

# Radioisotope Power Systems: Pu-238 and ASRG status and the way forward

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**10<sup>th</sup> Meeting of the  
NASA Small Bodies  
Assessment Group**

*8 January 2014*  
*Capitol Ballrooms A and B*  
*Embassy Suites Washington DC*  
*Convention Center Hotel*

**3:45 PM - 4:00 PM EST**

**Outer Planet  
Assessment Group  
Meeting**

*13 January 2014*  
*Drake Building*  
*University of Arizona*  
*Tucson, AZ*

**10:40 AM - 11:00 AM**

**APL**

*The Johns Hopkins University*  
**APPLIED PHYSICS LABORATORY**

# RPS use and infrastructure costs are still emerging from the Cold War years



- **Radioisotope Power Systems (RPS) are an enabling technology for providing power to satellite systems in cases for which solar power is impractical or absent altogether**
  - They have been used in space as well other applications, in the U.S. and in Russia
  - Many other applications have been phased out
- **Their technical origins stretch back to research on the Manhattan Project**
- **They were invented in the U.S. about 55 years ago and we have invested ~\$4.7 billion (FY2011) to date in perfecting this technology**
- **There are also in lightweight radioisotope heater units (LWRHUs) used to keep spacecraft components warm**

**First use: Transit 4A in 1961**



- **Bench check out and installation of the SNAP 3B7 radioisotope power supply**
- **Launch on Thor Able-Star 29 June 1961**

# Origin of RPSs in the U.S. was with Po-210 fuel



- **Research began at Mound Facility in Miamisburg, Ohio**
  - Operated from 1948 to 2003
  - 182 acres
- **Polonium-210 was investigated as an intense source of alpha particles beginning in 1942**
  - 1954 – program to generate electricity from Po-210
  - 1956 - conceptual design using a mercury boiler
  - 1958 - RTG powered by polonium-210
- **Po-210**
  - 120 watts per gram
  - Half-life of 138 days limited usefulness for space probe missions
  - Research and production at Mound phased out in 1971
- **Gadolinium polonide (GdPo) developed as fuel**





# Switch from Po-210 to Pu-238 for Long-Lived Missions

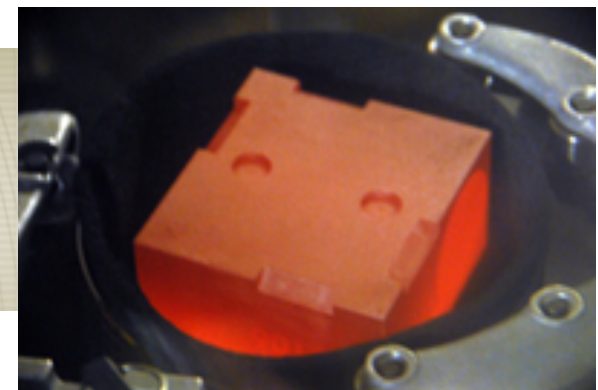
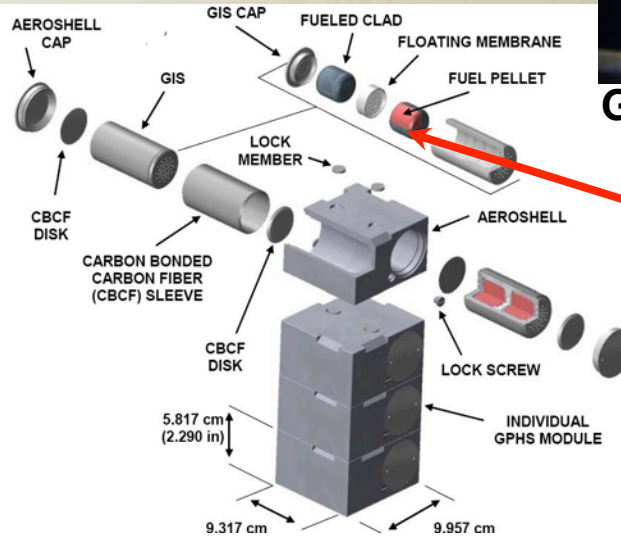


- Mid 1950s – Plutonium-238 research and development activity began at Mound
- 1959 – Initial research concerning plutonium-238 was transferred to Mound from Lawrence Livermore National Laboratory
- 1960 – First reduction of metallic plutonium-238 achieved at Mound Research and development relating to the application of plutonium-238 as a radioisotopic heat source material followed
  - Materials research
  - Development of processes for the production of heat source materials
  - Development of fabrication and metallurgical technology to ensure the containment and stability of heat source materials
  - Research and development activities were on the design of RTG systems for the various applications of this technology

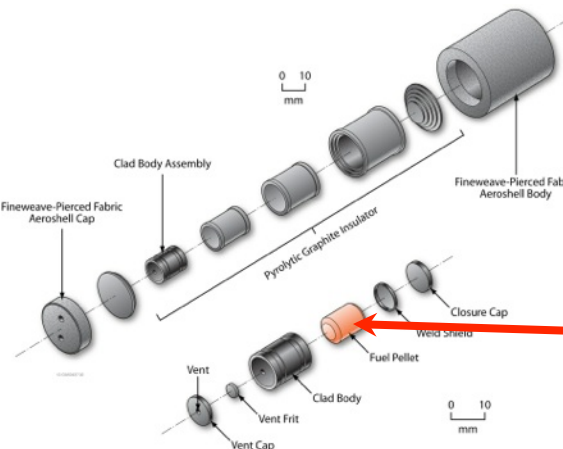
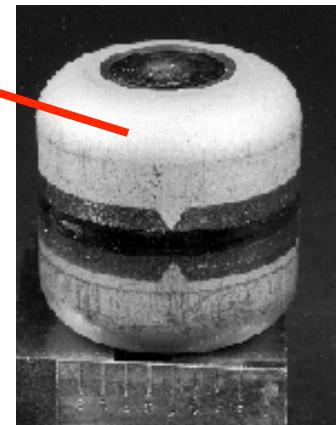
# Pu-238 usage in space – U.S. standard packaging **is a given**

