/71	G. VanHoorde	Time Sensitive Cycle Items - LSG.
097 1097A	4/19/72 D. 5/17/72	Toelle	Lunar Mass Spectrometer Design Verification Thermal Vacuum Test	1116	10/16/72	G. Pearos	ALSEP Array E LSG Therma
11	9/15/72	D. Toells	LMS Qualification & Flight Accept- ance Thermal Vacuum Test Sum- mary & Thermal Design Final Report,				Control Design Analysis & Te Final Report.

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were issued to a complete distribution list, e.g., each PI desired all system ATMS and those on all experiments other than his own. Full distribution of ATM logs should allow recipients to review each ATM topic and request only those items he knows will be of interest to him.

5.3 CONFIGURATION MANAGEMENT

Configuration control, the processing of drawing changes, specification changes, etc., was based on the use of the existing contractor's configuration management procedures which met the intent of NASA regarding design baseline configuration accounting and change control.

Change Board Membership, approval signatures required for various types of drawings, etc., was established and maintained via contractor program directives coordinated with contractor in-house policies and procedures.

PDR and CDR scheduling for ALSEP was determined by program management and supported by configuration control as appropriate.

5.4 SYSTEM TEST

Test Procedures, Types I and II for ALSEP consisted of dozens of documents generated for each type of functional or environmental test performed at component, integration system and end-item levels of assembly, during separate stowage mission modes, with variation procedure documents for MSFN test, KSC operational check, etc. The rather large variety of customer controlled tests and individual test-unique test procedures results in a costly test document program, justified only by man-rated safety programs, but not by Experimenters. Therefore it is suggested that emphasis be placed on meeting the experiment end item requirements in customer approved procedures and reduction in the number of formally controlled procedures and reports to those which deal with end item testing.

5.5 RELIABILITY AND QUALITY ASSURANCE

Quality Assurance activities which include Quality Engineering, Inspection, Testing, etc. was based on existing contractor's quality



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assurance procedures which met the intent of NASA requirements. The reliability program was conducted in accordance with the Reliability Program Plan developed to respond to NASA requirements as specified in the contract. Specific Reliability and Quality Assurance documentation and recommendations for future programs are included in the Reliability and Quality Assurance section of Appendix B.

5.6 SYSTEM SUPPORT/LOGISTICS

The System Support and Logistics activities were tailored to meet NASA requirements as specified in the contract in the areas of Launch Complex Operations, Safety, Training, Human Factors, Maintenance, Spares, Launch Complex Quality Programs, etc.

In areas such as familiarization manuals and maintenance manuals documentation costs can vary widely depending on their depth. Scope limited to external features, input/output functions and external test interfaces minimizes user cost and complexity. Recoverable or nonrecoverable hardware may be maintained at the end-item level which is generally more efficient for both the user and the supply contractor (who willsperform lower-level maintenance.)

5.7 MANUFACTURING

Manufacturing documentation consisted of a Manufacturing Program Plan, Schedules, Make or Buy Plan, Processes (existing contractor processes where applicable), Work Order Operation Sheets (WOOS), Workmanship Standards, and Tool Drawings. Documentation cost reduction in this area could best be achieved through the reduction of the number of separate documents to be maintained, reducing all document classification to Type III, and using existing contractor procedures to the maximum extent possible.

6. DOCUMENTATION VALUE ASSESSMENT

Appendix B presents a matrix evaluation of ALSEP documentation. Of the 81 line items, 48 or 59% of them are Type I and Type II which require approval/disapproval action.



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It would seem advisable to eliminate most Type I documents by negotiating the baseline equivalent during contract negotiation, all subsequent changes would be processed through contract change channels which can fully evaluate and control delta costs.

Documentation Requirements for Space Experiment System Programs

Objective Accomplishment, Relative Cost, Program Impact and Justified Cost/Result columns in the Table score effectiveness in these areas on a basis of one-to-ten, from low to highly effective.

Items which are scored as low as 7 or 8 in the "Justified Cost/ Result" column generally may be replaced by lower-cost alternate concepts or eliminated. Requirements for the lower cost alternates should be reduced to meet program intent for each contract; established government standards should only represent a guideline.

7. SUMMARY AND RECOMMENDATIONS

To realize a documentation cost reduction for a space experiment systems program it is necessary to minimize cost drivers while maintaining sufficient requirements to assure hardware performance, reliability, and safety. This can be achieved by the systematic application of the following actions:

- a. Reduce, to the maximum extent possible, the number and variety of documents for all program areas.
- b. Limit the number of Type I and Type II, requiring approval/ disapproval action.
- c. Reduce frequency of document submittals.
- d. Limit distribution lists to those areas where the document is needed.
- e. Reduce the number of document updates.
- f. Utilize a lower cost alternate document for the high cost item where possible.



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Appendix B summarizes the documentation reviewed and presents recommendations for future space experiment applicability and lower cost alternate documentation items. The rationale for the recommendation is included in the remarks column.

Appendix C was prepared for use as a guideline to a cost effective payload experiment documentation which can still satisfy the intent of NASA requirements for performance, reliability, and safety.



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APPENDIX A

SCHEDULE III EXHLBIT D

ALSEP FLIGHT SYSTEM NO. 6 (ARRAY E)

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DOCUMENTATION SCHEDULE

		Delivery	Dates		
Item No.	<u>Item Description</u>	Initial	Revisions	Type	Copies
Part A	Program Control Data		•		•
A1	Nanagement Control Plan	v/Proposal	2 weeks after contract award; thereafter 5 days after revision	п	•
12	Cost Proposal	w/Proposal	<i>t</i> •••	Π	•
A3	Nake or Buy Flan	w/Proposal		I	4
A li	Financial Mgt. Reports a. Monthly (533A)	SHOLD BE : MATHLY (ON a BORNS 35TH CAY OF 17 - 17 - 17 BENE XEN 100	Monthly (15th day of sonth reported)	п	مد
	b. Quarterly (533)	antiplation on allow	Quarterly		10
A5	Nonthly Letter	Profit Car of register as Filencial car frage of Carton Ocine Automation)	Monthly (15th working day following period being reported)	п	15
NG '	Photographic Requirement				

	e.	Piçtures	20 days after significant event	· •••		original and one work print
-	Minutes, Re	views*	5 days after review	As Required	п	•

