



Venus investigations after VENUS EXPRESS: mission VENERA-D

L.V. Zasova (1,6), Korablev O.I. (1,6), Zelenyi L.M. (1), Pichkhadze K.M. (2), Vorontsov V.A. (2), Marov M.Ya. (3), Basilevsky A.T.(3), Gavrik A.L. (4), Gerasimov M.V. (1), Gotlib V.M.(1), Grigoriev A.V. (1), Gromov V. (1), Ekonomov A.P. (1), Ignatiev N.I. (1,6), Khavroshkin O.B.(5) Ksanfomaliti L.V. (1), Lipatov A.N. (1), Liash A.(1), Mitrofanov I.G. (1), Moshkin B.E. (1), Petrukovich A.A. (1), Piccioni G. (7), Rodin A.V.(1,6), Skalsky A.A. (1), Vaisberg. O.L. (1), Zastenker G.N.(1).

(1)IKI RAS, (2) Lavochkin Assoc., (3) GEOKHI RAS, (4)FIRE RAS, (5) IFZ RAS , (6) MIPT, (7) IAPS-INAF

Venera-D. Baseline Mission Overview

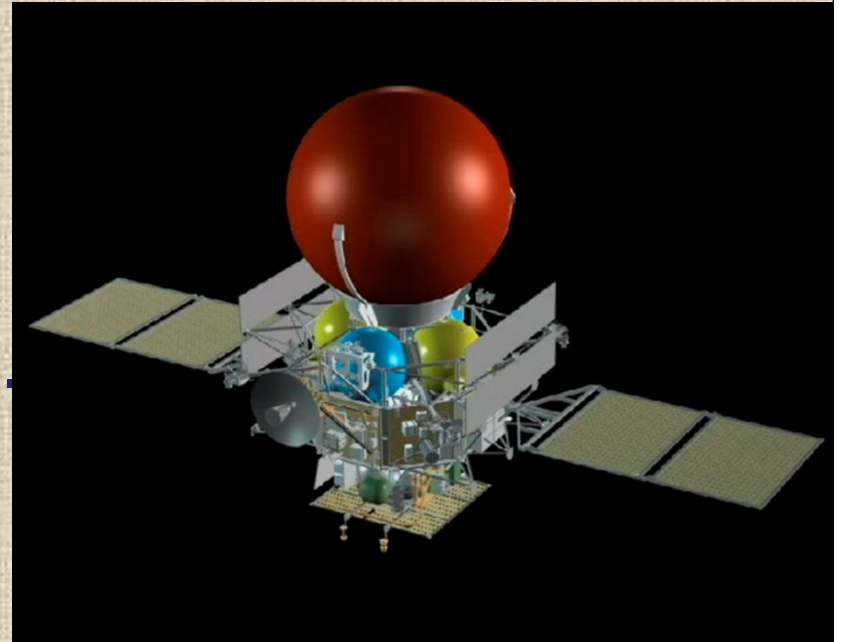
- ROSCOSMOS-led mission.
It was included in the Russian Federal Space Program (RFSP) 2006-2015.
RFSP is revised after PH-G crash.

**Objectives of Venera-D mission:
surface, atmosphere, plasma
environment**

**In 2013- phase A will be continued with laboratory
mock-ups of some elements**

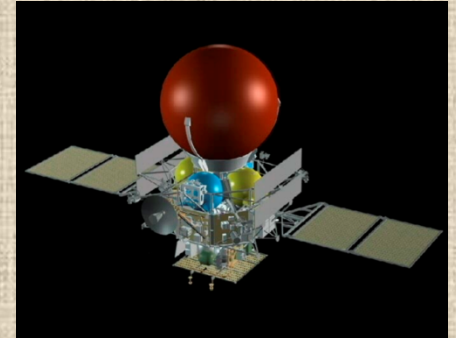
RFSP for 2016 – 2025 will be approved in 2013

Phase B for Venera-D may be open at 2016 earliest



*General view of Venera-D
(Lavochkin Association)*

Elements of mission (current)



- **Lander (VEGA-type, updated)**
 - 2 – 3 hours on the surface

- **Orbiter (Phobos-Grunt, updated to Venus)**
 - on the polar 24 hours orbit , lifetime > 3 years