Usage has been standardized largely due to rigorous and comprehensive safety analyses

- **Power:** General Purpose Heat Source (GPHS)  
Step-2, each containing pellets of Pu-238 in the chemical form  $\text{PuO}_2$  (nominal 150 g)
- **Heating:** Light Weight Radioisotope Heating Unit (LWHRU), each containing 1 pellet of Pu-238 in the chemical form  $\text{PuO}_2$  (nominal 2.7 g)



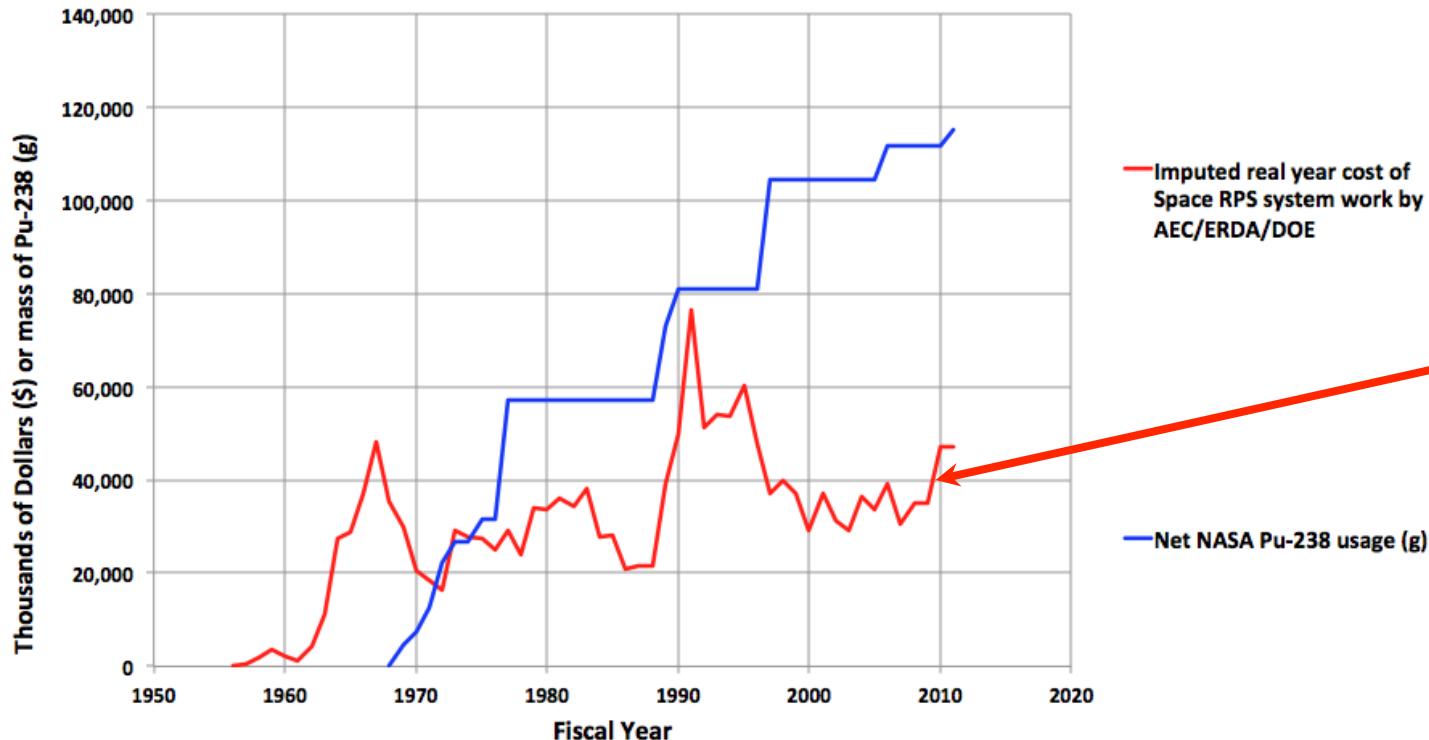
GPHS for Curiosity (from INL)



# Pu-238 usage in space – Quantity



AEC / ERDA / DOE Outlays for Space RPS and NASA Pu-238 Usage



- No other isotope has been used by the U.S. to power spacecraft

N.B. The costs directly supplied by DOD and NASA to these programs are **\*not\*** captured in these numbers

NASA usage: Nimbus B-1 through Curiosity 115 kg in 44 years = 2.6 kg/yr on average