Iten No.	Itea Description	Delive <u>Initial</u>	ry Dates Revisions	Туре	Copies
A 8	, Final Report	Not I	Required		
A9	New Technology Report	As Required		<u> </u>	<u> </u>
Part B	Design/Systems Engineering Dat	<u></u>		•	•
B1	Interface Control Specifi- cations	7/1/71	5 days after revision	Ĩ	7
B 2	Contract End Item (CET) Specification	TBD	5 days after revision	I	7
13	Crev Training Model Specification	TBD	5 days after revision	I	7
ві	Drawings (Per Para. 3.1.9 Exhibit A)	As Requested .	••	III	1
B5	Design Analysis Reports	As Available	5 days after revision	Π	4
36	Interface Control Documenta- tion	Submit 14 days prior to CDR	5 days after revision	I	3
B7	Subcontractor Drawings and Engineering Orders	As Requested	**	111	l plus l repro (sepin)
36	Specification and Source Control Drawings	As Requested		III	l plux l repro (sopia)

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B9 Operations Flam input TBD 9 days after revision I 22 B10 Design Cortification Review TBD II 9 B11 Neasurements Requirements Decement Freilaisary: 5 days after revision I 5 B11 Neasurements Requirements Decement Freilaisary: 5 days after revision I 5 B13 SpherestoryVendor Aspect and Data As Requested III 1 B13 Spherest Operations Data address of CH*o 30 days prior to college are review Coll close out forms, attract address of the review II 30 B13 Design Review Minutes (FM*s and CH*o) 16 days after collinger of fit. Coll close out forms, attract addres of the review II 30 B14 Design Review Minutes (FM*s and CH*o) 16 days after collinger of fit. As required II 30 B15 ALERCS As Reparted I As required I 10 Part C Test Plan (dystem Law) Col days prior to stat As Required I A C4 Quiffection Test Plan (dystem Javes) Col days prior to 5 days after revision I A C5 Test Procedures and Specifi- cationa-Brevised 20 days prior to 5 days after revision I A		3	•				•		
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Continuing Revision (Manu-				•	ta	ty and Quality Control Dat	Reliability s	<u>D</u> 1	Part
	сору ВхА	1 to MS rep., c 51 of B MP Hanu	II .	s Issuel	[ssued As	g Revision (Manu-	Coatinuing Re		Dl
D2 Quality Handling and In- 20 days prior to 5days after revision II 1 to spection Procedures each test rep.	(SC	l to MS rep.	. II	isys after revision					D2
D3 Contractor Parts List Preliminary Pinal: Concurrent II 4 As Available with delivery of flight hardware		h,	II	th delivery of	vailable vi		Contractor Pa	(D3
p4 Subcontractor Parts List Concurrent with III 1 subcontractor delivery of fit. hardware		1	111		very of fit.	subc deli	Subcontractor	1	Dŕ

		•	Delivery	Dates		
	Ites No.	Item Description	Initial	Revisions	Type	Copies
	Part E	System Surport/Logistics Date				
	E)	Storage Procedures for End Items (PriESI) Revised TM's wherever feasible	20 days prior to delivery of flt. bardware	5 days after revision	. 11	4 (RSC) Bra Gile Office 1 Ralpo
	5 2	Manual Updating	•			
	PC			••••••		
			•	•		
		Femiliarization Course Handout	Concurrent with course	:	11	Conrie re- quirtents plus 4 copie ani 1 rejto.
	E 3	Ground Safety Plan				
•		A. Draft]	90 days prior to flight hdw.	10 M	IT	5
		•	delivery		_	
		B. Final	Concurrent with flight hardware	' 	п	20
	Et	Crev/Missions Operation Hazerd Analysia	14 days prior to FTR	As Required	I	10 plus repro.
	B 5	Support Material List	As Required	· .	п	Ъ.
:	ణ	System Safety Plan	/Per Proposel *	As Beguired	I	5 plus repre
. :	E7	Quality Progres Flan	Per Proposal MCP's	As Required	I	5 fles forse.
		•			•	
		•			•	
			······································		•	,
			Delivery ()ates		5
	<u>[two No</u> .	Iten Description		Dates Revisions	IZP:	5 Copies
•	<u>Itum No</u> . 05	Item Description A. Preliminary ADP	Delivery (Type II	
•	05	······································	Delivery p Initial 2 weeks prior to hardware			Copies
I	o5 06	A. Preliminary ADP B. Acceptance Data Package (ADP), including in part: Qualification Status and Equipment Log Failure Reports	Delivery I <u>Initial</u> 2 weeks prior to hardware delivery Concurrent with delivery of fit, hardware	<u>Revisions</u>	11	Copies 2 1 (with bdw.
I	o5 06	A. Preliminary ADP B. Acceptance Data Package (ADP), including in part: Qualification Status and Equipment Log	Delivery I Initial 2 weeks prior to hardware delivery Concurrent with delivery of fit.	<u>Revisions</u>	11	Copies 2 1 (with bdw.
I	o5	A. Preliminary ADP B. Acceptance Data Package (ADP), including in part: Qualification Status and Equipment Log Failure Reports	Delivery I Initial 2 weeks prior to hardware delivery Concurrent with delivery of fit. hardware Within 24 hrs.	<u>Revisions</u>	I	Copies 2 1 (with bdw.
1	o5 06	A. Preliminary ADP B. Acceptance Data Package (ADP), including in part: Qualification Status and Equipment Log Pailure Reports A. TWX C. Pailure Analysis and	Delivery I <u>Initial</u> 2 weeks prior to hardware delivery Concurrent with delivery of fit, hardware Within 24 hrs. of failure isolation	<u>Revisions</u> 5 days after revision 	II I I	Copies 2 1 (with bdw.
I	95 96 97 98	A. Preliminary ADP B. Acceptance Data Package (ADP), including in part: Qualification Status and Equipment Log Failure Reports A. TWX C. Failure Analysis and corrective action reports Failure Node, Effects and	Delivery I Initial 2 weeks prior to hardware delivery Concurrent with delivery of fit. hardware Within 24 hrs. of failure isolation As Required Submit 14 days	<u>Revisions</u> 5 days after revision 	II II II	Copies 2 1 (with hdw. 1 MSC
1	95 96 97 98	A. Preliminary ADP B. Acceptance Data Package (ADP), including in part: Qualification Status and Equipment Log Failure Reports A. TWX C. Failure Analysis and corrective action reports Failure Node, Effects and Criticality Analysis Final Single Point Failure	Delivery I <u>Initial</u> 2 weeks prior to hardware delivery Concurrent with delivery of flt. hardware Within 24 hrs. of failure isolation As Required Submit 14 days prior to CDR Concurrent with	<u>Revisions</u> 5 days after revision 	II I II II II	Copies 2 1 (with hdw. 1 MSC
I I I I I I I I I I I I I I I I I I I	95 96 97 98	A. Preliminary ADP B. Acceptance Data Package (ADP), including in part: Qualification Status and Equipment Log Failure Reports A. TWX C. Failure Analysis and corrective action reports Failure Node, Effects and Criticality Analysis Final Single Point Failure Summary	Delivery I Initial 2 weeks prior to hardware delivery Concurrent with delivery of fit, hardware Within 24 hrs. of failure isolation As Required Submit 14 days prior to CDR Concurrent with FMECA	Revisions 5 days after revision 5 days after revision 	II II II II II II	Copies 2 1 (with hdw. 1 MSC

