



Lunar Exploration Initiative

Briefing Topic:

Lunar Mobility Review

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Lunar Mobility

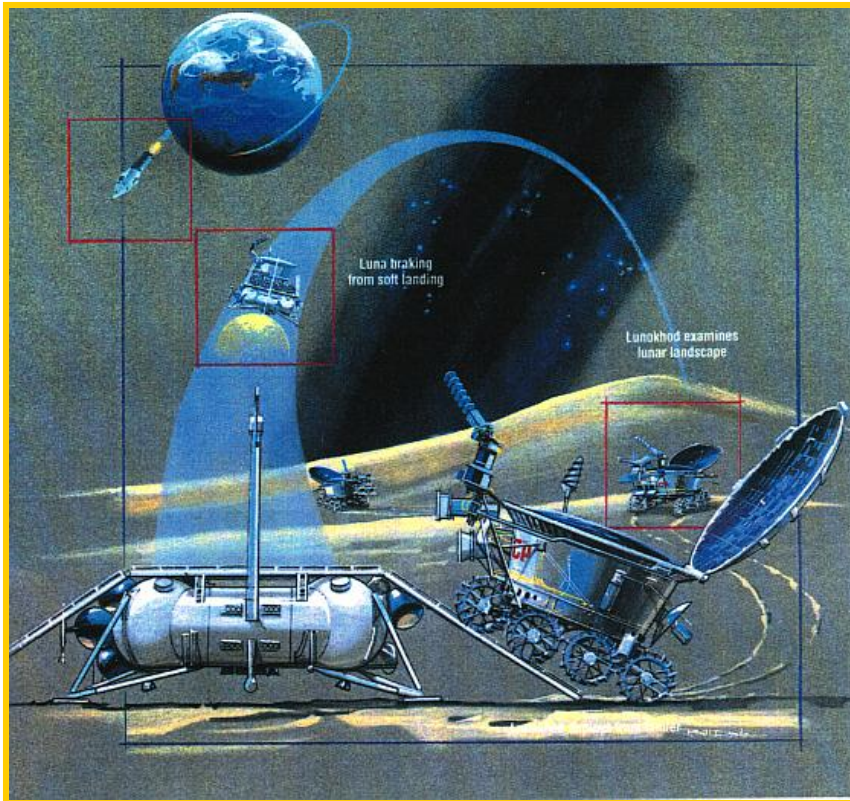
- **Robotic Vehicles**
 - Lunokhod 1 (Luna 17)
 - Lunokhod 2 (Luna 21)
- **Human Exploration Vehicles**
 - MET (Apollo 14)
 - LRV (Apollo 15, 16, and 17)
 - Lunar Motorcycle (for Apollo 15, but not flown)

