DAWN AT CERES

Vesta: 2011-2012


Presentation to the 9th Small Bodies Assessment Group Meeting
Washington, DC
11 July 2013
Ceres – A Unique Asteroid

...and a dwarf planet

- Twice the size of Enceladus (~475 km radius)
- Surface temperature > 200K
- Hydrostatic shape
- No craters > 100 km
- Density: 2088 kg/m³
- Rock/Ice mass fraction: ~70/30
- Surface dominated by brucite \([\text{Mg(OH)}_2]\) and carbonates (Milliken and Rivkin 2009)

There are no meteorites that are linked to Ceres
Only telescopic data are available to understand Ceres’ composition

Ceres is the third solar system object after Earth and Mars, on which carbonates have been observed.
Planning Constraints

• Loss of second reaction wheel at the end of the Vesta mission
  – Resulted in the need for a plan that could achieve the science objectives using only jet control (hydrazine reaction control system)
  – Hybrid control using jets with remaining two wheels will be used to extend operational lifetime
  – Hydrazine and jet pulses became the limiting consumable

• Hydrazine conservation is key to maximizing science return
  – Studied and implemented hydrazine conservation measures resulting in increase in hydrazine available for science operations
    • Reduced number of turns, slower turns
    • Pointing control deadbands tuned for observing conditions
  – Performed analyses and simulations to ensure efficient safe operation under hybrid control
Ceres Planning Status

• Concept study completed in December
  – Demonstrated a viable Ceres mission concept that met L1 Requirements within initial first-order hydrazine constraints

• Science Orbit Requirements completed
  – Includes Survey, High-Altitude and Low-Altitude orbits similar to Vesta
  – Low-Altitude Mapping Orbit (LAMO) has equivalent geometry in terms of solid angle to Vesta
  – Detailed navigation studies are underway

• Science Plan is being elaborated
  – Initial draft of activity-level time-ordered listing have been generated consistent with the L1 requirements, science objectives and higher-fidelity hydrazine constraints
## Ceres Science Plan Overview

<table>
<thead>
<tr>
<th>Phase</th>
<th>Radius*</th>
<th>FC Res</th>
<th>Observations</th>
</tr>
</thead>
<tbody>
<tr>
<td>Approach</td>
<td>163,000 km</td>
<td>15 km</td>
<td>OpNavs (9) FC &amp; VIR</td>
</tr>
<tr>
<td></td>
<td>80,000 km</td>
<td>7.4 km</td>
<td>RC1 FC &amp; VIR (pole &amp; spin rate)</td>
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<tr>
<td></td>
<td>&lt; 50,000 km</td>
<td>n/a</td>
<td>RC2 FC &amp; VIR (pole &amp; spin rate)</td>
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<td></td>
<td></td>
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<td>FC Satellite search</td>
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<td><strong>OpNavs (9) FC &amp; VIR</strong></td>
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<td><strong>RC1 FC &amp; VIR (pole &amp; spin rate)</strong></td>
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<td><strong>RC2 FC &amp; VIR (pole &amp; spin rate)</strong></td>
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<td></td>
<td></td>
<td></td>
<td><strong>FC Satellite search</strong></td>
</tr>
<tr>
<td>RC3</td>
<td>14,028 km</td>
<td>1,262 m</td>
<td>Three full rotations (north, equator, south) FC &amp; VIR</td>
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<td></td>
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<td></td>
<td>FC High Phase observations (2 rotations)</td>
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<tr>
<td>Survey</td>
<td>4725 km</td>
<td>395 m</td>
<td>VIR 80% Global coverage nadir</td>
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<td></td>
<td>FC Global coverage clear + 7 filters nadir</td>
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<td>FC Limb observations (3)</td>
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<tr>
<td>HAMO</td>
<td>1950 km</td>
<td>136 m</td>
<td>FC Global clear nadir + 4 off-nadir global maps</td>
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<td></td>
<td></td>
<td></td>
<td>FC Global color coverage (minimum 3 filters)</td>
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<td></td>
<td>VIR Low resolution L1 requirement</td>
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<tr>
<td>LAMO</td>
<td>841 km</td>
<td>33 m</td>
<td>GRaND 75% nadir duty cycle</td>
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<td></td>
<td></td>
<td>Gravity HGA + LGA coverage</td>
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<td></td>
<td>VIR High resolution L1 requirement</td>
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<td>FC Global coverage for chronology</td>
</tr>
</tbody>
</table>