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APPENDIX B

	Value Assessment	e Classification ssification	tive nplishment	lve	am :t	Justified Cost/Result	ASTP Documentation Requirements	Space Expmt Applicability	
Doc Iten	umentation	Type Class	Objective Accompli	Relative Cost	Program Impact	Justified Cost/Res	AST <i>P</i> Requir	Space Applic	Remarks
Pro	gram Management								
Man	agement Control Plan (MCP)	II	9	9	9	10 -	Yes	Yes	Submit with Proposal, no Update
MC	P Task Statement	II .	9	9	9	10	Yes	Yes.	Use as Major Contract Control. Submit with Proposal.
Cos	t Proposal	II ·	9	9	9	10	Yes	Yes	Submit with Proposal. Updated at Final Negotiations and by CCP's.
Mon	thly Letter Report	II	8	8	8	8	No	Yes	Reduce Frequency to Bi-Monthly or Quarterly informal Report.
Fina	ancial Management Reports	II	9	9	9	9	Yes	Yes	Reduce Frequency of Reporting.
Man	power/Overtime Reports	IÌ	7	8	7	7	Yes	No	
Tec Min	hnical and Management Review utes .	II	9	9	9	9	Yes	Yes	Reduce No. of Meetings and Participants.
Rev	iew Meeting Reports	II	7	7	7	7	No	No	Use Minutes as Only Document.
New	· Technology Reports	I	9	9	9	9	No	Yes	Define Areas. Submit Only Areas Verified.
Pho	tographic Documentation	II	9	5 .	9	7	No	No	Define Minimum Level.
Fun	ctional Flow Diagrams	п	8	7	8	7	No	No	
Fina	ll Program Reports	I	7	8	7	7	No	No	
Pro	gram Directives	III ·	10	10	10	10	No	Yes	

Value Assessment Documentation Item	Type Classification Classification	Objective Accomplishment	Relative Cost	Program Impact	Justified Cost/Result	ASTP Documentation Requirements	Space Expmt Applicability	Remarks
System and Design Engineering								
Specifications:						} .		
Contractor End Item Spec. (CEI)	I	8	8	9	8	Yes	Yes	Use CEI Spec. for all Design, Perform- ance, Interface, Construction, Verifica- tion, and Environmental Reqm'ts.
Interface Control Specs. (ICS)	I	8	8	8	8	No	No	Interface Reqm'ts per CEI Spec.
System Specs.	п	8	8	8	8	No	No	All System, Subsystem, and Component
Subsystem Specs. Component Specs.	II II	8	8 ⁻ 8	8 8	8 8	No No	No No	Specs, will be Defined by Experimenter and/or Contractor and not be Deliverable Documents.
Part/Device Specs.	III	9	8	9	9	No	Yes	As Applicable to Experimenter and/or Contractor Procurement Reqm'ts.
Training Model Specs.	ī	8	8	8	7	No	No	
Demonstration Model Spec.	11	8	8	8	7	No	No	
System Test Equipment Spec.	II	7	5	7	7	No	No	Interface Reqm'ts per CEI Spec.
Component Test Equip. Spec.	11	7	5	7	7	No	No	
GSE Spec.	11	7 .	5	7	7	No	No	
Special Fixture Specs.	11	7	5	7	7	No	No	· · · · · · · · · · · · · · · · · · ·
Shipping Container Specs.	11	7	5	7	7	No	No	
Human Factors Specs.	11	7	5	7	7	No	No	

Value Assessment Documentation Item	Type Classification Classification	Objective Accomplishment	Relative Cost	Program Impact	Justified Cost/Result	ASTP Documentation Requirements	Space Expmt Applicability	Remarks
Spec. Lists & Trees	III	8	7	8	8	No	No	
Structural/Thermal Specs.	п	8	7	8	8	No	No	
Material & Process Specs.	II	8	7	9	9	No	No	Defined by Program Reqm'ts and Con- tractor Approved Material.
Drawings:					• 1.			
Layouts Interface Assembly Details Subassemblies Specification Control Source Control Schematic Wiring List MFR Fixtures/Tools Trees Hardware Trees Design Analysis Reports	III I I III III	99999999999999999999999999999999999999	9 9 9 9 9 9 9 9 9 9 9 9	10 10 10 10 10 10 10 10 10 10 10 10	9 9 9 9 9 9 9 9 9 9 9 9 9 9 9 8	Yes Yes Yes Yes Yes Yes Yes Yes Yes Yes	Yes Yes Yes Yes Yes Yes Yes Yes Yes Yes	Contractor to Maintain As Built Configura- tion for each Model per Internal C/M System. Top Assembly, Interface, and Schematic Drawings Class I for Flight Configuration Only. No Deliverable Dwg. Pkg. Other Than Top Assy, ICD, and Schematics. Red-lining of Dwgs During Development Acceptable as Long as Con- figuration is Maintained. Parts & Ma- terials Used in Design Presented and Approved at CDR. All Non-preferred Parts & Materials Qual. Data Presented at CDR. No Formal Analysis Reporting Other Than
Technical Memorandum		0		0	ė			Provided in System Safety Assessment Report.
Interface Control Documents	I	8 8	8 7	8 8	8 9	Yes Yes	Yes Yes	Per Internal Requirements. Type I Controlled Document Approved at CDR.