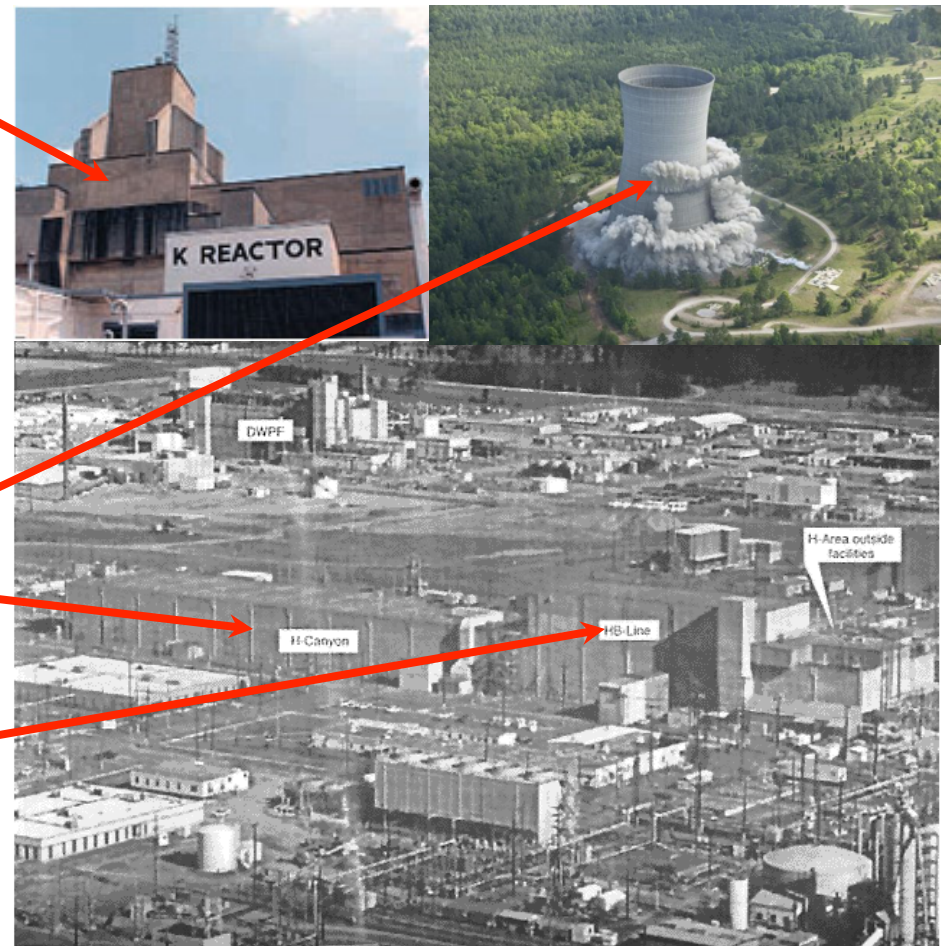
Other U.S. spacecraft have also used Pu-238



# Production and separation of Pu-238 were carried out at the Savannah River facility in South Carolina – Industrial Scale



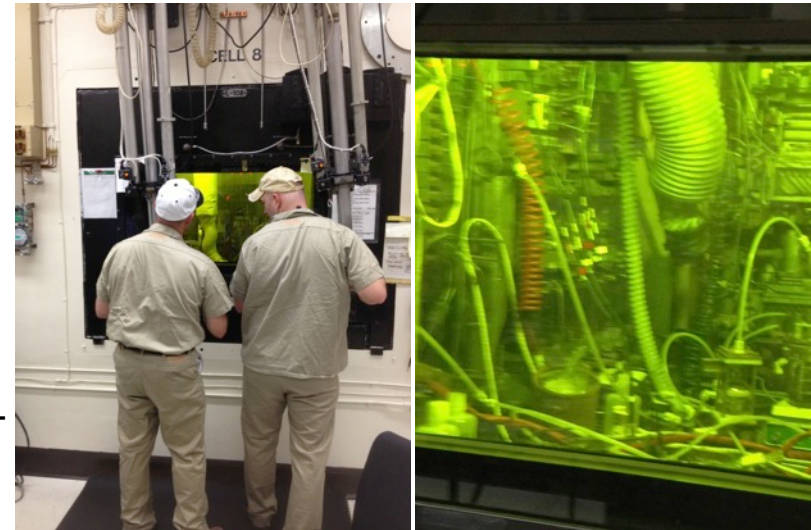
- **K-reactor used for production**
  - First went critical in 1954
  - To inactive status in 1988
  - Cooling tower built 1990
  - Operated with cooling tower in 1992
  - On cold standby 1993
  - Shutdown 1996
  - Reactor building converted to storage facility 2000
  - Cooling tower demolished 2010
- **H-canyon used for fuel reprocessing**
  - Only hardened nuclear chemical separations plant still in operation in the U.S.
  - Radioactive operations begin in 1955
- **HB-line**
  - Production begins of Pu-238 for NASA use 1985
- **~300 kg of Pu-238 produced 1959-1988**



# New Pu-238 Supply Project for NASA is more modest



- Production is targeted at ~1.5 kg “plutonium product” per year
- Facilities used include
  - Idaho National Laboratory (INL) – storage of  $\text{NpO}_2$  and irradiation of targets at ATR (see below)
  - Oak Ridge National Laboratory (ORNL)
    - Remove Pa-233 (312 keV  $\gamma$ -ray is worker-dose issue)
    - Fabricate reactor targets
    - Irradiate at High Flux Intensity Reactor (HFIR) – or ship to INL for irradiation at the Advanced Test Reactor (ATR) –
    - Process in hot cells at ORNL Radiochemical Engineering Development Center (REDC)
    - Remove and purify Pu; change to oxide; and do O-16 exchange for processing by Los Alamos National Laboratory (LANL) into fuel pellets for GPHSs or LWRHUs