- **Sub-satellite**
 - orbits with 48, 24, 12 hours are considered

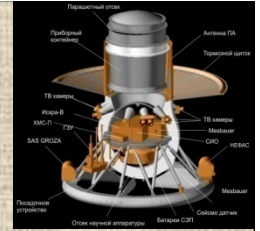
Scientific goals of the Venera-D mission



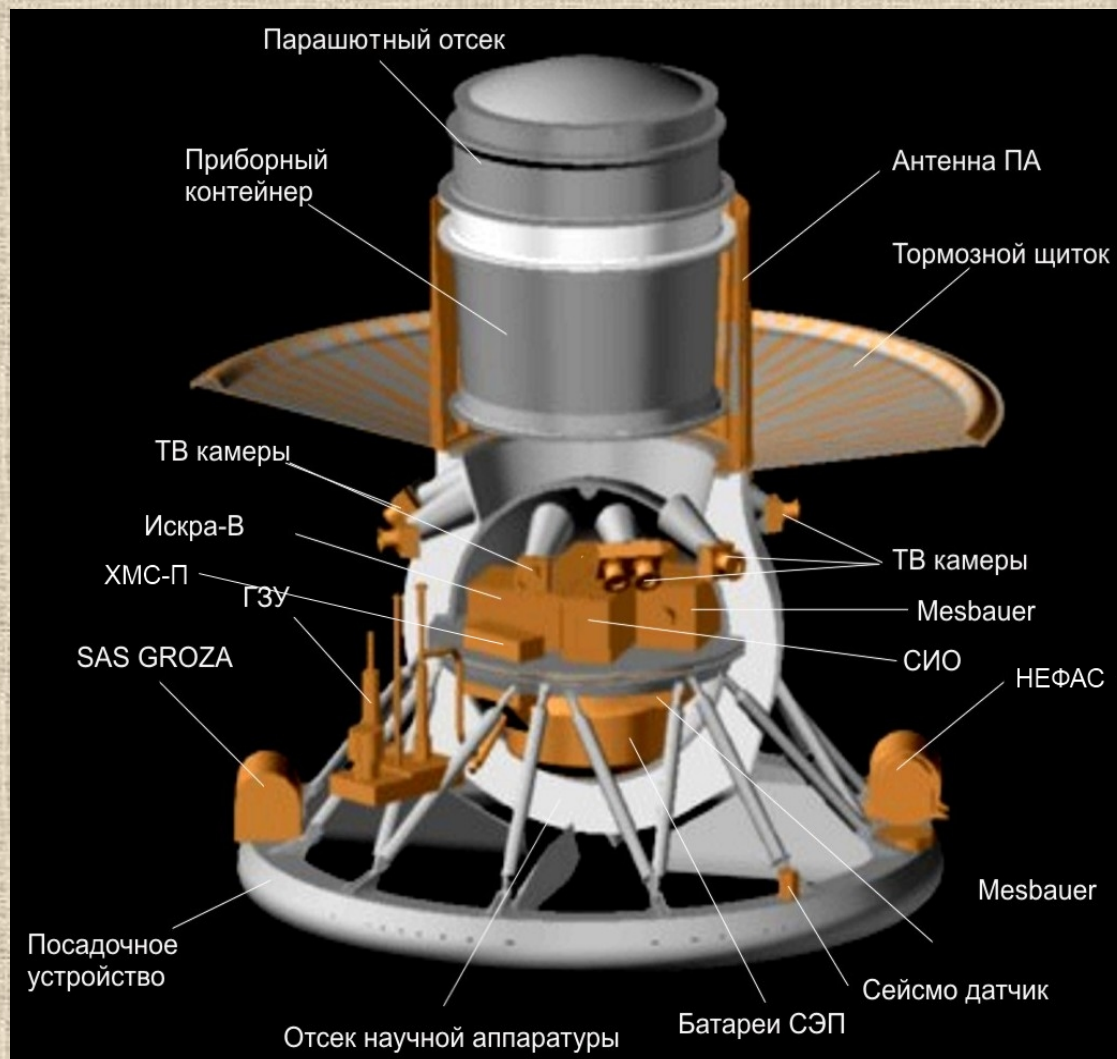
- Investigation of the structure and chemical composition of the atmosphere, including abundances and isotopic ratios of the light and noble gases;
- Thermal structure of the atmosphere, winds, thermal tides and solar locked structures;
- Clouds, structure, composition, microphysics, chemistry;
- Chemical analysis of the surface material, study of the elemental composition of the surface, including radiogenic isotopes;
- Study of interaction between the surface and atmosphere, search for volcanic and seismic activity; search for lightning;
- Study of the dynamics and nature of superrotation, radiative balance and nature of the enormous greenhouse effect;
- Investigation of the upper atmosphere, ionosphere, electrical activity, magnetosphere, escape rate