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	Value Assessment Documentation Item	Type Classification Classification	Objéctive Accomplishment	Relative Cost	Program Impact	Justified Cost/Result	ASTP Documentation Requirements	Space Exprnt Applicability	Remarks
	Subcontractor Documents and Engineering Orders	III	8	9	8	8	Yes	Yes	Per Contractor C/M Reqm'ts. No De- liverable Documentation other than Class I Interface or Safety Documents.
	Measurement Reqm'ts Document	I	8	9	8	8	No	No	
	Design Certificate Review Report	п	8	9	8	8	No	No	
29	Subcontractor/Vendor Reports & Data	ш	8	9	8	8	No	No	
	Design Review Documentation	11	9	9	9	9	Yes	Yes	
	Contingency Procedures	I	8	8	9	8	No	No	Include in Flight Operations Plan, Ref. Engineering Support Documentation.
	Measurement/Command List	ľ	8	8	9	8	No	No	
	Calibration Data - Listings, Curves, Mag. Tapes and Defini- tion Documents	II	9	9	9	9	No	Yes	Only as Required to Document Information Reqd for Mission Operations. Where PI Provides Self Contained Recording Capa- bility, no Calibration or Software Provided
	Software Program Listings, Flow	III	8	9	8	8	No	No	
	Software Utilization Documents	111	7	5 [,]	7	7	No	No	
	Design Description Documents	III	7	5	7	7	No.	No	
	System Weight Reports	п	7·	5	8	7	No	No	
	System Power Reports	п	7	5	8	7	No	No	
	Engineering Test Reports	III	7	5	7	7	No	No	

Documentati Item	Value Assessment	Type Classification Classification	Objective Accomplishment	Relative Cost	Program impact	Justified Cost/Result	ASTP Documentation Requirements	Space Expmt Applicability	Remarks
Design Data	Books	III	8	<u> </u>	8	8	No	S 4 No	
	int. Documents	11	7	8	7	7	No	Νο	No Formal Ste Manual Recommended Inter- face Schematics and the Assembly Draw- ings Provided to Document Design and Configuration. Any Interfaces which Effect Safety will be Class I and Engineering Changes Maintained. Calibration and Acceptance of Test Equipment is Experi- menter's Responsibility. No Formal R&QA Records of STE Design or Operation Recommended.
Contact/Trip Manpower P Task and Ha		III II II	7	- 8	7	7	No	No	Only as Dictated by the PI or Contractor Internal Requirements per Financial Report Requirements as Negotiated for Financial and Program Requirements.
Engineering		ш	10	10	10	10	No	Yes	
	n Management								
Configuratio	n Management Plan	I	9	7	9	9 .	No	Yes	Submit with Proposal. Use Contractor Inhouse System Modified per RFP.
Configuratio	n Documentation Index	II	8	8	9	8	No	No	Use Contractor Inhouse System.
Specification	Identification Log	II	· 8	8	9	8	No	No	Use Contractor Inhouse System.
Configuration Procedure	n Identification	II	8	8	9	8	No	No	Use Contractor Inhouse Procedure.

	Value Assessment Documentation Item	Type Classification Classification	Objective Accomplishment	Relative Cost		Justified Cost/Result	ASTP Documentation Requirements	Space Exprnt Applicability	Remarks
	Configuration Control Procedure	II	8	8	10	8	No	No	Use Contractor Inhouse Procedure.
	Engineering Change Proposal (ECP)	I	9	9	10	9	Yes	Yes	Submit for Class I Interface and Safety Changes only Using Contractor System and Forms.
	Specification Change Log	II	8	7	8	9	No	No	Use Contractor Inhouse System.
	Specification Change Notice	II	8	9	9	- 8	Yes	Yes	For Type I Changes only.
2]	Configuration Identification Index	II	8	9	8.	8	No	No	Use Contractor Inhouse System.
	Request for ECP (RECP)	I	[.] 8	9	8	8	Yes	Yes	Customer Form. Improve Response Time & Deligate Lower Level of Approval.
	Deviations and Waivers	Ι.	8.	7	8	. 8	Yes	Yes	For Safety, Materials Interface only.
	Equipment Serialization & Tracking	п	8	7	8 -	8	No	Yes	S/N & Dash No. of Major Assemblies only.
	Baseline Data Package	II	8	7	8	8	Yes	Yes	Informal Data Packages Parts and Materials Formal Type I Presented at CDR.
	Interface Controls	I	8	8	8	8	Yes	Yes	Class I-ICD's only. All others Class II Drawings.
	Drawing Controls	11	8	8	8	8	Yes	Yes	
	Subcontractor/Vendor Controls	111	8	7	8	8	Yes	Yes	At Contractor Discretion.
	Change Control Board (CCB)	II	8	9	8	8	Yes	Yes	Convene for Class I Changes Only.