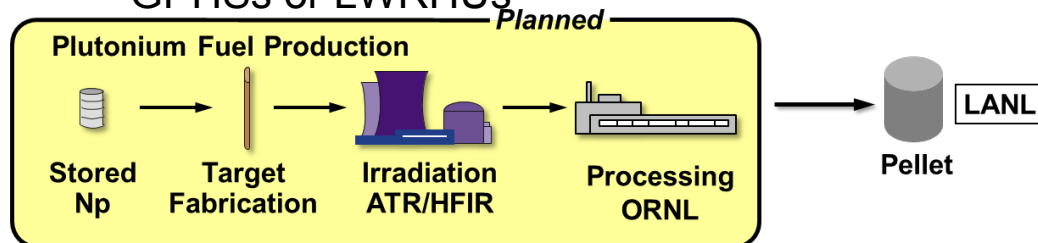


Hot Cell at ORNL REDC

10% conversion per campaign – to limit Pu-239 production

100 target per campaign to make 300 to 400 g of plutonium product

“Plutonium product” is **NOT** the same as Pu-238





# Nuclear Isotope Production Issues (Physics)



- When producing isotopes in a reactor, multiple channels as dictated by nuclear physics come into play – so no product is “clean”
- Only chemical processing of plutonium is “practical” – isotopic separation is not
- Once made, all isotopes begin decaying at physics-dictated rates and sometimes producing new radiological hazards
- Typical Pu-238 production at Savannah River – once reprocessed (Rinehart, 2001)
- The only “controls” are
  - Initial target composition
  - Reactor and target geometry
  - Exposure time
- Particular hazards in making Pu-238:
  - Protactinium-233 (Pa-233) – 312 keV  $\gamma$ , mitigate by chemical cleanup of Np-237 after removal from storage
  - Thallium-208 (Tl-208) – 2.61 MeV  $\gamma$ ; mitigate by minimizing Pu-236

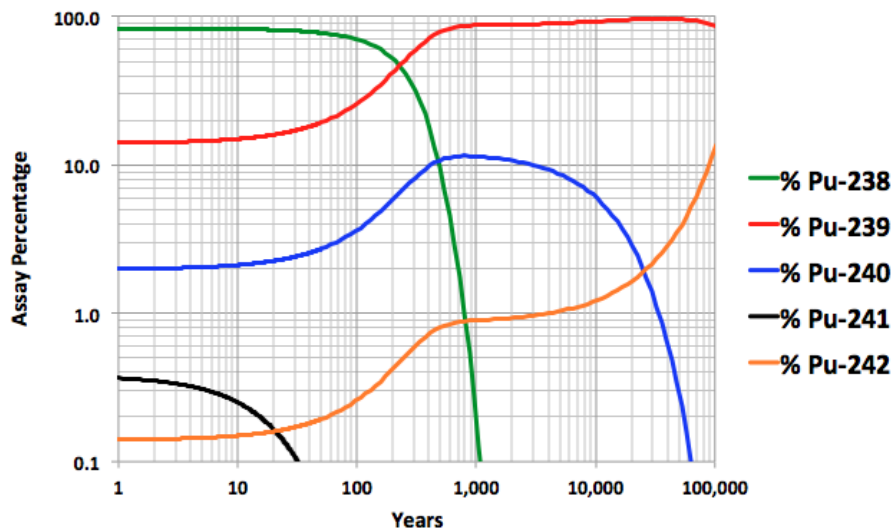
Isotope	Mass %
Pu-236	$\leq 1 \mu\text{g} / \text{g}$
<b>Pu-238</b>	<b>83.50</b>
<b>Pu-239</b>	<b>14.01</b>
Pu-240	1.98
Pu-241	0.37
Pu-242	0.14

# Older Fuel has less power density



- Pu-239 in particular decays less slowly than Pu-238
- Once the Pu is produced, the initial fractions are “frozen in”
- As the fuel ages, the relative fraction of Pu-238 decreases **and that cannot be changed**