LANDER PAYLOAD



- ❖ *Active Gamma and Neutron Spectrometric for Soil Analysis*
- ❖ *GC-MS*
- ❖ *Mossbauer spectrometer (MIMOSA-2)*
- ❖ *TV- cameras (landing, stereo, panoramic, high res. up to 0.1 mm)*
- ❖ *MTDLAS – Multi channel tunable diode laser spectrometer*
- ❖ *Nephelometer-particles counter*
- ❖ *Wave-package*
- ❖ *TPW- package*
- ❖ *Optical package*
- ❖ *Radio-science*

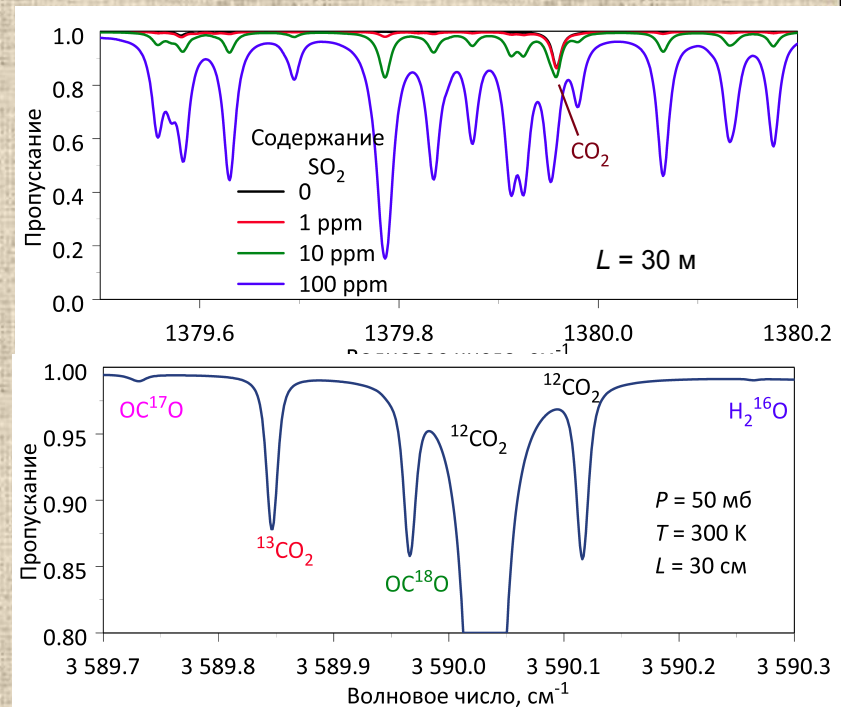
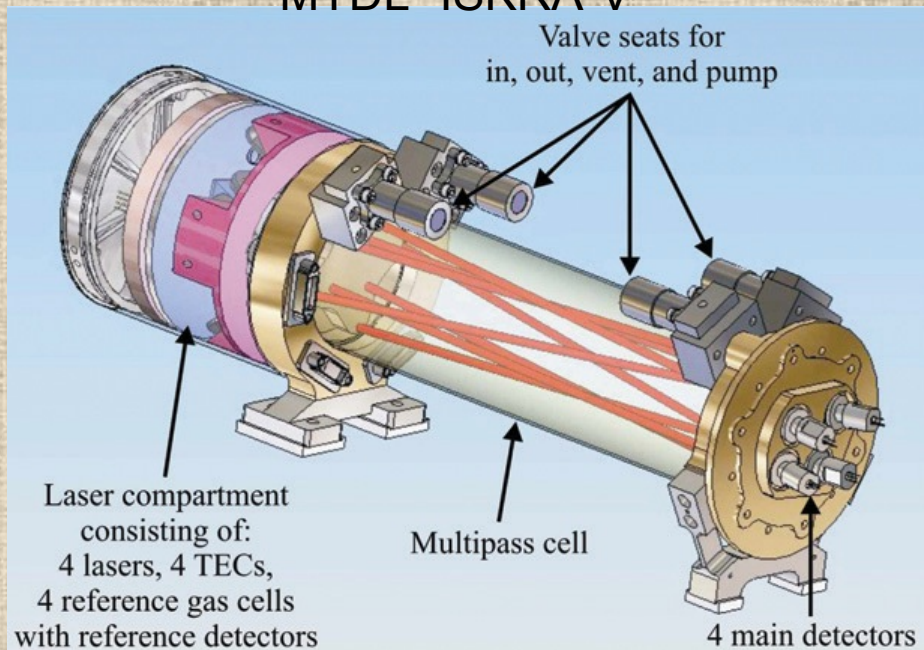