Value Assessment Documentation Item	Type Classification Classification	Objective Accomplishment	Relative Cost	Program Impact	Justified Cost/Result	ASTP Documentation Requirements	Space Expmt Applicability	Remarks
System Test								
Integrated System Test Plan	I	9	9	8	8	No	No	Submit Proposed Test Sequence and Levels with Proposal.
Qualification Test Plan	I	9	9	8	8	No	No	
Flight Acceptance Test Plan	I	8	9	8	. 8	No	No	
Spares Test Plan	I	8	9	8	8	No	No	•
End Item Test Plan	I	8	9	8	8	No	No	
Pre-Installation Test Procedures	п	7	8	7	7	Yes	No	
Qualification Test Procedure	I	9	8	9	9	Yes	Yes	Type II Procedures. Environmental and Test Requirements Provided to PI and Concurred upon for Qual. and Flight Tests.
Qualification Test Reports	11	8	8	8	9	Yes	Yes	As Run Copies of Test Procedures Records and Data.
Acceptance Test Procedures	I	9	8	9	9	Yes	Yes	Type II Procedures.
Acceptance Test Reports	II	8	8.	8	9	Yes	Yes	As Run Copies of Test Procedures and Data. ADP Includes Interface and Safety Test Results.
Test Equipment Documentation	11	8	8	8	8	No	No	
Pre- and Post-Test Meetings and Minutes	п	8	7	8	7	No	No	

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Value Assessment Documentation Item	Type Classification Classification	Objective Acconiplishment	Relative Cost	Program Impact	Justified Cost/Result	ASTP Documentation Requirements	Space Expmt Applicability	Remarks
Data Reduction Analysis	II	8	9	8	8	No	No	None other than Reqd by Contractor.
Qual Test Readiness Review (QTRR)	11	8	9	8	8	No	No	
Flight Test Readiness Review (FTRR)	II	8	9	8	8	No	No	
Qual Assessment Review (QAR)	11	8	9	8	8	No	No	
First Article Configuration Review (FACR)	п	. 8	9 ·	8	8	No	No	
Customer Acceptance Readiness Review (CARR)	п	8	9	8	9	No	Yes	
Reliability & Quality Assurance							[
Reliability Program Plan	I	9	9 [.]	9	9	Yes	Yes	
Quality Assurance Program Plan	I	9	9	9	`9	Yes	Yes	Use Contractor Inhouse System Modified per RFP.
Process Control Procedures	11 _.	9	8	9	8	No	No	Specify on Drawings.
Handling and Inspection Procedure:	II	9	8	9	8	No	No	Specify on Drawings.
Certification Test Procedures	II	8	9	8	8	No	No	
Certification Test Plan	п	8	9	8	8	No	No	
Certification Test Report	п	8	.9	8	8	No	No	
Certification of Flight Worthiness (COFW)	II	8	8	8	8	Yes	No	

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Value Assessment Documentation <u>Item</u>	Type Classification Classification	Objective Accomplishment	Relative Cost	Program Impact	Justified Cost/Result	ASTP Documentation Requirements	Space Expmt Applicability	Remarks
Installation Test Procedures	II	8	9	8	8	No	No	Submit Information Required Through KSC Ground Operation Plan.
Acceptance Data Package	I	9	9	9	9	Yes	Yes	Provide Major I/F Dimensional, Weight, Cleanliness Data. Also all Interface, Materials and Safety Deviations and/or Waivers.
Failure Mode and Effects Analysis (FMEA)	II	9	9	9	8	No	No	Provide Engineering Matrices as Required. Maintain and Track Failures/Problems which Relate to Safety or Interface Con- siderations. Analysis Presented at CDR.
Time/Cycle Control Procedure	II	8	9	8	8	No	No	
Time/Cycle Equipment Logs	11	8	9	8	8	No	No	
Time/Cycle Sensitive Component List	II	8	9	8	9	Yes	No	
Alerts and Response Reports	I	.8	9	8	8	No .	No	Supplied by Program for Material and Safety use.
EEE Parts List	п	9	9	9	8	Yes	Үез	Use Mil Standard Parts. Eliminate Screening. Standardize Parts. Provide Parts and Materials Data to Experi- menters/Contractors. Provide Materials List at CDR. All Non-Conformance need Qual Data.
Failure Notices and Reports	п	9	8	9	9	No	Yes	

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Value Assessment Documentation Item	Type Classification Classification	Objective Accomplishment	Relative Cost	Program Impact	Justified Cost/Result	ASTP Documentation Requirements	Space Expmt Applicability	Remarks
Subcontractor Parts List	ш	8	8	8	9	Yes	Yes	
Non Metalic Material List	п	9	9	9	9	Yes	Yes	
Parts Application Report	ш	8	9	8	8	No	No	
EEE Parts Derating Report	п	8	9	8	8	No	No	
EEE Parts Changes Substitutions Deviation	11	8	9	8	. 8	No	No	Submit with QA Plan and Acceptance Test Results.
Contamination Control Plan	п	8	9	8	8	No	No	
Failure Report Closure Plan	п	8	9	7	7	No	No	
Subcontractors Q/C Plan	m	8	8	8	8	No	No	
Worst Case Analysis	lii	8	10	8	8	No	No	
Mean Time Between Failure (MTBF) Analysis	11	8	9	8	8	No	No	
Material Review Board	II	8	10	8	8	No	No	No Formal MRB. Informal Tracking and Engineering Close Out of all Safety and Interface Problems/Failures.
Quality Assurance Instruction Report (QAIR)	11	8	9.	8	8	No	No	
Workmanship Traceability	п	8	9	8	8	No	Yes	For Critical Safety Related Hardware Only. Contractor to Define.
Parts Traceability	11	8	9	8	8	No	No	-

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	Value Assessment Documentation Item	Type Classification Classification	Objective Accomplishment	Relative Cost	Program Impact	Justified Cost/Result	ASTP Documentation Requirements	Space Expmt Applicability	Remarks
	Qualification Status List	II	7	8	7	7	No	No	
	System Support/Logistics								
	Operations Support Handbook	I	8	9	8	8	No	No	
	KSC Failure Reports	II	8	9	8	8	No	No	
2	Overall Flight Hardware Checkout Flow Plan	11	8	9	8	`8	No	No	
	Site Activation/Revalidation Schedule	11	• 8 .	9	8	8	No	No	
	Flight Hardware Schedule	111	8	9	9	8	No	Yes	Informal Schedule Maintained by Exp Supplier for Inputs to Overall KSC Test Integration Schedule.
	Preflight Operations Procedure	11	8	9 [.]	8	8 -	No	No	
	KSC Management Report	п	9	9	8	8	No	No	
	KSC Manpower Report	11	9	9	8	8	No	No	
	GSE Open Items Status Report	II	9	9	8	8	No	No	
	Test and Checkout Plan	II	9	9	8	8	No	No	
	Test and Checkout Procedure	II	8	9	9	8	Yes	Yes	System Safety Tests During KSC Spacecraft Integration Supported by Exp Field Team Combine with Ground Safety Plan.
	Hazard Analysis Report	п	8	9	9	8	Yes	No	