Lunar Robotic Vehicles

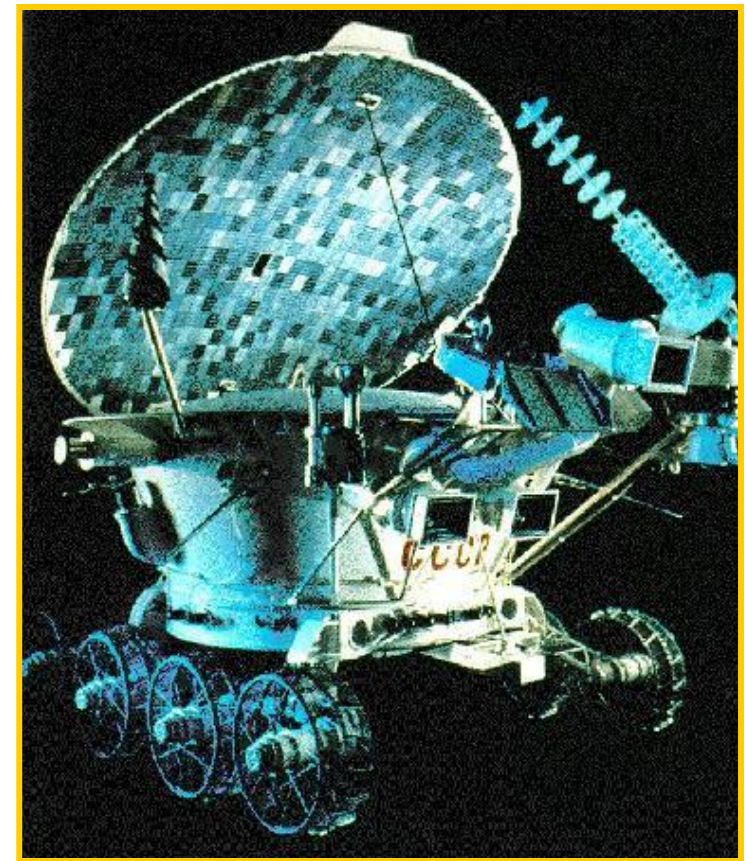
- **Robotic Rovers**
 - **Lunokhod 1 (Luna 17, Nov 1970)**
 - Explored Mare Imbrium
 - 756 kg
 - Rover had 1.7 m wheelbase and was ~1 m wide
 - Driven by 8 rigid spoked wheels with a wire mesh rim connected to three hoops
 - Wheel diameter ~51 cm and width ~20 cm
 - Operated on slopes up to 32°
 - 212-220 day lifetime (~7 lunar days) per Petrov (USSR, 1972) or 322 day lifetime (~11 lunar days) per National Space Science Data Center
 - Traversed 10.54 km
 - **Lunokhod 2 (Luna 21, Jan 1973)**
 - Explored Mare Serenitatis
 - 840 kg (1814 kg with lander)
 - 170 cm long, 160 cm wide, 135 cm high
 - Two-speeds: ~1 km/hr and ~2 km/hr
 - 139 day lifetime (~5 lunar days)
 - Traversed 37 km

NSSDC 1970-095A; 1973-001A

Lunokhod 2



Carried 3 TV cameras, one of which was high on rover for navigation, allowing real-time driving by 5-man team in USSR



Powered by batteries that were recharged by a solar panel on lid of payload bay & a Polonium-210 radiogenic heat source



Lunokhod 2 Mission Profile

- Earth parking orbit
- Translunar injection
- 90 x 100 km parking orbit around Moon
- Perilune lowered to 16 km and stabilized for 40 orbits
- Braking rocket puts lander in free fall
- Main thrusters fire 750 m above surface
- Main thrusters shut down 22 m above surface and secondary thrusters ignited
- Secondary thrusters shut down 1.5 m above surface
- Landing occurs from a free fall height of 1.5 m

- Surface operations
 - Dual-ramp roll-off
 - Navigated while on battery
 - Stopped occasionally to recharge battery with solar panel
 - Hibernated during lunar night, remaining warm with radiogenic heater



Lunokhod 2 Science

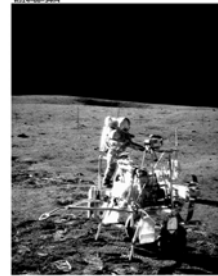
- **Science Goals**
 - Image lunar surface
 - Examine ambient light levels to assess suitability for astronomical observations
 - Perform laser ranging experiments from Earth
 - Observe solar x-rays
 - Measure local magnetic fields
 - Study mechanical properties of lunar surface materials
- **Instruments**
 - 4 panoramic cameras
 - Astrophotometer for VIS and UV light
 - Radiometer
 - Rubin-1 photodetector for laser detection & French-supplied laser corner-reflector
 - Solar X-ray detector
 - Magnetometer (at end of 2.5 m boom)
 - Soil mechanics device (penetrometer)

Human Exploration Vehicles

- **Modular Equipment Transporter (MET) for Human Exploration**

- **Apollo 14 (Jan-Feb 1971)**

- 75 kg (with instruments and samples)
- Hand-drawn
- 2 pneumatic tires
- 40 cm diameter tires, width of 10 cm



- **Apollo Lunar Roving Vehicle (LRV) for Human Exploration**

- **Apollo 15 (July-Aug 1971), Apollo 16 (April 1972), Apollo 17 (Dec 1972)**

- 708 kg (with astronauts, equipment, and samples; more than half of this mass was the astronauts and their life support systems)
- 4 wheels composed of a flexible mesh of woven zinc-coated piano wire and chevron-shaped titanium treads
- 82 cm wheel diameter and 23 cm width
- Battery-powered