Lander, preliminary assembling of scientific payload on the level of drawings(Lavochkin Association)

MTDLAS- ISKRA-V

- Abundance of isotopomers CO_2 , H_2O , CO , SO_2 , COS , HF , HCl
- Working interval : from 65 км to the surface with vertical resolution ~ 1 км
- The cell should be cleaned between measurements
- It is obtained a spectrum of atmospheric sample in the cell at $p=50$ мб и $T=300$ К for 4-6 spectral channels
- The width of the spectral intervals $\sim 0,5$ cm^{-1} , tuning of the wavelength in the range 5 cm^{-1}
- Spectral resolution $\lambda/\Delta\lambda \sim 10^7$, sensitivity of 10^{-5}
- Optical path length in the multipass cell from 30 cm to 30 м

MTDL- ISKRA-V



Example of construction (C. R. Webster et al., NASA)

Modeled spectra at 7,2 and 2,78 мкм

MTDLAS on the Lander

-measurements from 65 km to the surface

-vertical resolution of 1 – 2 km.

Measurements on Lander may give vertical profiles of measured components

The device for atmospheric sampling will take a probe and rarefy it (to 50 -100 mb)

Number of channels are limited only by existing laser diodes.

MTDL- ISKRA-V

The choice of molecules and spectral ranges within the parameters of existing diode lasers

Wavelength	Molecule	DFB laser
2,17 мкм (4608 cm ⁻¹)	¹² CO ₂ , ¹⁶ O ¹² C ¹⁸ O	GaInAsSb / GaAlAsSb
2,36 мкм (4220 cm ⁻¹)	CO	
2,43 мкм (4100 cm ⁻¹)	HF	
2,78 мкм (3590 cm ⁻¹)	CO ₂ , H ₂ O, ¹³ C/ ¹² C, ¹⁶ O/ ¹⁷ O/ ¹⁸ O (CO ₂) D/H, ¹⁶ O/ ¹⁷ O/ ¹⁸ O (H ₂ O)	
2,84 мкм (3514 cm ⁻¹)	¹³ CO ₂ , ¹² CO ₂	
3,35 мкм (2984 cm ⁻¹)	H ³⁵ Cl, H ³⁷ Cl	Difference-frequency generation with 2 lasers range 1.5 мкм
4,82 мкм (2073 cm ⁻¹)	CO, CO ₂ , OCS ¹³ C/ ¹² C (CO ₂) ³⁴ S/ ³³ S/ ³² S (in OCS)	QCL
7,2 мкм (1380 cm ⁻¹)	SO ₂	