DOCUMENTATION MATRIX

Value Assessment Documentation Item	Type Classification Classification	Objective Accomplishment	Relative Cost	Program Impact	Justified Cost/Result	ASTP Documentation Requirements	Space Expmt Applicability	Remarks
System Safety Plan		8	9	9	8	Yes	Yes	An extensive Safety Plan should be Pro- vided for each Experiment and be Re- viewed and Approved by NASA for Hazards as Pressure, Pyrotechnics, Electrical Hazards, Static Change, High- Tension, Radiation, Chemical Caustic Toxic Substances, Combustibles Grounding, etc.
KSC Operations Plan	I.	8	9	9	8	No	Yes	Conducted by Training Office.
Crew/Missions Operations Analysis	п	8	9	8	8.	No	No	
Support Material List	II	8	9	8	8	No	No	
Storage Procedures	11	8	9	8	8	No	No	
Familiarization Manual	11	8	9	8	8	No	No	
Training Course Documents	п	8	9	8	8	No	No	
KSC Quality Program Plan	I	8	9	8	8	No	No	
Support Equipment Analysis	iı	8	9	8	8	No	No	
Logistics/Spares Plan	I	8	9	9	8	Yes	Yes	Spares and Logistics Requirements need Identification for Quantity and Delivery if Experiment is Flown Several Times or Where Inflight Repair is Required.
Mission Support Data Book	I	8	9	8	8	No	No	

DOCUMENT ION MATRIX

Value Assessment Documentation Item	Type Classification Classification	Objective Accomplishment	Relative Cost	Program Impact	Justified Cost/Result	ASTP Documentation Requirements	Space Expmt Applicability	Remarks
Operation and Maintenance Manual		8	9	9	8	Yes	Yes	Operation and Maintenance of Experiments should be Combined with Logistics/ Spare Plan if Experiment Operation or Repair is Conducted by Shuttle Crew. Degree of Detail will Depend on Extent of PI or Crew Involvement.
Manufacturing]]		
Make or Buy Plan	I	8	9	8	[•] 8	No	No	
Manufacturing Processes	11	8.	9	9	8	Yes	Yes	Class III as Defined by Contractor.
Manufacturing Program Plan	I	8	9	8	8	No	No	
Manufacturing Work Order Operation Sheets	ш	8	9	9	8	No	Yes	Per Contractor System.
Workmanship Standards	II	8	9	8	8	No	No	No Traceability Required other than Defined by Contractor for Critical Parts only.
Tooling, Jigs, and Fixture Documentation	II	8	9	8	8	No	No	Informal.
Program Schedules (Tier I, II, III, and IV)	п	8	9	8	8	No	No	Other than Required for Program Control.
Manufacturing Directives	III	8	9	9	8	No	Yes	
·		<u> </u>	·					<u> </u>

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APPENDIX C

Documentation Requirements for Space Experiment System Programs

DOCUMENTATION REQUIREMENTS EXAMPLE



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 - 7.7 Operations Hazard Analysis

8.0 ACCEPTANCE DATA PACKAGE (ADP)



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EXHIBIT C

Documentation Requirements for Space Experiment System Programs

DOCUMENTATION REQUIREMENTS

1.0 GENERAL

The documentation requirements defined in this Exhibit indicate the scope of documentation effort of this contract. Specific documentation delivery requirements are presented in Schedule III attached.

The documentation requirements specified shall not be altered as a result of a make-or-buy decision, e.g., the contractor shall be responsible for the items he makes as well as those he buys. A conflict exists between the requirements of this document and the referenced specifications or documents, the requirements of this Exhibit apply.

1.1 CLASSIFICATION

Data required shall be of three categories. Type I shall be submitted to NASA for approval. Implementation of Type I documentation shall not proceed until after: (1) approval by NASA, or (2) until 7 days after receipt by NASA for procedures and 20 days for reports and plans. NASA approval is considered to be granted if the contractor has not received written notice of disapproval and identification of specific deficiencies within 7 days for procedures and 20 days for reports and plans. Type II data shall be submitted for coordination, surveillance, information, review and/or management control. Type III data shall be retained by the contractor and submitted to NASA only upon request. Insofar as practicable, the contractor's own internal documents shall not be retyped and duplicated on more expensive paper prior to submission.

1.2 DATA IDENTIFICATION

All contractor documentation shall be organized into a series of numbered documents. All documents delivered, except drawings, shall be clearly marked with the paragraph number which requires such delivery. Type I documents shall be clearly marked "Preliminary - NASA Approval Pending" or "Approved by NASA" as



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appropriate. The number of copies required will be one reproducible copy for all Type I and II documents except punch cards, tapes, and drawings, plus two additional copies unless otherwise noted in Schedule III.

1.3 REVISIONS, AMENDMENTS, AND ADDITIONS

In preparing Type I and Type II documentation which will require periodic revision, the contractor shall prepare initial documentation using a refastening method so that pages may be deleted and/or inserted. When the original document is so prepared, the contractor need not submit the entire document but shall submit revised, amended or additional pages as appropriate. Accompanying these pages will be an instruction page detailing the exact means for effecting the revision or amendment. The provision of this paragraph does not apply to specifications, drawings, etc., which have an established procedure for the processing of amendments and revisions.

2.0 APPLICABLE DOCUMENTS

Documents referenced in this exhibit are of the issue in effect on the data of contract effectivity and form a part of this exhibit to the extent specified herein.

3.0 MANAGEMENT DOCUMENTS

Management documents are to be top level documents which consist of the Management Control Plans and Function Plans. The plan shall detail the tasks by which the contractor intends to comply with the statement of work. The plan shall include master phasing charts and milestone charts for the overall program.

3.1 MANAGEMENT CONTROL PLANS

The documents shall consist of a series of plans, the total of which amounts for the activities of the contractor, subcontractor, and personnel on the program.