Items in blue are not yet in plan pending detailed hydrazine estimates

*Orbit radius will shift based on gravity model and technique used to develop the trajectory

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Proposed Ceres Mission Timeline

Proposed Ceres Science Mission

- Interplanetary Cruise
- CBE Ceres Capture Date (April 2)
- Ceres Approach
- Survey Science Orbit (~4250 km)
- Transfer
- High Altitude Science Orbit (~1475 km)
- Transfer
- Low Altitude Science Orbit (~365 km)
- Margin
- End Prime Mission

FY15
FY16
FY16
FY16

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Lighting conditions will not drive the Ceres operations timeline as it did at Vesta

- **RC3**
  - SS Lat = -3.9 deg

- **Survey**
  - SS Lat = -3.7 deg

- **HAMO**
  - SS Lat = -3.2 deg

- **LAMO**
  - SS Lat = -1.8 deg
Current timeline example for thrust end prior to RC3s
OpNav 8 Satellite Search

Satellite Search 2: Mar 28 22:00 – Mar 29 01:00 and Mar 29 03:00 – 06:00 (around OpNav 8)

Duration: Two 3 hr observations
Range: 46x10³ km
FC2: Each obs has 3 image stations with 4 sets of 16 images per station (384 images)
VIR: None
Pointing: FC2 pointed at two different explicit offsets from Ceres offsets
Global VIR coverage achieved with resolution of ~1200 km
Survey VIR Incidence Angle

BLUE >65
RED 35-65
YELLOW 0-35

Mercator map vs incidence angle
FC2 Limb Imaging Opportunity

C1 from south pole

C2 from north pole
Incomplete rotation

C6 from the south
Incomplete rotation
HAMO FC2 Stereo and SPC Coverage

Optimal Stereo Coverage

Usable Stereo Coverage

Optimal SPC coverage

Usable SPC coverage

Coverage estimates include all 5 cycles
HAMO VIR Coverage – All Cycles

C1: Purple
C2: Cyan
C3: Green
C4: Orange
C5: Blue

- Coverage reported is <20% (not including smear)
- Results for the preliminary design with 8 sec repetition time
LAMO Orbit Repeat Cycles

- Each cycle is 101 orbits divided into five 20-orbit segments plus one phasing orbit
  - Segment: 12 orbits imaging, 3 orbits for OMM or nadir, 5 orbits for playback,
  - Playback passes every 4.5 days for 1.1 day in duration

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FC LAMO Observations

- Longitude repeats in 93 orbits and the cycle would include 101 orbits (93/101)
  - Each cycle would be composed of 5 segments each containing 20 orbits
  - 12 imaging orbits, 3 non-imaging orbits, and 5 playback orbits (12-3-5)
  - ~25 hr playback passes occur every 4.5 days
- Nadir imaging: 3:00 m:ss cadence in north and south, 2:20 m:ss at equator
  - Images compressed at 3.6:1 will fit into FC internal memory and playback
  - 61 images per orbit initially
  - 3,660 images/cycle for a total of 14,640 images for 4-cycle LAMO
- Images acquired during turns from nadir to Earth could improve the topography model over part of the surface at no additional hydrazine cost
Ceres Hydrazine Budget

Post-Vesta ~30 kg of hydrazine available to complete the mission

- Project holds 5 kg as contingency to cover potential safe mode entries
- Anticipate ~5 kg to get to Ceres
- Leaving ~20 kg available to accomplish the Ceres mission

Estimates continue to be refined and uncertainty will be reduced

Available for Ceres: 20 kg

Used (Launch thru Vesta): 13 kg

Cruise to Ceres: 5 kg

Contingency: 5 kg

Unusable: 3 kg
Summary

• Initial plan for the Ceres science mission that achieves the science objectives and Level-1 Requirements has been completed
  – Mission plan is similar to Vesta
  – Lighting conditions will be nearly constant during encounter

• The hydrazine budget is limiting resource
  – Work is continuing to identify areas for additional hydrazine savings and to refine estimates of uncertainty

• Budget beyond FY15 is still uncertain