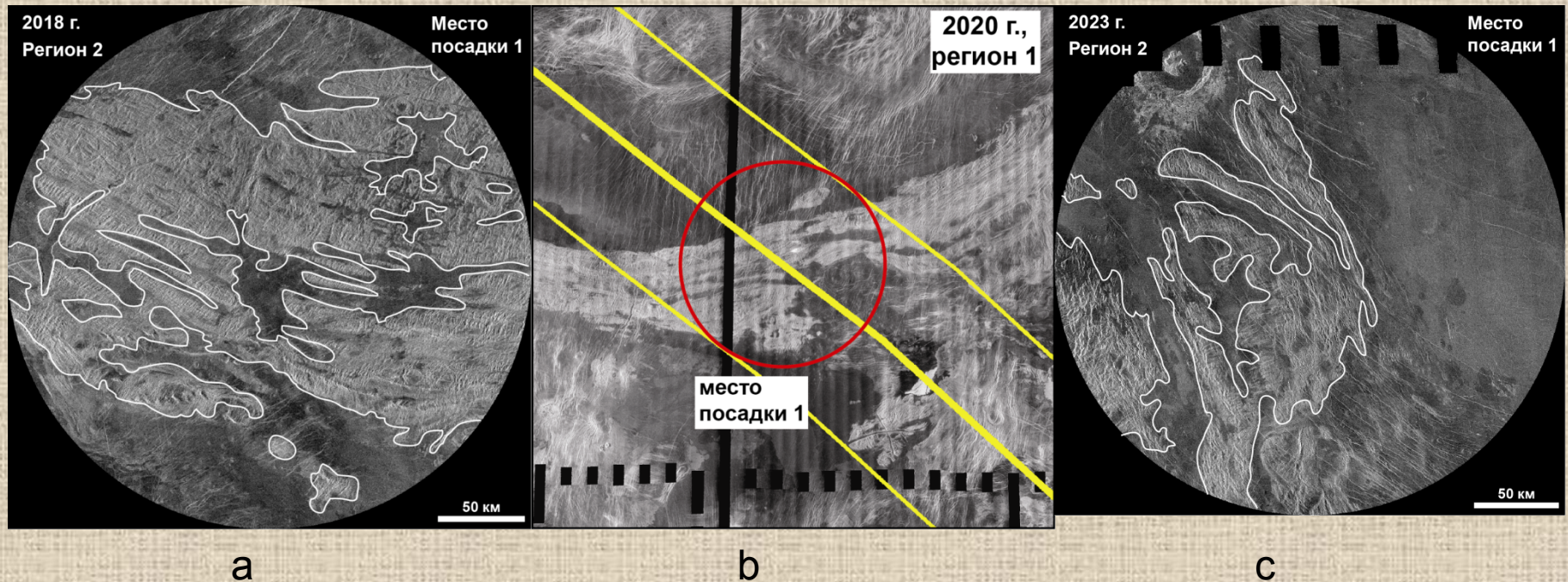
Important Lander scientific goals on the surface:

Chemical composition of the surface material, elemental composition of the surface, including radiogenic isotopes, oxidation state of the surface, geology, geochemistry

- Active Gamma and Neutron Spectrometric Soil Analysis**
- GC-MS with soil sampling**
- Mossbauer spectrometer**
- TV - cameras**

Landing sites – tesserae (A. Bazilevsky)

Landing ellipces for windows of launch 2018 (a), 2020 (b), 2023 (c), with tesserae inside



In ballistic reachable areas for windows of launch 2021 and 2024 the tesserae terrains are not found.

Experiments on the orbiter

**Spectrometer-Interferometer PFS-VD $\lambda = (1) 5-40 \mu\text{m}$,
 $\nu = (10000)2000-250 \text{ cm}^{-1}$, $\Delta\nu = 1 \text{ cm}^{-1}$**

Solar and star occultation UV (0.1-0.3 μm) and IR (2-4 μm)

MM-sounder $\lambda = 3-10 \text{ mm}$

UV-mapping spectrometer $\lambda = 0.2-0.5\mu\text{m}$, $\Delta\lambda = 0.0004 \mu\text{m}$

IR-mapping spectrometer $\lambda = 0.3-5.2 \mu\text{m}$, $\Delta\lambda = 2.4 \text{ nm}$

Multispectral monitoring camera

Radio science (L, S and X ranges)