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3.1.1 Financial Management Reports

A contractor Financial Management Report (NASA Form 533) shall be prepared each month.

3.1.2 Monthly Letter Progress Report

A monthly summary of schedule significant program progress and tasks shall be prepared.

3.2 COST PROPOSAL

The contractor shall prepare an integrated cost proposal for the implementation of this program and any modification thereto. The cost proposal shall be divided into the same organization units as the Management Control Plans and shall comply with standard Government policy.

- 4.0 FUNCTIONAL PLANS
- 4.1 INTEGRATED TEST PLAN

This document shall describe the hardware Qualification Test Program and the Flight Acceptance Test Program and shall be submitted in accordance with Schedule III.

4.2 RELIABILITY PLAN

Task statements included in the Management Control Plan:

- 4.2.1 Failure Reports on Qualification and Flight Models Shall Consist of:
 - (a) TWX notification to NASA
 - (b) Analysis and corrective action on non-GFE items.

4.2.2 Approved Materials List

The contractor and suppliers shall select parts, devices and materials for the contract hardware on the basis of suitability for their application(s). Initial selections may be based on good performance in



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prior comparable uses or its presence on an existing list, whether NASA-furnished or from another source. Whenever practicable, items selected shall be already qualified to pertinent specifications, and selection shall minimize the number of stypes of each generic type. When selecting items previously qualified, the contractor shall devote particular attention to currentness of data, applicability of basis of qualification and adequacy of specifications. The contractor and suppliers shall prepare and maintain project parts, devices, and materials lists for use in design of the contract hardware. Because these items are a limiting factor on the reliability of the design and hardware, every effort shall be made to select all necessary parts, devices, and materials as early in the project life as possible. The project lists should be complete (with the exception of a few items) and be submitted to the procuring NASA installation prior to detailed design of the hardware. After initial submittal, Contractor parts/ devices/materials lists shall be updated and submitted as specified in the contract.

4.3 QUALITY PLAN

Shall be defined in the basic contract and MCP tasks.

4.4 CONFIGURATION MANAGEMENT PLAN

Configuration Management as provided in the MCP tasks shall be implemented under this program.

5.0 SPECIFICATIONS

5.1 CFE SPECIFICATIONS

Using 6-section MIL-STD format as a guide, the contractor shall prepare CFE specifications for his hardware at the deliverable end item level. Equipment Specifications shall specify the detailed requirements of the particular equipment, identify the subsystem of which it is a part, and specify the quality and acceptance provisions designed to show that the requirements have been fulfilled.

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5.1.1 Interface Documentation

The preparation of interface documentation shall be the responsibility of the contractor. The contractor or other representative who is to provide the equipment on behalf of the Government shall co-sign the interface documentation prior to its submittal to NASA/MSC for approval. Interface differences between participants shall be resolved by NASA/MSC.

5.2 END ITEM SPECIFICATION

The contractor shall prepare specifications defining the technical requirements. In general, these specifications shall define such areas as functions, performance, design, configuration, interface, qualification, reliability and acceptance requirements for the module to be delivered.

5.3 TEST PROCEDURES AND SPECIFICATIONS

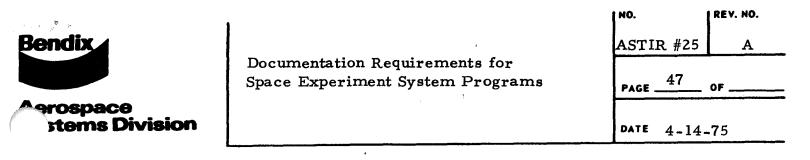
Test procedures shall be prepared for qualification and flight acceptance tests and will be utilized by test personnel to conduct such tests. As such, they will describe the step-by-step activities to be performed during the test operations. The activities will be listed in the sequence in which they are to be performed and keyed to a specific time reference. All safety or emergency procedures will be detailed for each hazardous condition.

5.3.1 Quality Test and Inspection Procedures

The contractor shall prepare test and inspection procedures in accordance with contractor standards for good practices.

5.3.2 Process Control Procedures

The contractor shall provide process control procedures in accordance with contractor standards for good practice.



5.3.3 Recommended Storage Procedures for End Items

The contractor shall provide adequate storage procedures. These procedures shall assure suitable protection against deterioration and damage for both short term and long term (in excess of two years) storage periods. Special handling and preventive maintenance considerations shall be included as necessary.

5.4 SYSTEM QUALIFICATION TEST REPORT

The contractor shall prepare a Qualification Test Report covering all such tests defined by the program.

5.5 FLIGHT ACCEPTANCE TEST REPORT

The contractor shall prepare a Flight Acceptance Test Report consisting of the following:

- 1) Pre-test meeting minutes (if applicable)
- 2) As-run procedure including DR's
- 3) Post test meeting minutes (if applicable)
- 4) Documentation of closing action items status.

6.0 ENGINEERING REPORTS AND DATA

6.1 TECHNICAL DATA, REPORTS AND ANALYSES

The contractor shall prepare technical reports which describe the studies, analyses, and results of the contractual effort. The reports shall be prepared at times when complete blocks of work have been accomplished, and if appropriate, as logical subdivisions thereof. Major technical areas shall not be combined in a single document, but shall be published individually. Format is contractors option.

6.2 DESIGN INFORMATION

The contractor shall submit preliminary design information to assist in expediting the interchange of design data and to keep NASA continually and currently appraised of the contractor's activities, philosophy, approaches, solutions, and design evaluations. See 6.1 for data media.



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6.2.1 Design Review Minutes

Minutes shall be prepared of the results of the Design Review. Action items incorporated shall include schedule dates.

6.3 SUBCONTRACTOR DRAWINGS AND ENGINEERING ORDERS

The contractor shall maintain a complete, up-to-date set of all drawings sufficient to describe each of the equipments, including those of his vendors, for which he is responsible. The contractor drawings shall be Type III. Drawings of non-deliverable modules or in-house equipment such as mockups, etc., shall be type III. These drawings shall be prepared using the contractor's internal drawing system, and shall conform to high professional standards.