Time Variation of Nominal Pu-238 Production Assay



GPFS fuel clad design is driven by metallurgy of the iridium alloy of the clads

Nominal “plutonium product” loading is 150 g

Design thermal output is 62.5 W

→  $62.5 \text{ W} / 150 \text{ g} = 0.42 \text{ W/g}$

Pu-238 isotope produces 0.56 W/g

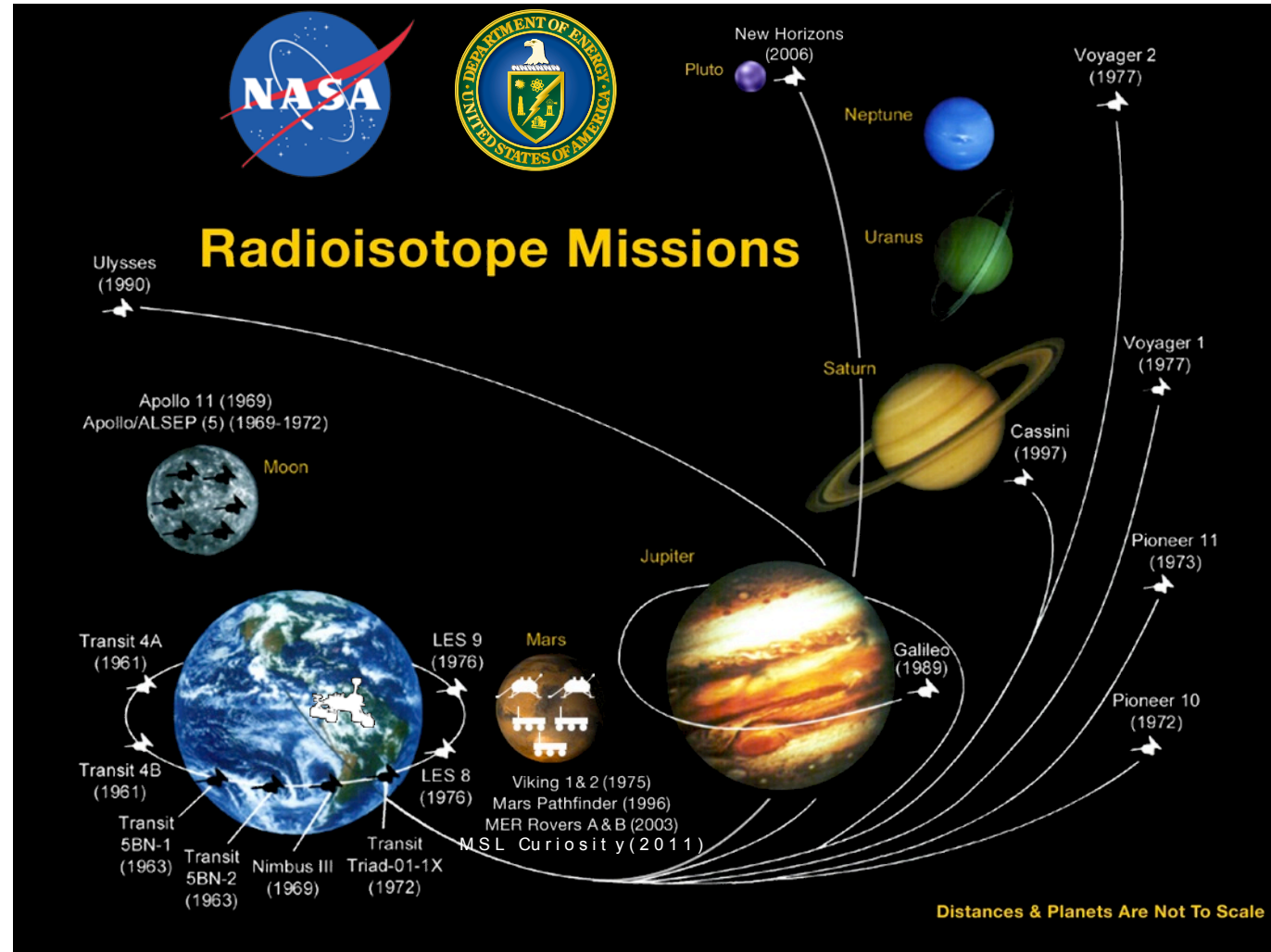
Hence, a fuel clad contains **roughly**  $0.42/0.56 \times 150 \text{ g} \sim 110 \text{ g}$  of Pu-238 isotope

**Details matter – this is the maximum thermal power available**

# U.S. RPS Missions



- The United States has launched 46 RTGs on 27 missions
- 35 RTGs have been used on 18 NASA missions
- No mission has failed due to an RTG



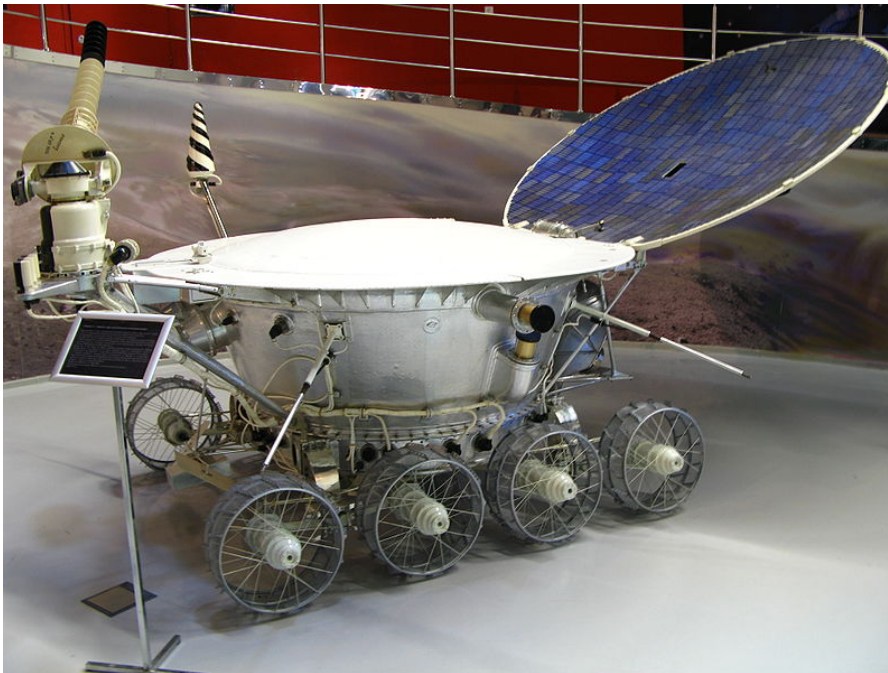


# Russian RPS Missions



- Lunokhod 1 and 2 (Yttrium polonide using Po-210)
- Mars – 96 (“Angel” RHU and RTG using Pu-238)

**RHUs ensure survival during lunar night and provide compact heater and power sources for small autonomous stations (SAS) and penetrators on planetary probes**