Plasma package

**On the sub-sattelite - scientific payload includes
plasma package and radio science**

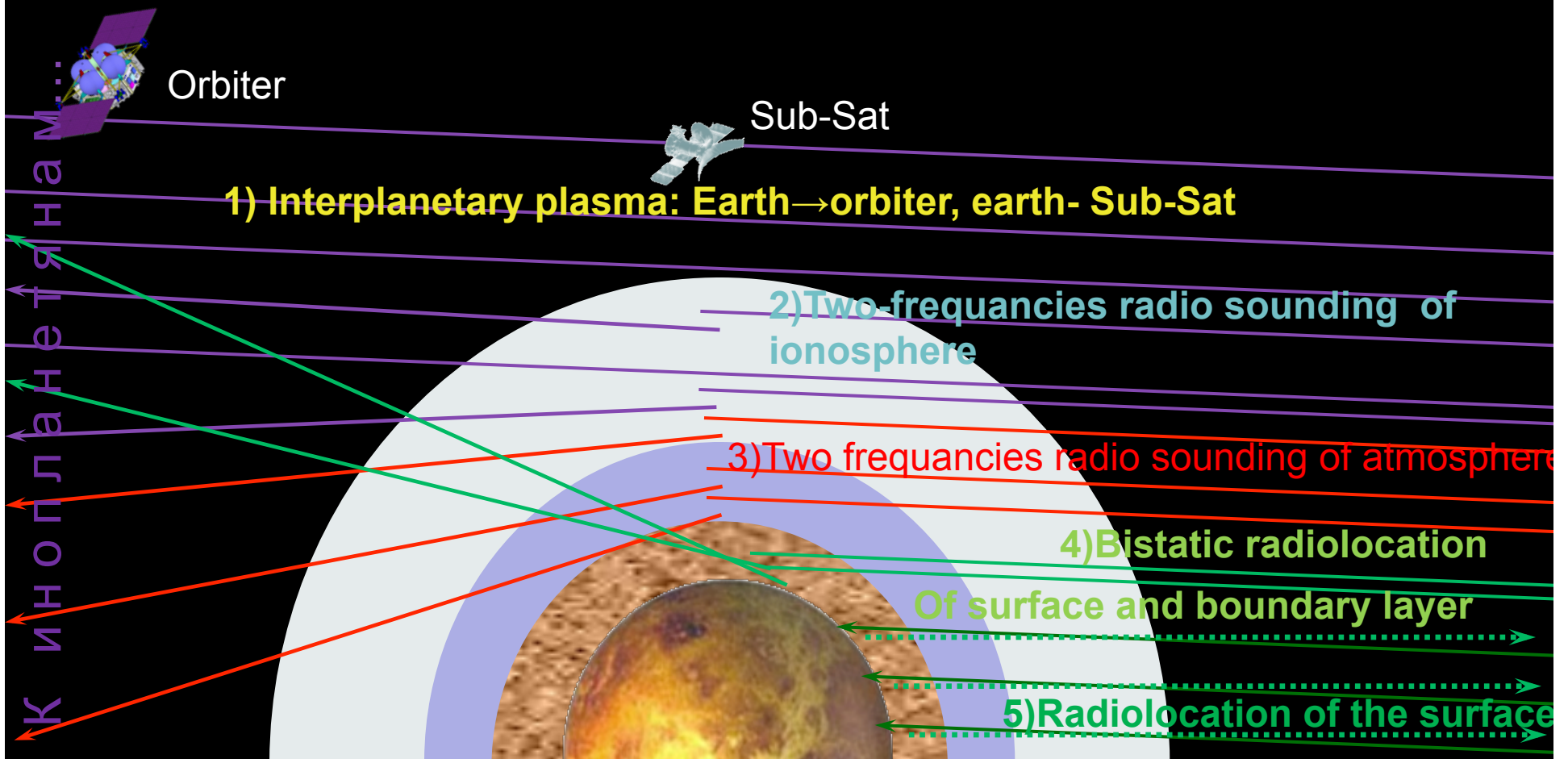
Important instruments on Venera-D, absent or not functioning on VEX (for thermal structure, superrotation, dynamics, thermal balance):

- Thermal IR Fourier spectrometer**
- UV imaging spectrometer**
- MM radiometer**

Other instruments, proposed for the orbiter are modernized experiments being installed on VEX or Veneras

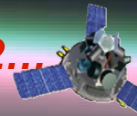
New conception of the RS experiment (FIRE RAN):

To emit signal by ground based transmitter and receive on board of orbiter and sub-sat. by three-frequency detector of L, S , X diapasons. Using 70-m antenna , the gain in power of signal transmitted from the Erath may be an order of magnitude comparing with traditional scheme. With high power of the signal five experiments may carried out simultaneously:

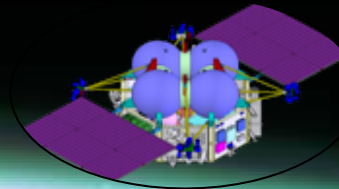


Measurements on orbiter : H, He, O, N, N₂, CO
 H⁺, He⁺, O⁺, N⁺, C⁺, CO₂⁺, N₂⁺, O₂⁺, NO⁺

Измерения на ОА: Эмиссии O₂, OH, O, CO, CO₂...