- **Lunar Motorcycle**

- **Designed for, but not flown on, Apollo 15**

- 2 pneumatic tires

Apollo LRV

Empty mass	218 kg
Payload mass	490 kg
Two astronauts	363 kg
Experiments, tools, & samples	127 kg
Gross Mass	790 kg
Dimensions	
Length	310 cm
Wheelbase	229 cm
Overall width	206 cm
Height	114 cm
Power supply	2 parallel, non-rechargable Ag-Zn batteries (36 V)
Drive	Independent motors on each wheel
Steering	Front and rear independent steering
Minimum turning radius	305 cm
Wheels	Woven Zn-coated piano wire with Ti-treads in chevron pattern (50% coverage)
Maximum speed	13 km/hr
Normal cruise speed	6 to 7 km/hr
Maximum slope	19 to 23 deg
Energy consumption	35 to 56 W-hr/km 0.05 to 0.08 W-hr/km/kg



Mobility

- **Wheeled vehicles**
 - Based on Apollo and Lunokhod mission results
 - Vehicles with round wheels work well on lunar surface if ground contact pressure does not exceed 7 to 10 kPa
 - Overcoming surface roughness and soil compaction consumes the energy equivalent to a 1 ½ degree climb up a smooth, rigid slope
 - Surface roughness, in a relatively low gravity situation, limits surface speed (otherwise, one bounces out of control)
 - The LRV was limited to 6-7 km/hr
 - Faster speeds require larger wheels, larger wheel base, greater mass, and/or softer suspension

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Mobility

- **Soft soils**

- **The Apollo 15 LRV spun its wheels (and got stuck) in soft soil**
- **The empty LRV weighed only 38 kg in lunar gravity, so the astronauts moved it to solve the problem. This solution is not possible in a completely robotic mission.**
- **Lunokhod 2 encountered soft soils on the inside walls of craters; the soil was particularly soft at the base of slopes**
 - **Normal wheel sinkage was 2 cm**
 - **Wheel sinkage was >20 cm near impact craters**

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Maneuvering

- **Cohesion varies as a function of geologic terrain**
 - **Cohesion on interior crater rims is less than that in intercrater areas**
 - **Cohesion in intercrater areas is less than that on crater rims**

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Slope Requirements

- **LExSWG (1995) findings for rover mobility**
 - **Impact-cratered terrains**
 - Old 100 m diameter crater (a common feature) has maximum slopes of 5 to 10°
 - Somewhat fresher craters have interior slopes of 15 to 20°
 - A very fresh crater, 500 m diameter South Ray Crater, has ejecta blanket and rim slopes of 7° or less; interior crater wall slopes can be as high as 35°, but routes to crater floors with slopes of 17 to 26° exist
 - Even large craters with diameters >10 km have average crater wall slopes <30°
 - **Conclusion: capability to ascend and descend slopes of ~25° is sufficient**

Slope Requirements

- **LExSWG (1995) findings**
 - **Volcanic terrains**
 - Near vertical walls will occur near rilles, but less steep routes to rille floors exist
 - Topographic study of Rima Prinz and Rima Mozart reveal numerous routes to rille floors with slopes of 15 to 20°; routes with slopes <15° also exist
 - Conclusion: capability to ascend and descend slopes of ~25° is sufficient

Trafficability

- Empirical equations for the slope-climbing ability and energy consumed by a wheeled vehicle moving through lunar soil were determined for Apollo's LRV (Bekker, 1969):
 - Wheel sinkage
 - Soil compaction resistance per wheel
 - Gross pull per wheel
 - Maximum trafficable slope
- These equations failed, however, to represent the trafficability of small rovers in lunar soils, as simulated in 1/6 G conditions on NASA 930 (KC-135A) flights (Carrier, 1994, summarizing Scott)
- A computational method (WHEEL-E) was developed to evaluate small rover wheel performance in lunar soils (Carrier, 1995). These solutions are for flexible, elastic wheels on a flexible, elastic surface, so they may potentially be modified to assess the trafficability of tracks.

Summary

- Human and robotic rovers operated on the lunar surface in the past.
- The latter operated for several lunar days & nights, enduring cold conditions without solar power.
- The lunar surface is covered with a soft soil that varies in depth and cohesion; a wheeled vehicle has been stuck in this soil.
- LExSWG (1995) recommended future rovers have the ability to climb slopes up to 25° for operations in both impact-cratered and volcanic terrains.



References

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