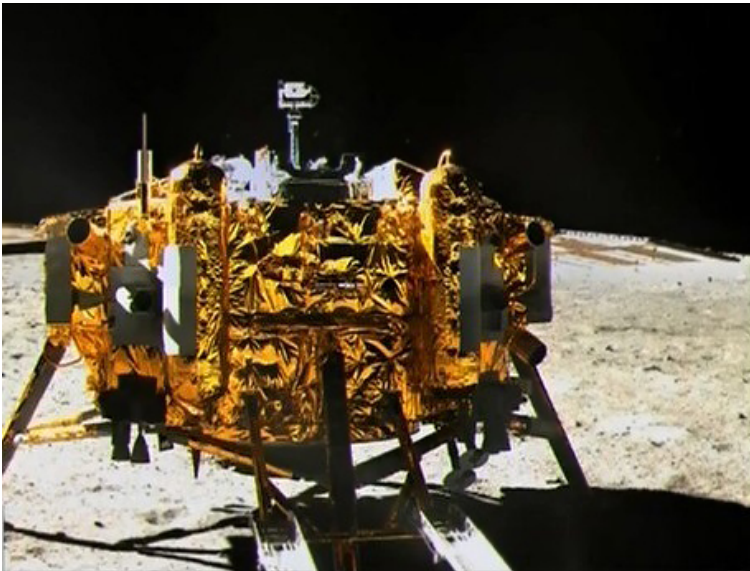
**8.5 W<sub>th</sub> and  
200 mW<sub>e</sub>  
«Angel»  
RHU and  
RTG  
employed on  
Mars-96**

# Chinese RPS Missions



- Chang'e-3 and Yutu (Pu-238 RHUs)
- Lunar Lander and Rover

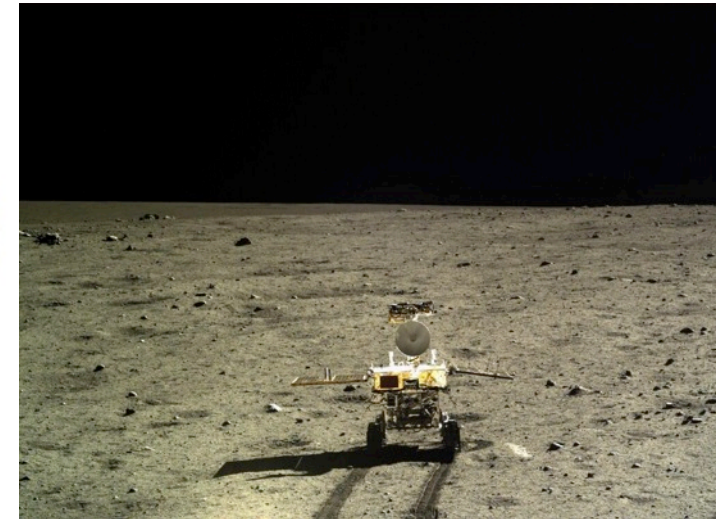
**RHUs ensure survival during lunar night**



**Chang'e-3 lander from Yutu rover**



**RHU with APXS on Yutu –**



**Yutu rover from Chang'e-3 lander**

image credited to CLEP at 2011-13  
[www.spaceflight101.com](http://www.spaceflight101.com) - Patrick Blau

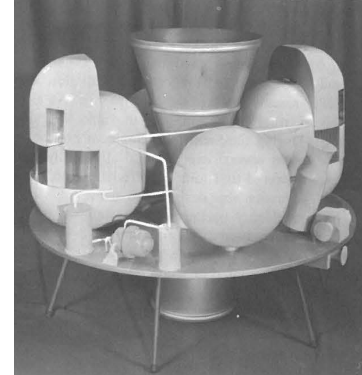


# Convertor Technologies Have Proven Difficult to Develop



- **Requirements are high reliability and high thermal-to-electrical energy conversion**
  - In the U.S. emergence of thermoelectric materials were chosen over dynamic systems (Rankine - cycle mercury boiler was baselined for SNAP-1) for reliability
  - PbTe and TAGS materials followed by higher efficiencies with SiGe couples operating at higher temperatures
- **Other approaches were abandoned due to material difficulties**
  - Selenide thermoelectrics
  - Alkali metal thermal-to-electric converter (AMTEC)
- **Still other approaches continue to show promise, but need larger infusions of research funds to further the technical readiness level of the the technology**
  - Skutterudites and other materials
  - Advanced Stirling Radioisotope Generator (ASRG) has been the most promising dynamic system to date