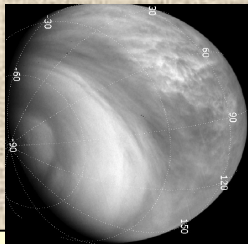
СС



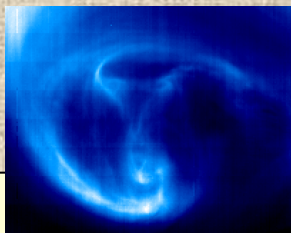
ОА

H=100 км

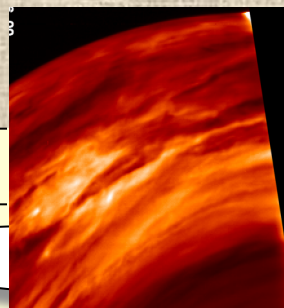
Upper clouds, VMC, UV



S-polar «dipole»
 VIRTIS, 5мкм



Lower clouds,
 VIRTIS, 1.7мкм



Measurements on orbiter :
 H₂O, SO₂, SO, CO, HF, HCl,
 HDO, ClO, H₂S, H₂O₂, OCS

H₂SO₄+УФ-поглотитель?

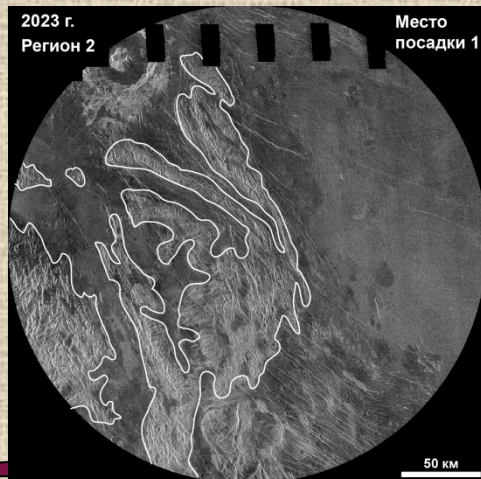
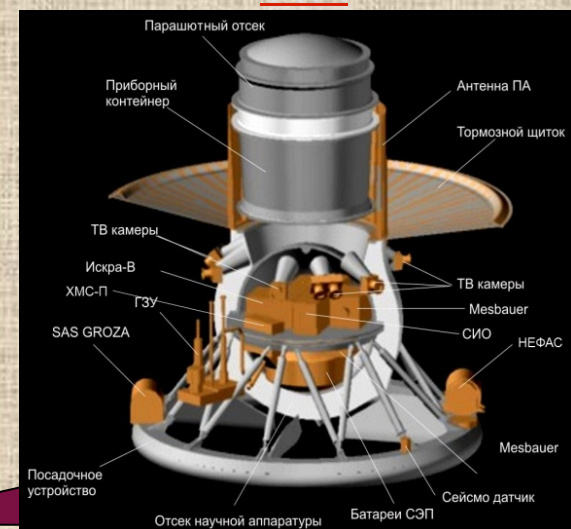
H₂SO₄+кристаллы?

H=50 км

Time of descent in atmosphere – 30 min, lifetime on the surface > 120 min

ПА

Landing site-
 tesserae terrain



Measurements on the descent :
 CO, H₂O, SO₂, OCS, HCl, HF,
 Ne, Ar, Kr, Xe, O, C, N

Measurements on the surface:
 Si, O, Ca, Na, Al, Mg, Fe (Fe²⁺,
 Fe³⁺, Fe⁶⁺), Ti, K, Th, U

Other works, connected to Venera-D:

- Technical description of the long-living station was developed. It was shown that for available at the moment technology, which uses traditional materials on the basis of silicon electronics, the lifetime of the station of mass of 100 kg on the surface of Venus is limited by 24 hours.
- Technical description of the high pressure camera (P up to 150 bar) is created. We hope to have this camera in IKI with working volume 10-15 liters inside to test the separate instruments, which needs to be installed outside

Summary

- Mission Venera-D is delayed to 20th
- Current conception is two orbiters and Lander
- Long living station (for 24 hours) with just simple measurements of PTW may be included
- Inclusion balloon may be considered
- International status of mission would accelerate the start of phase B of the project
- Dialog NASA with Roscosmos about cooperation would be very useful