6.4 SPECIFICATIONS/SOURCE CONTROL DRAWINGS

Contractor procured hardware shall not require submission of engineering drawings for suppliers. The contractor, in order to exercise control of suppliers, will prepare specification/source control drawings, as applicable, and will make these documents available to NASA upon request.

7.0 OTHER DOCUMENTATION

7.1 OPERATIONAL DATA BOOK INPUT

A Spacecraft Operational Data Book addendum shall be written and prepared to define configuration, operational data, system constraints and limitations and system command descriptions. This Type I data will be published and distributed by NASA for use in lunar operations. Inputs supplied by the Contractor will be published and distributed by NASA.

7.2 CONTRACTOR'S PARTS LIST

The contractor shall prepare a list of parts and materials selected and submit these data to scheduled design reviews.



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7.3 MINUTES OF MEETINGS

Meetings - NASA shall be notified 3 days in advance of meetings with other contractors, government agencies, and participants in the Program. Minutes (including all agreements and action items) of meetings shall be written by the contractor and the draft signed by representatives of the organizations involved prior to departure of the meeting parties. The minutes shall be typed and forwarded to all attendees and various designated NASA addresses.

7.4 MEASUREMENTS REQUIREMENTS DOCUMENT

Provide NASA data survey parameters applicable to each measurement point in tabular form to permit evaluation of the validity of each measurement.

7.5 GROUND SAFETY PLAN

The Ground Safety Plan shall include the purpose and description of the hardware. It shall include details of equipment design with special emphasis on the safety features. It shall also describe ordnance items, technical data sheets, and the KSC flight systems operations and safety management program precautions which apply.

7.6 OPERATIONS PLAN INPUT DATA

The Operations data shall define the operating sequences for the operational hardware. It shall contain the time lines for each experiment turn-on, and operations procedures. The operations data provides the guidelines for the development of the Science Operation Support Plan.

7.7 OPERATIONS HAZARD ANALYSIS

The details of the report shall be consistent with the complexity and inherent hazard potential of the hardware during test, checkout, and support for maintenance, training, simulation, and operations.



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8.0 ACCEPTANCE DATA PACKAGE (ADEP)

ADP contents to be as specified in the basic contract Exhibits. Two (2) copies of the ADP for each end item will be delivered with each Flight Model. NASA review and approval of each ADP is required prior to NASA final acceptance of the end item.

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SCHEDULE III

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DOCUMENTATION SCHEDULE

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Item No.	Item Description	<u>Delivery De</u> Initial	Revisions	Type	Copies
Part A	Program Control Data				
Al	Management Control Plan	w/Proposal	2 weeks after contract award; thereafter 5 days after revision	ц	4
A2	Cost Proposal	w/Proposal		п	4
A 3	Financial Mgt. Reports	. •			
	b. Quarterly (533)			ш	10
A4	Monthly Letter	•	Bi-Monthly	ш	15
A5	Minutes, Reviews*	5 days after review	As Required	п	4
A 6	New Technology Report	As Required		I	5
Part B	Design/Systems Engineering D	ata			
Bl	Contract End Item (CEI) Specification	TBD	5 days after revision	I	7
B2	Drawings	As Requested	•-	ш	1.
B 3	Interface Control Documenta- tion	Submit 14 days prior to CDR	5 days after revision	I	3
B4	Subcontractor Drawings and Engineering Orders	As Requested		ш	l plus l repro (sepia)
B5	Specification and Source Control Drawings	As Requested		ш	l plus l repro (sepia)
B6	Operations Plan Input	TBD	5 days after revision	1	25
B7	Spacecraft Operations Data Book Inputs	30 days prior to delivery of flt. sys. delivery	Revisions as Required	I .	10, inc. 1 repro
B8	Design Review Minutes (PDR's and CDR's)	14 days after design review	Chit close out forms, after signed, by MSC representative	п	10
Part C	Testing Data .				
Cl	Integrated System Test Plan (System Level)	60 days prior to test	As Required	I .	4
C2	Qualification Test Procedure	60 days prior to test	5 days after revision	I	4
C3	Flight Acceptance Test Procedure	60 days prior to test	5 days after revision	I	4

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Delivery Dates					
Item No.	Item Description	Initial	Revisions	Type	Copies
C4	Level A Spares Test Plan	60 days prior to test	5 days after revision	I	4
C5	Test Procedures and Specifi- cations-Revised	20 days prior to each test	5 days after revision	I or II per Table I	Type I-3 to MSC rep., Type II-1 to M3C rep.
C6	Acceptance and Qualification Test Reports and Data Tapes	4 weeks after test completion; data tapes as requested	Test report addenda published as required	II	3 to MSC rep., Orig- inal Mag. Tape to MSC
Part D	Reliability and Quality Control	Data			,
DI	Contactor Parts List	Preliminary As Available	Final: Concurrent with delivery of flight hardware	II	4
D2	Subcontractor Parts List	Concurrent with subcontractor delivery of flt. hardware		ш	1
D3	A. Preliminary ADP	2 weeks prior to hardware delivery	,	п	2
•	B. Acceptance Data Pack- age (ADP), including in part; Qualification Status and Equipment Log	Concurrent with delivery of flt. hardware	5 days after revision	I	l (with hdw. l MSC
D4	Failure Reports				
	A. TWX	Within 24 hrs. of failure isolation		Ш	4
	B. Failure Analysis and corrective action re- ports	As Required	5 days after revision	ш	4
D5	Acceptable Parts List	At CDR		II	4
D6	Approved Materials	14 days prior to CDR		II	4
Part E	System Support/Logistics Data				
El	Storage Procedures for End Items (FPHGGI) Revised TM's wherever feasible	20 days prior to delivery of flt. hardware	5 days after revision	п	4 (KSC) BxA Site Office 1 RALPO
E2	Manual Updating				
E3	Ground Safety Plan A. Draft	90 days prior to flight hdw. delivery		II	5
	B. Final	Concurrent with flight hardware		11 2	20
E4	System Safety Plan	Per Proposal	As Required	I	5 plus repro.
E5	Quality Program Plan	Per Proposal MCP's	As Required	I	5 plus repro.