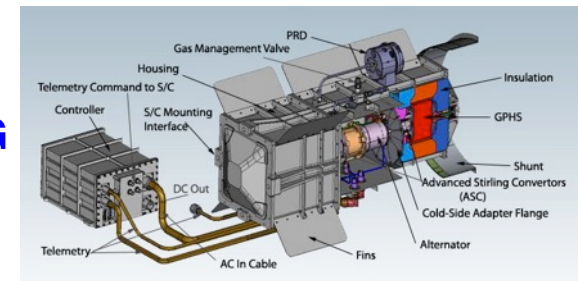
**SNAP – 1  
concept**



**AMTEC  
cell**



**ASRG**

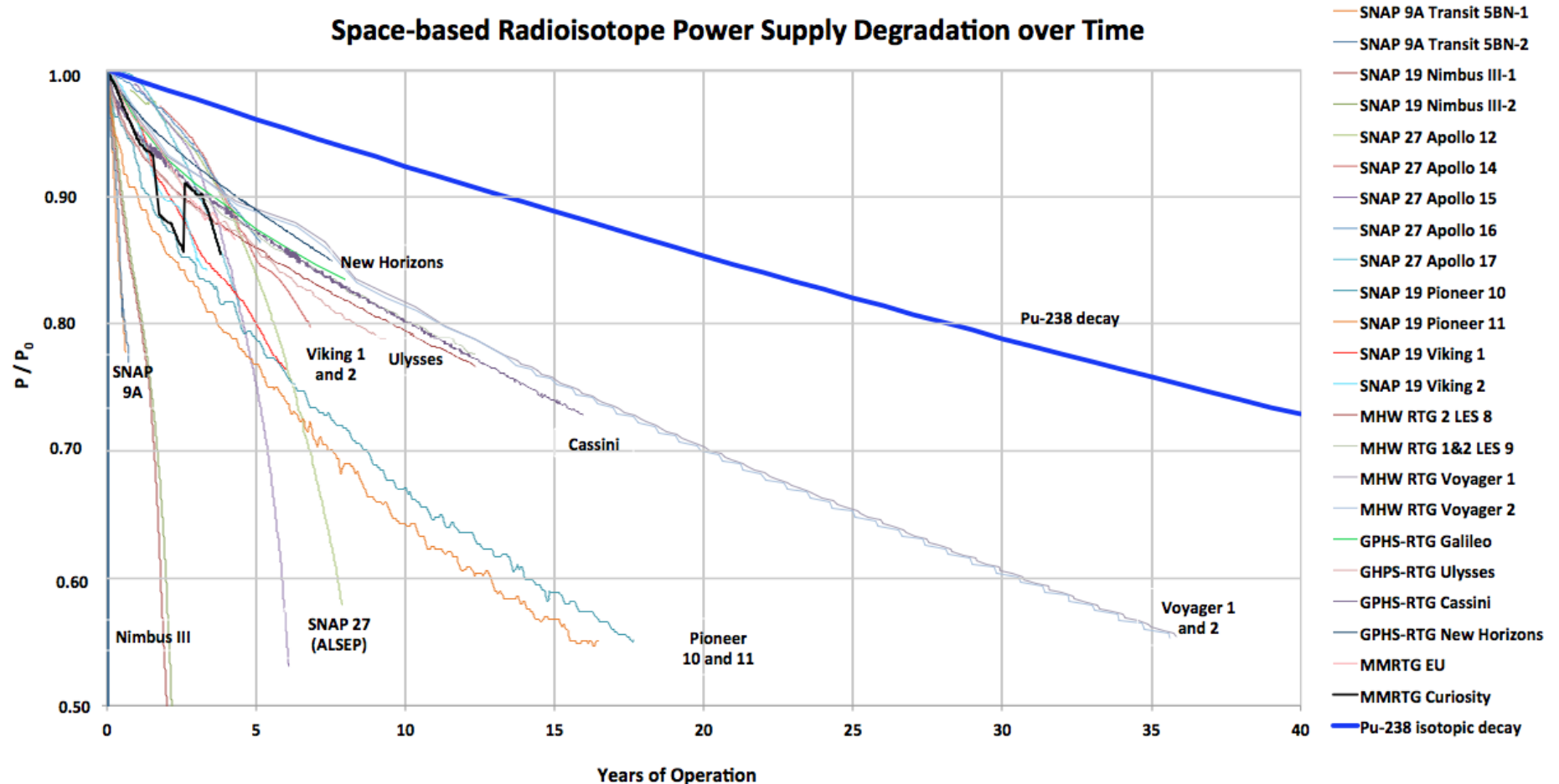




# Long-lasting Electrical Power – with No Maintenance



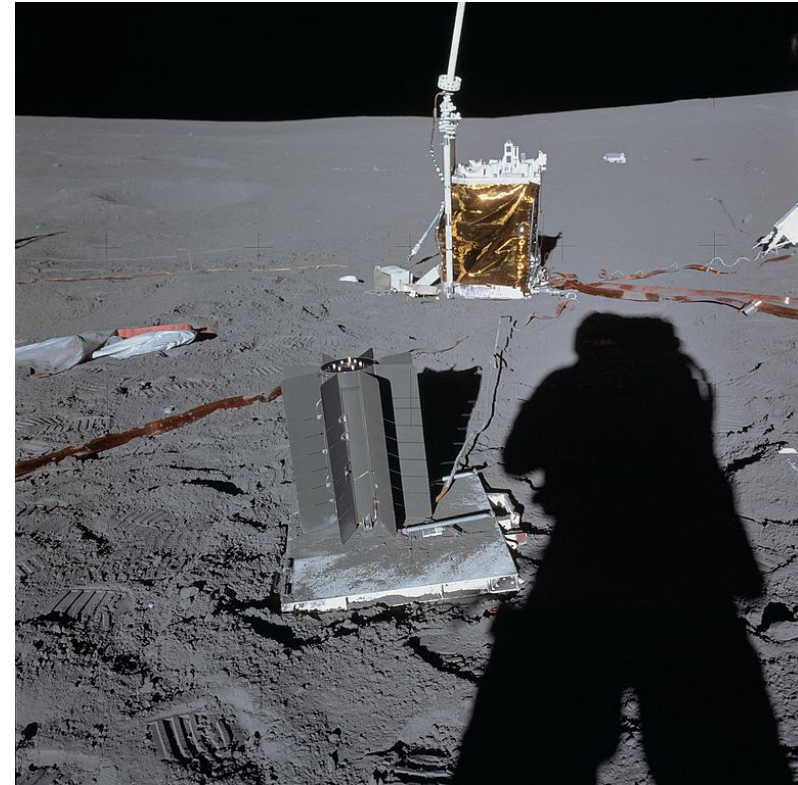
Space-based Radioisotope Power Supply Degradation over Time



# Missions Enabled: Long-Term Lunar Presence

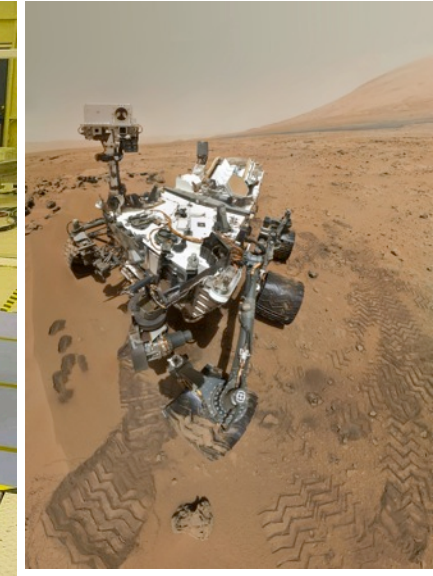
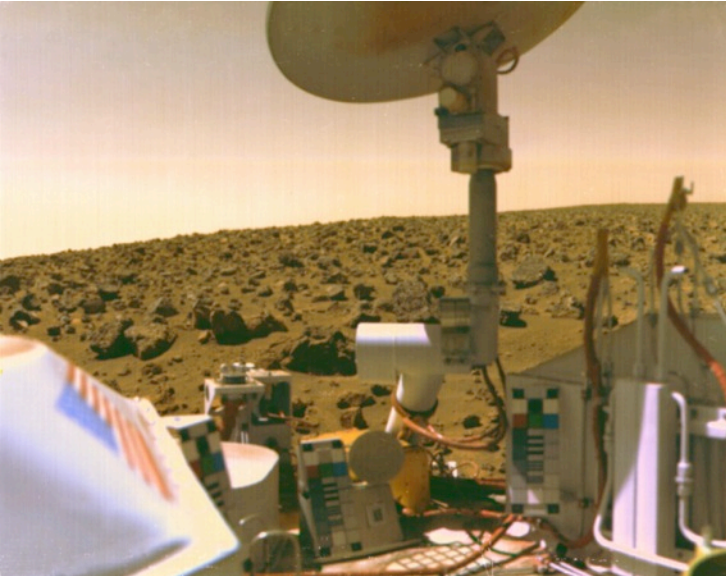


- **Surveyor was originally planned to employ RTGs so as to survive the lunar night**
  - The SNAP 11 was to use Curium-242 to allow the spacecraft to function for 130 days
  - Dropped due to cost
- **The Apollo Lunar Surface Experiment Package (ALSEP) was deployed on Apollo 12, 14, 15, 16, and 17**
  - The SNAP 27 used Plutonium-238
  - Assembly by an astronaut was required following landing
  - The units were turned off long after the last landing due to cost constraints (30 Sep 1977)



**ALSEP and SNAP 27 deployed on Apollo 14**

# Missions Enabled: The surface of Mars



**SNAP 19 RTGs  
for power:**

**Viking 1 and 2  
landers**

**RHUs for  
warmth:**

**Sojourner, Spirit,  
and Opportunity**

**MMRTG for  
mobility:**

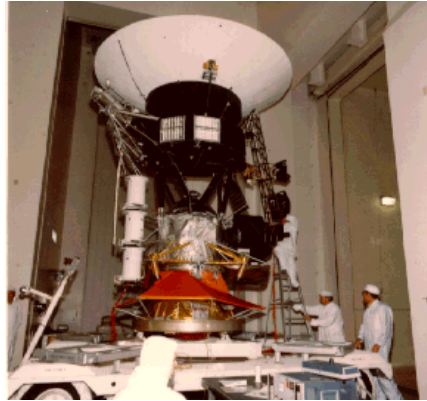
**Curiosity**



# Missions Enabled: The outer solar system ...and beyond



- Multi-hundred watt (MHW) RTGs systems and evolution to GPHS-RTGs



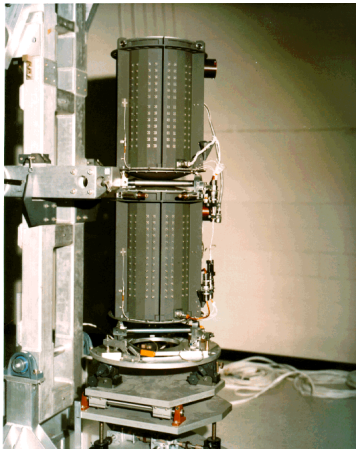
Voyager 1 and 2



Galileo



Ulysses w/ IUS



MHW RTGs for Voyager



Cassini GPHS RTGs



New Horizons



Cassini-Huygens

# Current (January 2014) Operations and Plans



- The President's proposed FY 2014 budget shifts fiscal responsibility and target budget for maintenance of NASA-required DOE infrastructure to NASA
- To improve transparency on DOE's planning basis to support NASA's mission DOE established in July 2013 an allocation of 35 kg of Pu-238 for Civil Space (NASA) use including both older U.S. supplies and previously purchased supplies from the Russian government
- In September 2013 NASA has deferred flight development of the ASRG
- Beginning in FY 2012 the Plutonium-238 Supply Project began at Oak Ridge National Laboratory to produce an average ~1 kg/yr of Pu-238 isotope (1.5 kg of PuO<sub>2</sub> product) by 2021
  - This effort is currently in a technology demonstration phase
- Any RPS-enabled flights for the next decade will use the flight-qualified MMRTG, as is the Mars 2020 mission – the only such future mission currently in Phase A study by NASA